

-SPACE SHUTTLE-

**BASIC SUBSONIC STATIC
AERODYNAMIC CHARACTERISTICS
FOR GRUMMAN H-33 ORBITER
CONFIGURATION (M=0.17)**

by

**W. Jung, GAC
F. Carlucci, GAC**

N72-12877

(NASA-CR-120006) SPACE SHUTTLE: BASIC
SUBSONIC STATIC AERODYNAMIC CHARACTERISTICS
FOR GRUMMAN H-33 ORBITER CONFIGURATION (M
EQUALS 0.17) W. Jung, et al (Chrysler
Corp.) Nov. 1971 98 p

CSCL 22B G3/31

Unclass
09319

**GRUMMAN 7 X 10-FOOT
LOW SPEED WIND
TUNNEL**

**GRUMMAN AEROSPACE
CORPORATION**

SADSAC SPACE SHUTTLE
AEROTHERMODYNAMIC
DATA MANAGEMENT SYSTEM

CONTRACT NAS8-401
MARSHALL SPACE FLIGHT CENTER

SPACE DIVISION



**CHRYSLER
CORPORATION**

THIS DOCUMENT SHOULD BE
REFERENCED AS NASA
CR-120,006



Reproduced by
**NATIONAL TECHNICAL
INFORMATION SERVICE**
U S Department of Commerce
Springfield VA 22151

A B S T R A C T

This report presents the results of an experimental aerodynamic investigation of the Grumman Aerospace Corporation H-33 Space Orbiter. The purpose of the investigation was to determine static aerodynamic characteristics of the orbiter at a Mach number of 0.17. These data were determined by employing a 1/25 scale model of the orbiter for pitch and yaw variations of -4 degrees to 24 degrees and -15 degrees to 15 degrees, respectively. This investigation was conducted in the Grumman Aerospace Corporation 7 - by - 10 Foot Wind Tunnel.

THIS PAGE INTENTIONALLY LEFT BLANK

NASA Series Number: S-0615

DMS-DR-1167
November, 1971

SADSAC/SPACE SHUTTLE

WIND TUNNEL TEST DATA REPORT

CONFIGURATION: GAC H-33 Orbiter

TEST PURPOSE: To Determine Basic Subsonic Aerodynamic Data on the H-33
Orbiter (M = 0.17)

TEST FACILITY: Grumman 7 x 10 Foot Low Speed Wind Tunnel

TESTING AGENCY: Grumman Aerospace Corporation

TEST NO. & DATE: GWTT 292, June 21 - July 6, 1971

FACILITY COORDINATOR: Mr. M. Quan - GAC

PROJECT ENGINEER(S): Mr. W. Jung - GAC

Mr. F. Carlucci - GAC

DATA MANAGEMENT SERVICES

LIAISON:

J. J. Glynn
for L. Mulkey

DATA OPERATIONS:

J. R. Ziller

RELEASE APPROVAL:

J. J. Glynn
for N. D. Kemp, Supervisor
Aero Thermo Data Group

CONTRACT NAS 8-4016

AMENDMENT 153

DRL 184-58

This report has been prepared by Chrysler Corporation Space Division under a Data Management Contract to the NASA. Chrysler assumes no responsibility for the data presented herein other than its display characteristics.

FACILITY COORDINATOR:

Mr. M. Quan
Grumman Aerospace Corporation
Bethpage, Long Island, New York 11714
Phone: (516) 575-7044

PROJECT ENGINEERS:

Mr. W. Jung
Grumman Aerospace Corporation
Bethpage, Long Island, New York 11714

Phone: (516) 575-7044

Mr. F. Carlucci
Grumman Aerospace Corporation
Bethpage, Long Island, New York 11714

Phone: (516) 575-7044

SADSAC LIAISON:

Mr. T. L. Mulkey
Chrysler Corporation Space Division
Ames Research Center
Mail Stop 229-1
Moffett Field, California 94035

Phone: (415) 961-2402

SADSAC OPERATIONS:

Mr. J. R. Ziler
Chrysler Corporation Space Division
P. O. Box 29200
New Orleans, Louisiana 70129

Phone: (504) 255-2304

TABLE OF CONTENTS

	<u>PAGE NUMBER</u>
LIST OF FIGURES & LIST OF TABLES	2
SUMMARY	3
CONFIGURATIONS INVESTIGATED	5
DATASET COLLATIONS	6
TEST FACILITY DESCRIPTION	8
TEST CONDITIONS	10
DATA REDUCTION	11
SUMMARY DATA PLOT INDEX	12
FIGURES	15
MODEL COMPONENT DESCRIPTION SHEETS	20
GENERAL NOMENCLATURE	26
NON-STANDARD NOMENCLATURE	32
TABULATED DATA LISTING	33
PLOTTED DATA	34

LIST OF FIGURES

<u>FIGURE NUMBER</u>	<u>TITLE</u>	<u>PAGE NUMBER</u>
1.	Axis System	16
2.	Grumman Orbiter Configuration H-33 Without External Tanks	17
3.	Full Scale Model of H-33 Orbiter Without External Tanks	18
4.	H-33 Orbiter Vertical Tail With Segmented Rudder (1/25 Scale)	19

LIST OF TABLES

<u>TABLE NUMBER</u>	<u>TITLE</u>	<u>PAGE NUMBER</u>
I	SADSAC Nomenclature of Aerodynamic Coefficients	4

S U M M A R Y

Presented herein are the results of an experimental static aerodynamic investigation of the Grumman Aerospace H-33 Space Orbiter configuration. The purpose of the investigation was to determine basic subsonic aerodynamic characteristics for pitch and yaw variations of -4 degrees to 24 degrees and -15 degrees to 15 degrees, respectively, for freestream Mach number of 0.17. Primary test parameters included a basic component buildup, and effects on the basic aerodynamics due to elevon and rudder deflections. Also investigated were the effects of the rudder speed brake at flare angles of 0 degrees, -10 degrees, and -15 degrees along with top rudder panel variations.

TABLE I. SADSAC NOMENCLATURE OF AERODYNAMIC COEFFICIENTS

COEFFICIENT	COEFFICIENT NAME	SADSAC NOMENCLATURE		
		BODY AXIS	STABILITY AXIS	WIND AXIS
C_A	Total Axial Force	CA	-	-
C_{AB}	Base Axial Force	CAB	--	-
C_{AF}	Forebody Axial Force	CAF	-	-
C_D	Total Drag Force	-	CD	CDTOTL
C_{DB}	Base Drag Force	-	CDB	CDBASE
C_{DF}	Forebody Drag Force	-	CDF	CDFORE
C_L	Lift Force	-	CL	CL
C_N	Normal Force	CN	-	-
C_Y	Side Force	CY	CY	CC
C_ℓ	Rolling Moment	CBL	CSL	CWL
C_m	Pitching Moment	CLM	CLM	CPM
C_n	Yawing Moment	CYN	CLN	CLN
L/D	Lift-To-Drag Force Ratio	-	L/D	CL/CD
L/D	Lift-To-Forebody Drag Force Ratio	-	L/DF	CL/CDF
N/A	Normal-To-Axial Force Ratio	N/A	-	-
N/A	Normal-To-Forebody Axial Force Ratio	CN/CAF	-	-

CONFIGURATIONS INVESTIGATED

Tests were conducted on the 1/25 Scale Grumman Orbiter Configuration H-33 without external tanks as shown in Figures 2 and 3. The delta wing (W4) had adjustable elevons and the vertical tail (V5) had upper and lower rudder segments. The lower segments were capable of splitting into a "V" section to form a speed brake while functioning as a rudder.

Base pressures were taken at 10 locations on the model. Boundary layer transition was fixed by the use of strips of pinked electrician's tape (0.007 thick) located 1.5 in. streamwise aft of the fuselage nose and 0.75 in. aft of the leading edges of the wing and tail surfaces.

TEST GWTT 292 DATA SET COLLATION SHEETLOW SPEED AERODYNAMIC CHARACTERISTICS OF A $\frac{1}{25}$
SCALE MODEL OF THE GRUMMAN CONF. H-33 ORBITER

PRETEST
 POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.	MACH NUMBERS		NO. OF RUNS	CONTROL DEFLECTIONS								
			a	b		0.17			δ_{EL}	δ_{ER}	δ_R	δ_{RT}		
RD1002	$B_5 W_4 V_5$	A	0	2	1				0	0	0	-	-	-
004	$B_5 W_4 V_5$	A	3	4					0	0	0	-	-	-
006	$B_5 W_4^{-5,-5} V_5$	A	0	6					-5	-5	0	-	-	-
007	$B_5 W_4^{-10,+10} V_5$	A	0	7					-10	-10	0	-	-	-
008	$B_5 W_4^{-15,-15} V_5$	A	0	8					-15	-15	0	-	-	-
009	$B_5 W_4^{-35,-35} V_5$	A	0	9					-35	-35	0	-	-	-
010	$B_5 W_4^{+5,-5} V_5$	A	0	10					-5	-5	0	-	-	-
011	$B_5 W_4^{+5,-15} V_5$	A	0	11					+5	-15	0	-	-	-
012	$B_5 W_4^{-5,-15} V_5$	A	0	12					-5	-15	0	-	-	-
013	$B_5 W_4^{0,-10} V_5$	A	0	13					0	-10	0	-	-	-
014	$B_5 W_4 V_5^{0,+15,-15}$	A	0	14					0	0	-	0	+15	-15
015	$B_5 W_4 V_5^{0,+30,-30}$	A	0	15					0	0	-	0	+30	-30
016	$B_5 W_4 V_5^{-10,+5,-25}$	A	0	16					0	0	-	-10	+5	-25
017	$B_5 W_4 V_5^{-15,0,-30}$	A	0	17					0	0	-	-15	0	-30
018	$B_5 W_4 V_5^{-15,+25,-35}$	A	0	18					0	0	-	-15	+25	-35
019	$B_5 W_4 V_5^{-5}$	A	0	19					0	0	-5	-	-	-
020	$B_5 W_4 V_5^{-10}$	A	0	20					0	0	-10	-	-	-
021	$B_5 W_4 V_5^{-15}$	A	0	21					0	0	-15	-	-	-
023	B_5	A	0	23				Y	-	-	-	-	-	-

1 7 13 19 25 31 37 43 49 55 61 67 75 76

CL	ICD	KY	CLM	CLN	CSL							
----	-----	----	-----	-----	-----	--	--	--	--	--	--	--

COEFFICIENTS: $\alpha A = -4^\circ \text{ to } 24^\circ \text{ by } 2^\circ$ $\Rightarrow IDPVAR(1) | IDPVAR(2) | NDV$

α or β
SCHEDULES $\beta B = 0, \pm 1, \pm 2, \pm 3, \pm 5, \pm 7, \pm 9, \pm 12, \pm 15$

TEST GWTT 292 DATA SET COLLATION SHEET

PRETEST
 POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.	MACH NUMBERS			NO. of RUNS	CONTROL DEFLECTIONS					
			a	b	0.17		δ_{el}	δ_{er}	δ_R	δ_{RT}	δ_{RL}	δ_{RLR}
RD1025	$B_5 W_4$	O B	25			1	0	0	-	-	-	-
027	$B_5 W_4 V_5$	O B	27				0	0	0	-	-	-
028	$B_5 W_4^{0,-10} V_5$	O B	28				0	-10	0	-	-	-
029	$B_5 W_4^{-10,-10} V_5$	O B	29				-10	-10	0	-	-	-
030	$B_5 W_4^{-5,-15} V_5$	O B	30				-5	-15	0	-	-	-
031	$B_5 W_4^{+5,-15} V_5$	O B	31				+5	-15	0	-	-	-
032	$B_5 W_4^{+5,+5} V_5$	O B	32				+5	-5	0	-	-	-
033	$B_5 W_4^{+5,+5} V_3$	O B	33				-5	-5	0	-	-	-
034	$B_5 W_4 V_5^{0,+30,-30}$	O B	34				0	0	-	0	+30	-30
035	$B_5 W_4 V_5^{-10,+25,-35}$	O B	35				0	0	-	-10	+25	-35
036	$B_5 W_4 V_5^{-15,+25,-35}$	O B	36				0	0	-	-15	+25	-35
037	$B_5 W_4 V_5^{-15,0,-30}$	O B	37				0	0	-	-15	0	-30
038	$B_5 W_4 V_5^{-10,+5,-25}$	O B	38				0	0	-	-10	+5	-25
039	$B_5 W_4 V_5^{0,+15,-15}$	O B	39				0	0	-	0	+15	-15
040	$B_5 W_4 V_5^{-10,-10}$	O B	40				0	0	-10	-	-	-
041	$B_5 W_4 V_5^{-15,-15}$	O B	41				0	0	-15	-	-	-
042	$B_5 W_4 V_5^{-5,+25,-35}$	O B	42				0	0	-	-5	+25	-35
054	$B_5 W_4^{-25,+25} V_5$	A O	54				-25	-25	0	-	-	-
055	$B_5 W_4 V_5^{-5,+25,-35}$	A O	55			Y	0	0	-	-5	+25	-35

1 7 13 19 25 31 37 43 49 55 61 67 75 76

CL CD ICY CLM CLN CSL

COEFFICIENTS:

a or b $\alpha A = -4^\circ \text{ to } 24^\circ \text{ by } 2^\circ$

SCHEDULES $\beta B = 0, \pm 1, \pm 2, \pm 3, \pm 5, \pm 7, \pm 9, \pm 12, \pm 15$

IDPVAR(1) | IDPVAR(2) | NDV

TEST FACILITY DESCRIPTION

This is a continuous flow, open circuit, closed throat tunnel. The test section is 7 feet by 10 feet by 20 feet. A 12-foot diameter propeller is driven at speeds up to 1200 rpm by a 1750 HP induction electric motor.

PERFORMANCE PARAMETERS:

Speed Range (mph):	134 maximum
Reynolds Number ($\times 10^6$ /ft):	1.7 maximum
Operating Pressure:	Atmospheric
Dynamic Pressure (psf):	46 maximum
Stagnation Temperature:	Ambient
Power (HP):	1750

TESTING CAPABILITIES:

The GAC Low Speed Wind Tunnel is equipped for Aerodynamic force and pressure testing on 3-dimensional, 2-dimensional, and reflection plane models. Capability also exists for conducting powered model, flutter, jet flap, flow visualization, and wake survey tests.

A variable frequency power unit having a range from 0 to 660 HZ, and a maximum output of 100 KW is available for the 3 phase induction motors used in powered models.

Available model supports include: two and three point supports, a single mount with fixed linkages for setting model pitch attitude, wire supports, and a

TEST FACILITY DESCRIPTION (Continued)

pedestal mount for half model tests. Special installations can be provided to sting support models. Image systems are available for evaluating model support tare and interference effects.

The tunnel primary balance is a six-component mechanical external yoke type. The balance incorporates a pitch arm with $\pm 45^{\circ}$ pitch capability and a yaw table which can provide $\pm 45^{\circ}$ of yaw. Capability also exists for conducting tests with internal strain gage balance installations.

Removable wall inserts are available for 2-dimensional tests and a stationary ground plane is available for investigating ground effects.

An IBM 1800 computer system serves as the facilities data acquisition and reduction center. It is dedicated to the facility and provides on-line data reduction capability. On-site plotting is also available on a Calcomp drum-type plotter.

The facility can provide photographic, closed-circuit and play-back television documentation. Manometers and scanivalves are available for pressure recording.

TEST CONDITIONS

TEST GWTT 292

BALANCE UTILIZED: YOKE-TYPE LOW SPEED WIND TUNNEL

CAPACITY:

ACCURACY.

COEFFICIENT TOLERANCE:

NF	<u>+ 4000 lb.</u> to <u>-2000 lb.</u>	<u>.5 lb.</u>
SF	<u>+ 500 lbs.</u>	<u>.2 lb.</u>
AF	<u>± 500 lbs.</u>	<u>.2 lb.</u>
PM	<u>+ 1200 ft.-lbs.</u>	<u>1 ft. lb.</u>
YM	<u>+ 1200 ft.-lbs.</u>	<u>1 ft. lb.</u>
RM	<u>+ 1200 ft.-lbs.</u>	<u>1 ft. lb.</u>

$\pm .002$
 $\pm .001$
 $\pm .0005$
 $\pm .005$
 $\pm .005$
 $\pm .005$

COMMENTS:

DATA REDUCTION

All force data collected by the six component mechanical balance were reduced to standard NASA coefficient form along and about a system of stability axis. This axis system originated from a nominal c.g. located at F.S. 1285, W.L. 403 and B.L. 0.

The data were reduced with the following reference values:

$$S_{ref} = 7.74 \text{ ft.}^2$$

$$b_{ref} = 3.78 \text{ ft.}$$

$$\ell_{ref} = 5.40 \text{ ft.}$$

Corrections were applied to the data for effects due to tunnel flow alignment and model support tare and interference. The values of these corrections were determined from GWTT 290 which was run in April, 1971.

The data was corrected for tunnel wall effects by employing the following equations:

$$\Delta a = 1.04 C_L$$

$$\Delta C_D = .0182 C_L^2$$

$$\Delta C_M = .00055 C_L^2$$

The free stream dynamic pressure was corrected for model blockage by the following equation:

$$q_\infty = 1.030 q_{INDICATED}$$

Transition was fixed with strips of pinked electrician's tape, 1.5 inches streamwise aft of the fuselage nose and 0.75 inches aft of the leading edges of the wing and tail surfaces.

SUMMARY DATA PLOT INDEX

TITLE	PLOTTED COEFFICIENTS SCHEDULE	CONDITIONS VARYING	PAGES
Elevator Effectiveness	A	Elevon Deflection Angle	1
Elevator Effectiveness	B	Elevon Deflection Angle	2-3
Effect of Body	C	Configuration	4-6
Yaw Polar, Beta = 3. Degrees	D	Beta Angle	7-12
Aileron Effectiveness in Pitch, Elevtr = 0. Degrees	D	Aileron Deflection Angle	13-18
Aileron Effectiveness in Pitch, Elevtr = -5. Degrees	D	Aileron Deflection Angle	19-24
Aileron Effectiveness in Pitch, Elevtr = -10. Degrees	D	Aileron Deflection Angle	25-30
Speed Brake Effect in Pitch - Brakes at -15., 15. and -30., 30. Degrees	C	Speed Brake Deflection Angle	31-33
Rudder Effectiveness in Pitch - Speed Brakes at 15. and -15. Degrees	D	Rudder and Speed Brake Deflection Angle	34-39
Rudder Effectiveness in Pitch - Speed Brakes at 30. and -30. Degrees	D	Rudder and Speed Brake Deflection Angle	40-45
Rudder Effectiveness in Pitch - No Speed Brakes	D	Rudder Deflection Angle	46-51
Effect of Vertical Tail in Yaw	E	Configuration	52-54

SUMMARY DATA PLOT INDEX
(CONTINUED)

TITLE	PLOTTED COEFFICIENTS SCHEDULE	CONDITIONS VARYING	PAGES
Effect of Speed Brakes in Yaw - Brakes at 15., -15. and 30., -30. Degrees	F	Speed Brake Deflection Angle	55-60
Aileron Effectiveness in Yaw, Elevtr = 0. Degrees	F	Aileron Deflection	61-66
Aileron Effectiveness in Yaw, Elevtr = -5. Degrees	F	Aileron Deflection Angle	67-72
Aileron Effectiveness in Yaw, Elevtr = -10. Degrees	F	Aileron Deflection Angle	73-78
Rudder Effectiveness in Yaw - Speed Brakes at 30. and -30. Degrees	F	Rudder Deflection Angle	79-84
Rudder Effectiveness in Yaw - Speed Brakes at 15. and -15. Degrees	F	Rudder and Speed Brake Deflection Angle	85-90
Rudder Effectiveness in Yaw - No Speed Brakes	F	Rudder Deflection Angle	91-96

Note: Plotted Coefficients Schedule on following page.

SUMMARY DATA PLOT INDEX
(CONTINUED)

PLOTTED COEFFICIENTS SCHEDULE:

SCHEDULE A

CL vs. α

SCHEDULE B

CL vs. CD
CL vs. CLM

SCHEDULE C

CL vs. α
CD vs. α
CLM vs. α

SCHEDULE D

CL vs. α
CD vs. α
CLM vs. α
CLN vs. α
CSL vs. α
CY vs. α

SCHEDULE E

CLN vs. β
CSL vs. β
CY vs. β

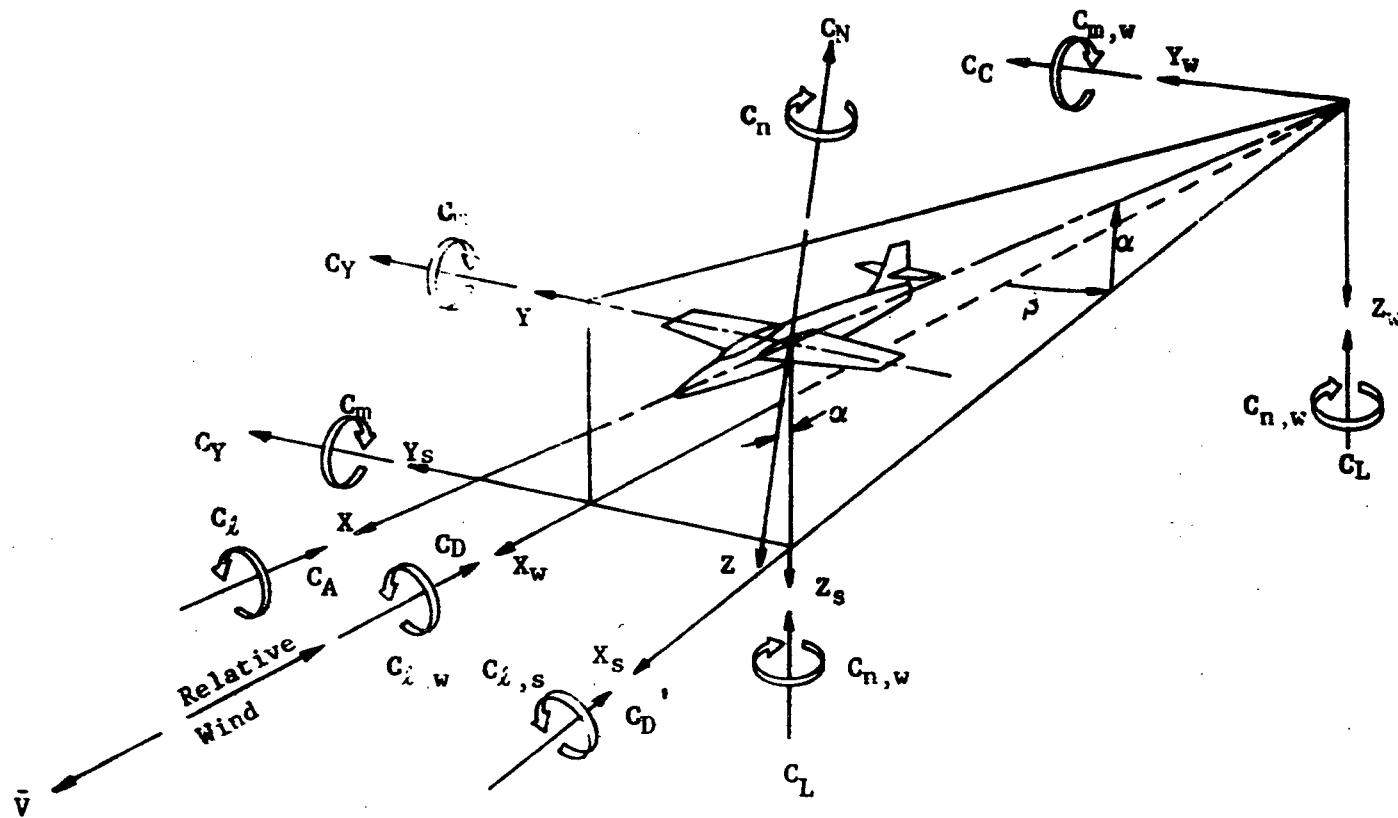
SCHEDULE F

CLN vs. β
CSL vs. β
CY vs. β
CL vs. β
CD vs. β
CLM vs. β

F I G U R E S

Notes:

1. Positive directions of force coefficients moment coefficients, and angles are indicated by arrows.
2. For clarity, origins of wind and stability axes have been displaced from the center of gravity.



16

Figure 1. Axis systems, showing direction and sense of force and moment coefficients, angle of attack, and sideslip angle

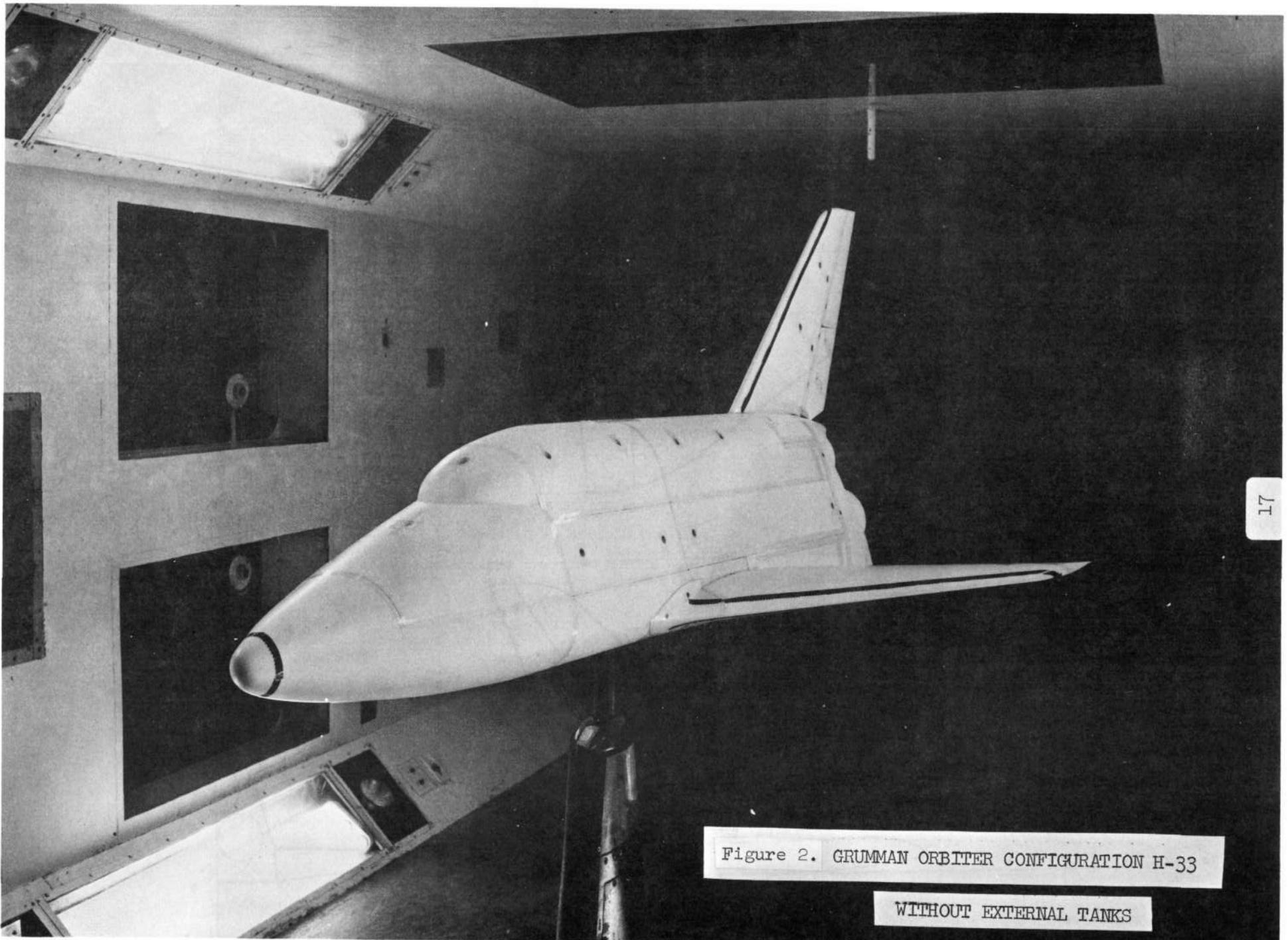
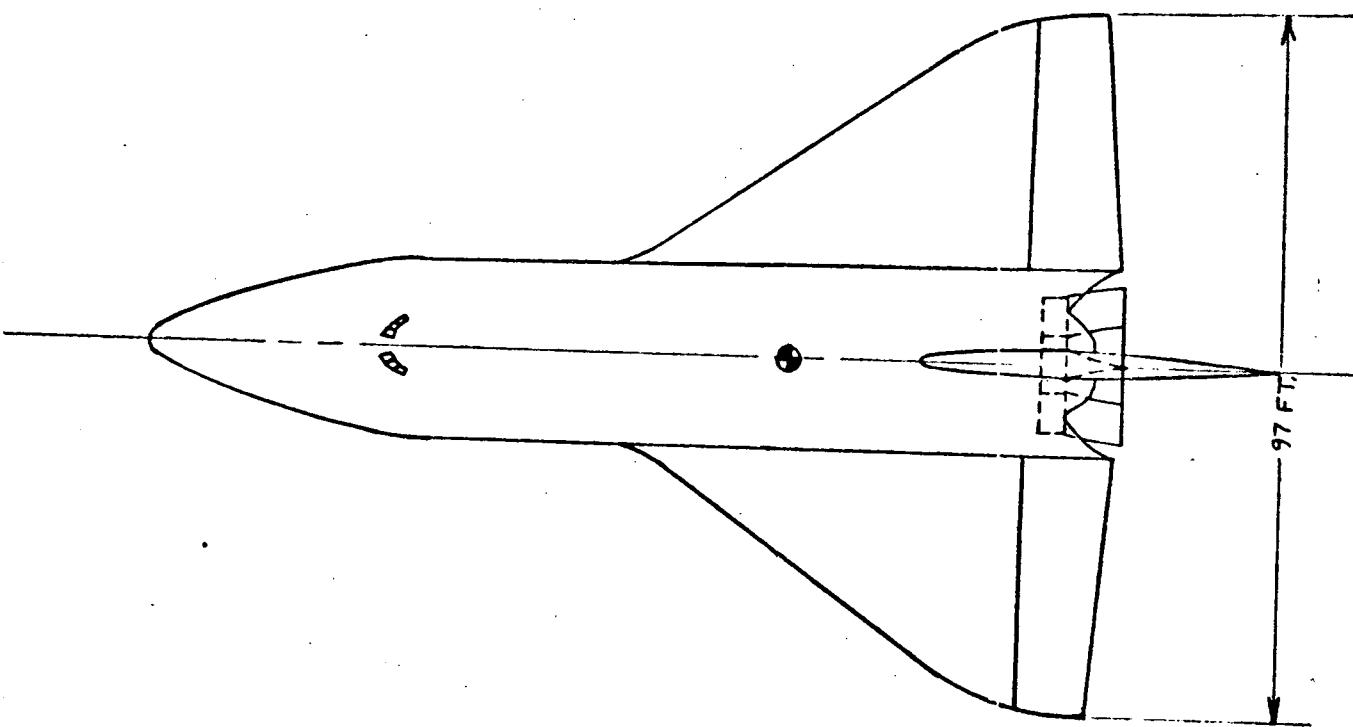


Figure 2. GRUMMAN ORBITER CONFIGURATION H-33

WITHOUT EXTERNAL TANKS



18

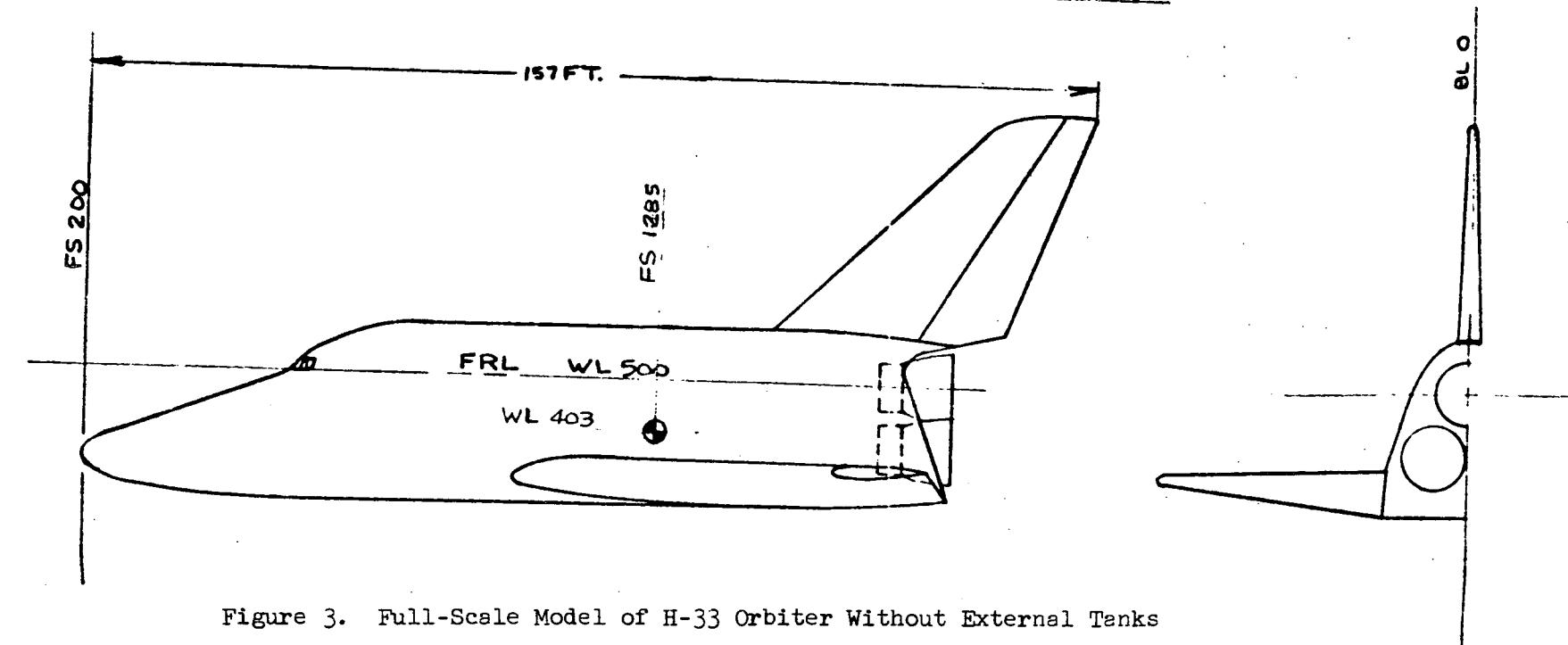


Figure 3. Full-Scale Model of H-33 Orbiter Without External Tanks

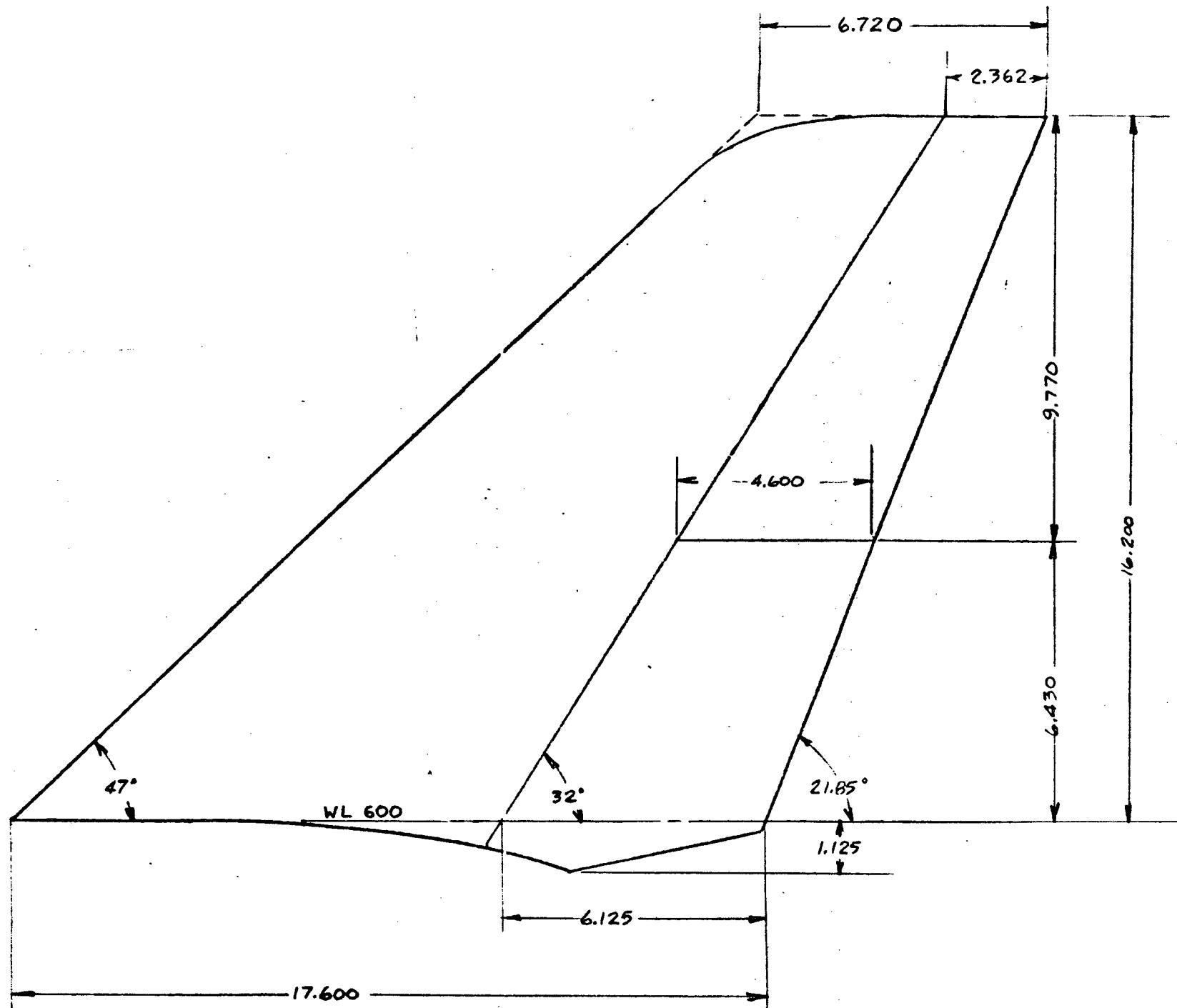


Figure 4. H-33 Orbiter Vertical Tail With Segmented Rudder (1/25 Scale)

MODEL COMPONENT DESCRIPTION SHEETS

MODEL COMPONENT: BODY - B₅

GENERAL DESCRIPTION: BASIC H-33 ORBITER BODY

DRAWING NUMBER: 518 MOD 826

DIMENSIONS:

	<u>FULL-SCALE</u> (ft. or ft. ²)	<u>MODEL SCALE</u> (in. or in. ²)
Length	<u>135</u>	<u>64.80</u>
Max. Width	<u>25</u>	<u>12.00</u>
Max. Depth	<u>27.5</u>	<u>13.20</u>
Fineness Ratio	<u>4.92</u>	<u>4.92</u>
Area		
Max. Cross-Sectional	<u>530</u>	<u>122.11</u>
Planform	<u>3120</u>	<u>718.85</u>
Wetted	<u>10,345</u>	<u>2383.49</u>
Base	<u>461</u>	<u>106.21</u>

MODEL COMPONENT: WING - W₁

GENERAL DESCRIPTION: BASIC H-33 ORBITER WING

DRAWING NUMBER:

518 MOD 827

DIMENSIONS:

TOTAL DATA

Area

Planform

4840

1115.14

Wetted

5940

1368.58

Span (equivalent)

94.5

45.36

Aspect Ratio

1.845

1.845

Rate of Taper

.178

.178

Taper Ratio

5°

5°

Dihedral Angle, degrees

20@body-30@tip

20@body-30@ tip

Incidence Angle, degrees

Aerodynamic Twist, degrees

Toe-In Angle

Cant Angle

Sweep Back Angles, degrees

Leading Edge

55°

55°

Trailing Edge

-5°

-5°

0.25 Element Line

46.32°

46.32°

Chords:

Root (Wing Sta. 0.0)

86.96

41.74

Tip, (equivalent)

15.48

7.43

MAC, ~~inches~~

59.5

28.56

Fus. Sta. of .25 MAC

W.P. of .25 MAC

Airfoil Section

Root

t/c=9.5% cambered sect.

Tip

" "

" "

EXPOSED DATA

Area

2900

668.16

Span, (equivalent)

69.5

33.36

Aspect Ratio

1.666

1.666

Taper Ratio

.228

.228

Chords

Root

67.98

32.63

Tip

15.48

7.43

MAC

47.2

22.66

Fus. Sta. of .25 MAC

W.P. of .25 MAC

MODEL COMPONENT: ELEVON (FOR W₄ WING)

GENERAL DESCRIPTION: INDIVIDUAL MOVABLE CONTROL SURFACE ASSOCIATED WITH THE W₄ WING

DRAWING NUMBER: 518 MOD 827

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u> (ft. or ft. ²)	<u>MODEL SCALE</u> (in. or in. ²)
Area	<u>410</u>	<u>94.46</u>
Span (equivalent)	<u>34.75</u>	<u>16.68</u>
Inb'd equivalent chord	<u>13.6</u>	<u>6.53</u>
Outb'd equivalent chord	<u>10.0</u>	<u>4.80</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord		
At Outb'd equiv. chord		
Sweep Back Angles, degrees		
Leading Edge	<u>0°</u>	<u>0°</u>
Tailing Edge	<u>-5°</u>	<u>-5°</u>
Hingeline	<u>0°</u>	<u>0°</u>
Area Moment (Normal to hinge line)		

MODEL COMPONENT: C VERTICAL TAIL - V5

GENERAL DESCRIPTION: BASIC H-33 ORBITER VERTICAL TAIL

DRAWING NUMBER: 518 MOD 825

DIMENSIONS:

	<u>FULL-SCALE</u> (ft. or ft. ²)	<u>MODEL SCALE</u> (in. or in. ²)
Area	<u>855</u>	<u>196.99</u>
Span (equivalent)	<u>33.75</u>	<u>16.20</u>
Inb'd equivalent chord	<u>36.66</u>	<u>17.60</u>
Outb'd equivalent chord	<u>14.0</u>	<u>6.72</u>
Ratio Elevator chord/horizontal tail chord		
At Inb'd equiv. chord	<u>.348</u>	<u>.348</u>
At Outb'd equiv. chord	<u>.351</u>	<u>.351</u>
Sweep Back Angles, degrees		
Leading Edge	<u>47</u>	<u>47°</u>
Tailing Edge	<u>21.85°</u>	<u>21.85°</u>
Hingeline	<u>32</u>	<u>32°</u>
Area Moment (Normal to hinge line)		
ASPECT RATIO	<u>1.33</u>	<u>1.33</u>
TAPER RATIO	<u>.38</u>	<u>.38</u>
MAC	<u>27</u>	<u>12.96</u>
AIRFOIL SECTION	<u>NACA 64AO10</u>	<u>NACA 64AO10</u>

MODEL COMPONENT: RUDDER (FOR V₅ VERTICAL TAIL)

GENERAL DESCRIPTION: MOVABLE CONTROL SURFACE ASSOCIATED WITH THE V₅ VERTICAL TAIL

DRAWING NUMBER: 518 MOD 825

DIMENSIONS:

	<u>FULL-SCALE</u> (ft. or ft. ²)	<u>MODEL SCALE</u> (in. or in. ²)
Area	<u>292</u>	<u>67.28</u>
Span (equivalent)	<u>34.75</u>	<u>16.68</u>
Inb'd equivalent chord	<u>12.76</u>	<u>6.12</u>
Outb'd equivalent chord	<u>4.92</u>	<u>2.36</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord		
At Outb'd equiv. chord		
Sweep Back Angles, degrees		
Leading Edge	<u>32</u>	<u>32°</u>
Tailing Edge	<u>21.85°</u>	<u>21.85°</u>
Hingeline	<u>32</u>	<u>32°</u>
Area Moment (Normal to hinge line)		

NOMENCLATURE

(General)

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
α	ALPHA	angle of attack, angle between the projection of the wind X_w -axis on the body X, Z-plane and the body X-axis; degrees
β	BETA	sideslip angle, angle between the wind X_w -axis and the projection of this axis on the body X-Z-plane; degrees
ψ	PSI	yaw angle, angle of rotation about the body Z-axis, positive when the positive X-axis is rotated toward the positive Y-axis; degrees
ϕ	PHI	roll angle, angle of rotation about the body X-axis, positive when the positive Y-axis is rotated toward the positive Z-axis; degrees
ρ		air density; kg/m^3 , slugs/ ft^3
a		speed of sound; m/sec , ft/sec
V		speed of vehicle relative to surrounding atmosphere; m/sec , ft/sec
q	$Q(\text{PSI})$ $Q(\text{PSF})$	dynamic pressure; $1/2\rho V^2$, psi, psf
M	MACH	Mach number; V/a
RN/L	RN/L	Reynolds number per unit length; million/ ft
p		static pressure; psi
P		total pressure; psi
C_p	CP	pressure coefficient; $(p-p_\infty)/q$

NOMENCLATURE (Continued)

Reference & C. G. Definitions

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
S		wing area; m^2 , ft^2
S	SREF	reference area; m^2 , ft^2
\bar{c}		wing mean aerodynamic chord or reference chord; m, ft, in (see ℓ_{ref} or LREF)
ℓ_{ref}	LREF	reference length; m, ft, in.; (see \bar{c})
b_{ref}	BREF	wing span or reference span; m, ft, in
A_b		base area; m^2 , ft^2 , in^2
c. g.		center of gravity
MRP	MRP	abbreviation for moment reference point
	XMRP	abbreviation for moment reference point on X-axis
	YMRP	abbreviation for moment reference point on Y-axis
	ZMRP	abbreviation for moment reference point on Z-axis

NOMENCLATURE (Continued)

Axis System General

<u>SYMBOL</u>	<u>DEFINITION</u>
F	force; F, lbs
M	moment; M, in-lb

<u>Subscript</u>	<u>Definition</u>
N	normal force
A	axial force
L	lift force
D	drag force
Y	force or moment about the Y axis
Z	moment about the Z axis
X	moment about the X axis
s	stability axis system
w	wind axis system
ref	reference conditions
∞	free stream conditions
t	total conditions
b	base

NOMENCLATURE (Continued)

Body Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
C_N	CN	normal force coefficient; F_N/qS
C_A	CA	axial force coefficient; F_A/qS
C_{A_b}	CAB	base axial force coefficient; $[-1] \left[(p_b - p_\infty)/q \right] (A_b/S)$
C_{A_f}	CAF	forebody axial force coefficient; $C_A - C_{A_b}$
C_m	CLM	pitching moment coefficient; $M_Y/qS l_{ref}$
C_y	CY	side force coefficient; F_Y/qS
C_n	CYN	yawing moment coefficient; $M_Z/qS b_{ref}$
C_l	CBL	rolling moment coefficient, $M_X/qS b_{ref}$

NOMENCLATURE (Continued)

Stability Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
C_L	CL	lift force coefficient; F_L/qS
C_D	CD	drag force coefficient; F_D/qS
C_{D_b}	CDB	base drag coefficient
C_{D_f}	CDF	forebody drag coefficient; $C_D - C_{D_b}$
C_m	CLM	pitching moment coefficient; $M_{Y,s}/qS l_{ref}$
C_y	CY	side force coefficient; F_y/qS
C_n	CLN	yawing moment coefficient; $M_{Z,s}/qS b_{ref}$
C_ℓ	CSL	rolling moment coefficient; $M_{X,s}/qS b_{ref}$
L/D	L/D	lift-to-drag ratio; C_L/C_D
L/D_f	L/DF	lift to forebody drag ratio; C_L/C_{D_f}

NOMENCLATURE (Continued)

Surface Definitions

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
i_t	HORIZT	horizontal tail incidence; positive when trailing edge down; degrees
δ		symmetrical surface deflection angle; degrees; positive deflections are:
	AILRDN	aileron - total aileron deflection; (left aileron - right aileron)/2
	CANARD	canard - trailing edge down
	ELEVON	elevon - trailing edge down
	ELEVTR	elevator - trailing edge down
	FLAP	flap - trailing edge down
	RUDDER	rudder - trailing edge to the left
	SPOILR	spoiler - trailing edge down
	TAB	tab - trailing edge down with respect to control surface
δ		antisymmetrical surface deflection angle, degrees; positive trailing edge down:
	AIL-L	left aileron - trailing edge down
	AIL-R	right aileron - trailing edge down
	ELVN-L	left elevon - trailing edge down
	ELVN-R	right elevon - trailing edge down
	SPLR-L	left spoiler - trailing edge down
	SPLR-R	right spoiler - trailing edge down

<u>SURFACE SUBSCRIPTS</u>	<u>DEFINITION</u>
a	aileron
b	base
c	canard
e	elevator or elevon
f	flap
r	rudder or ruddervator
s	spoiler
t	tail

NON-STANDARD NOMENCLATURE

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
δ_{EL}	LELEVN	Left wing elevon deflection angle, positive is trailing edge down, in degrees.
δ_{ER}	RELEVN	Right wing elevon deflection angle, positive is trailing edge down, in degrees.
δ_{RT}	TRUDDR	Top portion of segmented rudder, positive is trailing edge left, in degrees.
δ_{RLL}	LLRUDD	Left (pilot's view) part of split lower rudder segment in the V-shaped, speed brake configuration. Positive is trailing edge left, in degrees.
δ_{RLR}	LRRUDD	Right (pilot's view) part of split lower rudder segment in the V-shaped speed brake configuration. Positive is trailing edge left, in degrees.

$w_4^{x,y}$

Superscripts x and y indicate the deflections of the left and right elevons, respectively, where positive deflections are T.E. down.

v_5

The rudder is segmented into upper and lower panels. The lower panel splits open along its chord line to become a V-shaped speed brake.

Superscripts: 1) a single superscript indicates the deflection of the entire full span rudder. 2) three superscripts, x, y, z, denotes the segmented rudder with speed brakes flared. x denotes upper segment deflection, y the left half of the lower segment, and z the right half of the lower segment.

For any of these surfaces, positive deflection is T.E. left.

TABULATED DATA LISTING

A tabulated data listing, consisting of all aero data sets, both original and those created in arriving at the plotted material to be presented subsequently, is available as an addendum to this report. The tabular listing is made up in two sections:

- (a) a brief summary list of all data sets containing the identifier, the descriptor, and the resident dependent variables.
- (b) a full list of all data sets containing all resident or selected aerodynamic coefficients of the data sets as well as the above mentioned information.

The listing is currently sent on limited distribution to the following organizations:

NASA AMES	Mr. V. Stevens
GAC	Mr. M. Quan

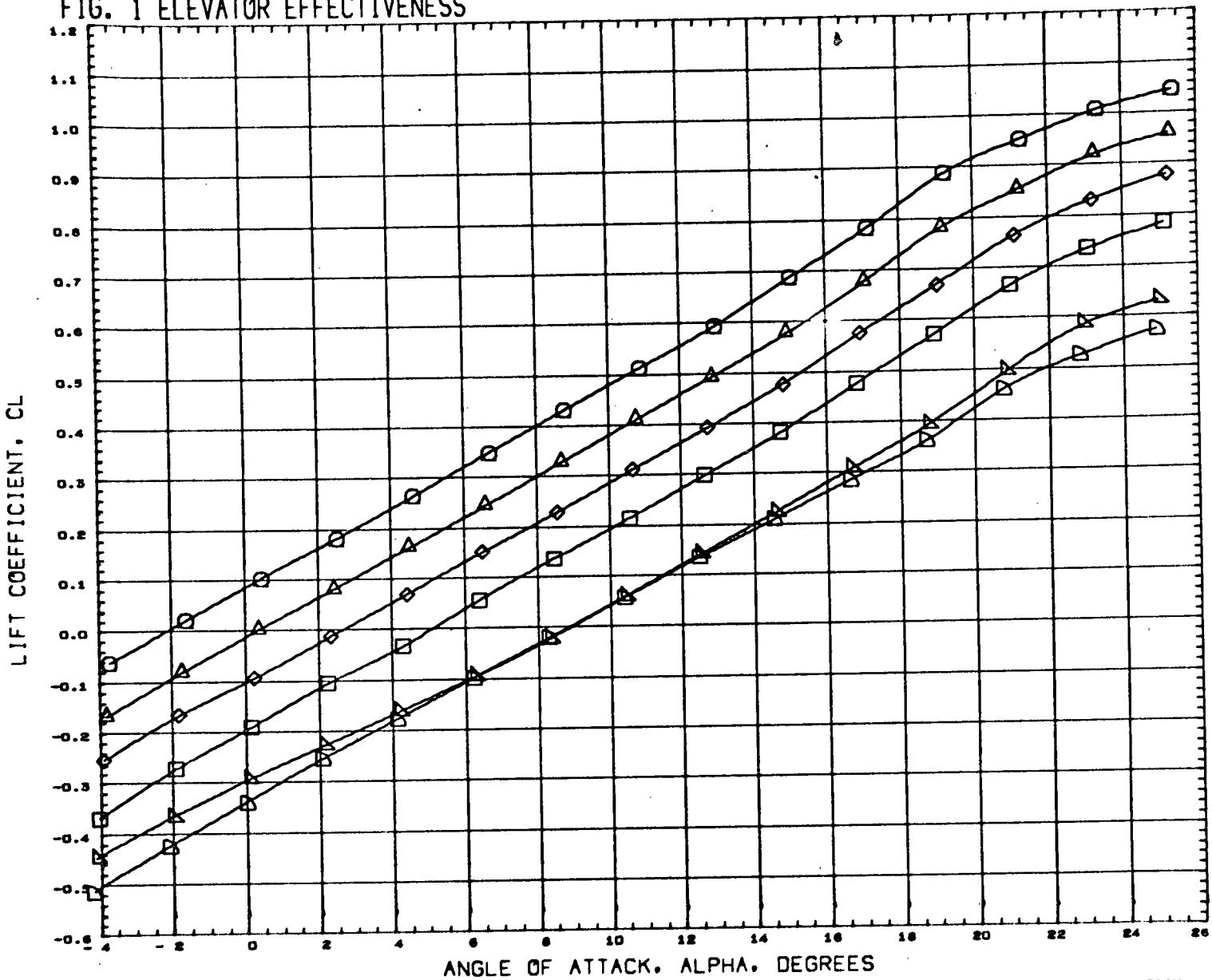
If copies of this listing are desired, please contact the above or the cognizant SADSAC personnel who, for this data, is:

Mr. J. R. Ziler
Department 2780
Chrysler Corporation Space Division
New Orleans, La. 70129

(504) 255-2304

P L O T T E D D A T A

FIG. 1 ELEVATOR EFFECTIVENESS



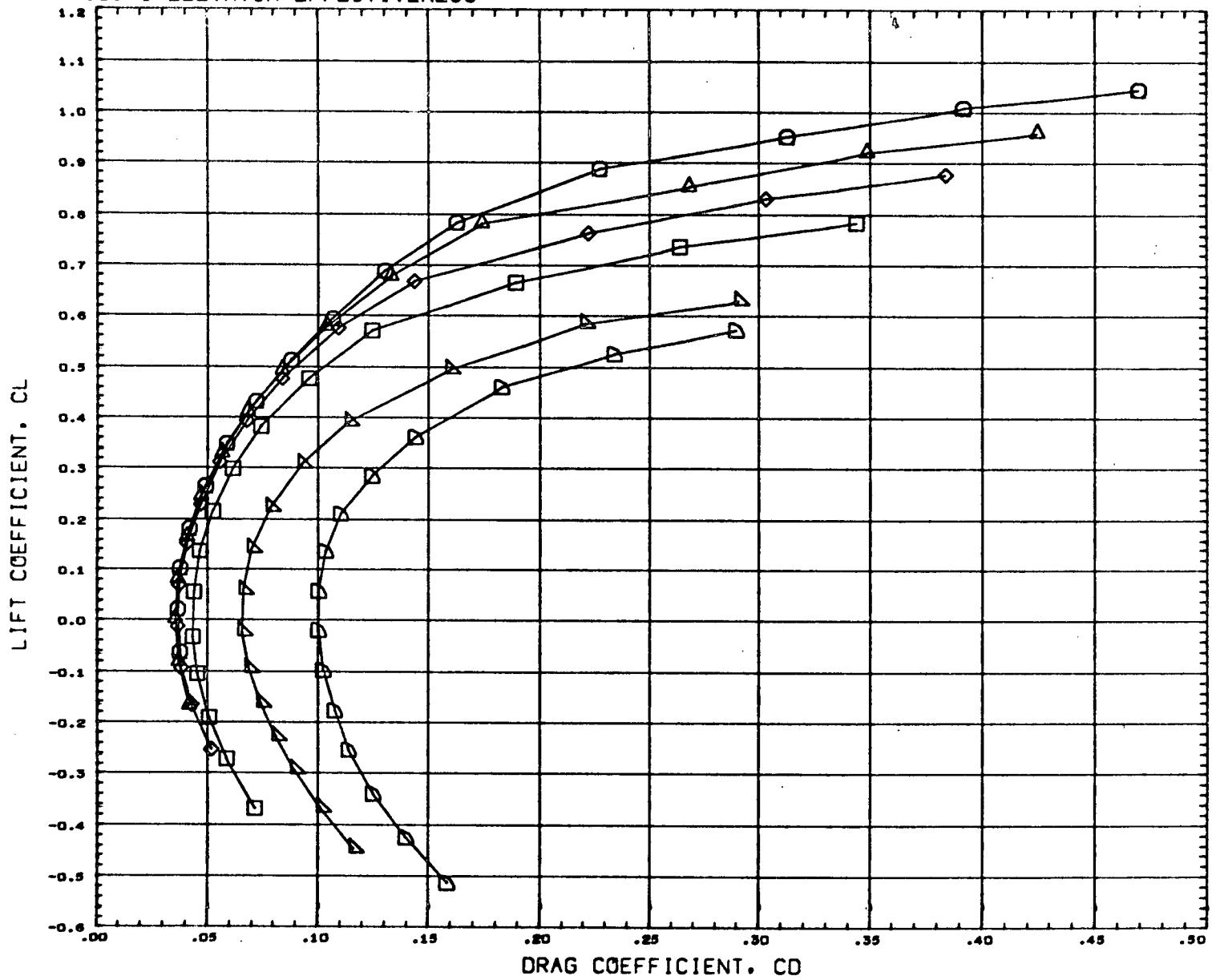
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RD1002)	GWTT 292 CCNF.H-33 CRBITER B5W4V5
(RD1006)	GWTT 292 CCNF.H-33 CRBITER B5W4(-5,-5)V5
(RD1007)	GWTT 292 CCNF.H-33 CRBITER B5W4(-10,-10)V5
(RD1008)	GWTT 292 CCNF.H-33 CRBITER B5W4(-15,-15)V5
(RD1054)	GWTT 292 CCNF.H-33 CRBITER B5W4(-25,-25)V5
(RD1009)	GWTT 292 CCNF.H-33 CRBITER B5W4(-35,-35)V5

MACH 0.170

LELEVN	RELEVN	BETA
0.000	0.000	0.000
-5.000	-5.000	0.000
-10.000	-10.000	0.000
-15.000	-15.000	0.000
-25.000	-25.000	0.000
-35.000	-35.000	0.000

REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7800 FT.
XMRP 1285.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

FIG. 1 ELEVATOR EFFECTIVENESS



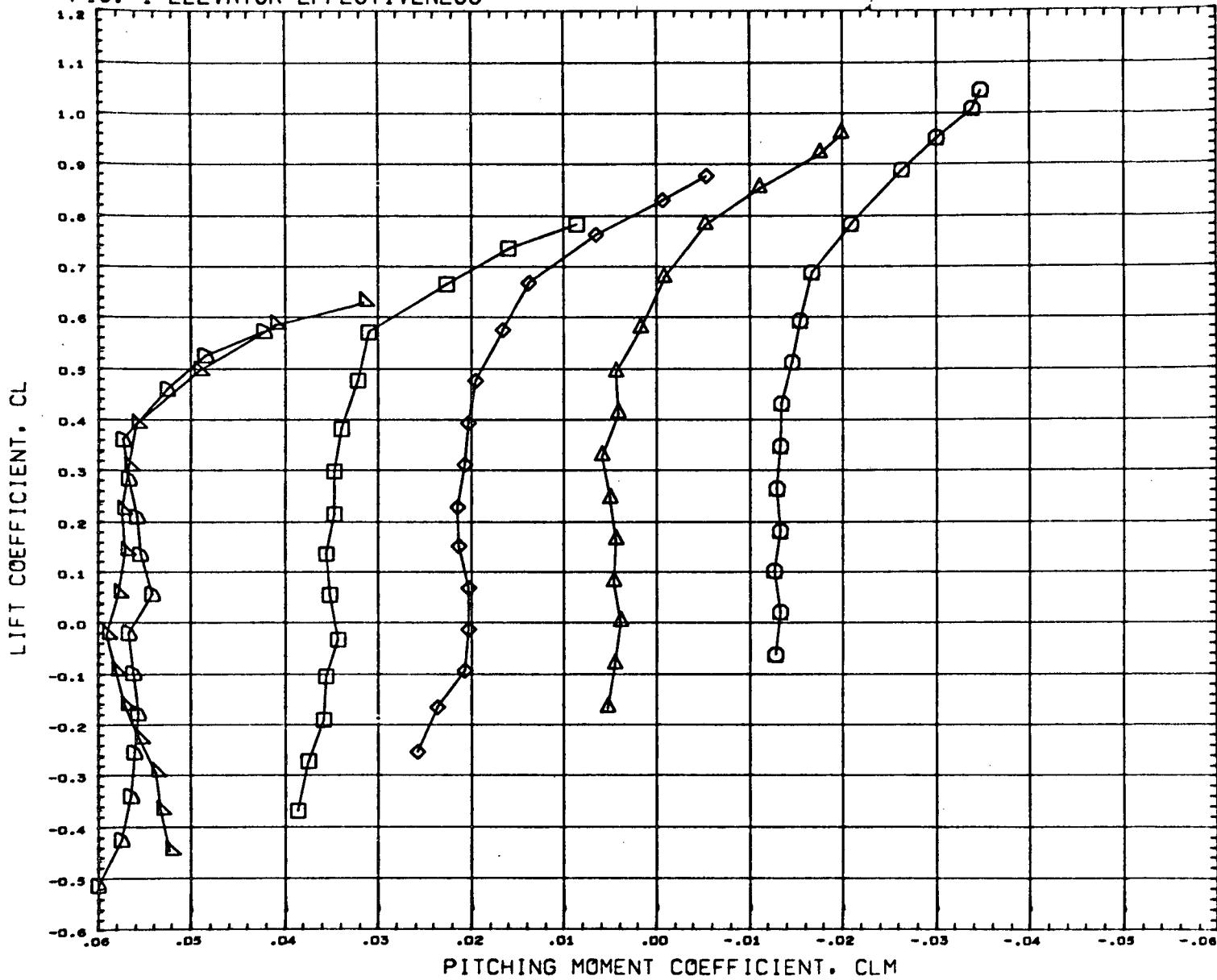
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RD1002)	GWTT 292 CONF.H-33 ORBITER B5W4V5
(RD1006)	GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-5)V5
(RD1007)	GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5
(RD1008)	GWTT 292 CONF.H-33 ORBITER B5W4 (-15,-15)V5
(RD1034)	GWTT 292 CONF.H-33 ORBITER B5W4 (-25,-25)V5
(RD1009)	GWTT 292 CONF.H-33 ORBITER B5W4 (-35,-35)V5

MACH 0.170

LELEVN	RELEVN	BETA
0.000	0.000	0.000
-5.000	-5.000	0.000
-10.000	-10.000	0.000
-15.000	-15.000	0.000
-25.000	-25.000	0.000
-35.000	-35.000	0.000

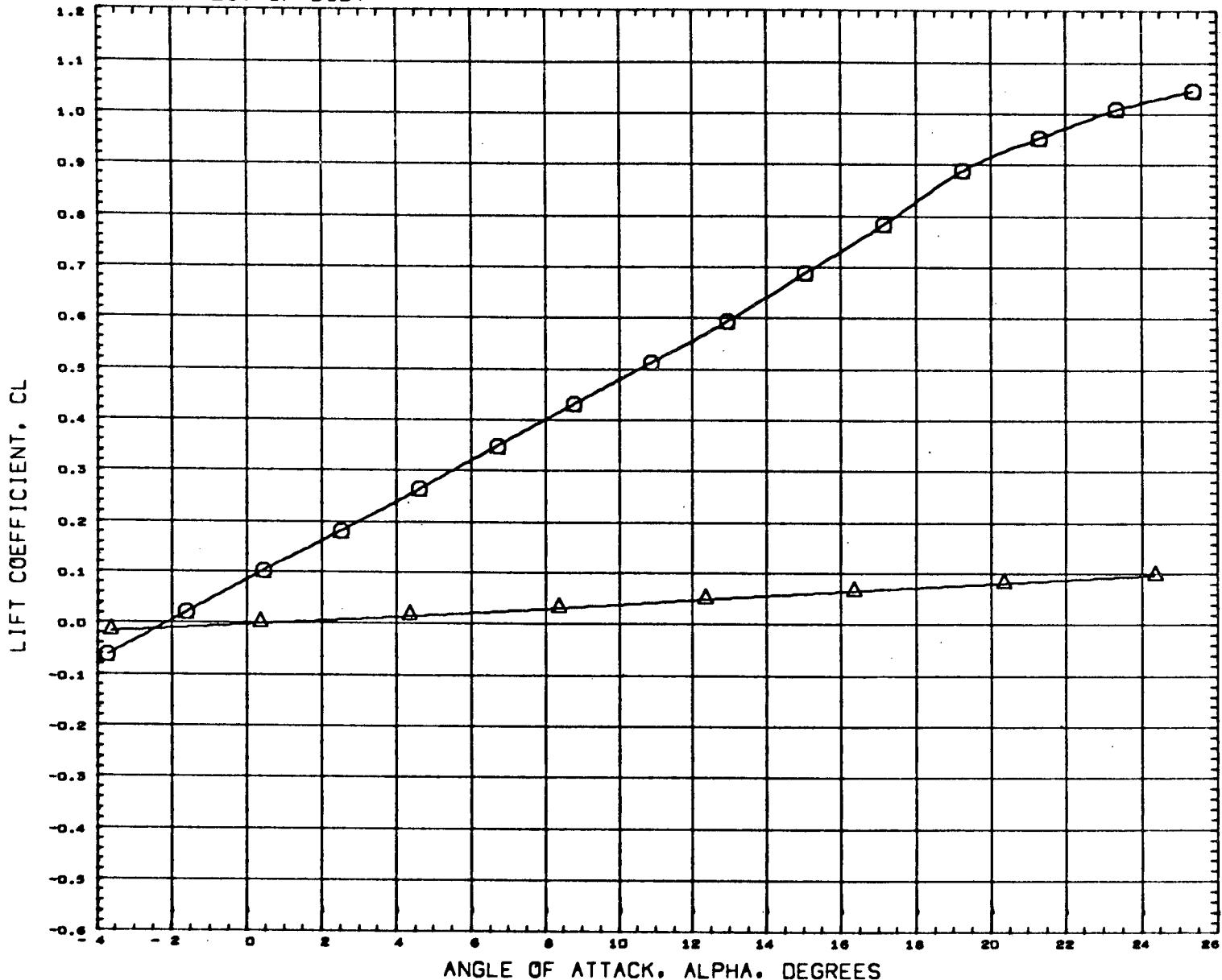
REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7600 FT.
XMRP 1285.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

FIG. 1 ELEVATOR EFFECTIVENESS



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	LELEVN	RELEVN	BETA	REFERENCE INFORMATION
(RD1002)	GWTT 292 CONF.H-33 ORBITER B5W4V5	0.000	0.000	0.000	SREF 7.7440 SQ FT
(RD1006)	GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-5)V5	-5.000	-5.000	0.000	LREF 5.4000 FT.
(RD1007)	GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5	-10.000	-10.000	0.000	BREF 3.7800 FT.
(RD1008)	GWTT 292 CONF.H-33 ORBITER B5W4 (-15,-15)V5	-15.000	-15.000	0.000	XMRP 1285.0040 IN.
(RD1054)	GWTT 292 CONF.H-33 ORBITER B5W4 (-25,-25)V5	-25.000	-25.000	0.000	YMRP 0.0000 IN.
(RD1009)	GWTT 292 CONF.H-33 ORBITER B5W4 (-35,-35)V5	-35.000	-35.000	0.000	ZMRP 403.0004 IN.
	MACH 0.170				SCALE 0.0400

FIG. 2 EFFECT OF BODY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT 292 CONF.M-33 ORBITER B5W4V5
 (RD1023) GWT 292 CONF.M-33 ORBITER B5

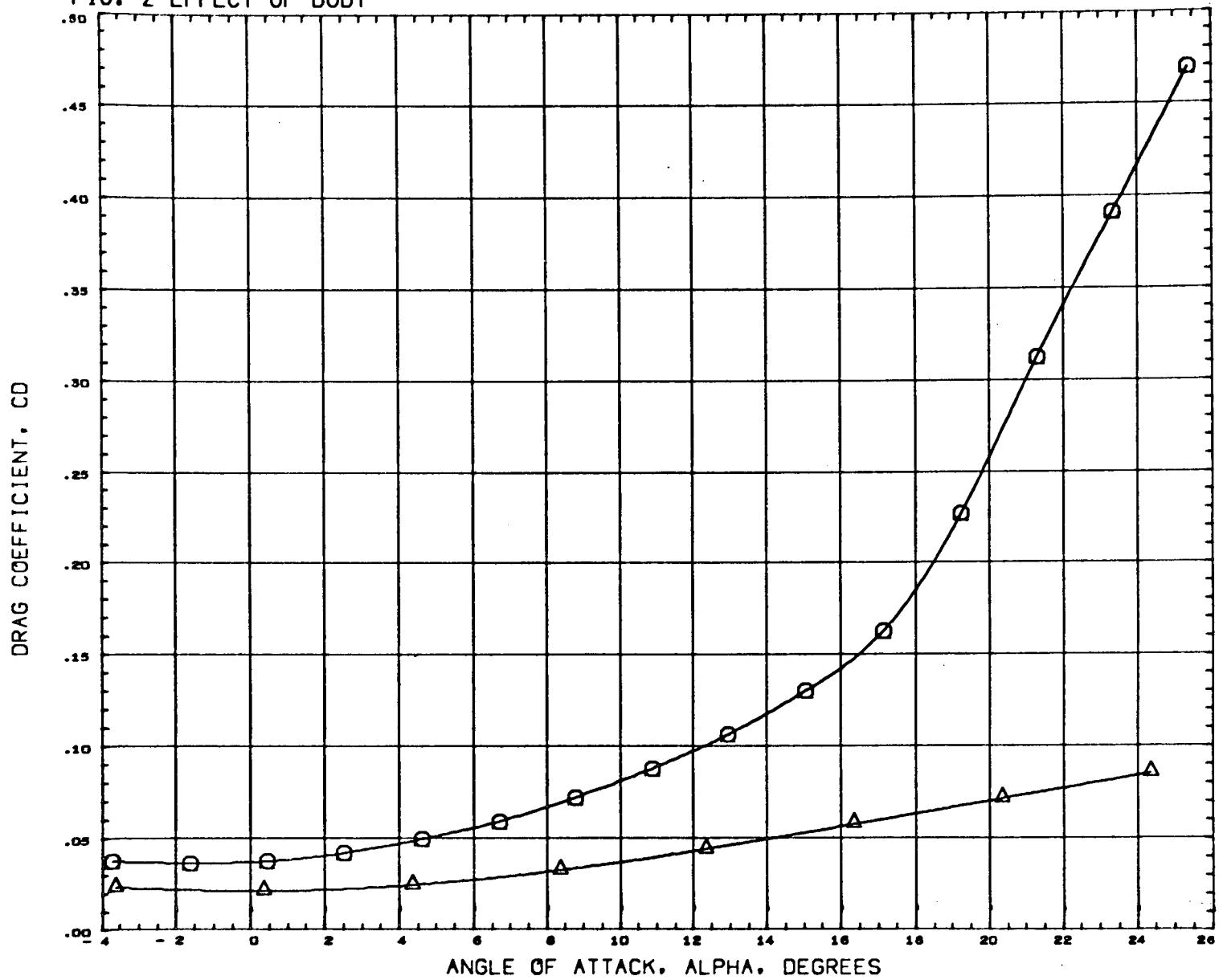
LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 4

FIG. 2 EFFECT OF BODY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) O GWT 292 CONF.H-33 ORBITER B5W4V5
 (RD1023) A GWT 292 CONF.H-33 ORBITER B5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000

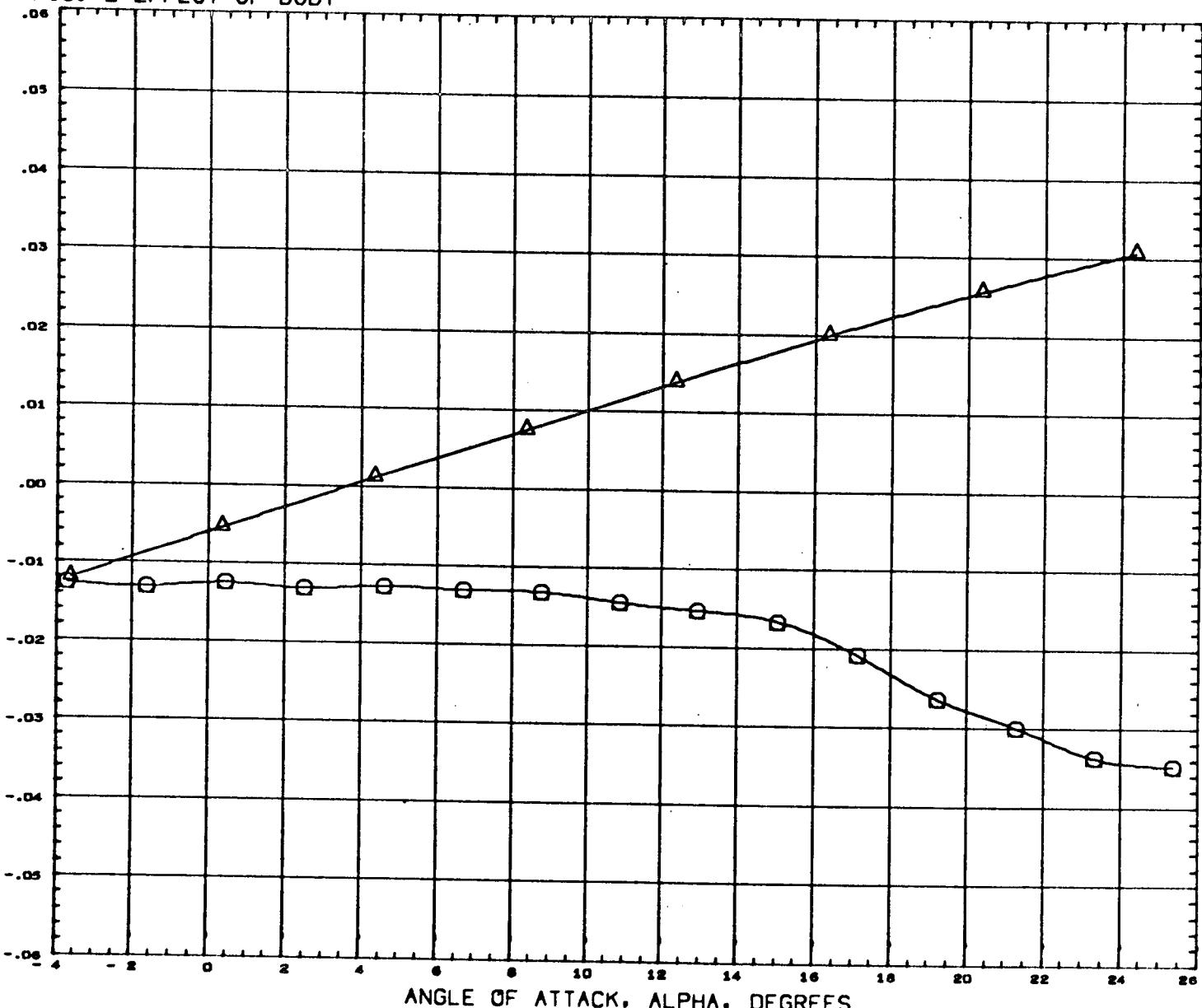
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMNP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMNP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 5

FIG. 2 EFFECT OF BODY

PITCHING MOMENT COEFFICIENT, CLM



ANGLE OF ATTACK, ALPHA, DEGREES

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWTT 292 CONF.H-33 ORBITER B5W4VS
 (RD1023) GWTT 292 CONF.H-33 ORBITER B5

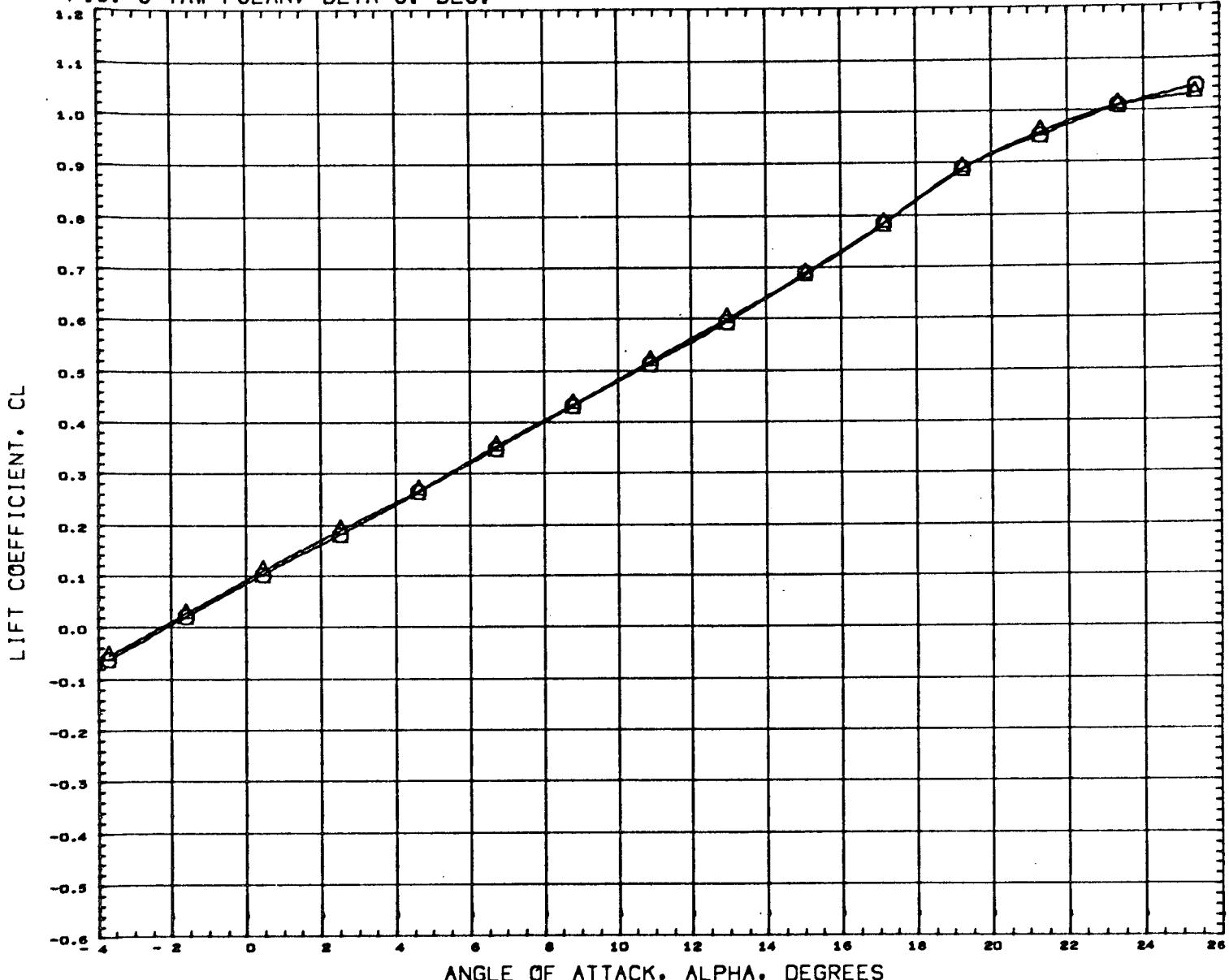
LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 6

FIG. 3 YAW POLAR, BETA=3. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT 292 CCNF.H-33 ORBITER B5W4V5
 (RD1004) GWT 292 CCNF.H-33 ORBITER B5W4V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

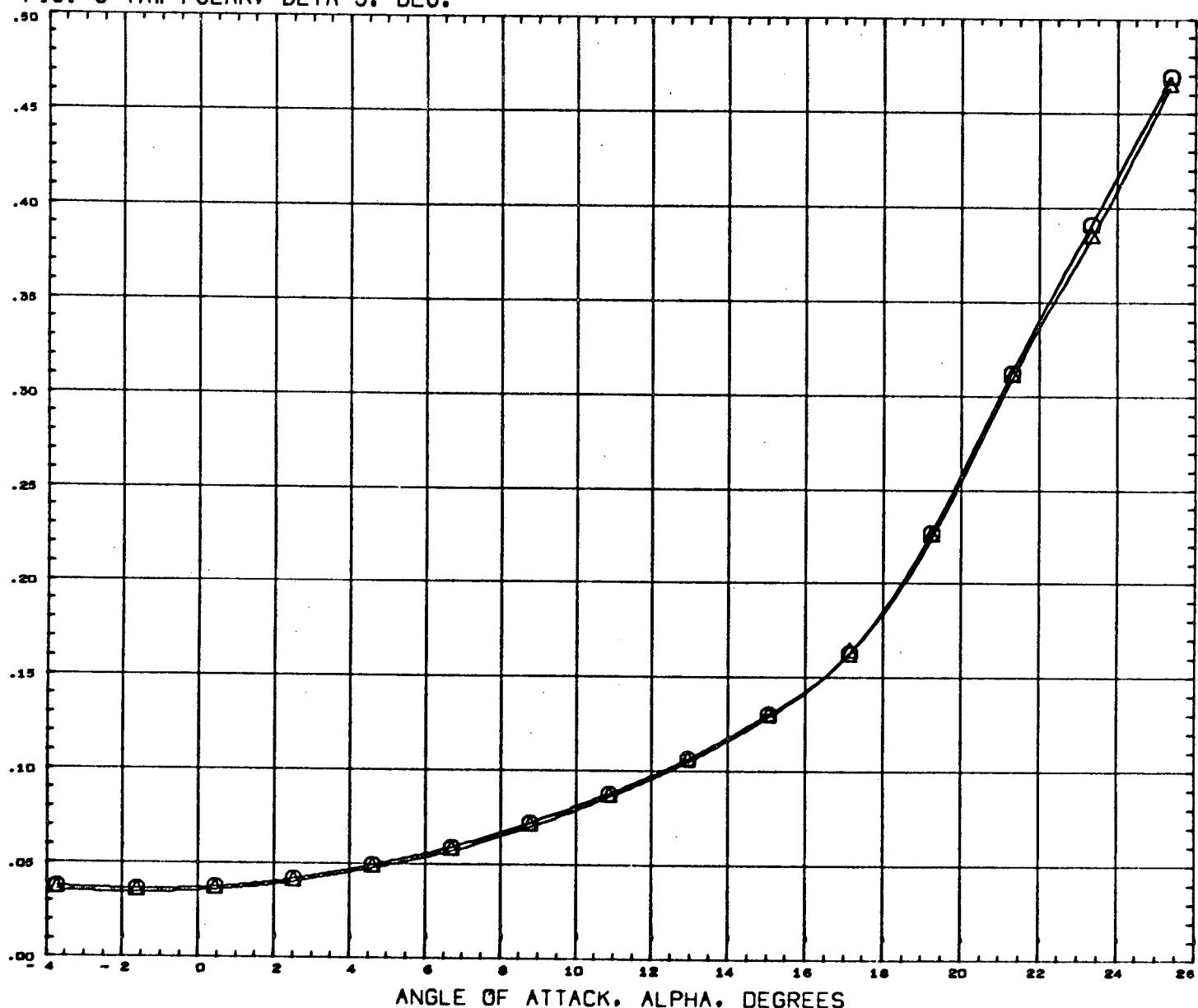
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 7

FIG. 3 YAW POLAR, BETA=3. DEG.

DRAG COEFFICIENT, CD



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT 292 CONF.H-33 ORBITER B5W4V5
 (RD1004) GWT 292 CONF.H-33 ORBITER B5W4V5

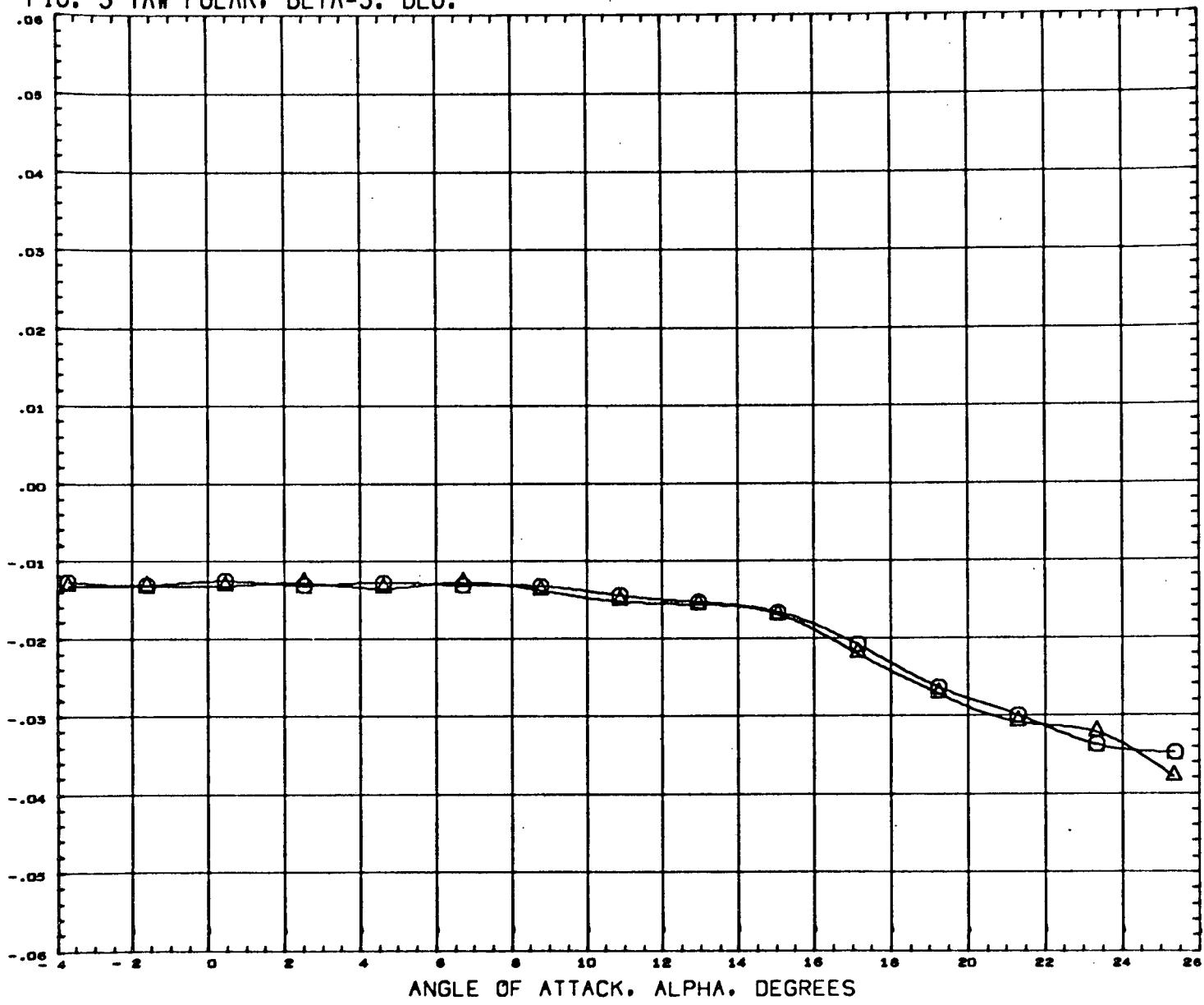
MACH 0.170

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

FIG. 3 YAW POLAR, BETA=3. DEG.

PITCHING MOMENT COEFFICIENT. CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT 292 CONF.M-33 CRBITER B5W4V5
 (RD1004) GWT 292 CONF.M-33 CRBITER B5W4V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

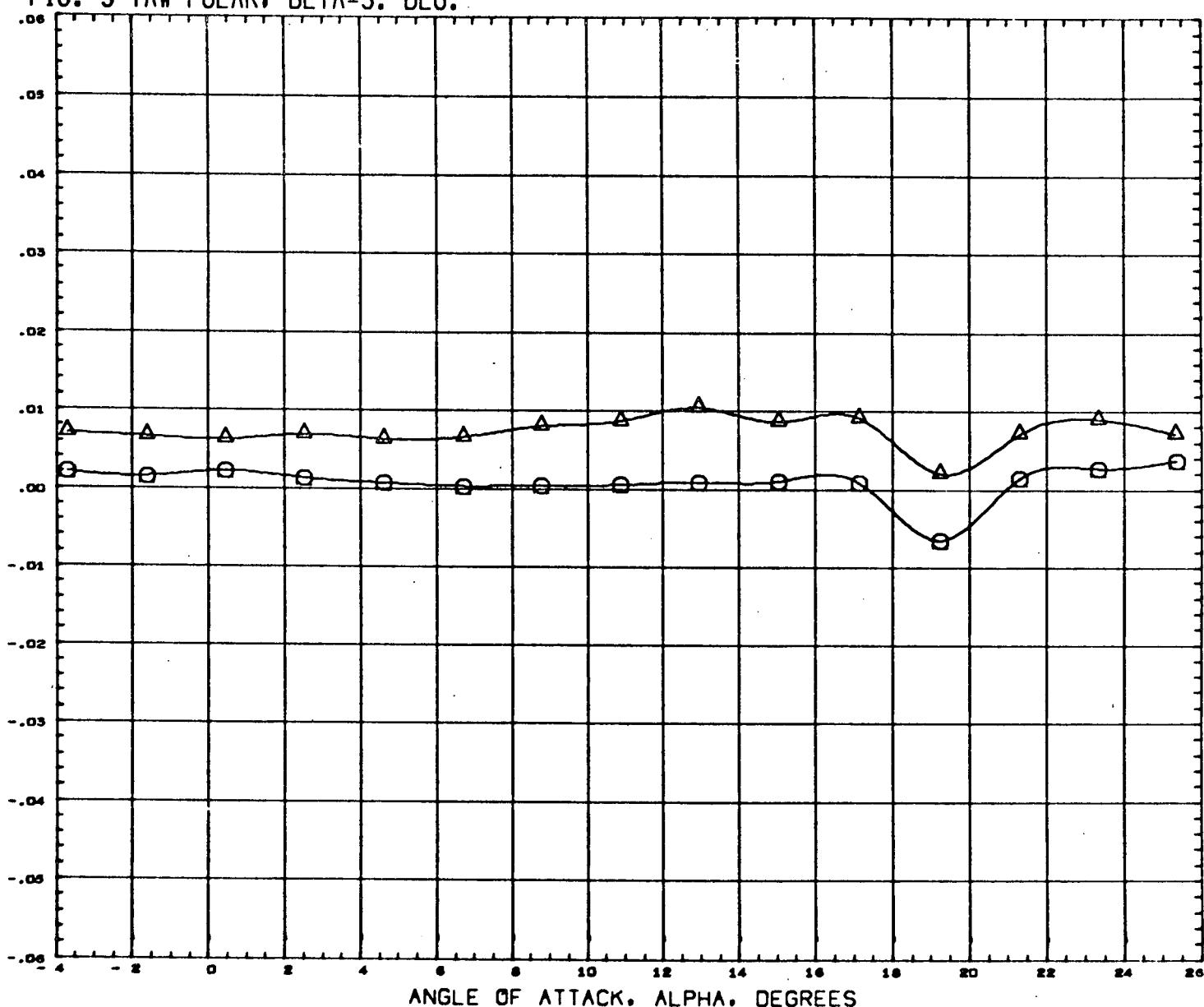
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 9

FIG. 3 YAW POLAR, BETA=3. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) Q GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1004) Q GWTT 292 CONF.H-33 ORBITER B5W4V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

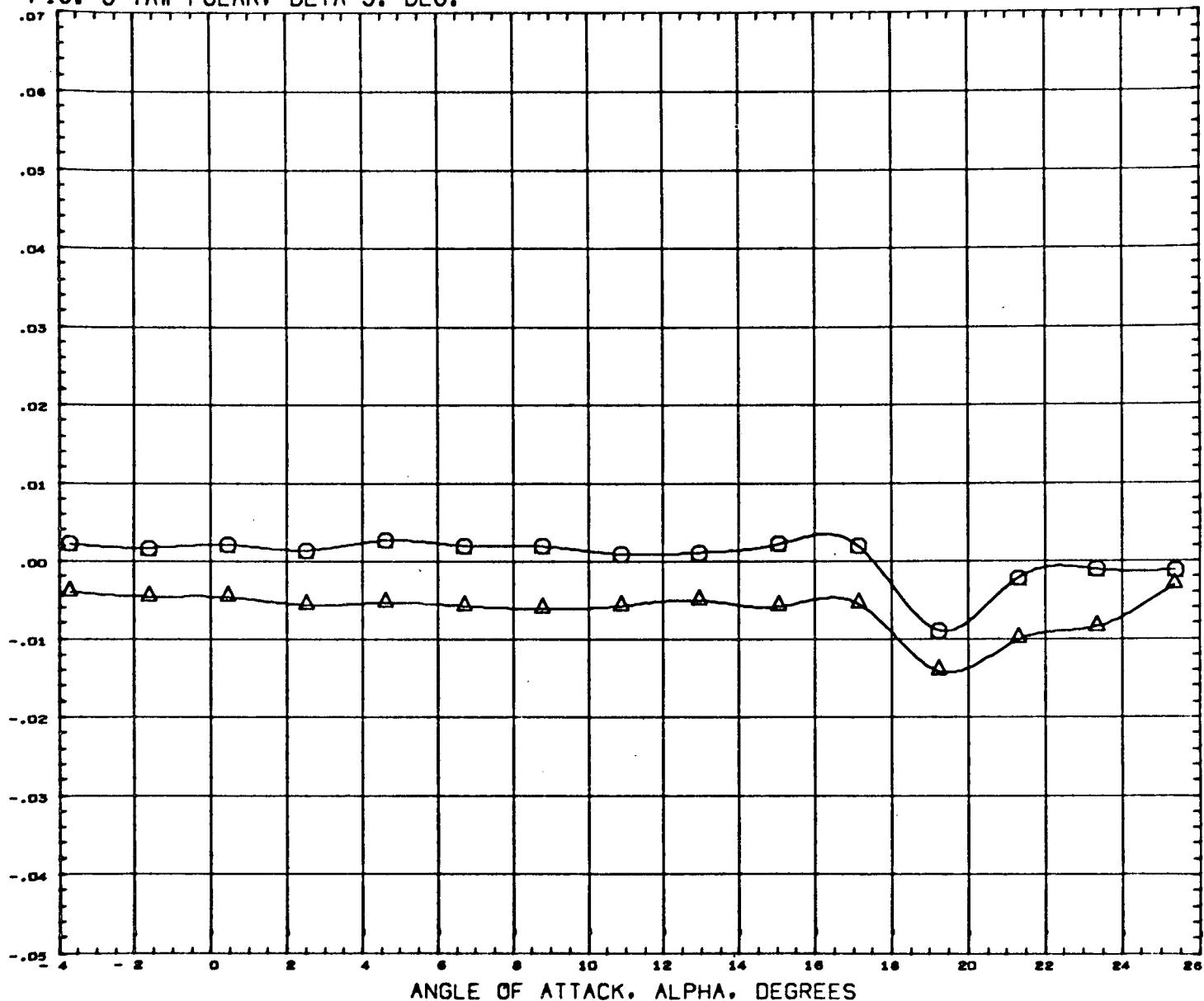
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 10

FIG. 3 YAW POLAR, BETA=3. DEG.

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT 292 CONF.H-33 ORBITER 85W4V5
 (RD1004) GWT 292 CONF.H-33 ORBITER 85W4V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

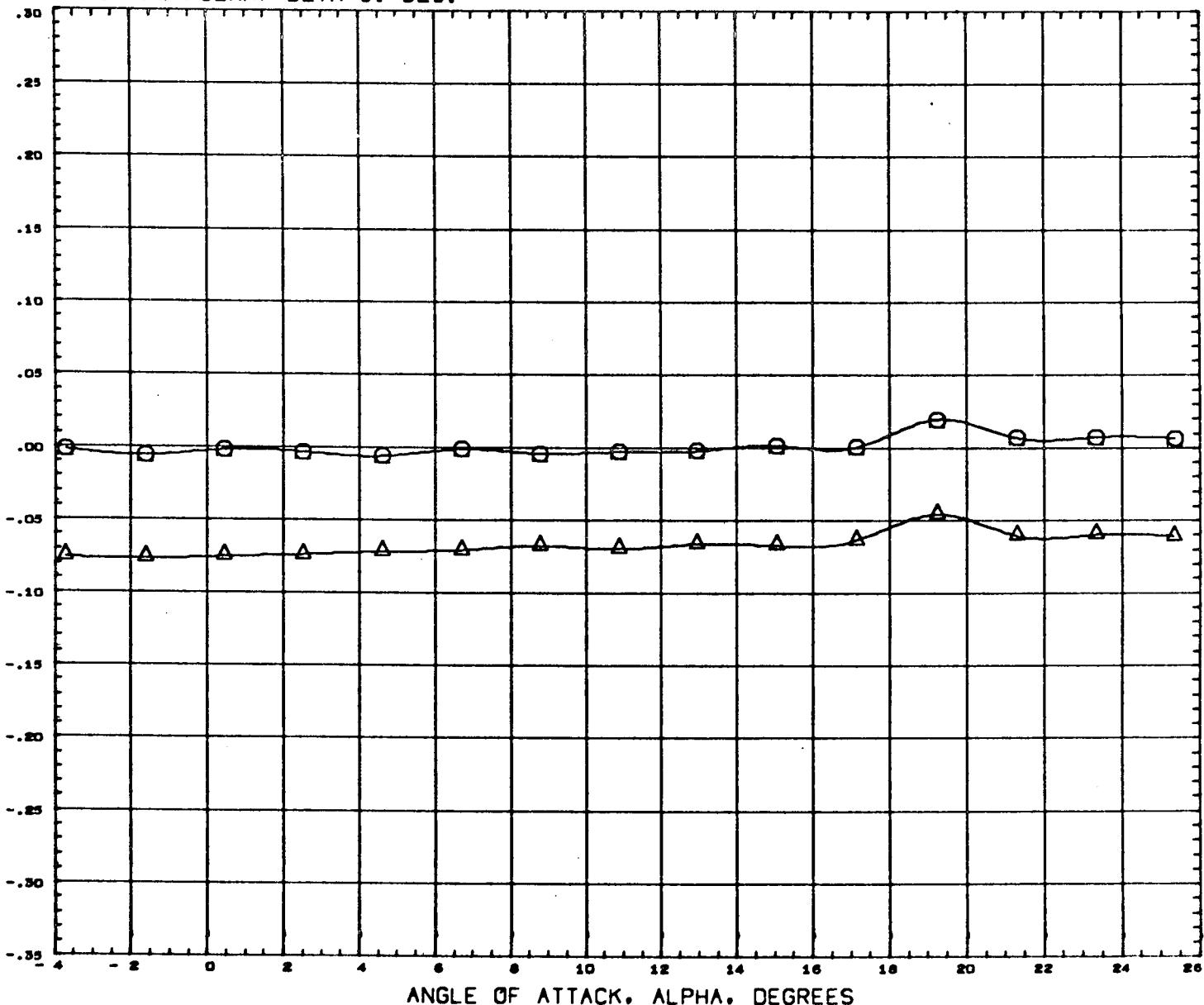
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 11

FIG. 3 YAW POLAR, BETA=3. DEG.

LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWTT 292 CONF.H-33 ORBITER BSW4V5
 (RD1004) GWTT 292 CONF.H-33 ORBITER BSW4V5

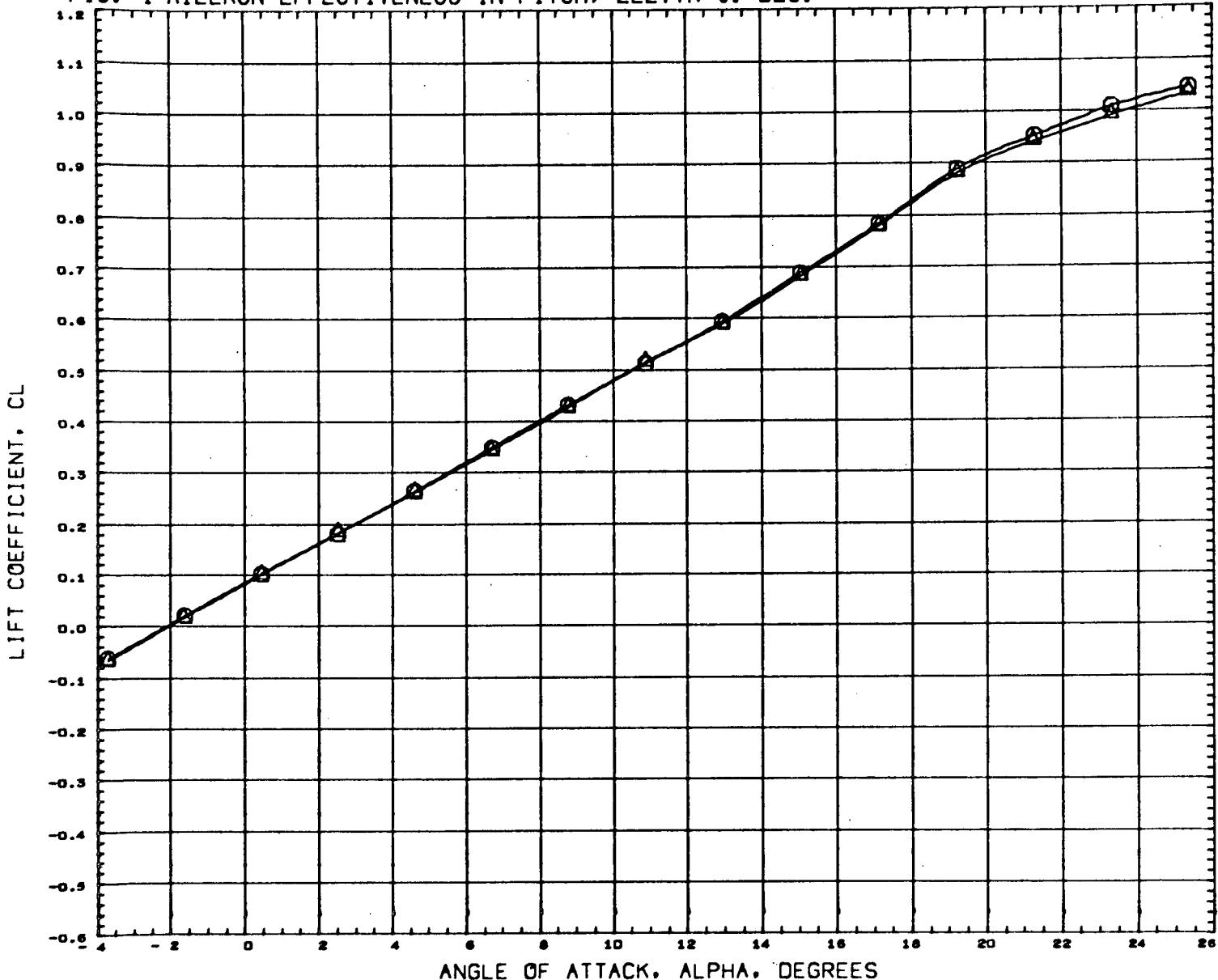
LELEVN RELEVN BETA
 0.000 0.000 0.000
 0.000 0.000 3.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 12

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) Q GWTT 292 CONF.H-33 ORBITER 85W4V5
 (RD1010) Q GWTT 292 CONF.H-33 CRBITER 85W4 (+5,-5)V5

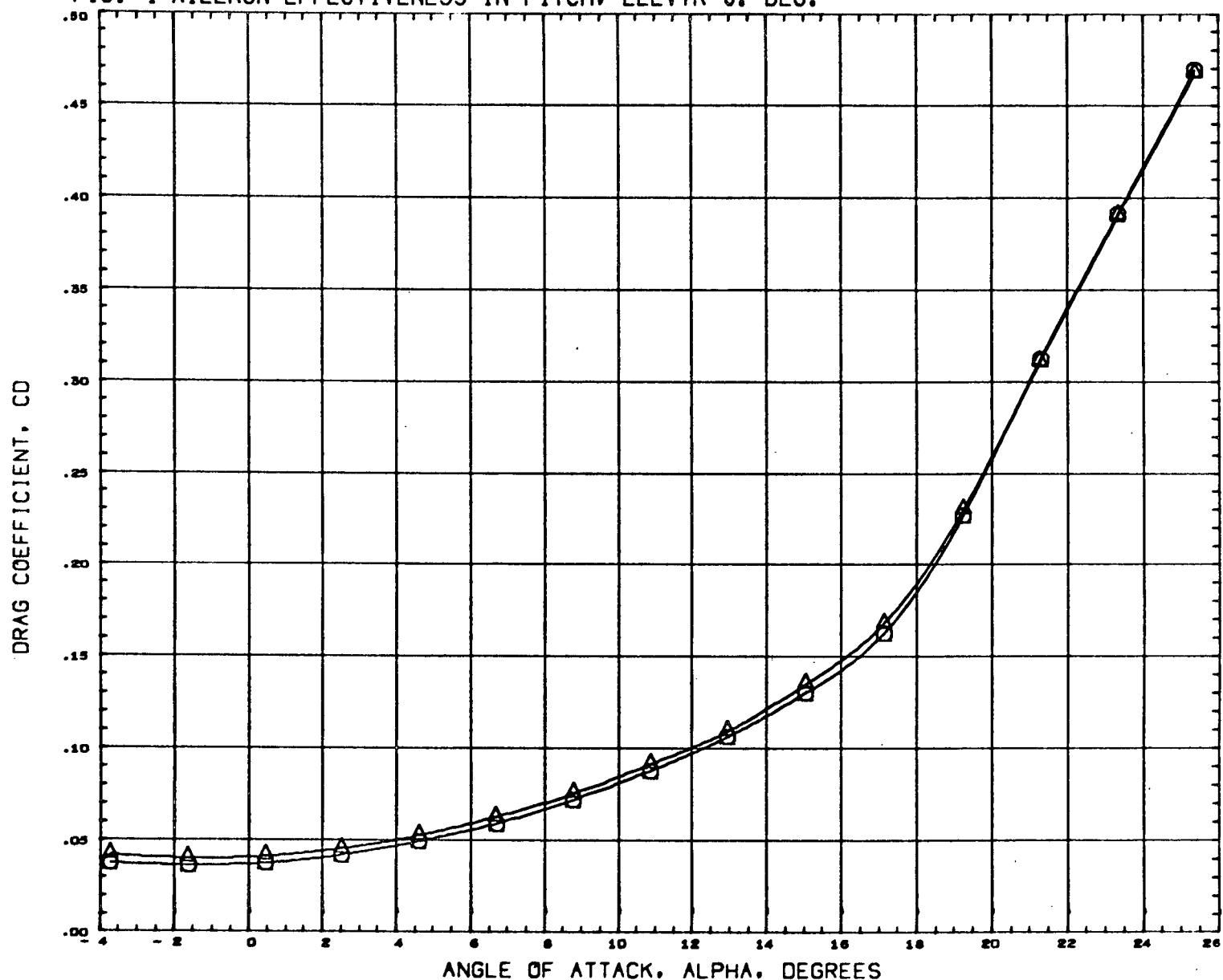
LELEVN RELEVN BETA
 0.000 0.000 0.000
 5.000 -5.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 13

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) O GWTT 292 CONF.M-33 ORBITER BSW4V5
 (RD1010) □ GWTT 292 CONF.M-33 ORBITER BSW4 (+5,-5)V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 5.000 -5.000 0.000

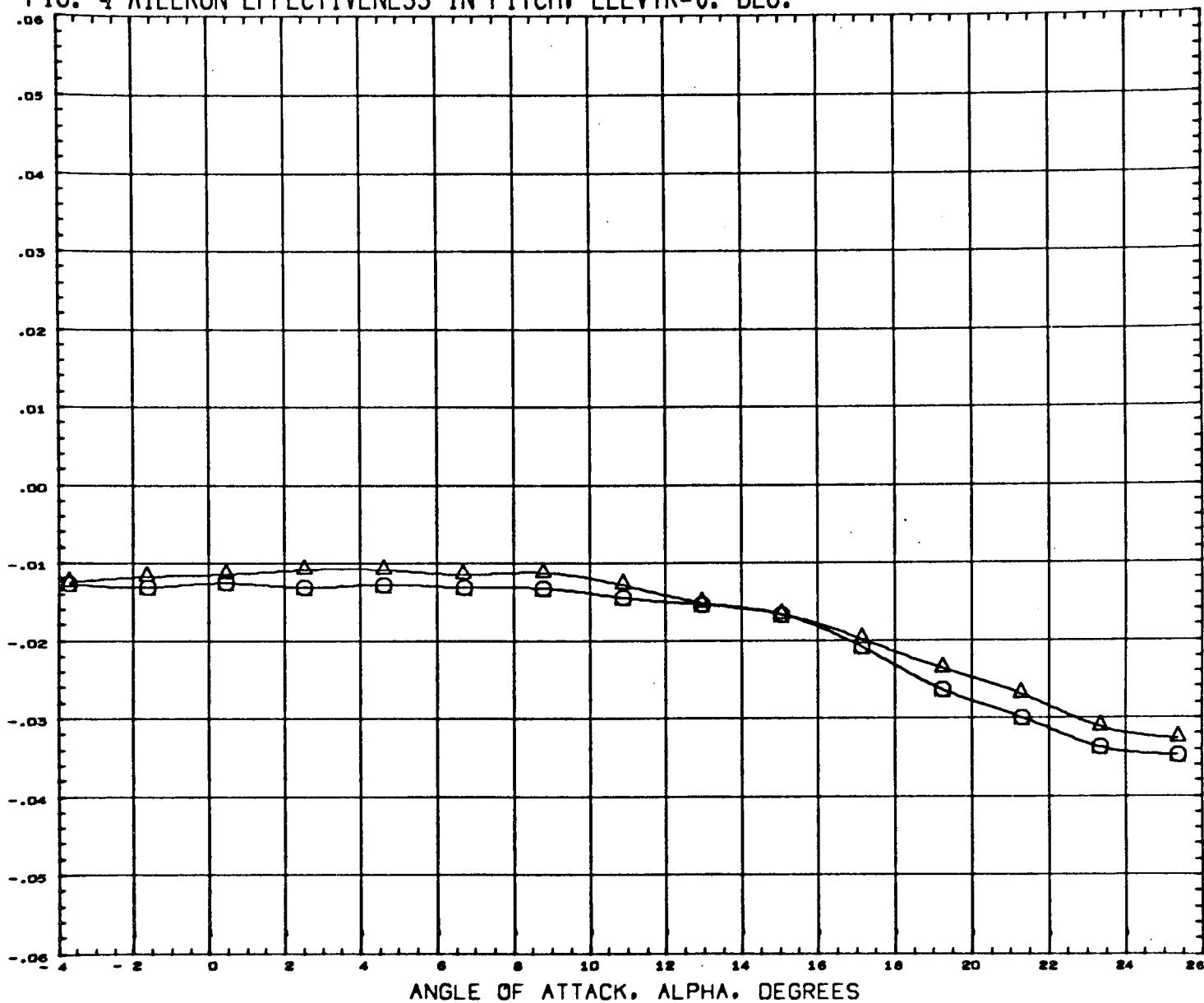
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 14

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0. DEG.

PITCHING MOMENT COEFFICIENT, CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) Q GWTT 292 CONF.M-33 CRBITER BSW4V5
 (RD1010) Q GWTT 292 CONF.M-33 CRBITER BSW4 (+5, -5)V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 5.000 -5.000 0.000

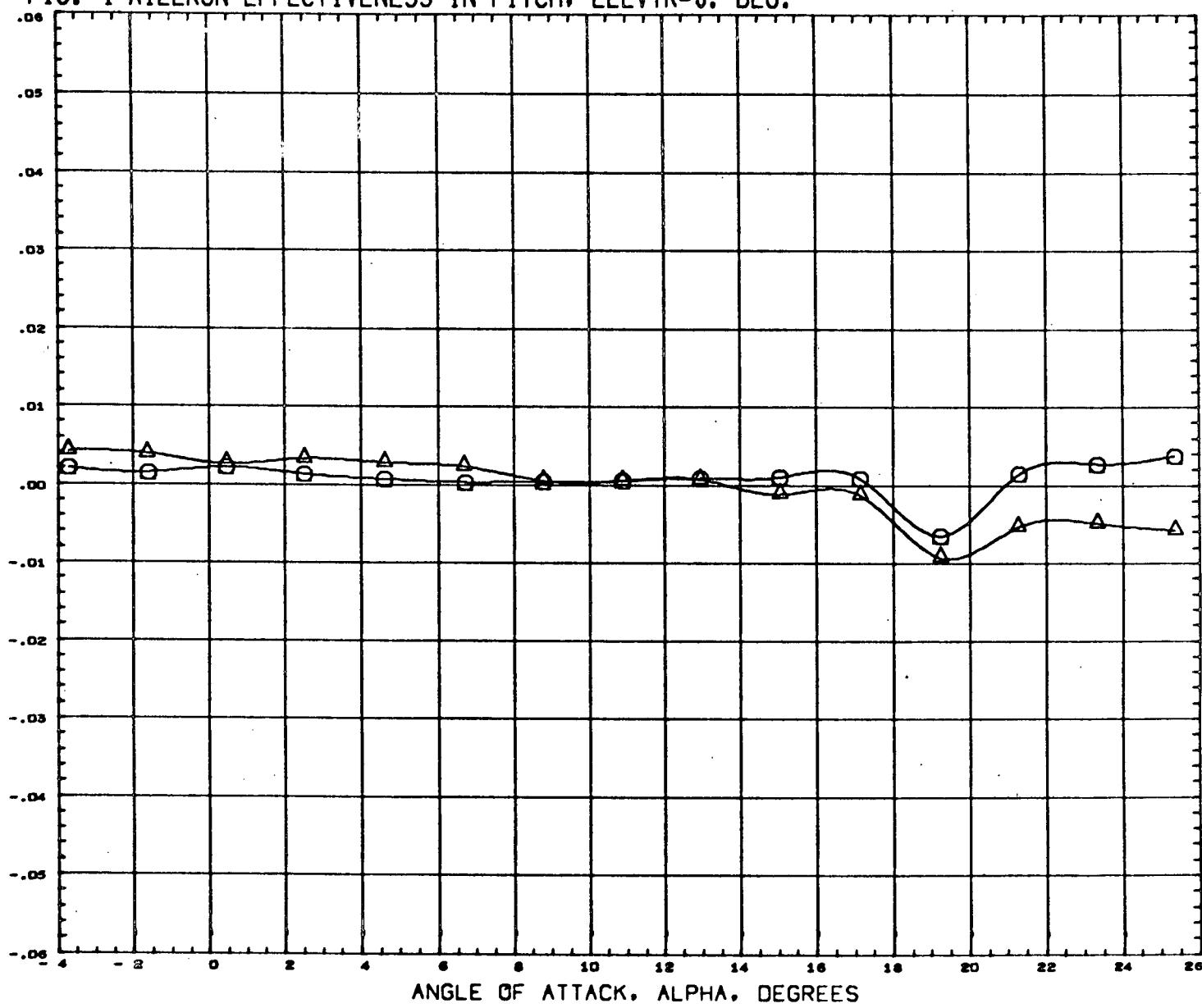
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 15

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1002) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1010) GWTT 292 CONF.H-33 ORBITER B5W4(+5,-5)V5

LELEVN RELEVN BETA
 0.000 0.000 0.000
 5.000 -5.000 0.000

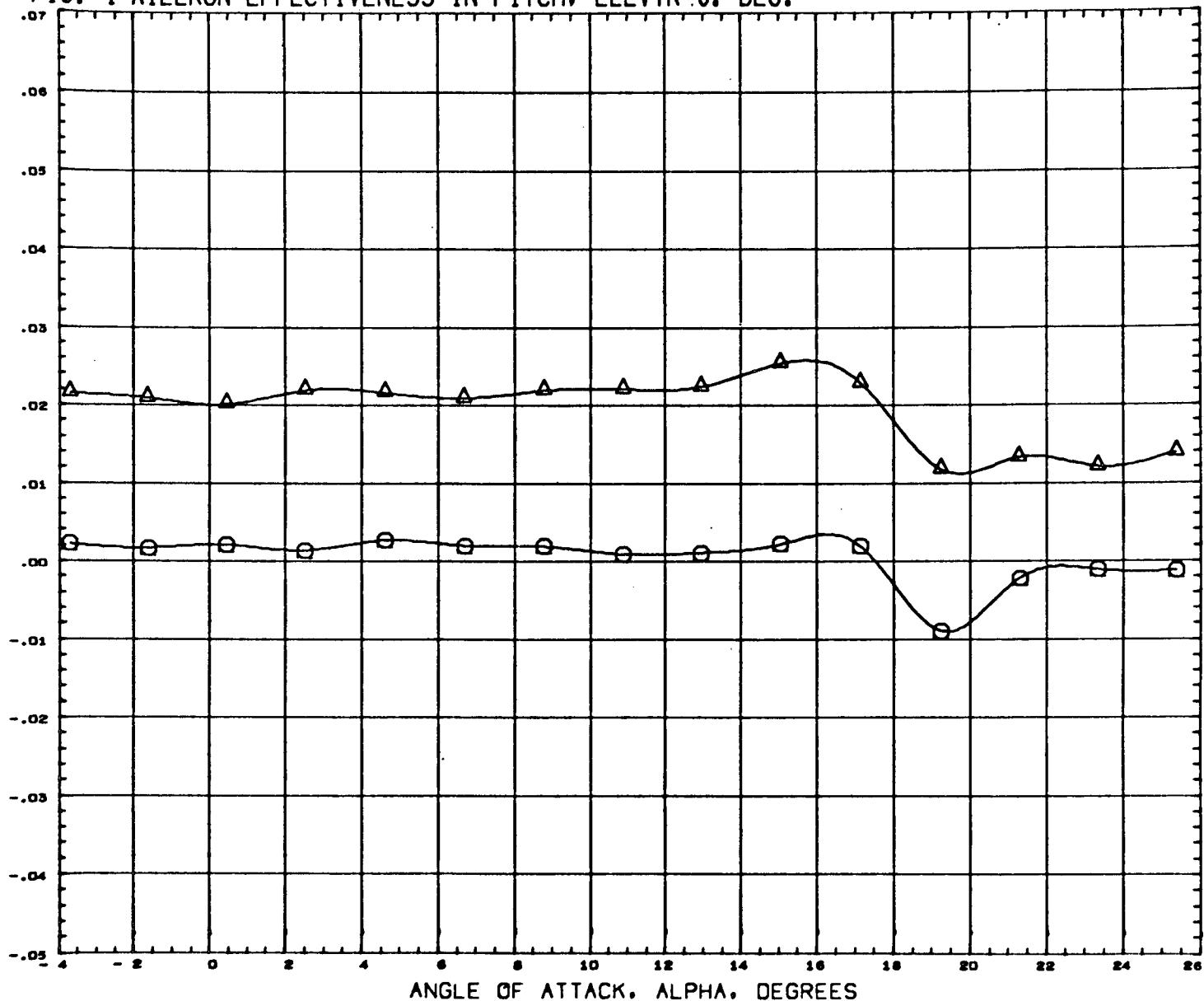
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 16

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1002) GWT 292 CONF.H-33 ORBITER B5W4V5
(RD1010) GWT 292 CONF.H-33 ORBITER B5W4 (+5,-5)V5

LELEVN RELEVN BETA

0.000 0.000 0.000
5.000 -5.000 0.000

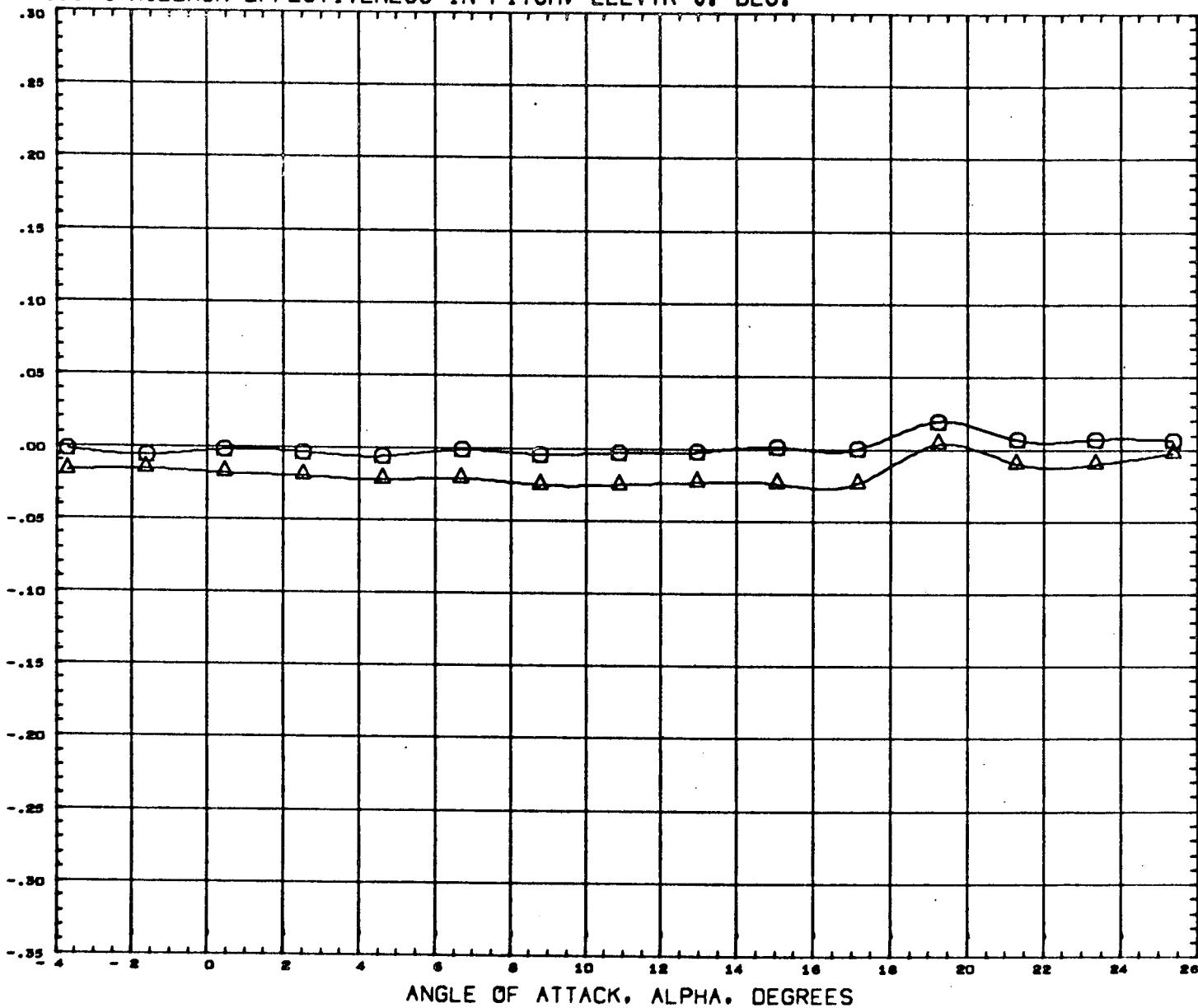
REFERENCE INFORMATION

SREF	7.7440	SQ FT
LREF	5.4000	FT.
BREF	3.7800	FT.
XMRP	1285.0040	IN.
YMRP	0.0000	IN.
ZMRP	403.0004	IN.
SCALE	0.0400	

MACH 0.170

FIG. 4 AILERON EFFECTIVENESS IN PITCH, ELEVTR=0, DEG.

LATERAL FORCE COEFFICIENT, CY



ANGLE OF ATTACK, ALPHA, DEGREES

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) Q GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1010) A GWTT 292 CONF.H-33 ORBITER B5W4(+5,-5)V5

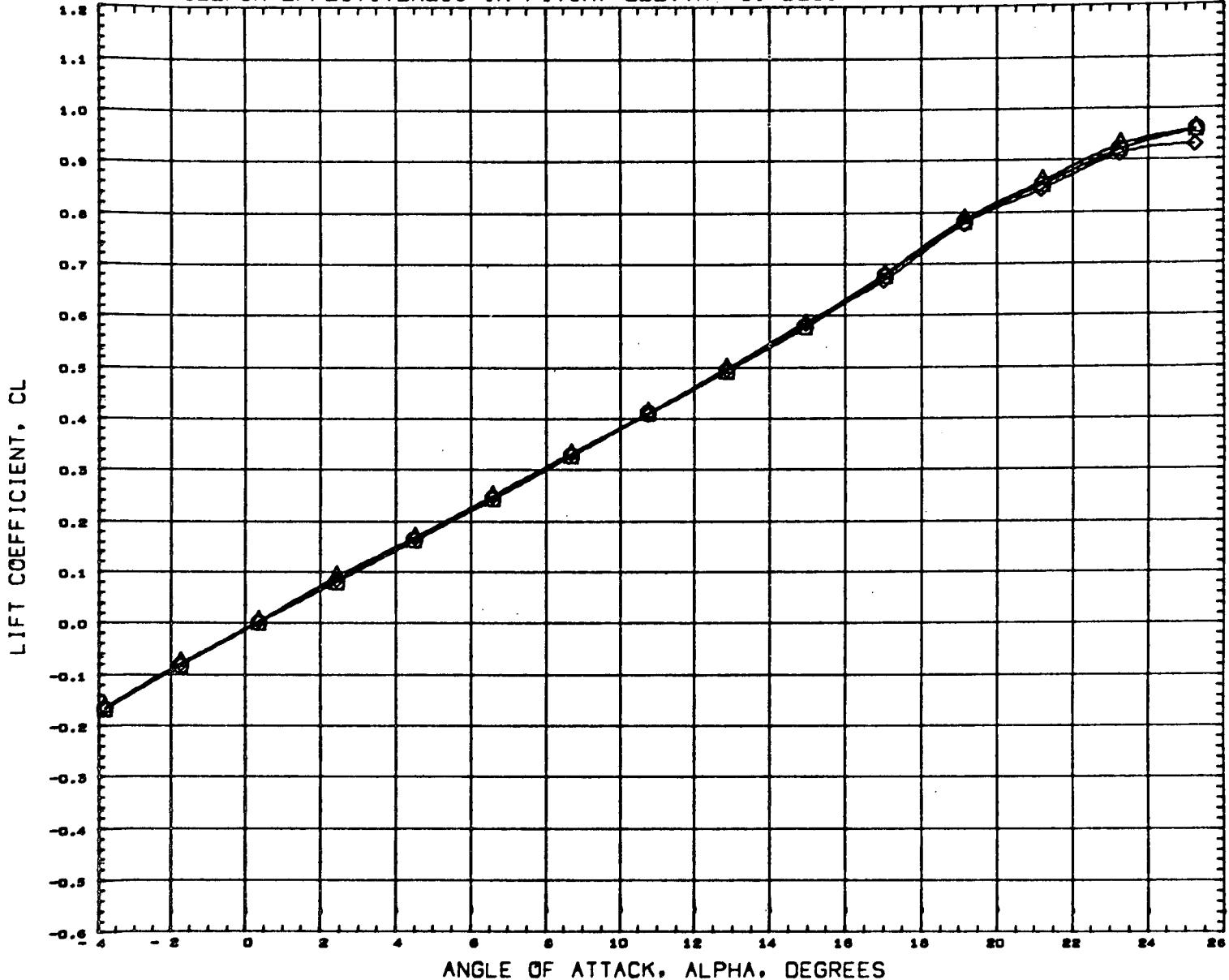
LELEVN RELEVN BETA
 0.000 0.000 0.000
 5.000 -5.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 18

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1006) GWTT 292 CONF.M-33 ORBITER B5W4 (-5,-5)V5
 (RD1013) GWTT 292 CONF.M-33 ORBITER B5W4 (0,-10)V5
 (RD1011) GWTT 292 CONF.M-33 ORBITER B5W4 (+5,-15)V5

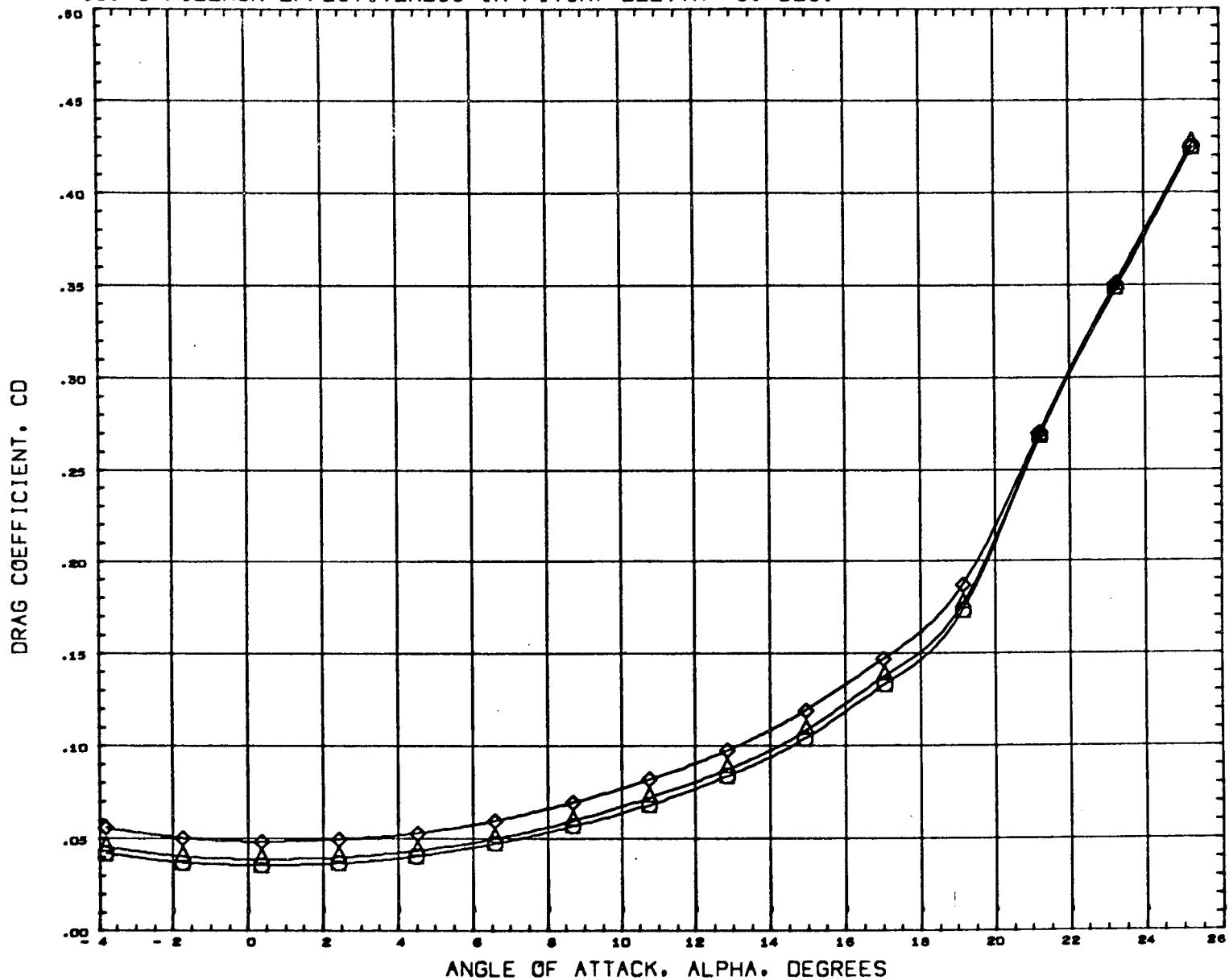
LELEVN RELEVN BETA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 19

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1006) GWTT 292 CONF.M-33 CRBITER BSW4 (-5,-5)V5
 (RD1013) GWTT 292 CONF.M-33 CRBITER BSW4 (0,-10)V5
 (RD1011) GWTT 292 CONF.M-33 CRBITER BSW4 (+5,-15)V5

LELEVN RELEVN BETA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

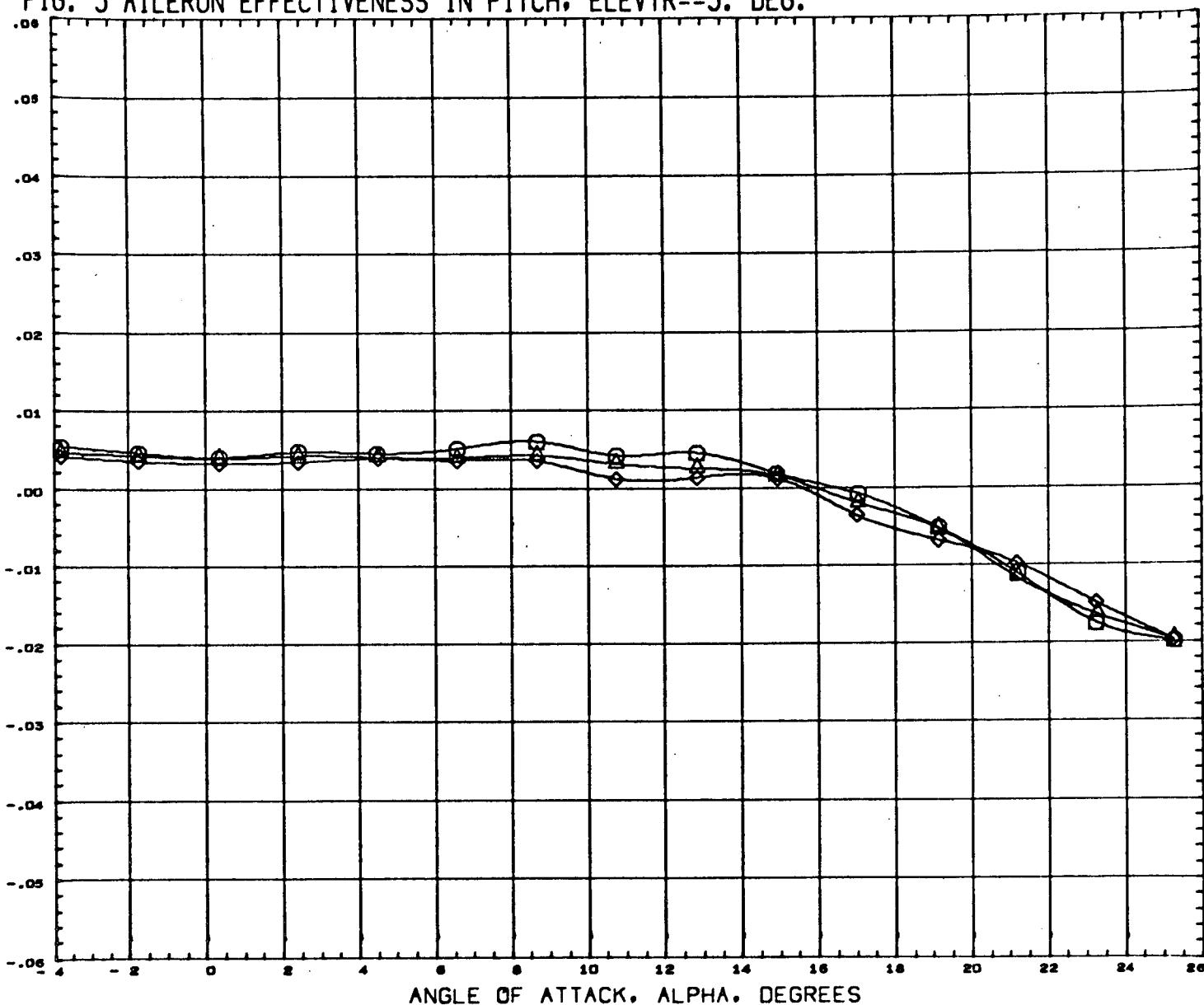
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRF 1285.0040 IN.
 YMRF 0.0000 IN.
 ZMRF 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 20

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.

PITCHING MOMENT COEFFICIENT. CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1006) GWTT 292 CONF.H-33 CRBITER B5W4 (-5,-5)V5
 (RD1013) GWTT 292 CONF.H-33 CRBITER B5W4 (0,-10)V5
 (RD1011) GWTT 292 CONF.H-33 CRBITER B5W4 (+5,-15)V5

LELEVN RELEVN BETA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

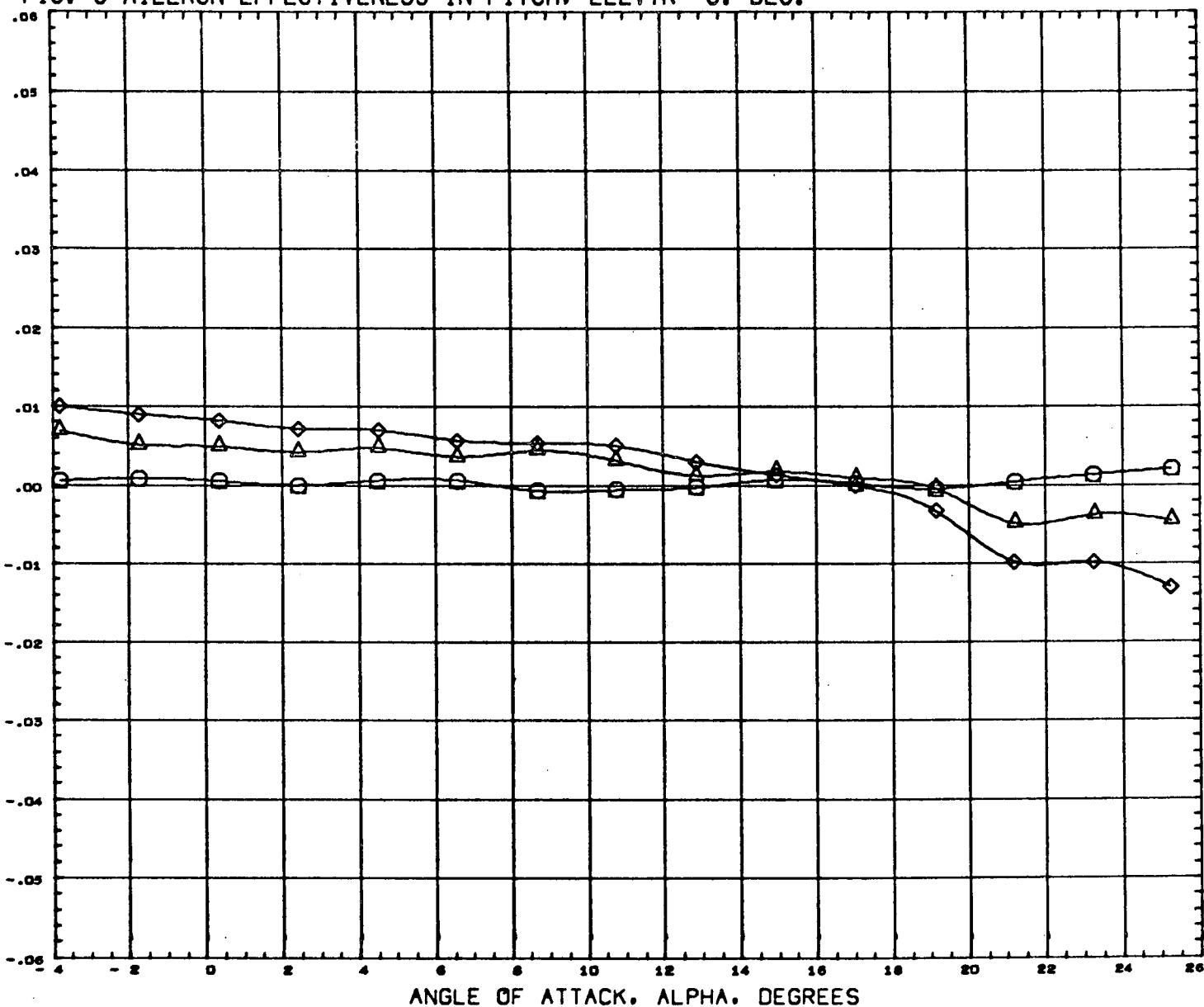
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 21

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1006) GWTT 292 CONF.H-33 ORBITER 85W4 (-5,-5)V5
 (RD1013) GWTT 292 CONF.H-33 ORBITER 85W4 (0,-10)V5
 (RD1011) GWTT 292 CONF.H-33 ORBITER 85W4 (+5,-15)V5

LELEVN RELEVN BETA

-5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

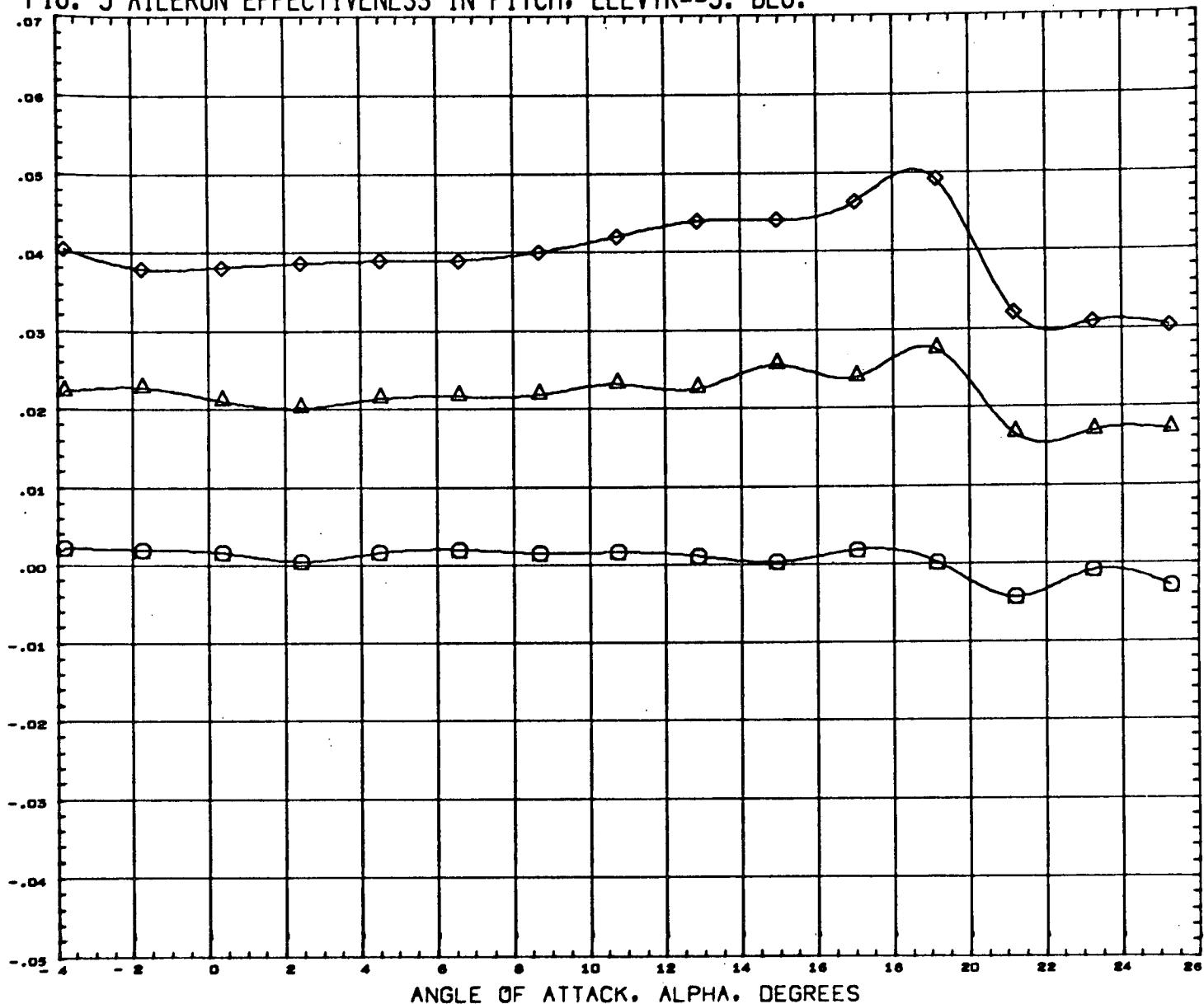
REFERENCE INFORMATION

SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)



ANGLE OF ATTACK, ALPHA, DEGREES

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1006) GWTT 292 CONF.M-33 ORBITER BSW4 (-5,-5)V5
 (RD1013) GWTT 292 CONF.M-33 ORBITER BSW4 (0,-10)V5
 (RD1011) GWTT 292 CONF.M-33 ORBITER BSW4 (+5,-15)V5.

ELEVN RELEVN BETA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

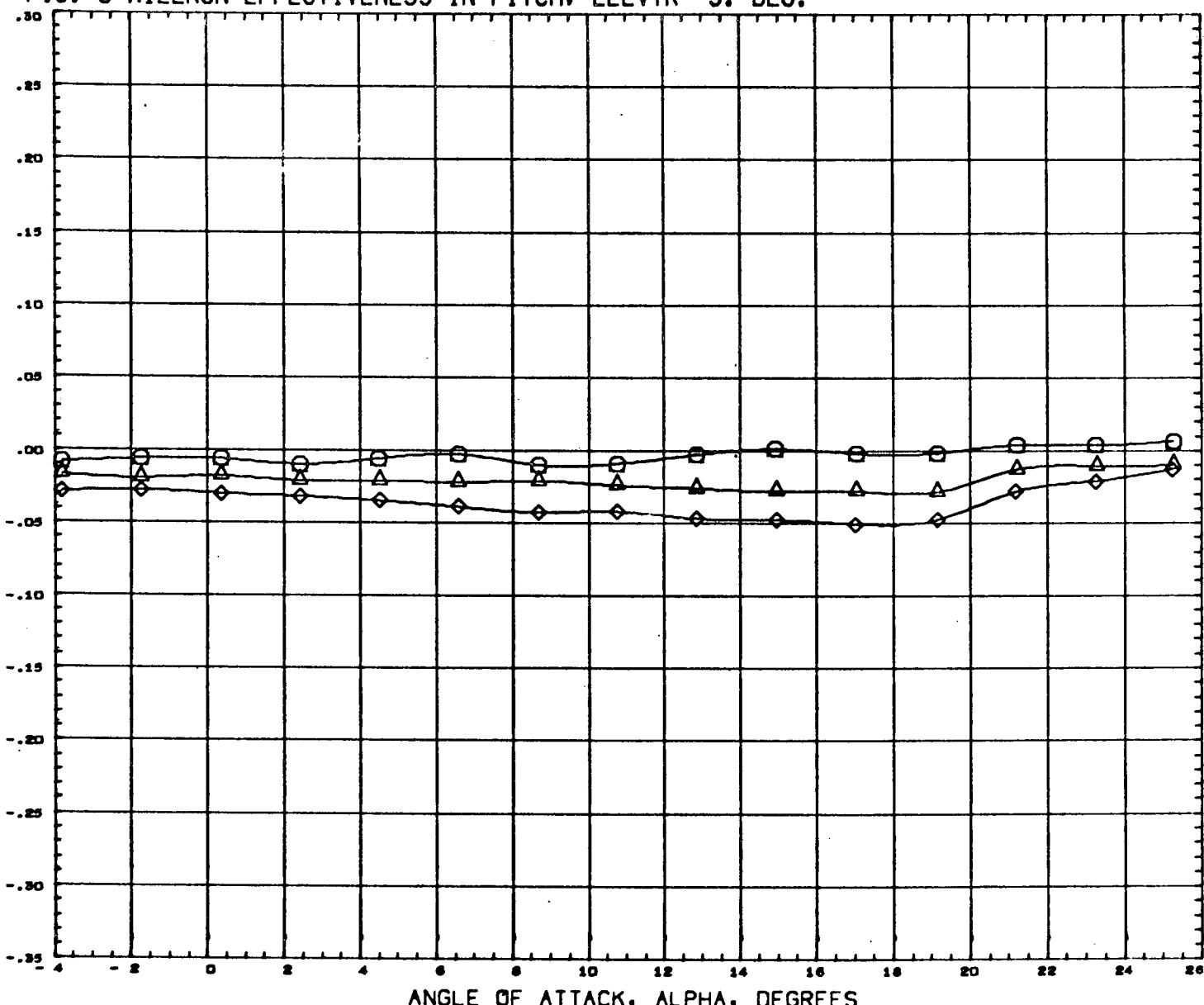
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 23

FIG. 5 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-5. DEG.

LATERAL FORCE COEFFICIENT, CY



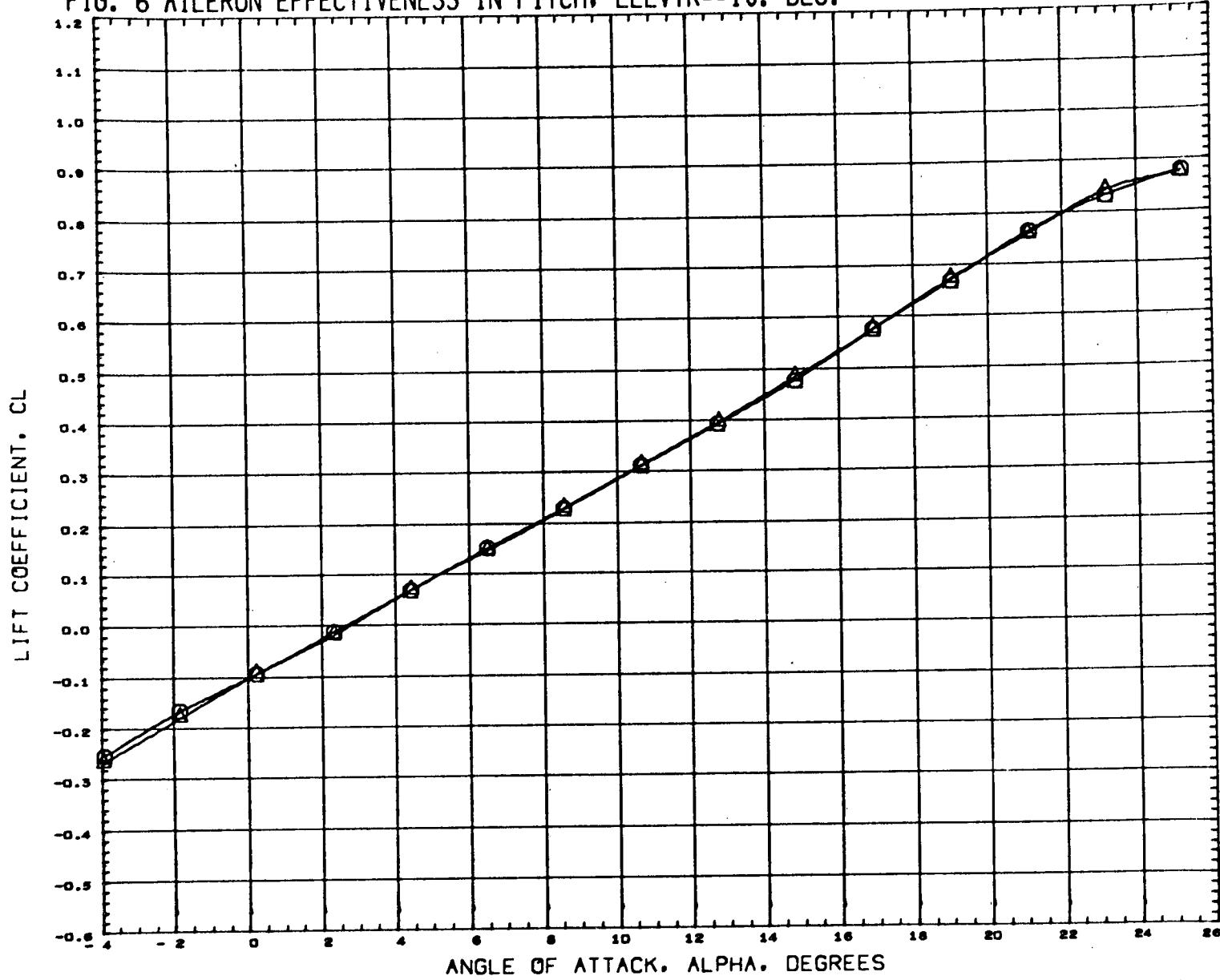
ANGLE OF ATTACK, ALPHA, DEGREES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	LELEVN	RELEVN	BETA	REFERENCE INFORMATION
(RD1008)	GWTT 292 CCNF.H-33 ORBITER B5W4 (-5,-5)V5	-5.000	-5.000	0.000	SREF 7.7440 SQ FT
(RD1013)	GWTT 292 CCNF.H-33 ORBITER B5W4 (0,-10)V5	0.000	-10.000	0.000	LREF 5.4000 FT.
(RD1011)	GWTT 292 CCNF.H-33 ORBITER B5W4 (+5,-15)V5	5.000	-15.000	0.000	BREF 3.7800 FT.

MACH 0.170

PAGE 24

FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1007) GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5
 (RD1012) GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-15)V5

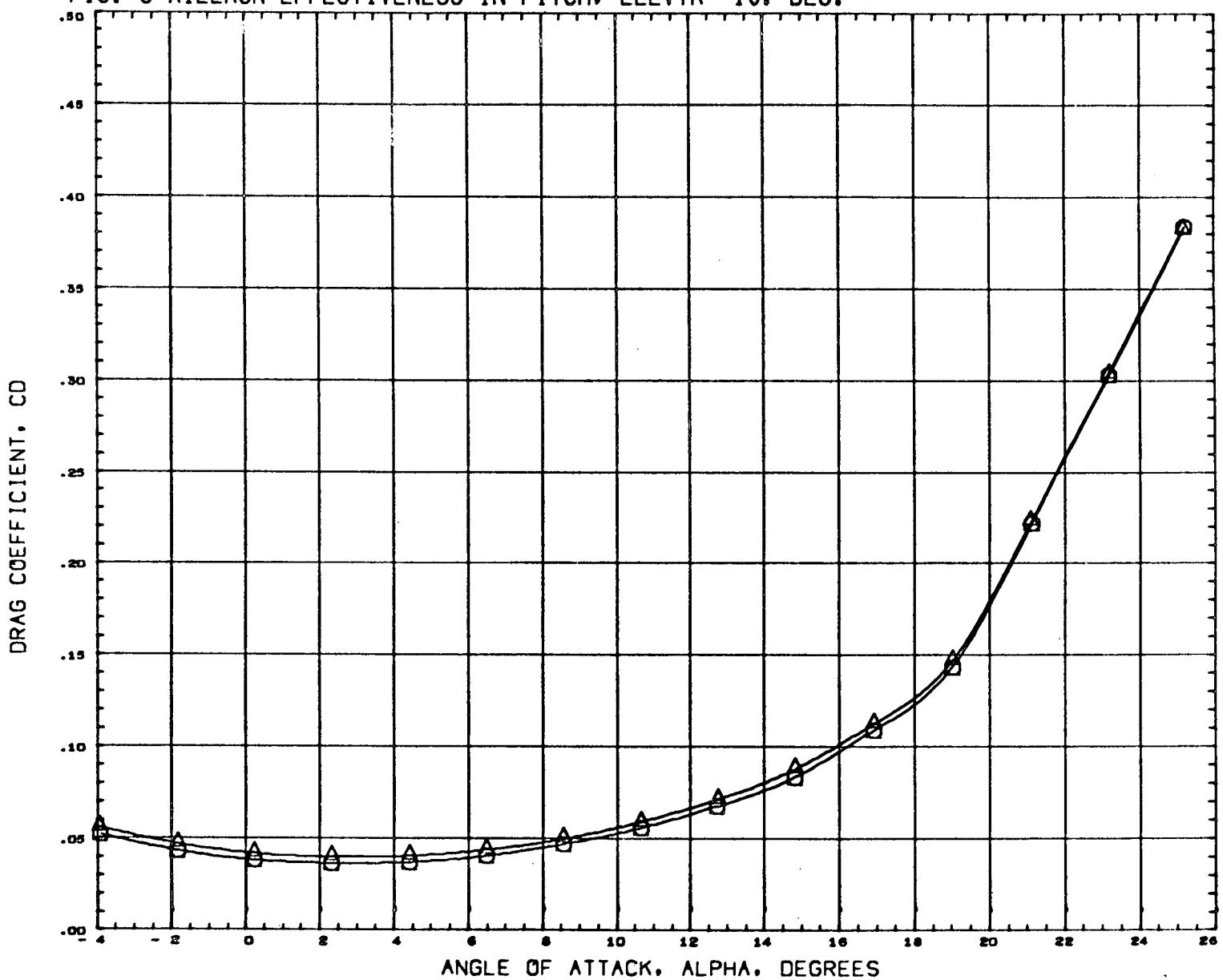
LELEVN RELEVN BETA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 25

FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1007) GWTT 292 CONF.M-33 ORBITER 85W4(-10,-10)V5
 (RD1012) GWTT 292 CONF.M-33 ORBITER 85W4(-5,-15)V5

LELEVN RELEVN BETA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.

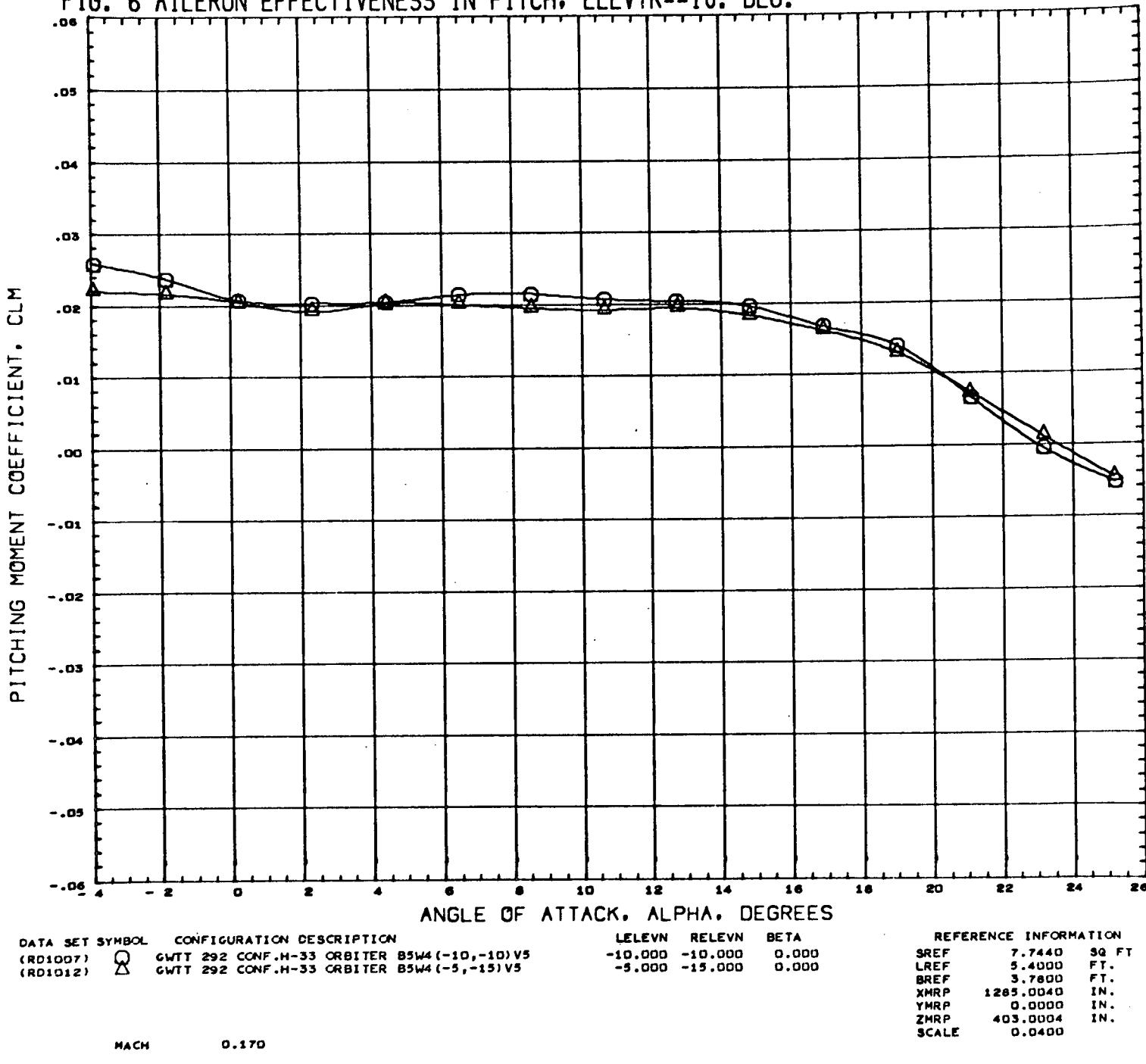
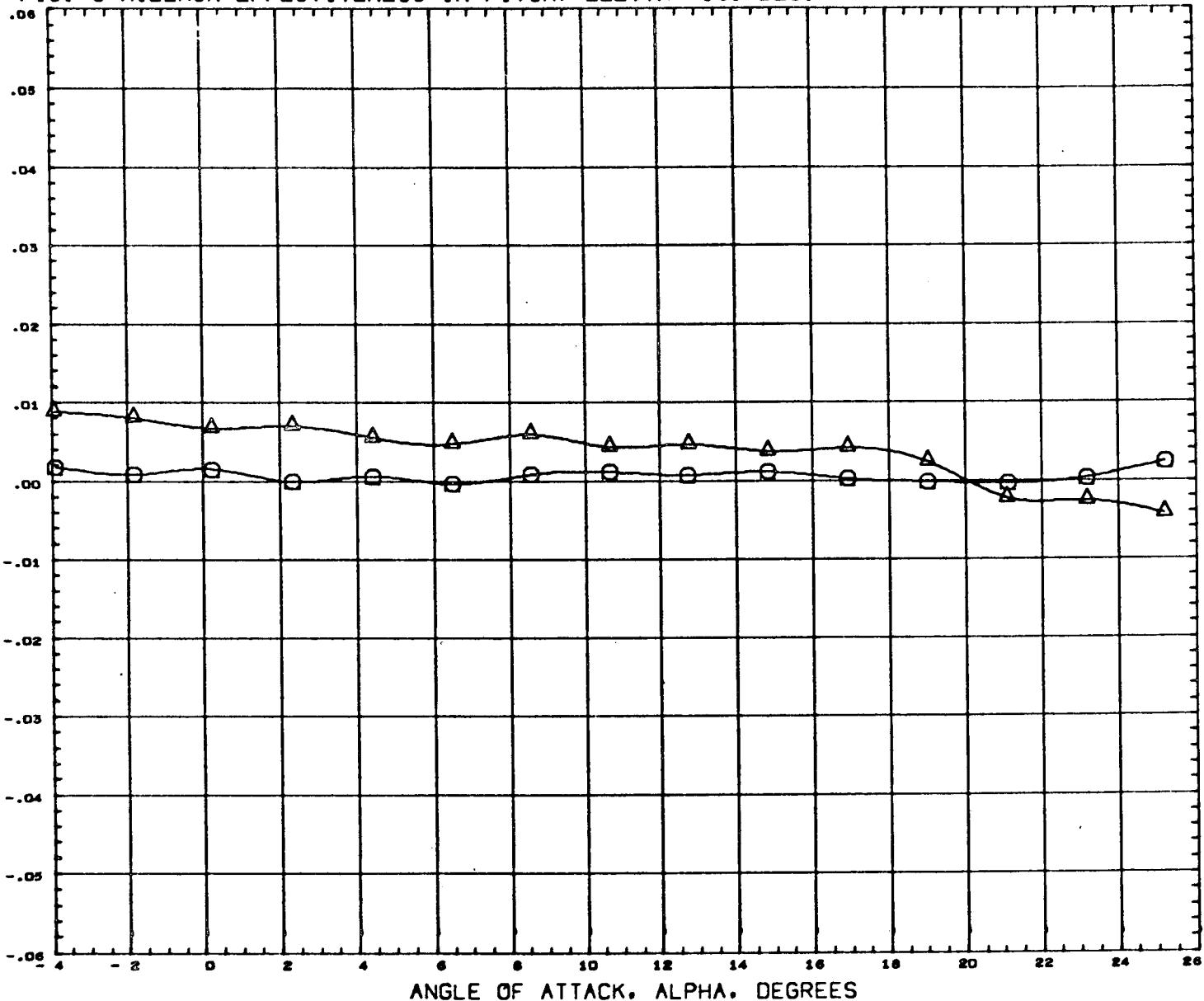


FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1007) GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5

(RD1012) GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-15)V5

ELEVN RELEVN BETA

-10.000 -10.000 0.000
-5.000 -15.000 0.000

REFERENCE INFORMATION

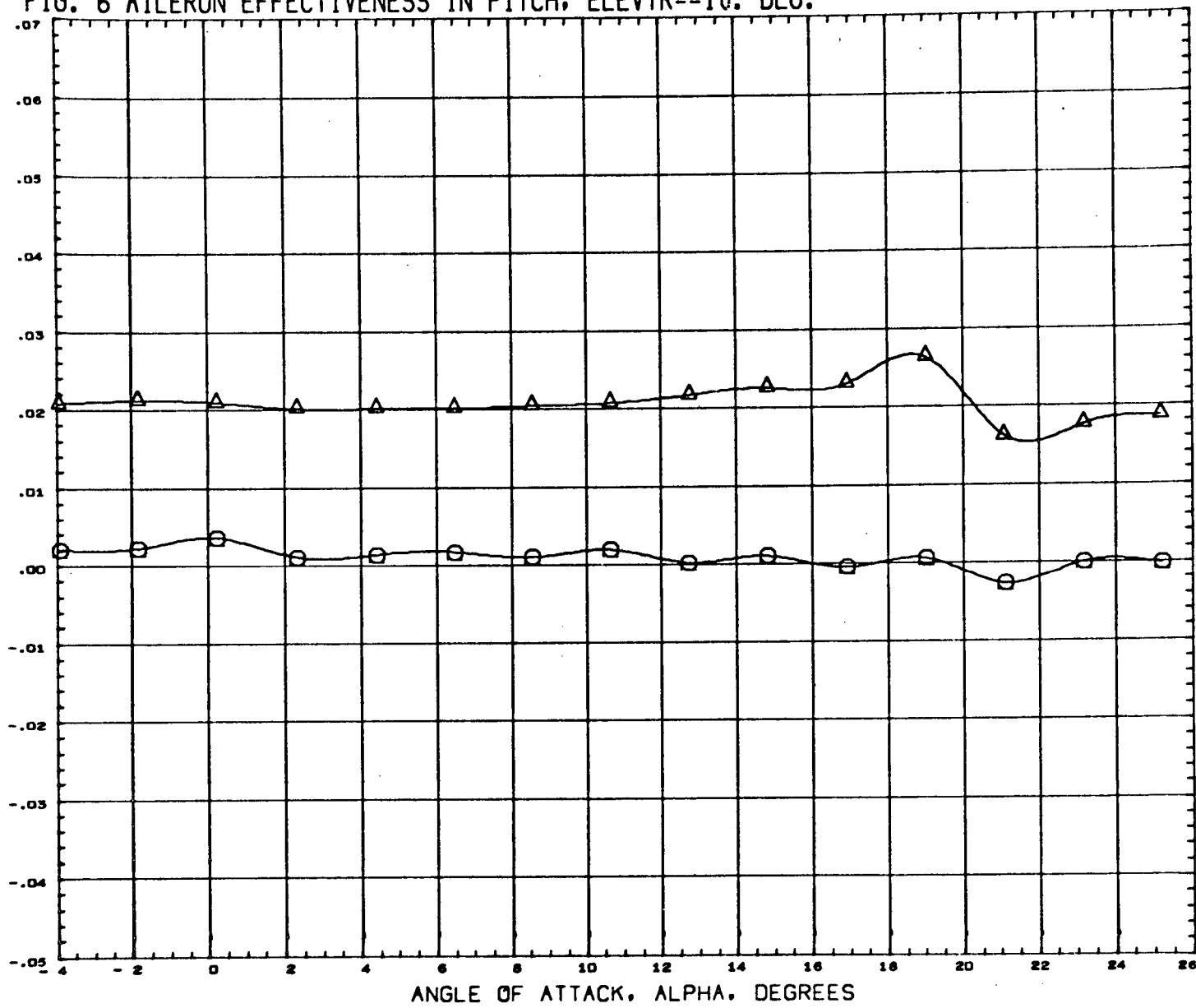
SREF	7.7440	SQ FT
LREF	5.4000	FT.
BREF	3.7600	FT.
XMRP	1285.0040	IN.
YMRP	0.0000	IN.
ZMRP	403.0004	IN.
SCALE	0.0400	

MACH 0.170

PAGE 28

FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



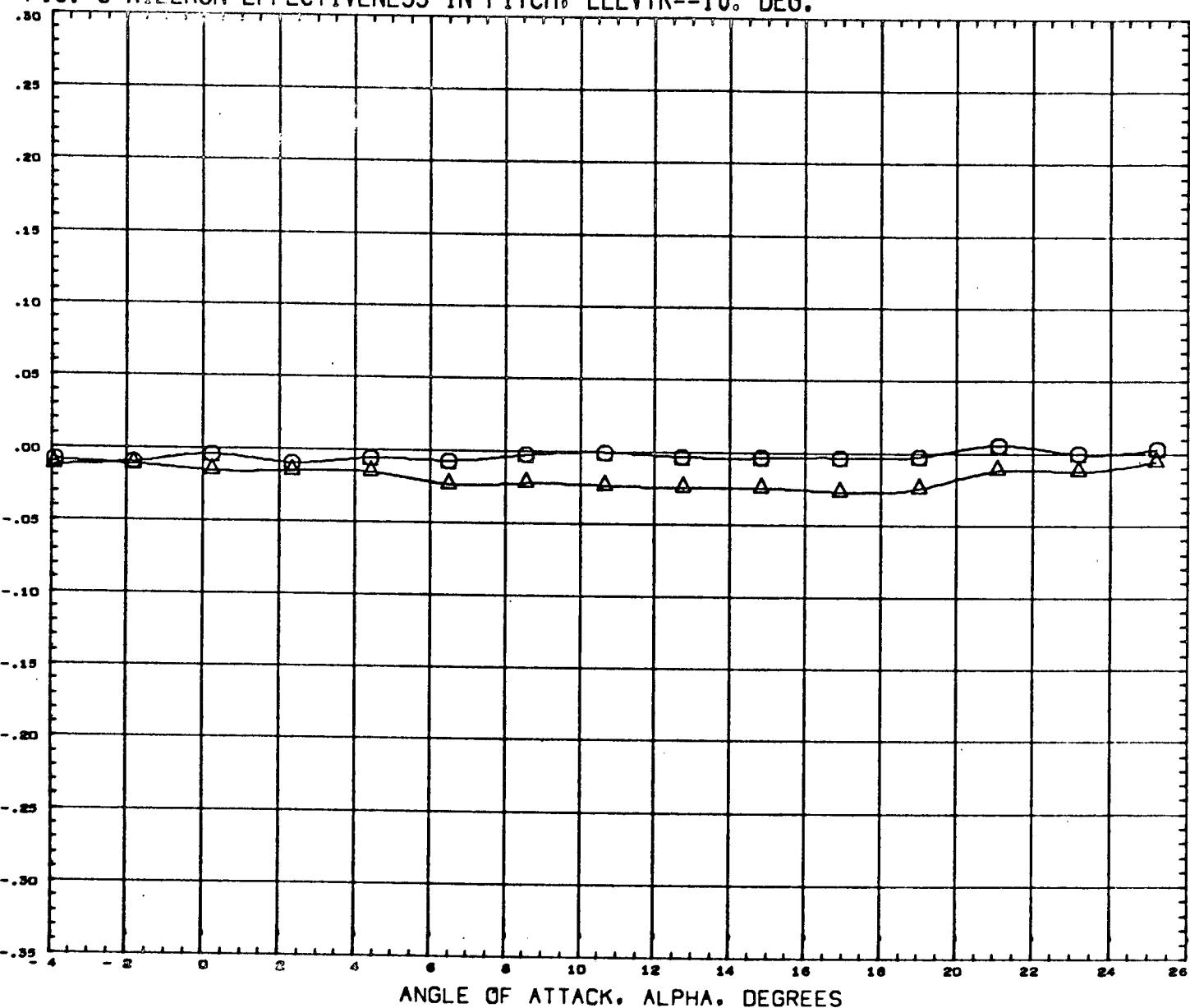
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	LELEVN	RELEVN	BETA	REFERENCE INFORMATION
(RD1007)	GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5	-10.000	-10.000	0.000	SREF 7.7440 SQ FT
(RD1012)	GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-15)V5	-5.000	-15.000	0.000	LREF 5.4000 FT.
				BREF 3.7800 FT.	
				XMRP 1285.0040 IN.	
				YMRP 0.0000 IN.	
				ZMRP 403.0004 IN.	
				SCALE 0.0400	

MACH 0.170

PAGE 29

FIG. 6 AILERON EFFECTIVENESS IN PITCH, ELEVTR=-10. DEG.

LATERAL FORCE COEFFICIENT, CY



ANGLE OF ATTACK, ALPHA, DEGREES

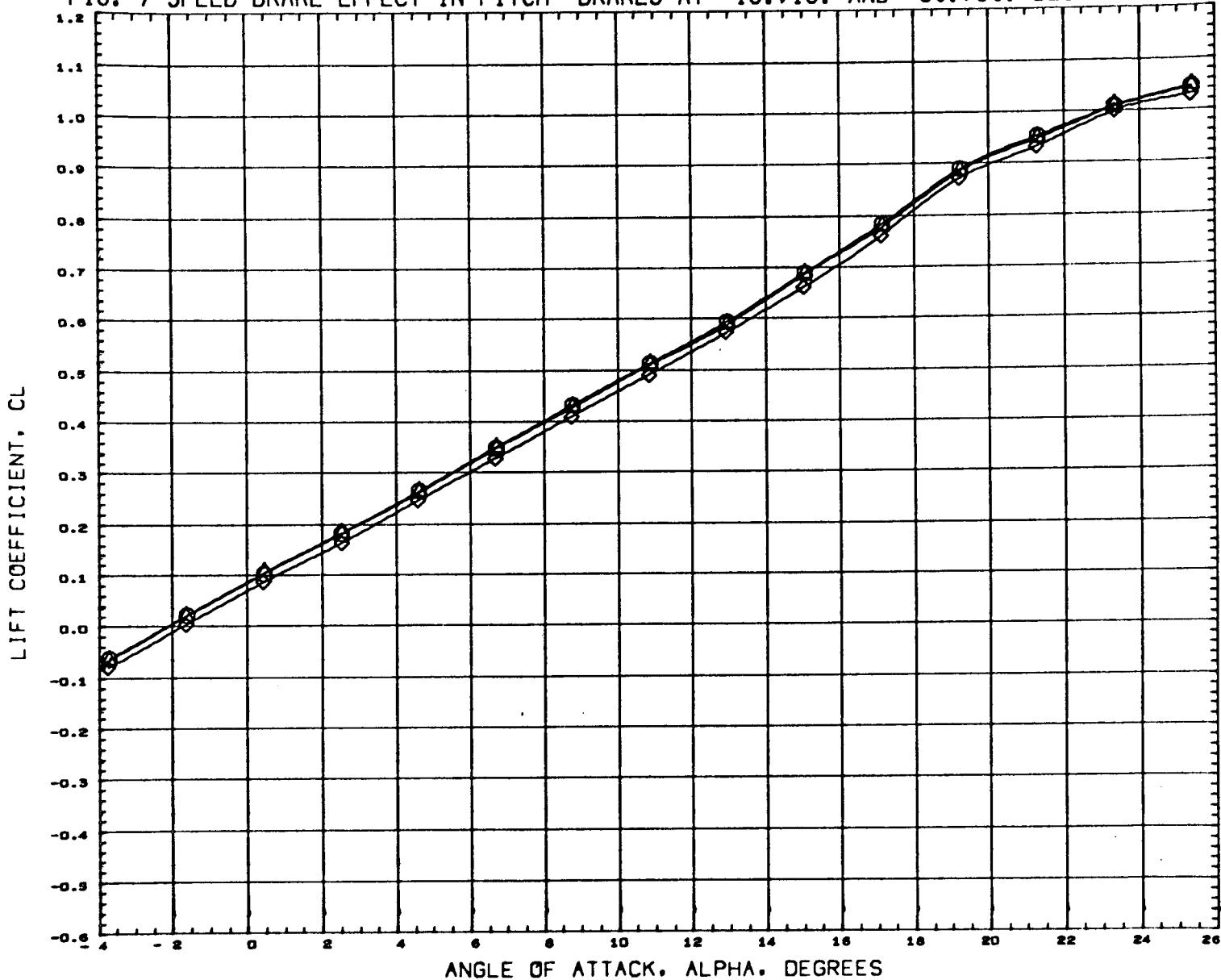
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1007) GWTT 292 CONF.M-33 ORBITER B5W4 (-10,-10)V5
 (RD1012) GWTT 292 CONF.M-33 ORBITER B5W4 (-5,-15)V5

LELEVN RELEVN BETA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMNP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMNP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 7 SPEED BRAKE EFFECT IN PITCH- BRAKES AT -15., 15. AND -30., 30. DEG.

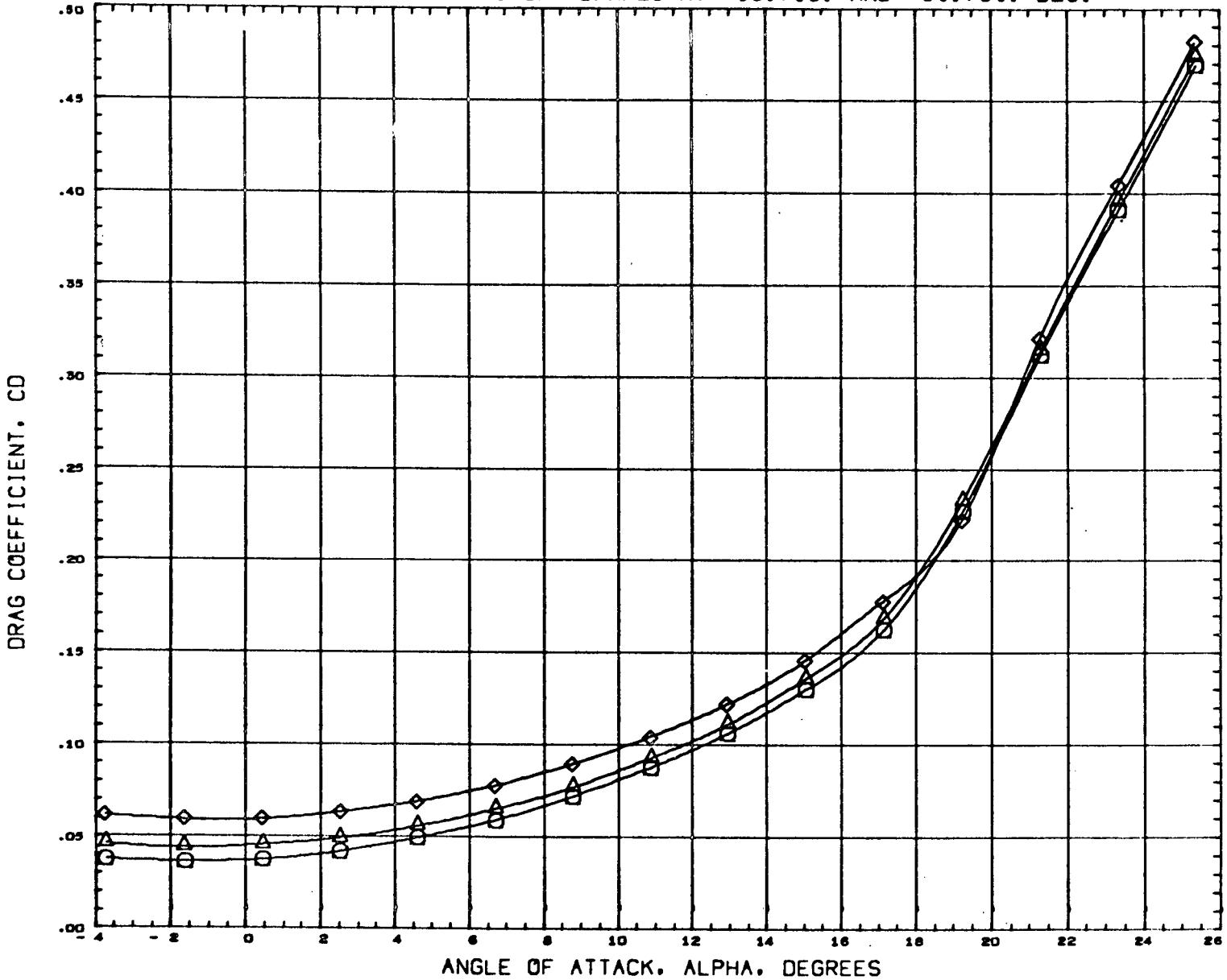


DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1014) GWTT 292 CONF.H-33 ORBITER B5W4V5 (D,+15,-15)
 (RD1015) GWTT 292 CONF.H-33 ORBITER B5W4V5 (D,+30,-30)

TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 7.7440 SQ FT
0.000	15.000	-15.000	0.000	LREF 5.4000 FT.
0.000	30.000	-30.000	0.000	BREF 3.7800 FT.
				XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

FIG. 7 SPEED BRAKE EFFECT IN PITCH- BRAKES AT -15., 15. AND -30., 30. DEG.



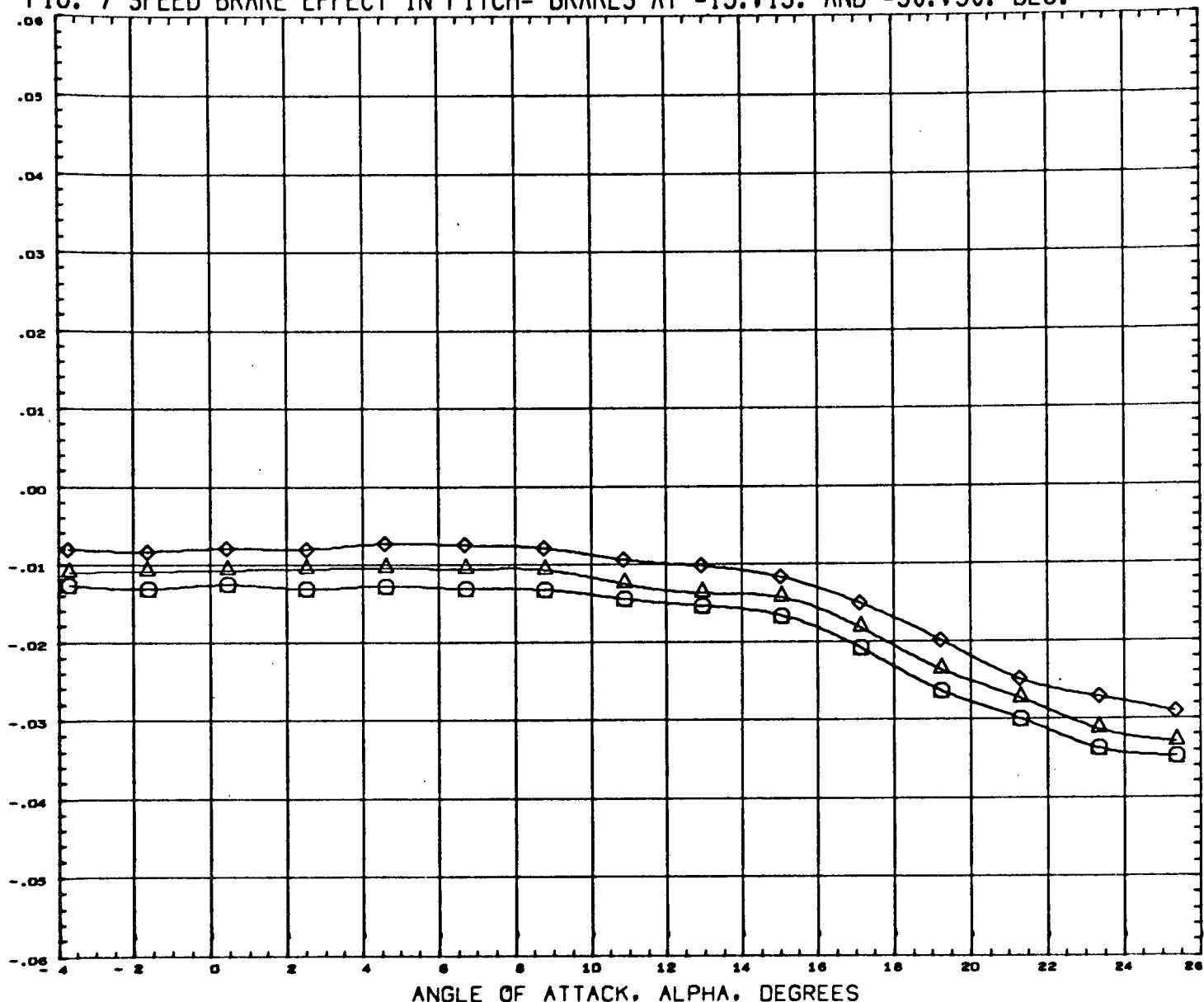
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1002)	GWT 292 CONF.H-33 ORBITER B5W4V5	0.000	0.000	0.000	0.000	SREF 7.7440 SQ FT
(RD1014)	GWT 292 CONF.H-33 ORBITER B5W4V5 (D,+15,-15)	0.000	15.000	-15.000	0.000	LREF 5.4000 FT.
(RD1015)	GWT 292 CONF.H-33 ORBITER B5W4V5 (D,+30,-30)	0.000	30.000	-30.000	0.000	BREF 3.7800 FT. XMRP 1285.0040 IN. YMRP 0.0000 IN. ZMRP 403.0004 IN. SCALE 0.0400

MACH 0.170

PAGE 32

FIG. 7 SPEED BRAKE EFFECT IN PITCH- BRAKES AT -15., 15. AND -30., 30. DEG.

PITCHING MOMENT COEFFICIENT, CLM

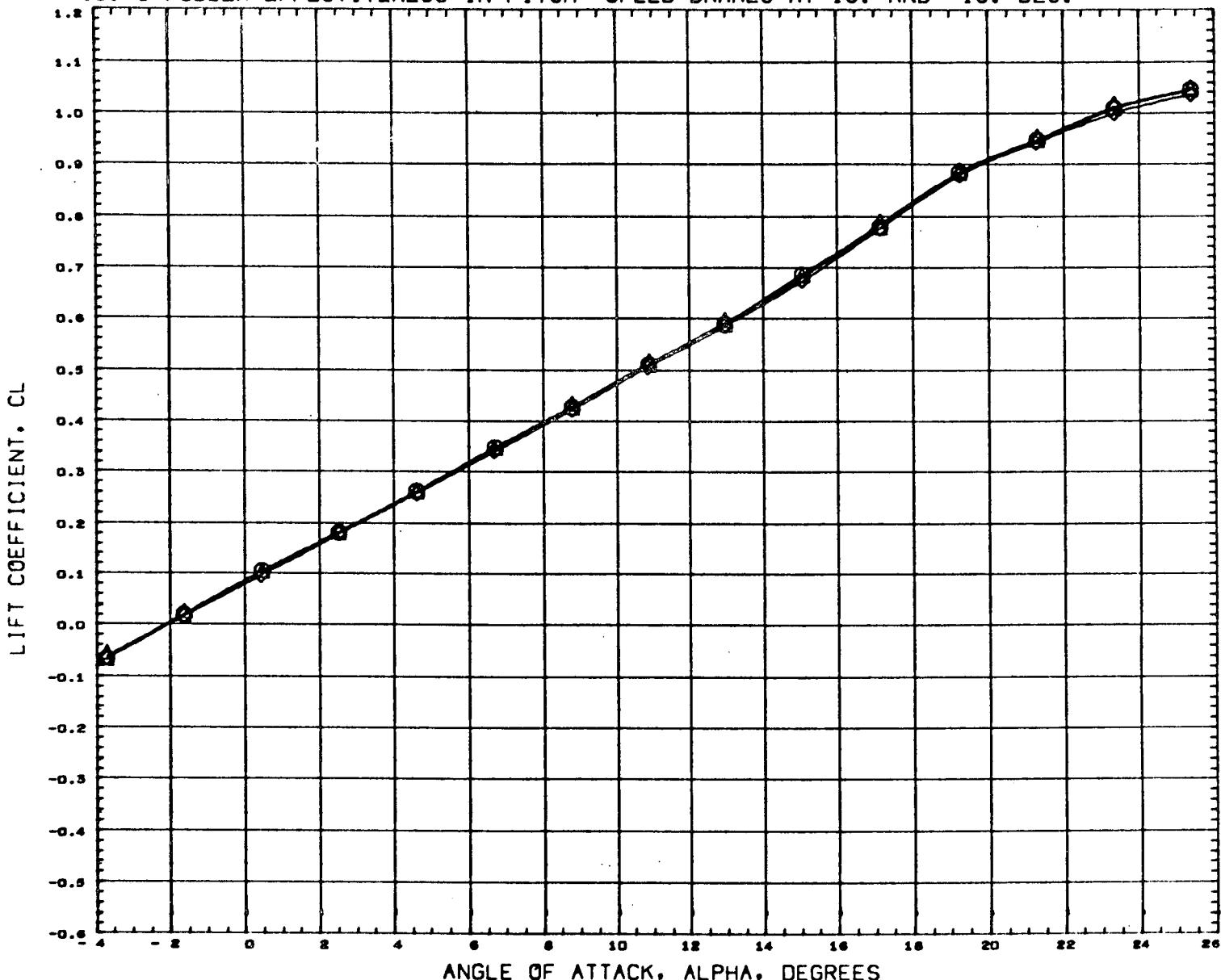


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1002)	GWTT 292 CCNF.H-33 ORBITER B5W4V5	0.000	0.000	0.000	0.000	SREF 7.7440 SQ FT
(RD1014)	GWTT 292 CCNF.H-33 ORBITER B5W4V5(0,+15,-15)	0.000	15.000	-15.000	0.000	LREF 5.4000 FT.
(RD1015)	GWTT 292 CCNF.H-33 ORBITER B5W4V5(0,+30,-30)	0.000	30.000	-30.000	0.000	BREF 3.7600 FT.
					XMRP 1285.0040 IN.	
					YMRP 0.0000 IN.	
					ZMRP 403.0004 IN.	
					SCALE 0.0400	

MACH 0.170

PAGE 33

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.

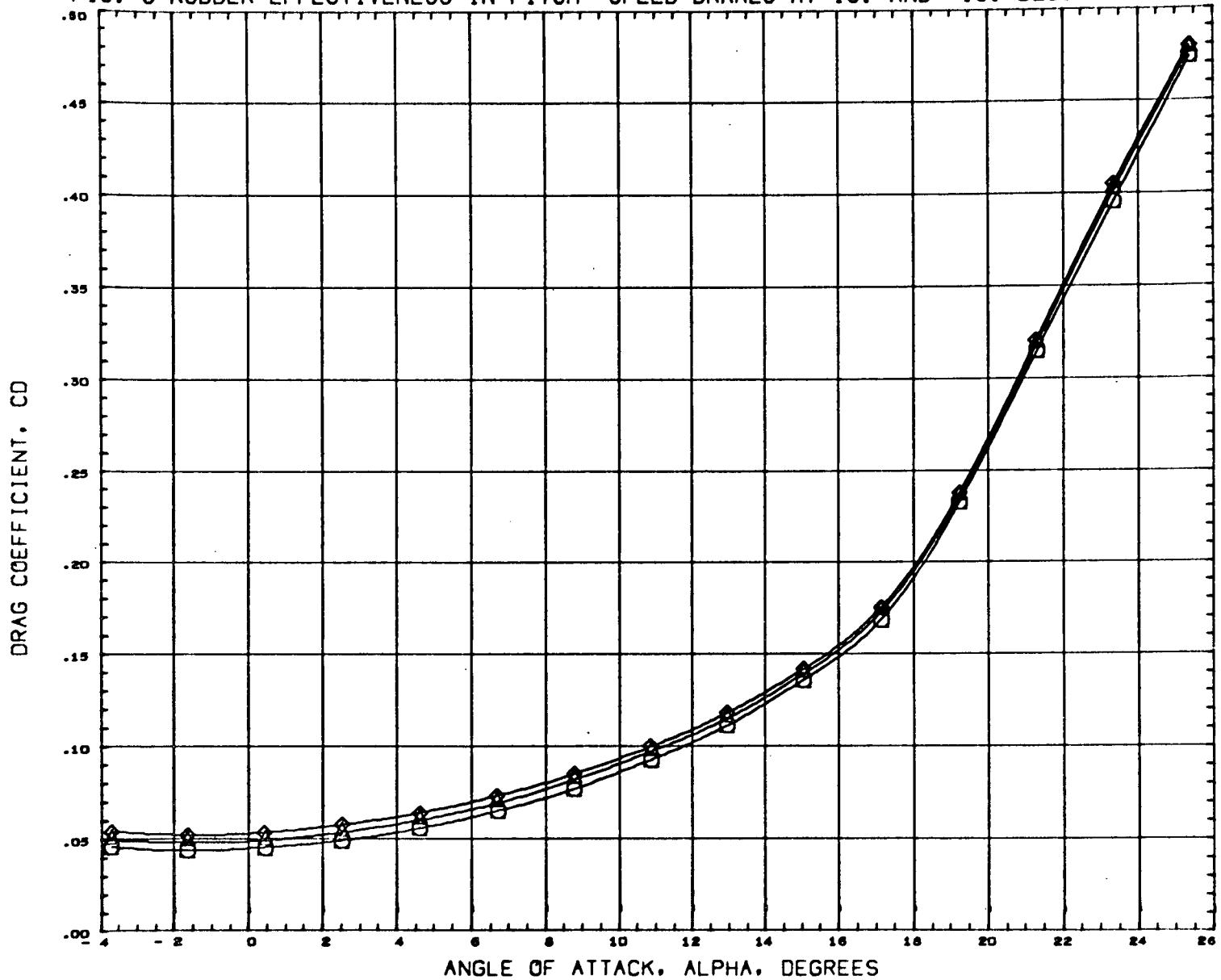


DATA SET SYMBOL		CONFIGURATION DESCRIPTION	TRUDDOR	LLRUDD	LRRUDD	BETA	REFERENCE	INFORMATION
(RD1014)	Q	CWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF	7.7440 3Q FT
(RD1016)	△	CWTT 292 CONF.H-33 CRBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF	5.4000 FT.
(RD1017)	◇	CWTT 292 CONF.H-33 CRBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF	3.7800 FT.
							XMRP	1285.0040 IN.
							YMRP	0.0000 IN.
							ZMRP	403.0004 IN.
							SCALE	0.0400

MACH 0.170

PAGE 34

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.



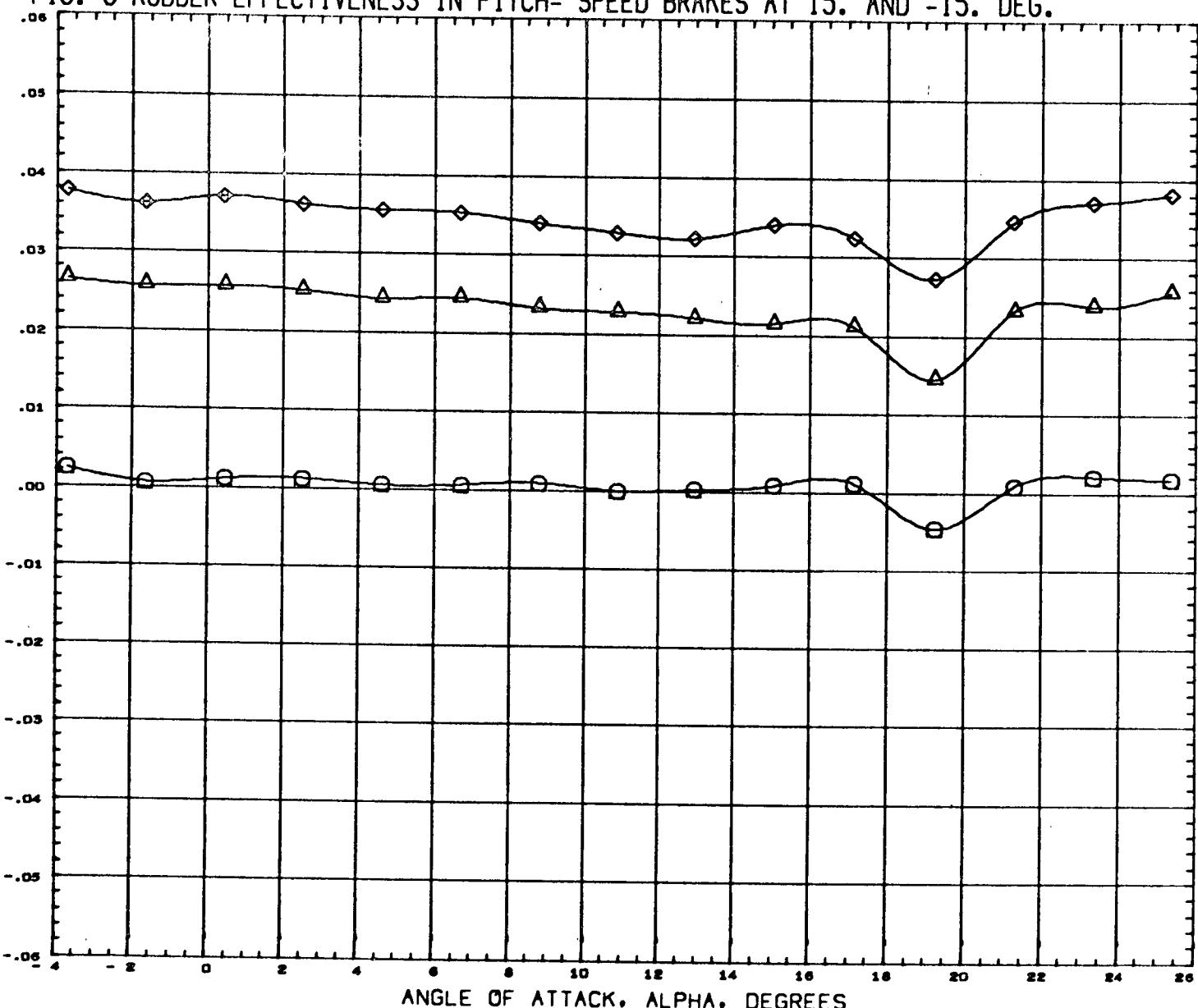
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1014)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
(RD1016)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
(RD1017)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF 3.7800 FT.

MACH 0.170

PAGE 35

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.

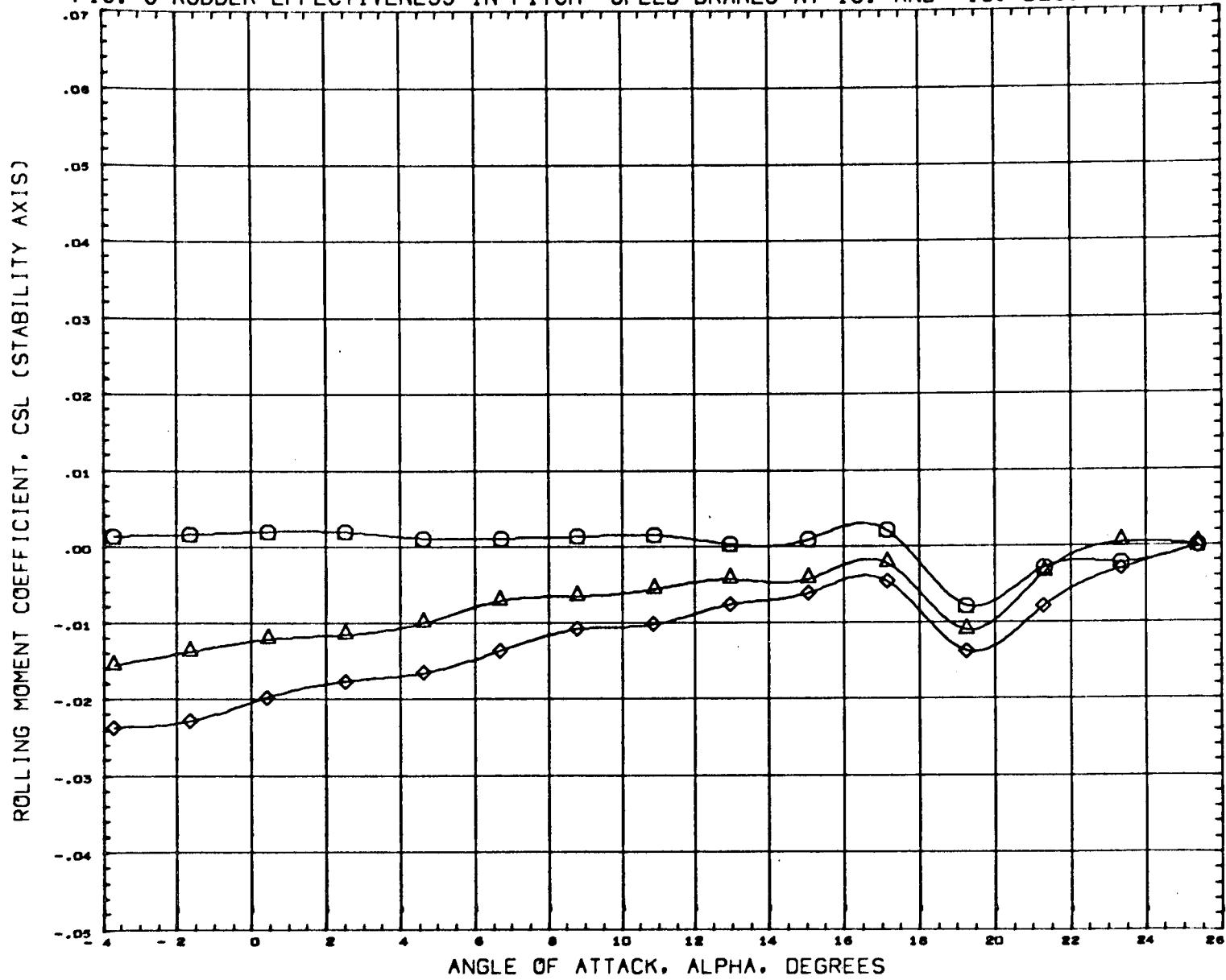
YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL		CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION	
(RD1014)	○	GWT T 292 CONF.H-33 CRBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF	7.7440 SQ FT
(RD1016)	△	GWT T 292 CONF.H-33 CRBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF	5.4000 FT.
(RD1017)	◇	GWT T 292 CONF.H-33 CRBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF	3.7800 FT.
							XMRP	1265.0040 IN.
							YMRP	0.0000 IN.
							ZMRP	403.0004 IN.
							SCALE	0.0400

MACH 0.170

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.

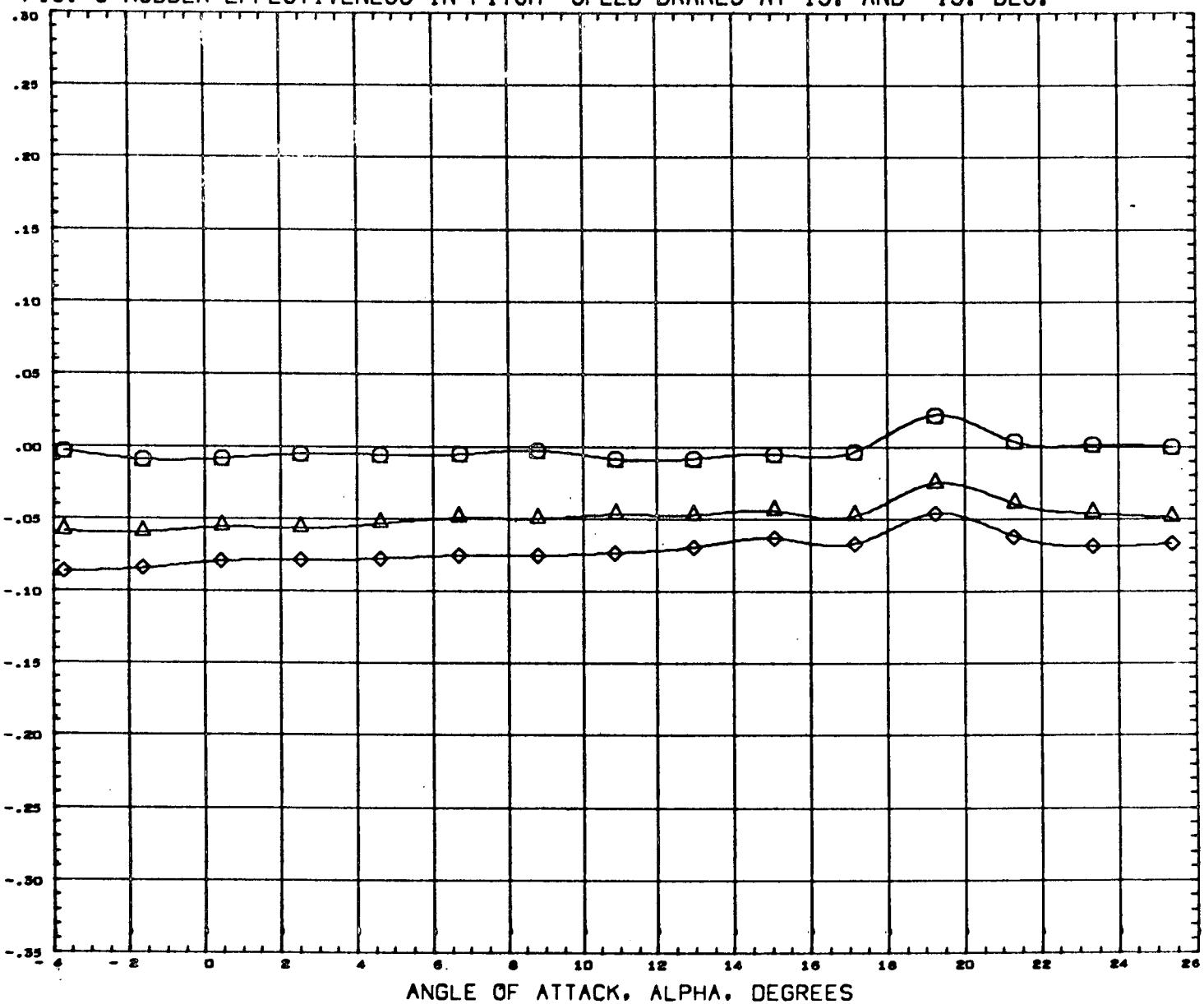


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1014)	GWTT 292 CONF.M-33 ORBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
(RD1016)	GWTT 292 CONF.M-33 ORBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
(RD1017)	GWTT 292 CONF.M-33 ORBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF 3.7800 FT.
						XMRP 1285.0040 IN.
						YMRP 0.0000 IN.
						ZMRP 403.0004 IN.
						SCALE 0.0400

MACH 0.170

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.

LATERAL FORCE COEFFICIENT, CY



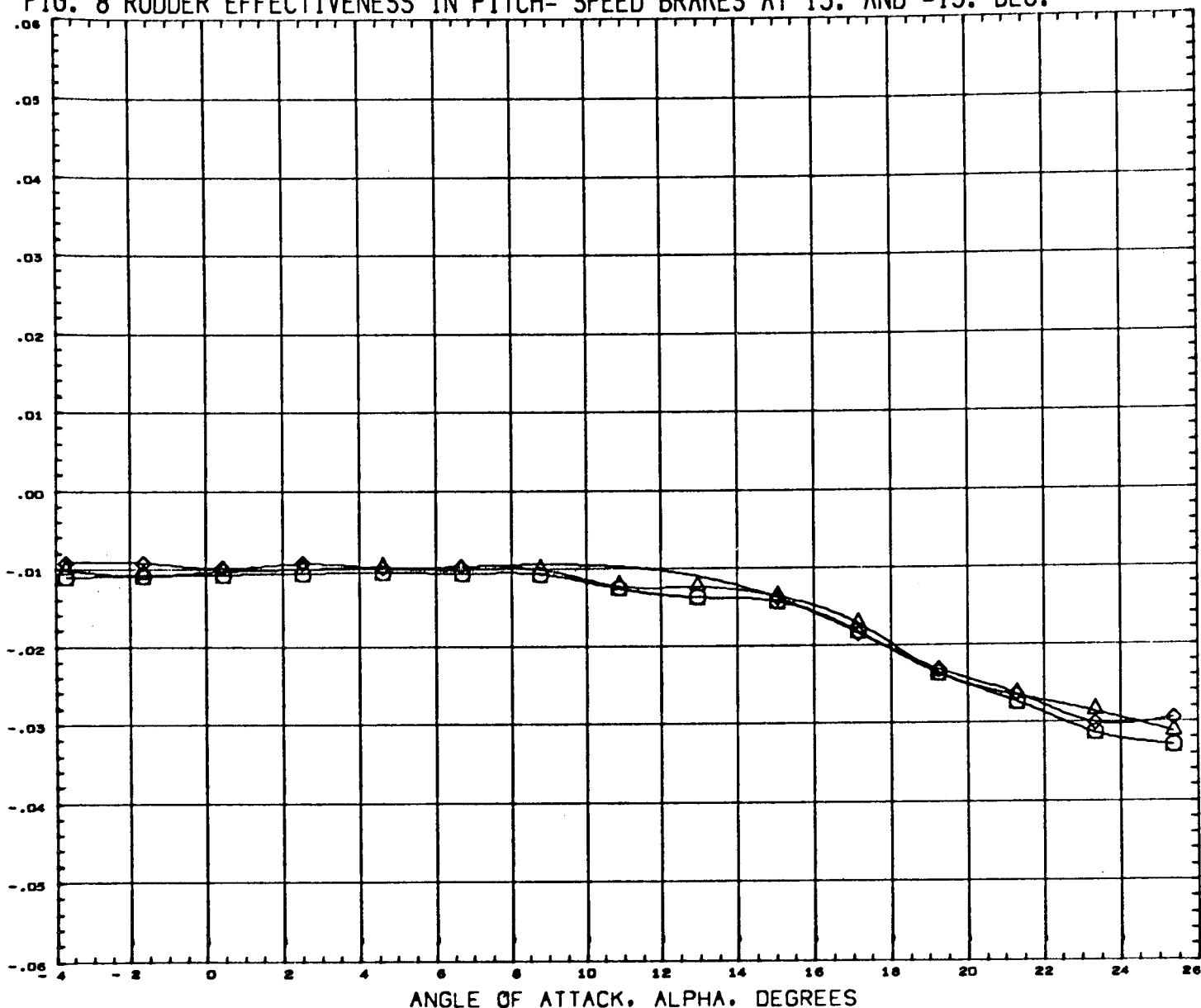
ANGLE OF ATTACK, ALPHA, DEGREES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RDI014)	GWTT 292 CONF.M-33 CRBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
(RDI016)	GWTT 292 CONF.M-33 CRBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
(RDI017)	GWTT 292 CONF.M-33 CRBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF 3.7800 FT. XMRP 1285.0040 IN. YMRP 0.0000 IN. ZMRP 403.0004 IN. SCALE 0.0400

MACH 0.170

FIG. 8 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 15. AND -15. DEG.

PITCHING MOMENT COEFFICIENT, CLM

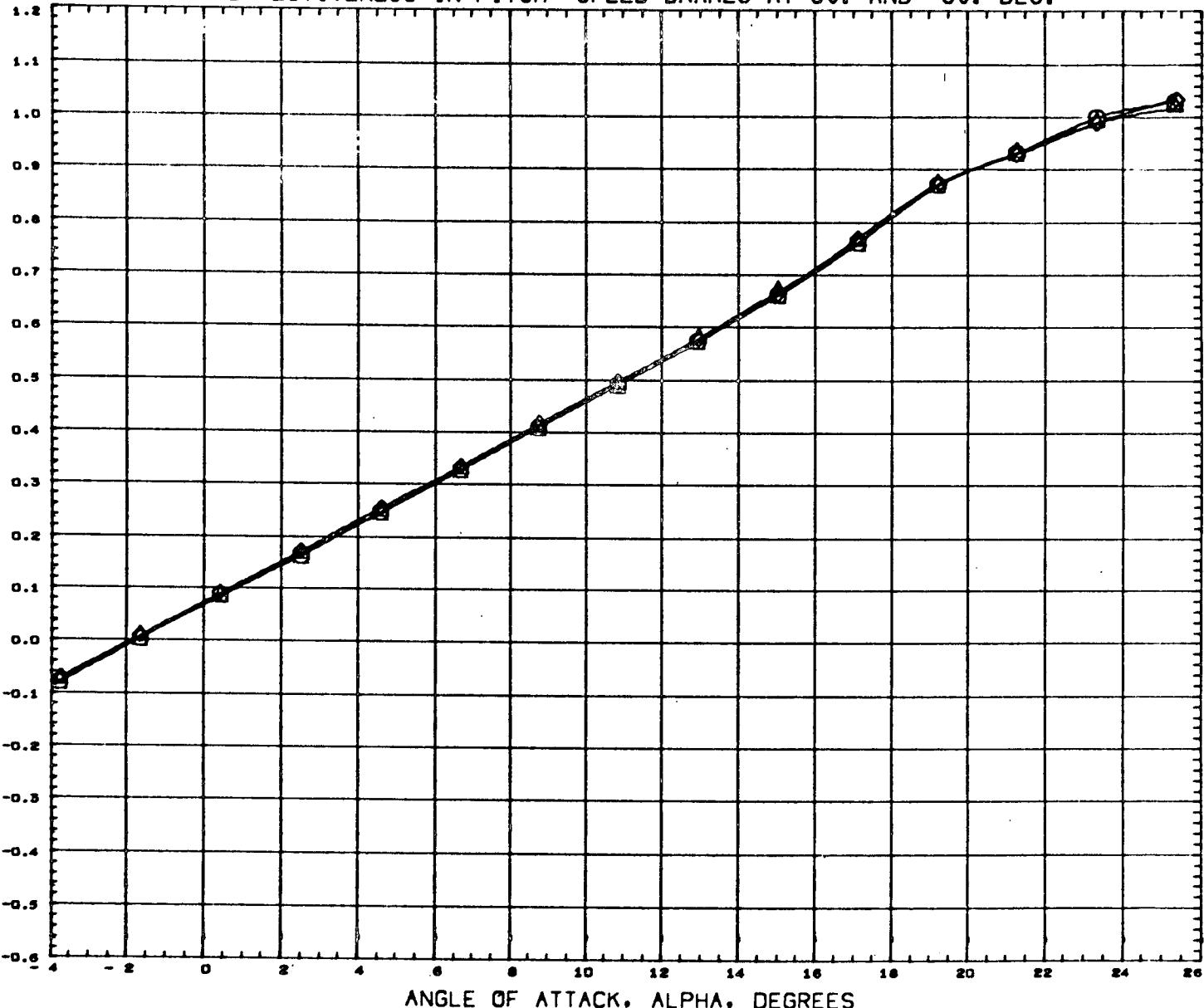


DATA SET SYMBOL		CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION		
(RD1014)	○	GWT 292 CONF.H-33 ORBITER B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF	7.7440	SQ FT
(RD1016)	□	GWT 292 CONF.H-33 ORBITER B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF	5.4000	FT.
(ZD1017)	◊	GWT 292 CONF.H-33 ORBITER B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF	3.7800	FT.
							XMRP	1285.0040	IN.
							YMRP	0.0000	IN.
							ZMRP	403.0004	IN.
							SCALE	0.0400	

MACH 0.170

PAGE 39

FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.

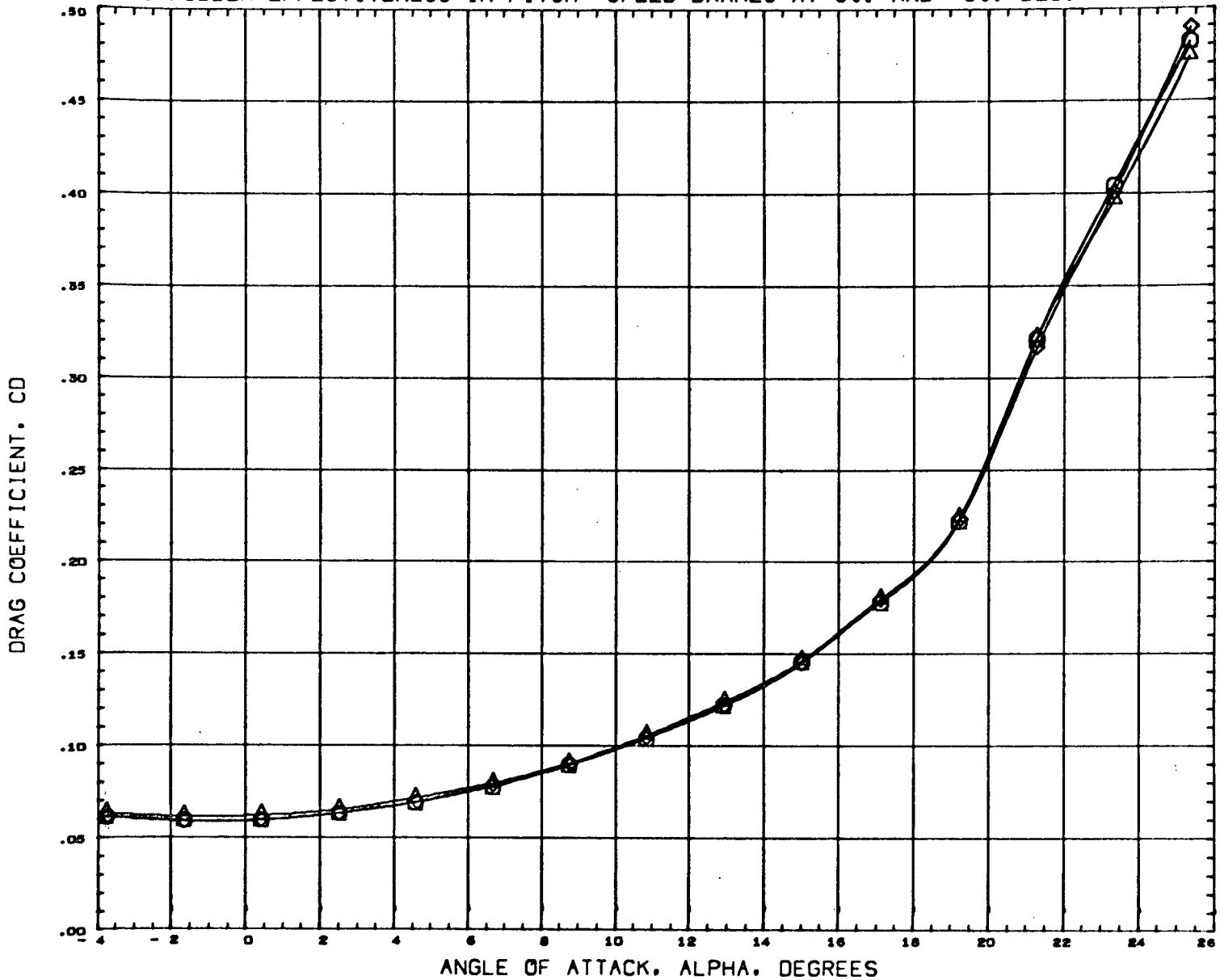


DATA SET SYMBOL CONFIGURATION DESCRIPTION

RD1015	○	GWTT 292 CONF. H-33 ORBITER BSW4V5 (0,+30,-30)	TRUDDR	LLRUDD	LRRUDD	BETA	SREF	7.7440	SQ FT
(RD1016)	□	GWTT 292 CONF. H-33 ORBITER BSW4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	LREF	5.4000	FT.
(RD1055)	◇	GWTT 292 CONF. H-33 ORBITER BSW4V5 (-5,+25,-35)	-5.000	25.000	-35.000	0.000	BREF	3.7800	FT.
							XMRP	1285.0040	IN.
							YMRP	0.0000	IN.
							ZMRP	403.0004	IN.
							SCALE	0.0400	

MACH 0.170

FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1015) GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+30,-30)
 (RD1018) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15,+25,-35)
 (RD1055) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5,+25,-35)

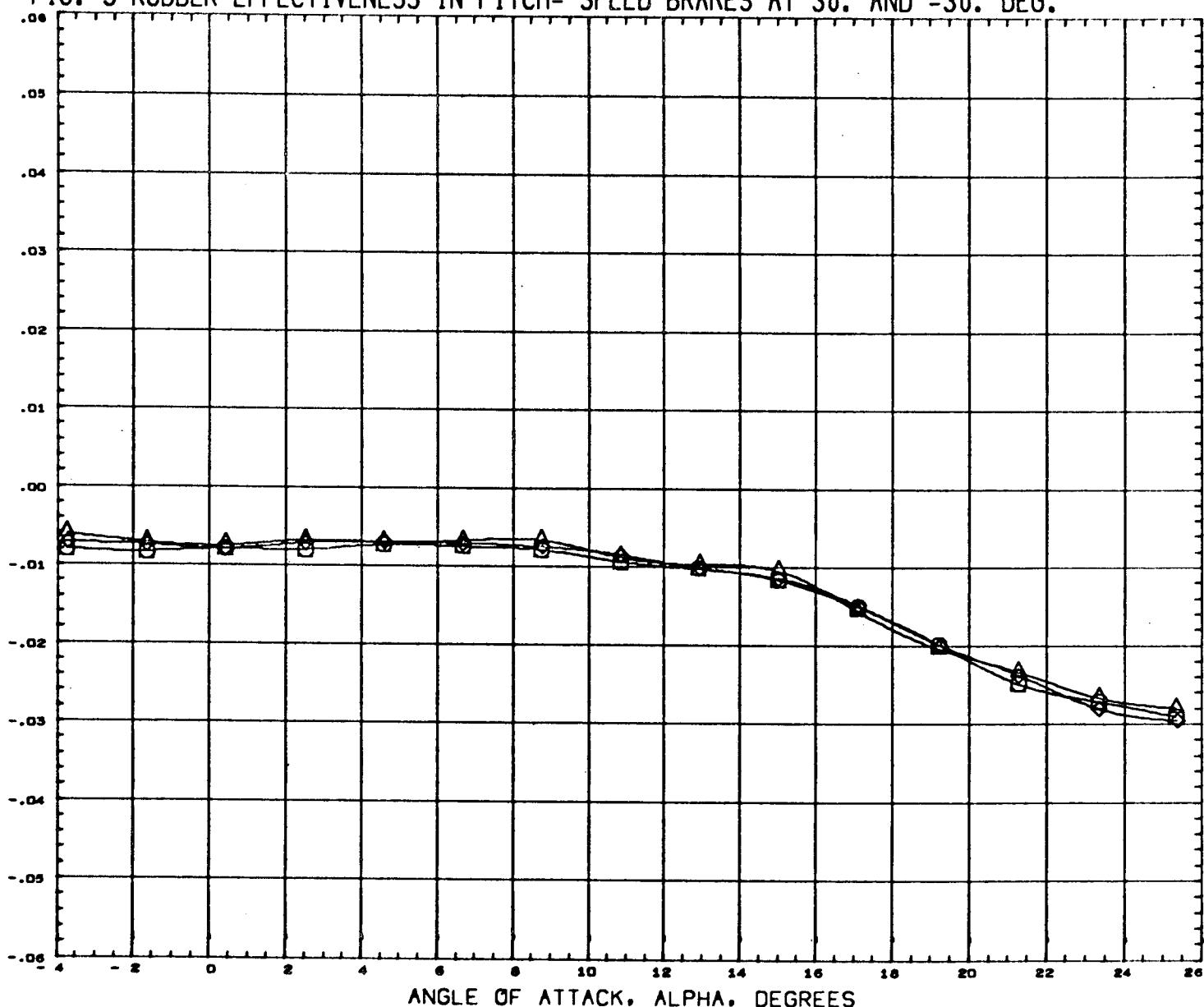
TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
-15.000	25.000	-35.000	0.000	LREF 5.4000 FT.
-5.000	25.000	-35.000		BREF 3.7800 FT.
				XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

PAGE 41

FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.

PITCHING MOMENT COEFFICIENT, CLM



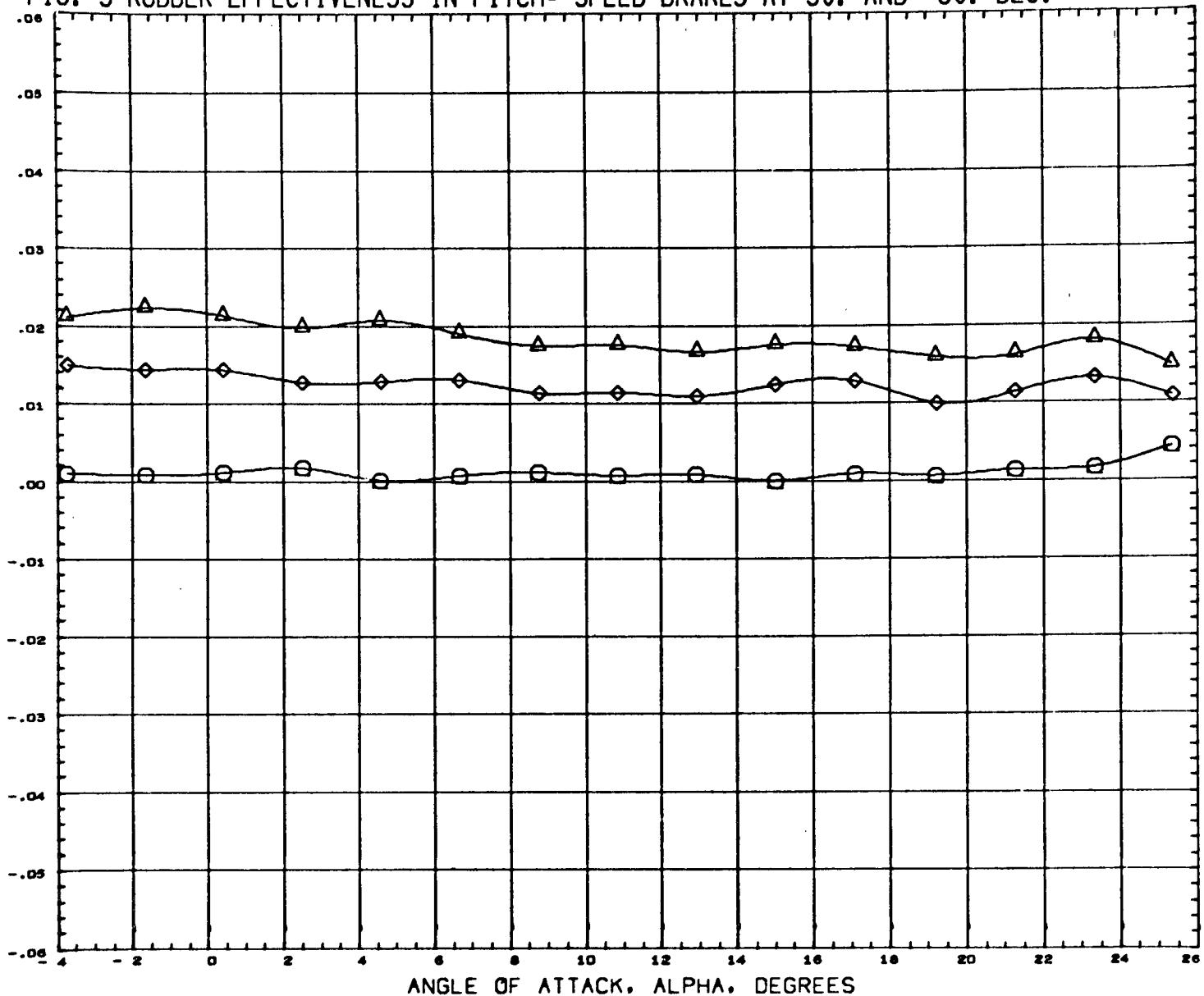
DATA SET SYMBOL		CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE	INFORMATION
(RD1015)	○	GWTT 292 CONF.H-33 CRBITER B5W4V5 (0,+30,-30)	0.000	30.000	-30.000	0.000	SREF	7.7440 SQ FT
(RD1018)	△	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	LREF	5.4000 FT.
(RD1055)	◇	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-5,+25,-35)	-5.000	25.000	-35.000		BREF	3.7800 FT.
							XMRP	1285.0040 IN.
							YMRP	0.0000 IN.
							ZMRP	403.0004 IN.
							SCALE	0.0400

MACH 0.170

PAGE 42

FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.

YAWING MOMENT COEFFICIENT. CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1015) GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+30,-30)
 (RD1018) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15,+25,-35)
 (RD1055) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5,+25,-35)

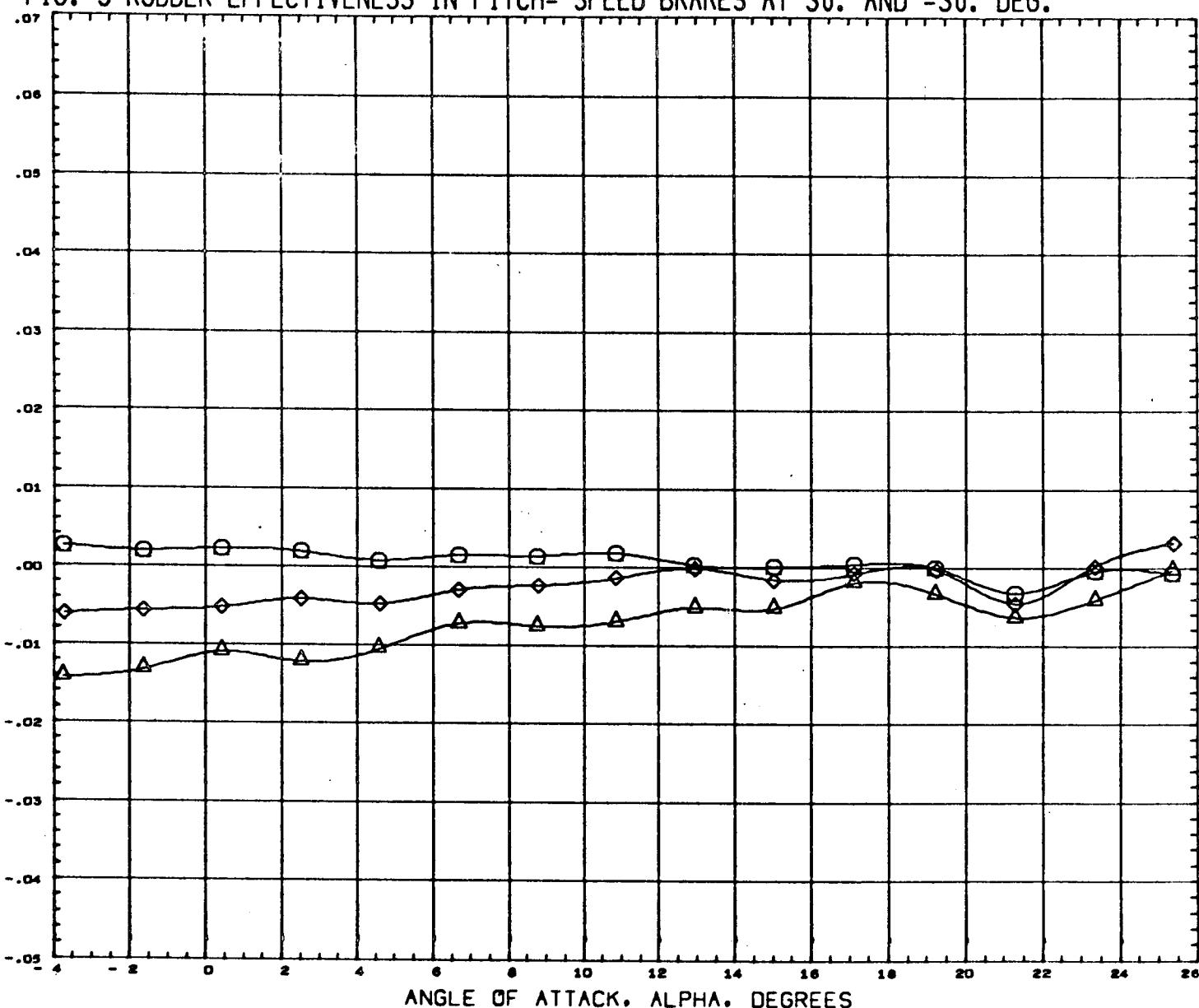
TRUDDR	LLRUDD	LRRUDD	BETA	SREF	7.7440	SQ FT
0.000	30.000	-30.000	0.000	LREF	5.4000	FT.
-15.000	25.000	-35.000	0.000	BREF	3.7800	FT.
-5.000	25.000	-35.000		XMRP	1265.0040	IN.
				YMRP	0.0000	IN.
				ZMRP	403.0004	IN.
				SCALE	0.0400	

MACH 0.170

PAGE 43

FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)

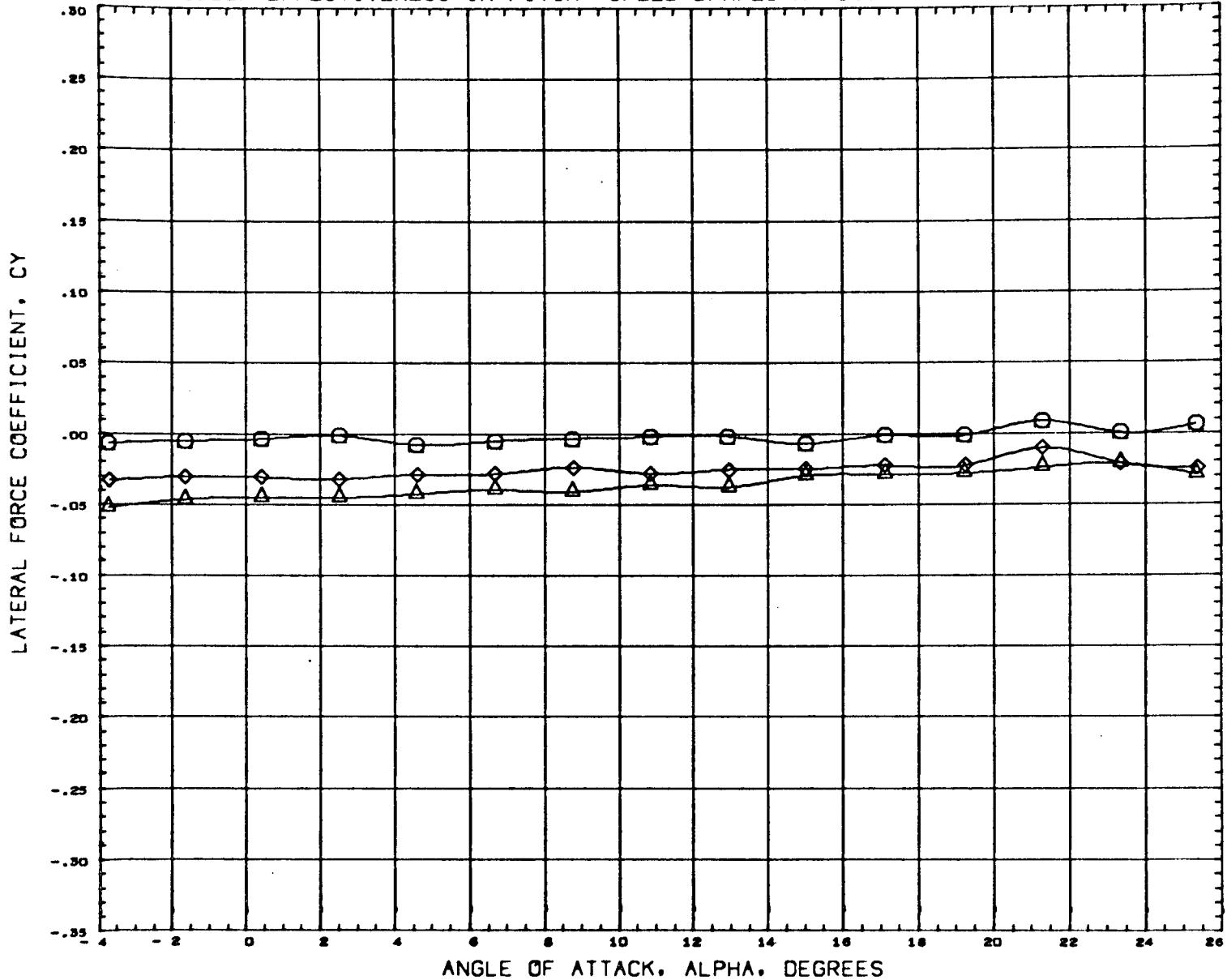


ANGLE OF ATTACK. ALPHA. DEGREES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1015)	GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+30,-30)	0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
(RD1018)	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	LREF 5.4000 FT.
(RD1055)	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5,+25,-35)	-5.000	25.000	-35.000		BREF 3.7800 FT. XMRP 1285.0040 IN. YMRP 0.0000 IN. ZMRP 403.0004 IN. SCALE 0.0400

MACH 0.170

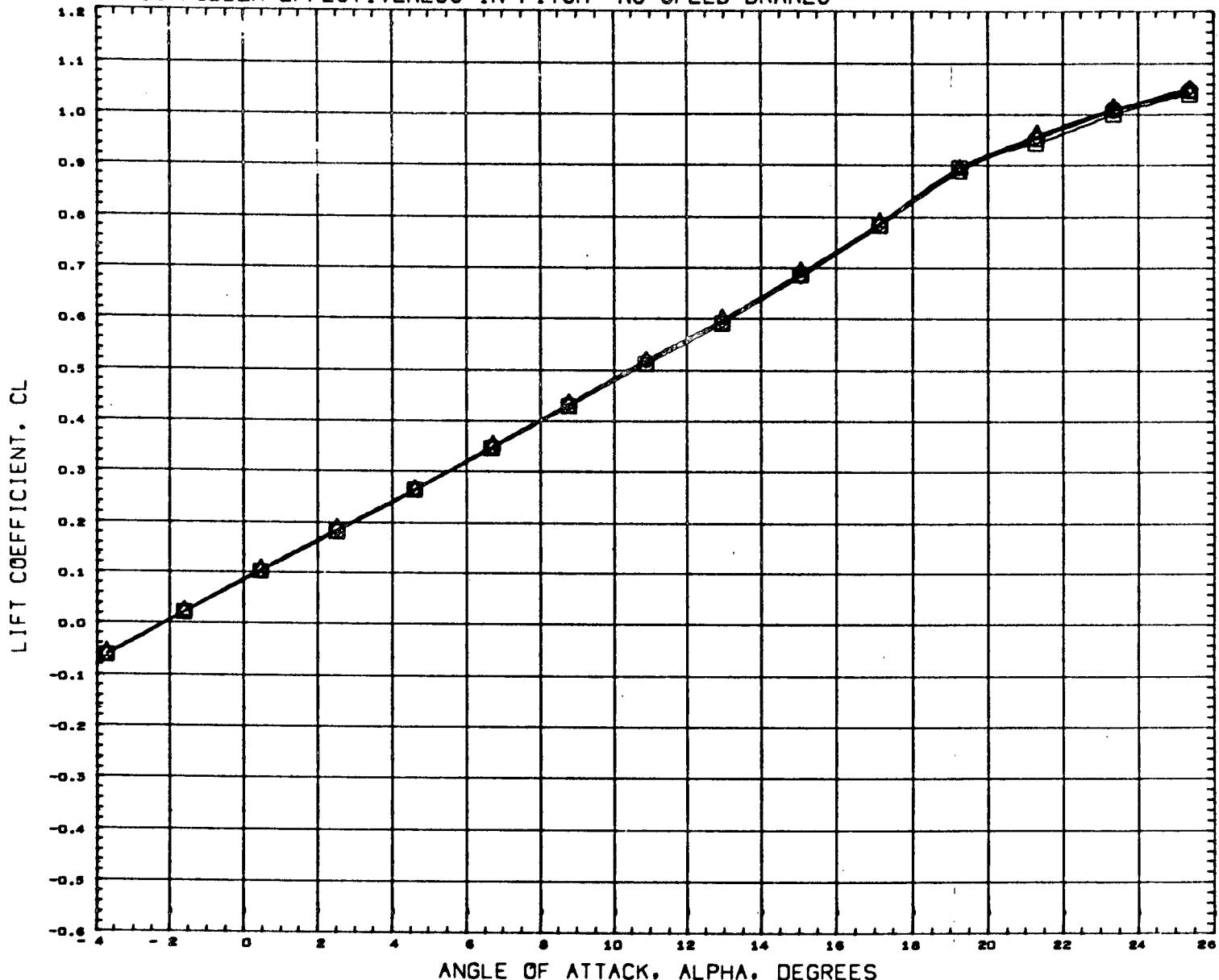
FIG. 9 RUDDER EFFECTIVENESS IN PITCH- SPEED BRAKES AT 30. AND -30. DEG.



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDD	LLRUDD	LRRUDD	BETA	REFERENCE INFORMATION
(RD1015)	GWTT 292 CONF.H-33 ORBITER 85W4V5 (0,+30,-30)	0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
(RD1018)	GWTT 292 CONF.H-33 ORBITER 85W4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	LREF 5.4000 FT.
(RD1055)	GWTT 292 CONF.H-33 ORBITER 85W4V5 (-5,+25,-35)	-5.000	25.000	-35.000	0.000	BREF 3.7800 FT.
					XMRP 1285.0040 IN.	
					YMRP 0.0000 IN.	
					ZMRP 403.0004 IN.	
					SCALE 0.0400	

MACH 0.170

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) Q GWTT 292 CONF.H-33 CRBITER BSW4V5
 (RD1019) □ GWTT 292 CONF.H-33 CRBITER BSW4V5 (-5)
 (RD1020) ◇ GWTT 292 CONF.H-33 CRBITER BSW4V5 (-10)
 (RD1021) ▨ GWTT 292 CONF.H-33 CRBITER BSW4V5 (-15)

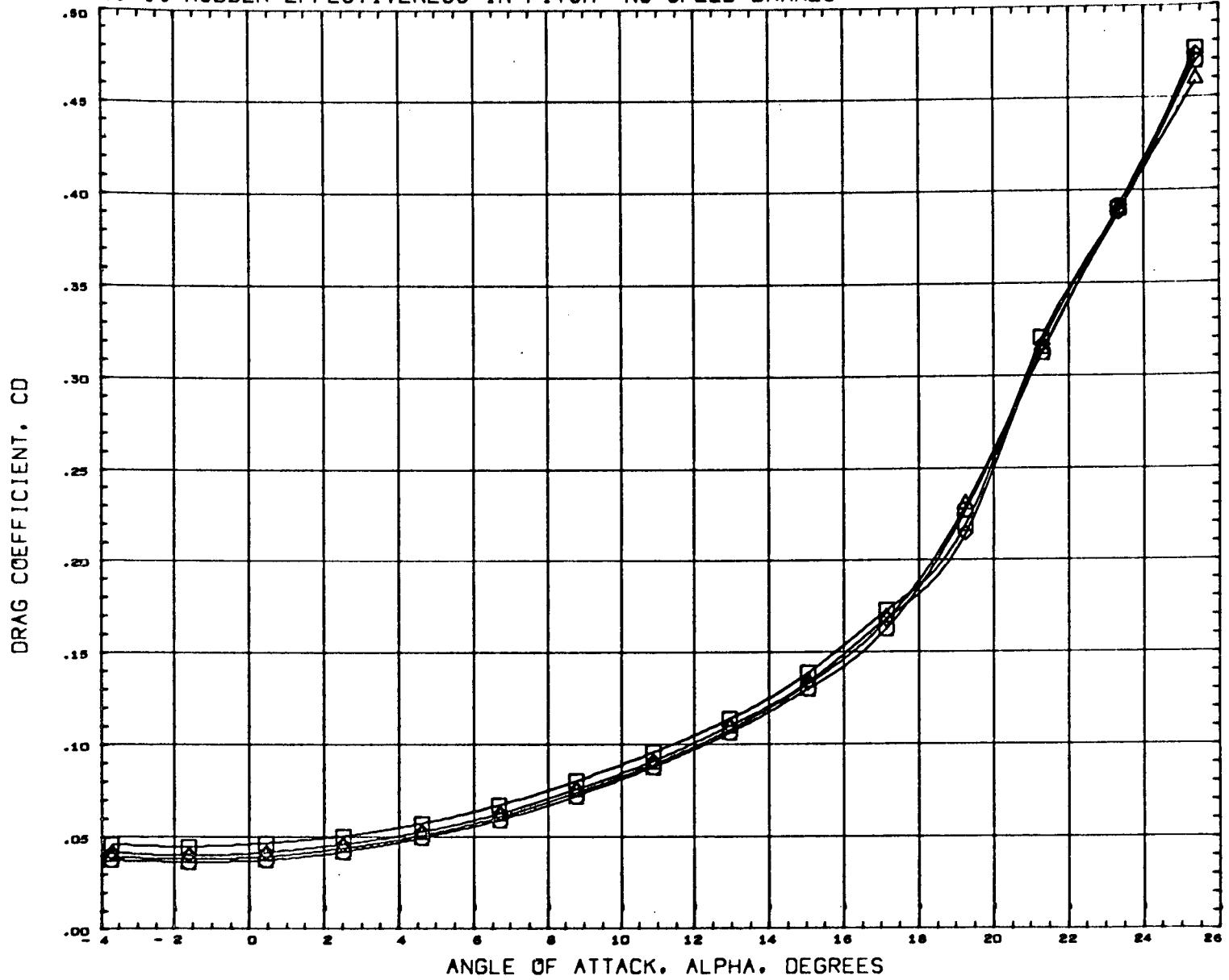
RUDDER BETA
 0.000 0.000
 -5.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 BREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 46

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1002) GWT T 292 CONF.H-33 CRBITER B5W4V5
 (RD1019) GWT T 292 CONF.H-33 CRBITER B5W4V5 (-5)
 (RD1020) GWT T 292 CONF.H-33 CRBITER B5W4V5 (-10)
 (RD1021) GWT T 292 CONF.H-33 CRBITER B5W4V5 (-15)

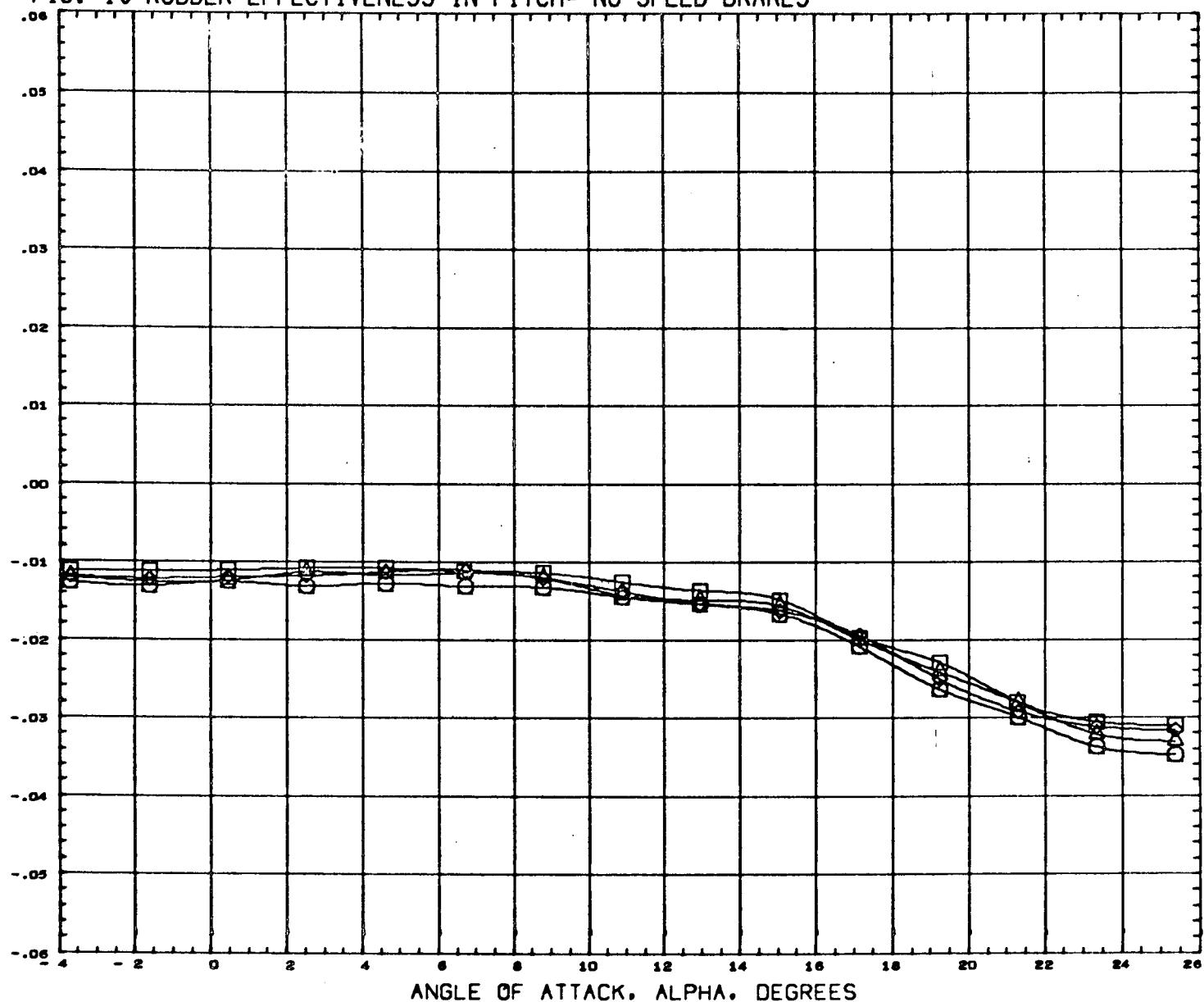
RUDDER BETA
 0.000 0.000
 -5.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES

PITCHING MOMENT COEFFICIENT, CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION

- (RD1002) GWTT 292 CONF.H-33 ORBITER BSW4V5
- (RD1019) GWTT 292 CONF.H-33 CRBITER BSW4V5 (-5)
- (RD1020) GWTT 292 CONF.H-33 CRBITER BSW4V5 (-10)
- (RD1021) GWTT 292 CONF.H-33 CRBITER BSW4V5 (-15)

RUDDER BETA
0.000 0.000
-5.000 0.000
-10.000 0.000
-15.000 0.000

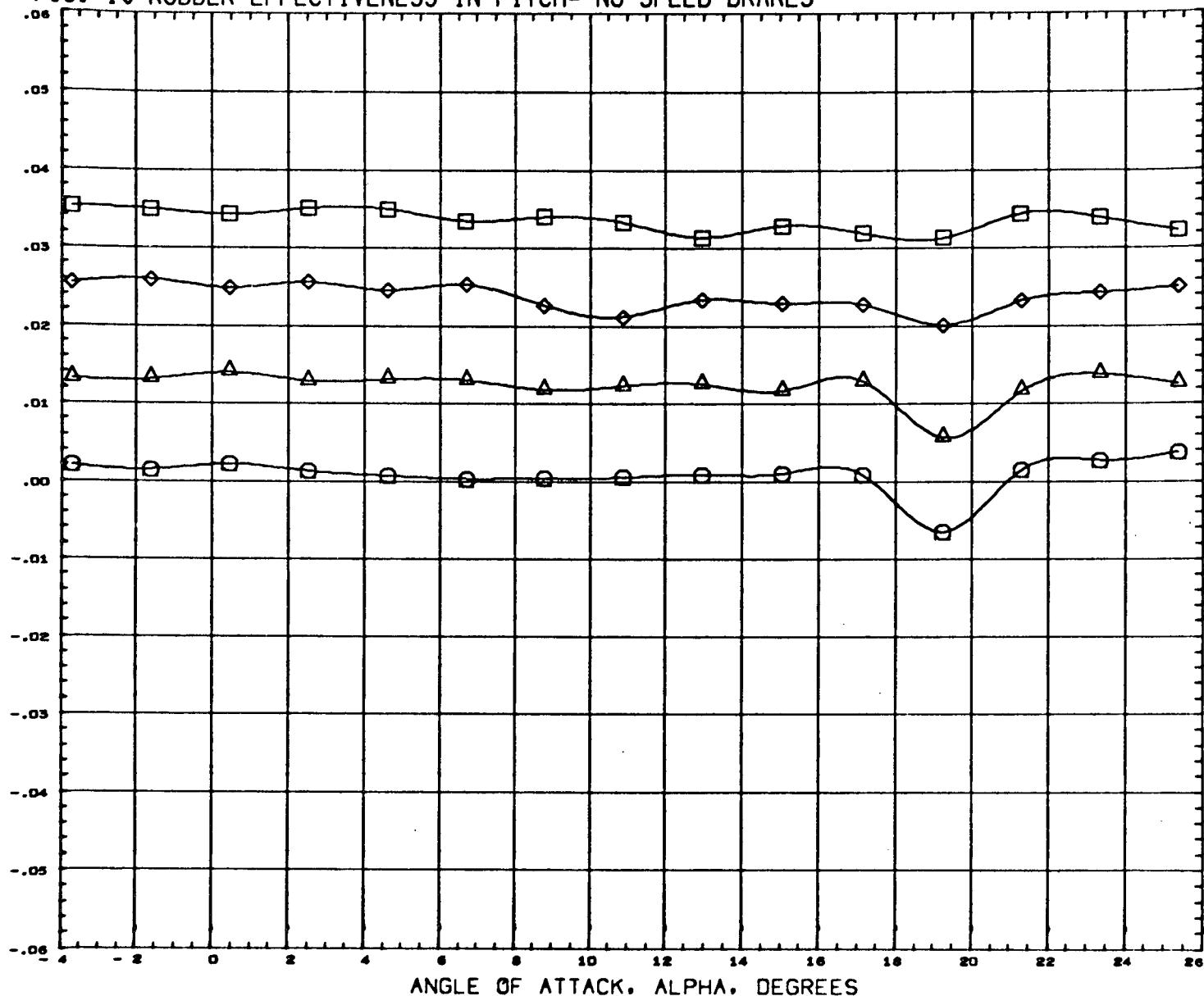
REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7800 FT.
XMRP 1285.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

MACH 0.170

PAGE 48

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES

YAWING MOMENT COEFFICIENT. CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1002)	○	GWTT 292 CONF.H-33 ORBITER B5W4V5
(RD1019)	△	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5)
(RD1020)	◇	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-10)
(RD1021)	□	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15)

RUDDER BETA

0.000	0.000
-5.000	0.000
-10.000	0.000
-15.000	0.000

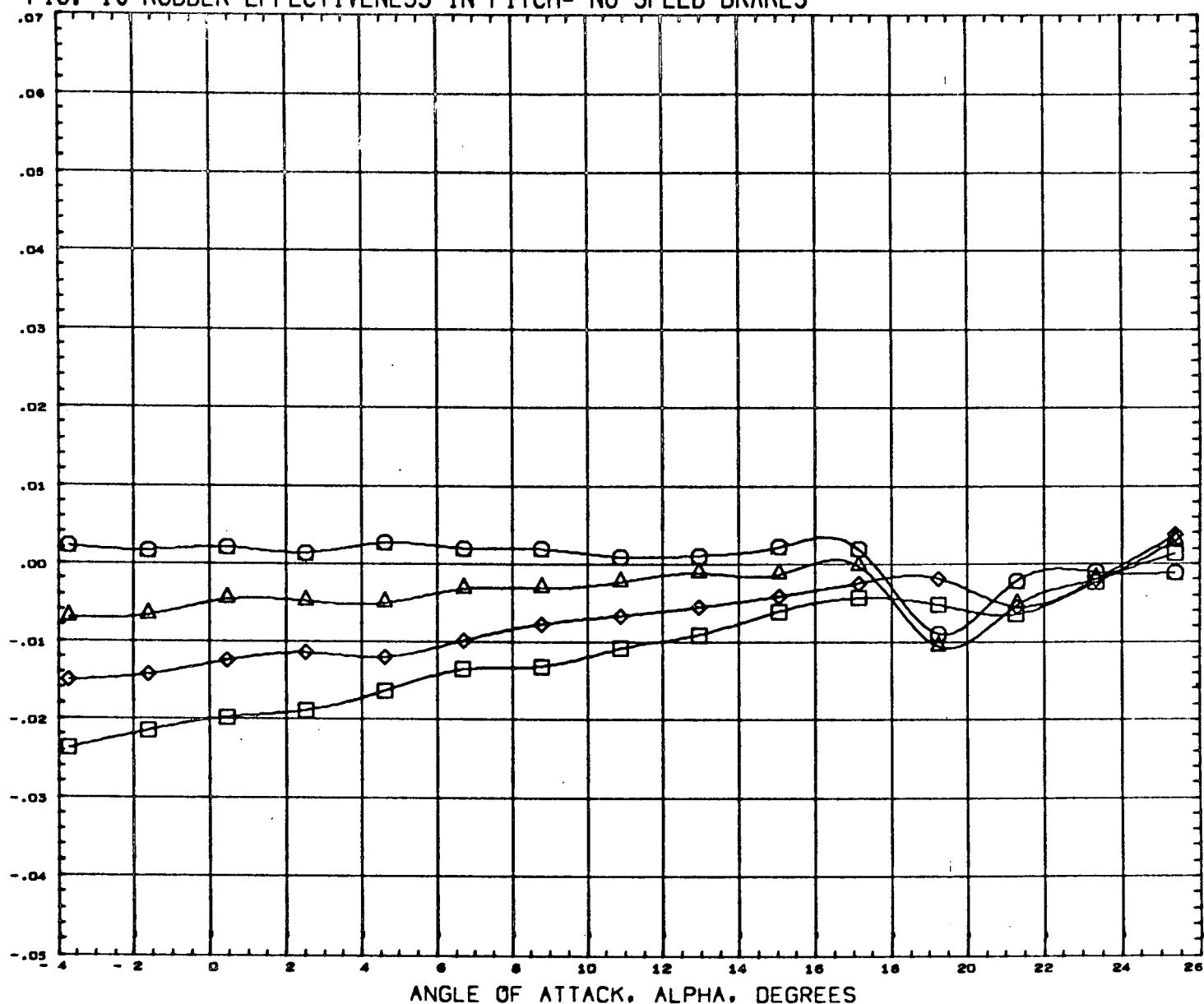
REFERENCE INFORMATION

SREF	7.7440	SQ FT
LREF	5.4000	FT.
BREF	3.7800	FT.
XMRP	1285.0040	IN.
YMRP	0.0000	IN.
ZMRP	403.0004	IN.
SCALE	0.0400	

MACH 0.170

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RD1002)	GWTI 292 CONF.H-33 ORBITER 85W4V5
(RD1019)	GWTI 292 CONF.H-33 ORBITER 85W4V5 (-5)
(RD1020)	GWTI 292 CONF.H-33 ORBITER 85W4V5 (-10)
(RD1021)	GWTI 292 CONF.H-33 ORBITER 85W4V5 (-15)

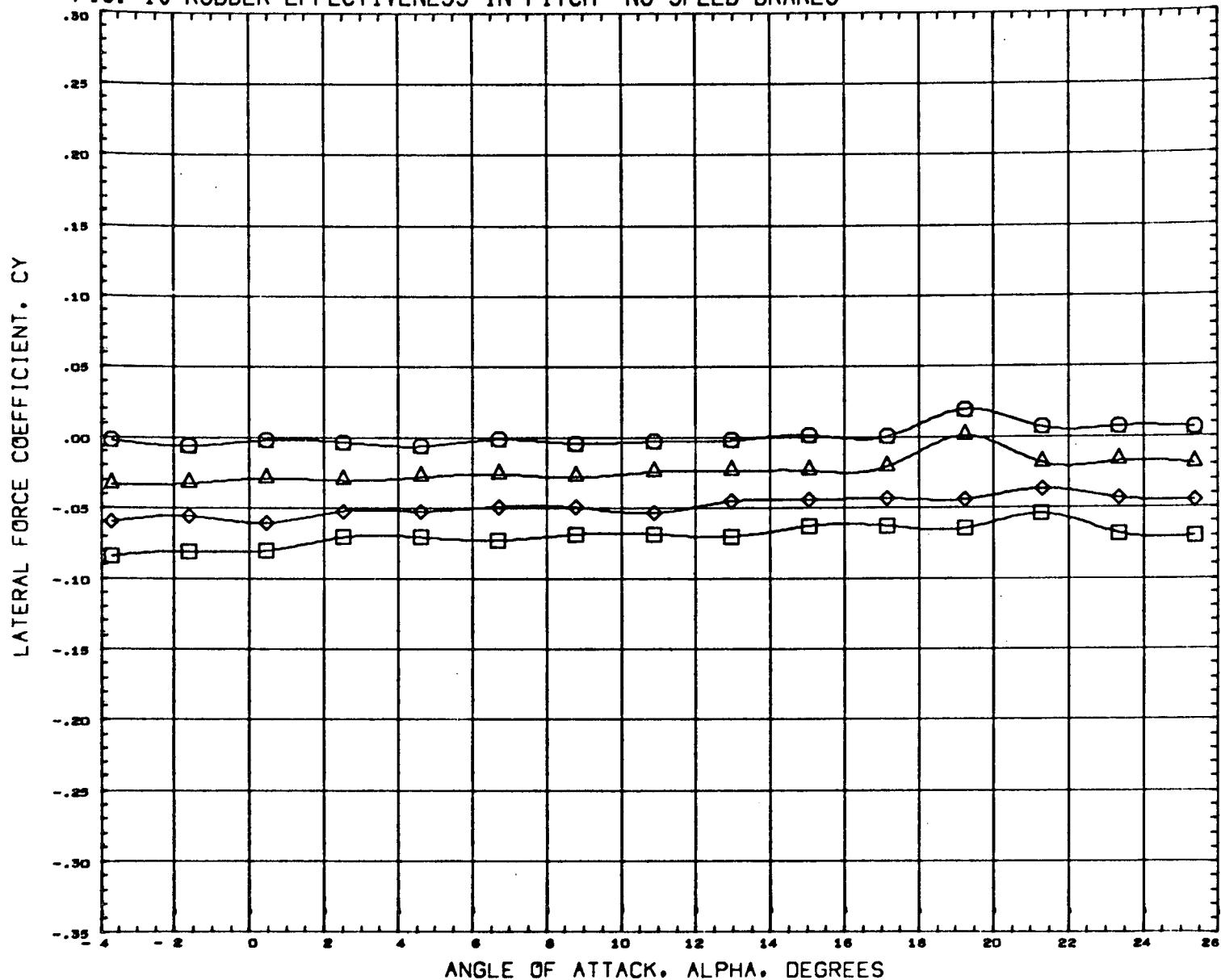
RUDDER	BETA
0.000	0.000
-5.000	0.000
-10.000	0.000
-15.000	0.000

REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7800 FT.
XMRP 1265.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

MACH 0.170

PAGE 50

FIG. 10 RUDDER EFFECTIVENESS IN PITCH- NO SPEED BRAKES



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1002)	○	GWTT 292 CONF.H-33 ORBITER B5W4V5
(RD1019)	△	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5)
(RD1020)	◇	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-10)
(RD1021)	□	GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15)

RUDDER BETA

0.000	0.000
-5.000	0.000
-10.000	0.000
-15.000	0.000

REFERENCE INFORMATION

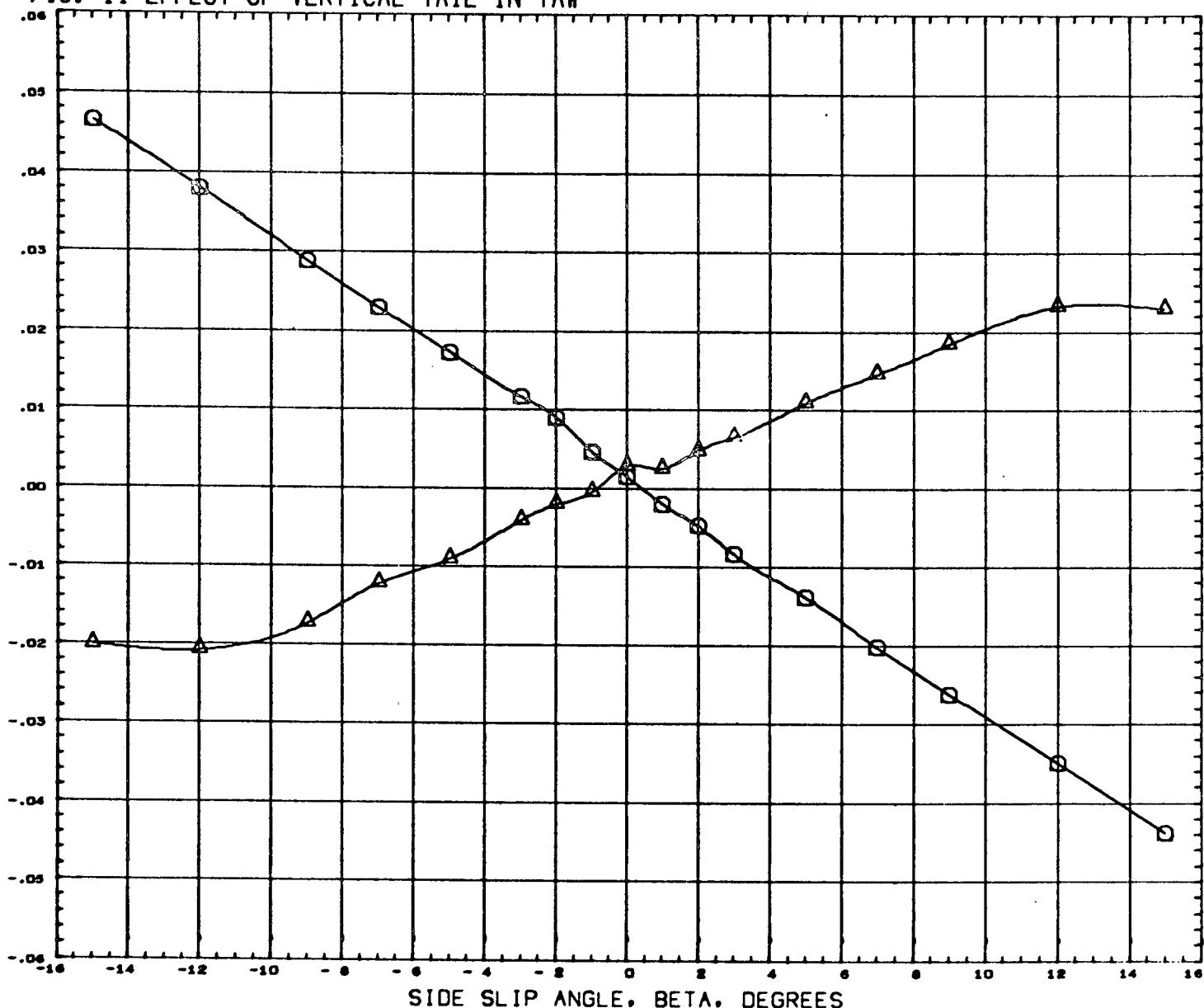
SREF	7.7440	SQ FT
LREF	5.4000	FT.
BREF	3.7600	FT.
XMRP	1285.0040	IN.
YMRP	0.0000	IN.
ZMRP	403.0004	IN.
SCALE	0.0400	

MACH 0.170

PAGE 51

FIG. 11 EFFECT OF VERTICAL TAIL IN YAW

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1025) GWT 292 CONF.H-33 ORBITER B5W4
 (RD1027) GWT 292 CONF.H-33 ORBITER B5W4V5

ALPHA
0.000
0.000

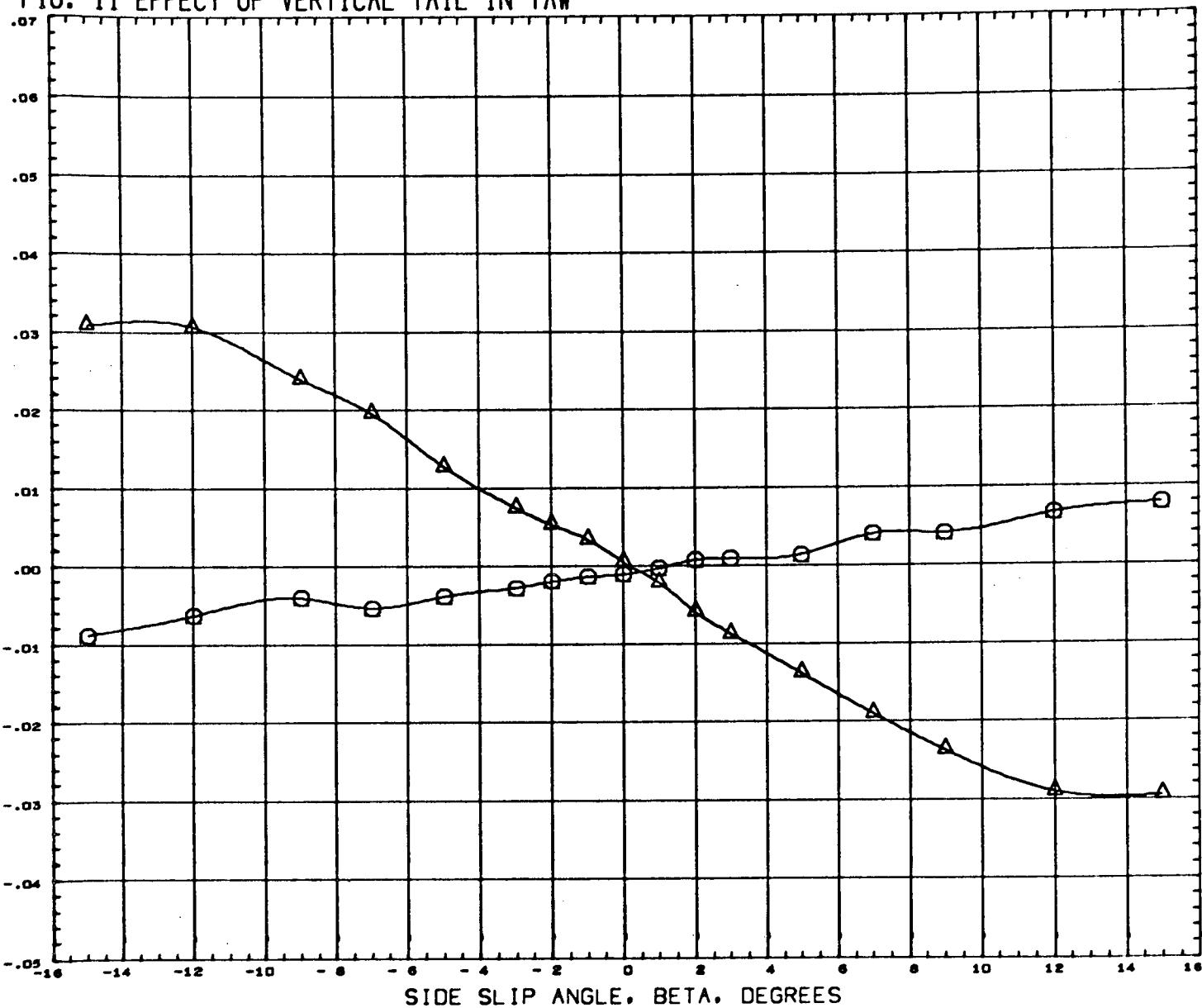
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1205.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 52

FIG. 11 EFFECT OF VERTICAL TAIL IN YAW

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1025) GWTT 292 CONF.H-33 ORBITER BSW4
 (RD1027) GWTT 292 CONF.H-33 ORBITER BSW4VS

ALPHA
 0.000
 0.000

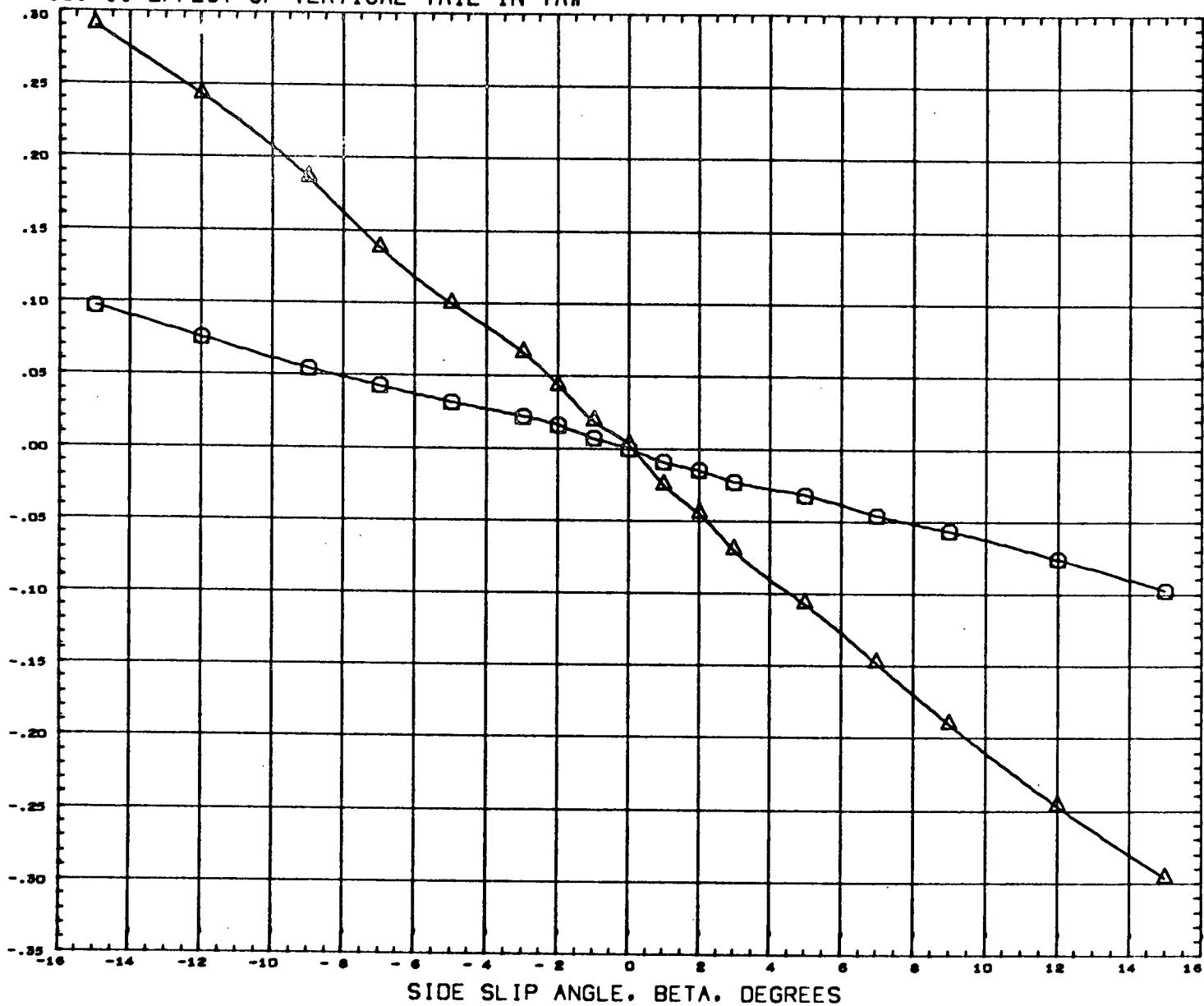
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 53

FIG. 11 EFFECT OF VERTICAL TAIL IN YAW

LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1025) GWTT 292 CONF.H-33 ORBITER B5W4
 (RD1027) GWTT 292 CONF.H-33 CRBITER B5W4V5

ALPHA
 0.000
 0.000

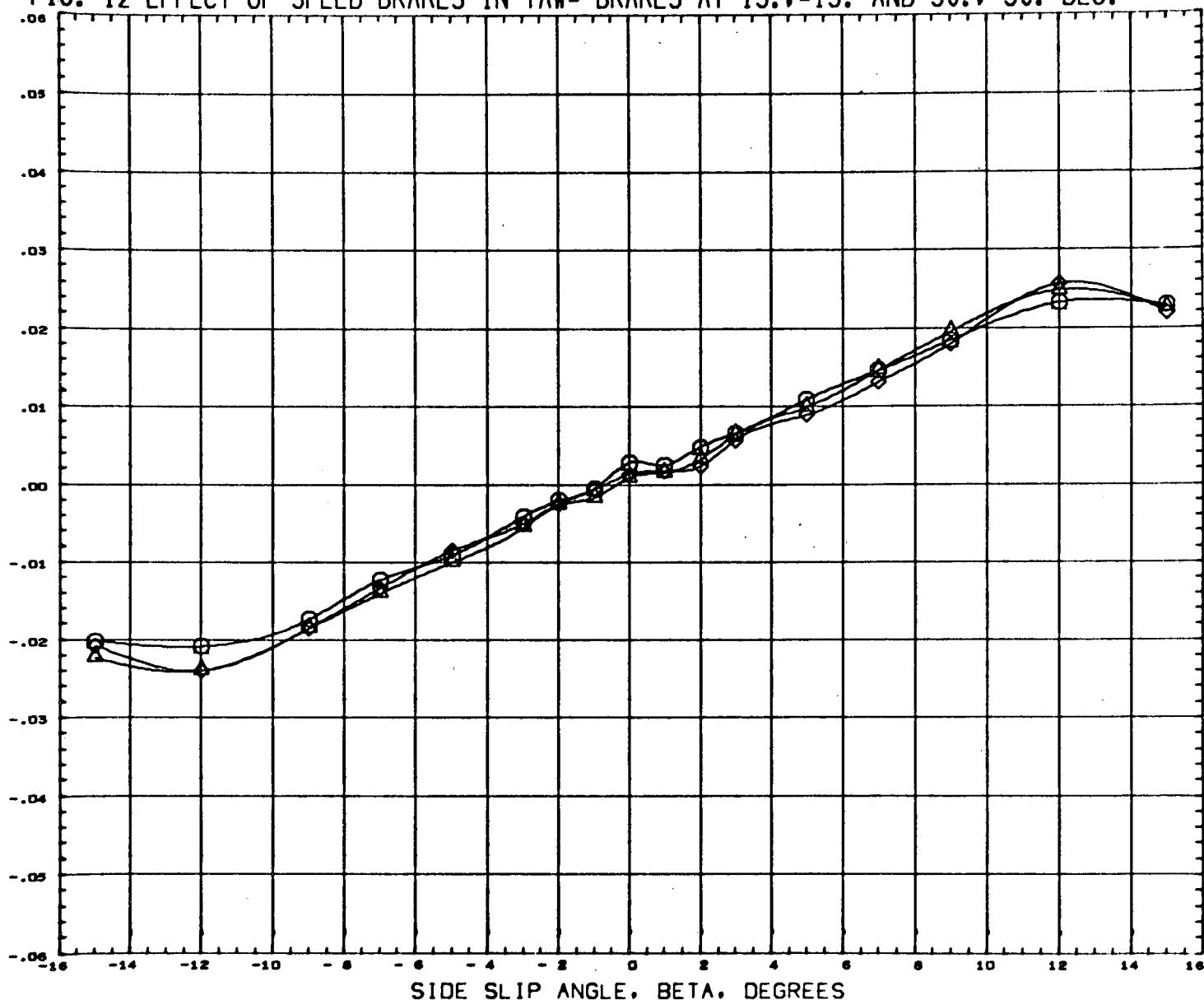
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 54

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15.,-15. AND 30.,-30. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1039) GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+15,-15)
 (RD1034) GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+30,-30)

ALPHA
0.000
0.000
0.000

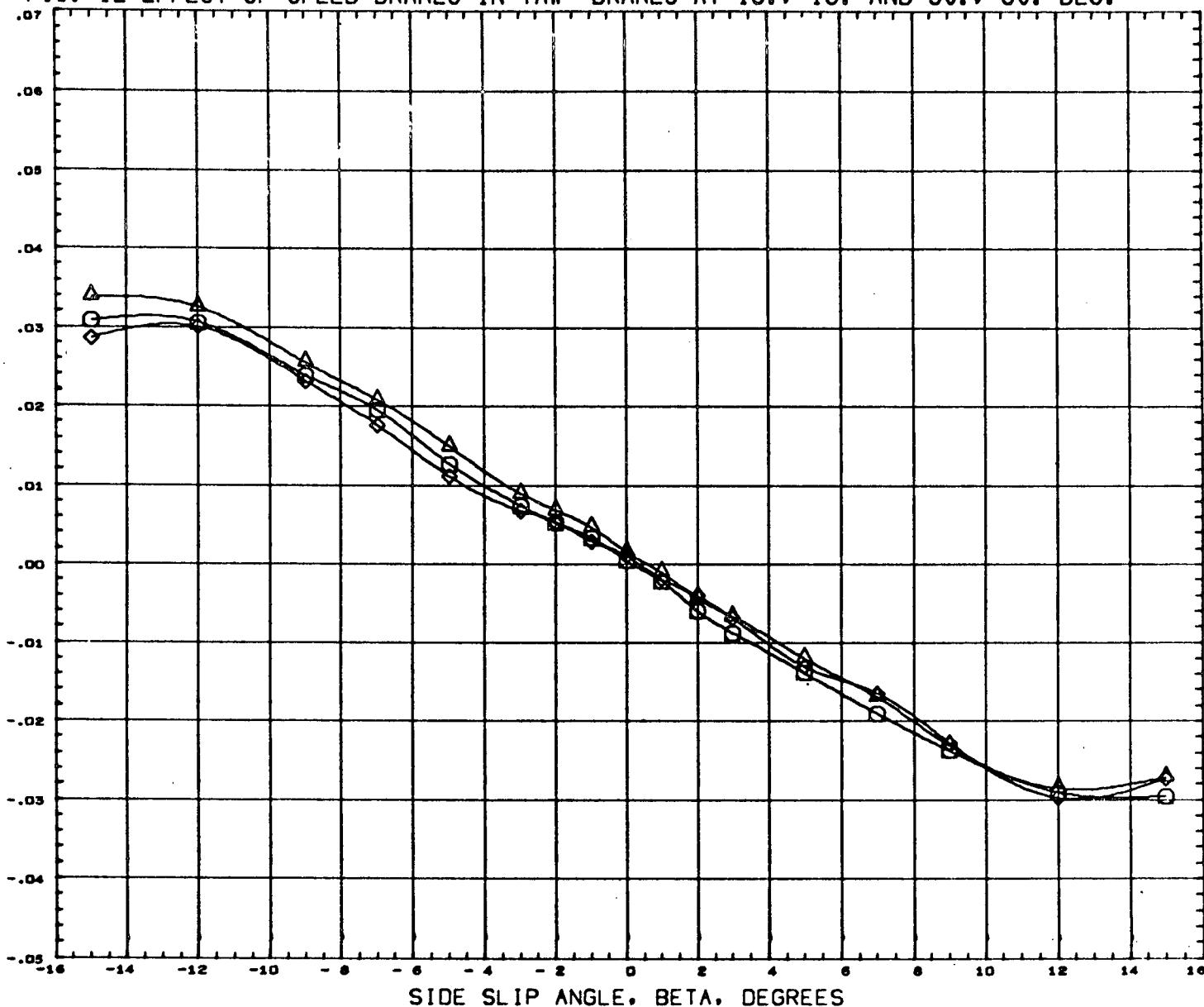
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 55

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15., -15, AND 30., -30. DEG.

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)

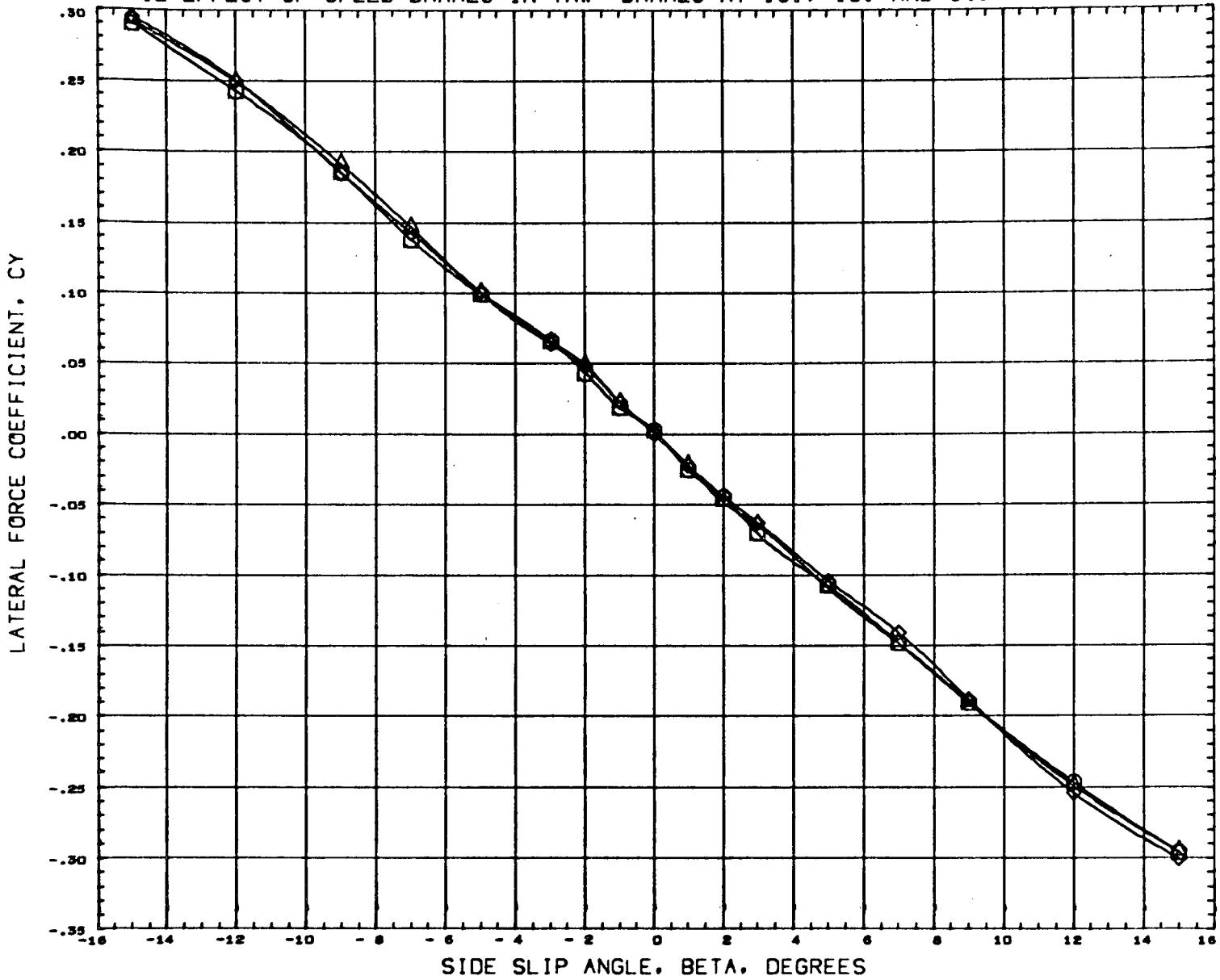


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ALPHA
(RD1027)	GWT 292 CNF.H-33 ORBITER B5W4V5	0.000
(RD1039)	GWT 292 CNF.H-33 ORBITER B5W4V5(0,+15,-15)	0.000
(RD1034)	GWT 292 CNF.H-33 ORBITER B5W4V5(0,+30,-30)	0.000

MACH 0.170

REFERENCE INFORMATION		
SREF	7.7440	SQ FT
LREF	5.4000	FT.
BREF	3.7800	FT.
XMRP	1285.0040	IN.
YMRP	0.0000	IN.
ZMRP	403.0004	IN.
SCALE	0.0400	

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15., -15. AND 30., -30. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWT2 292 CONF.M-33 ORBITER BSW4V5
 (RD1039) GWT2 292 CONF.M-33 ORBITER BSW4V5 (0,+15,-15)
 (RD1034) GWT2 292 CONF.M-33 ORBITER BSW4V5 (0,+30,-30)

ALPHA
0.000
0.000
0.000

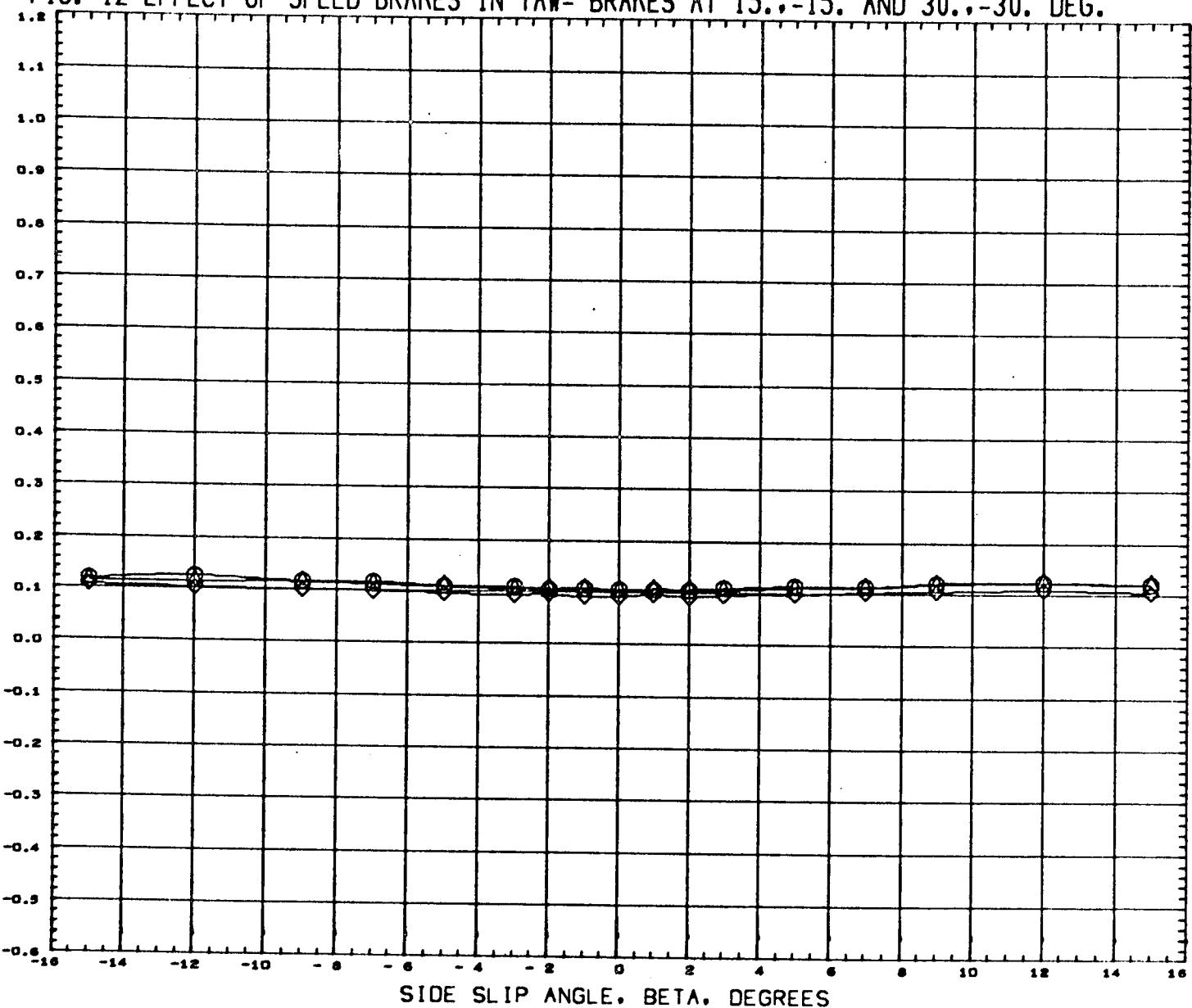
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMNP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMNP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 57

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15.,-15. AND 30.,-30. DEG.

LIFT COEFFICIENT. CL



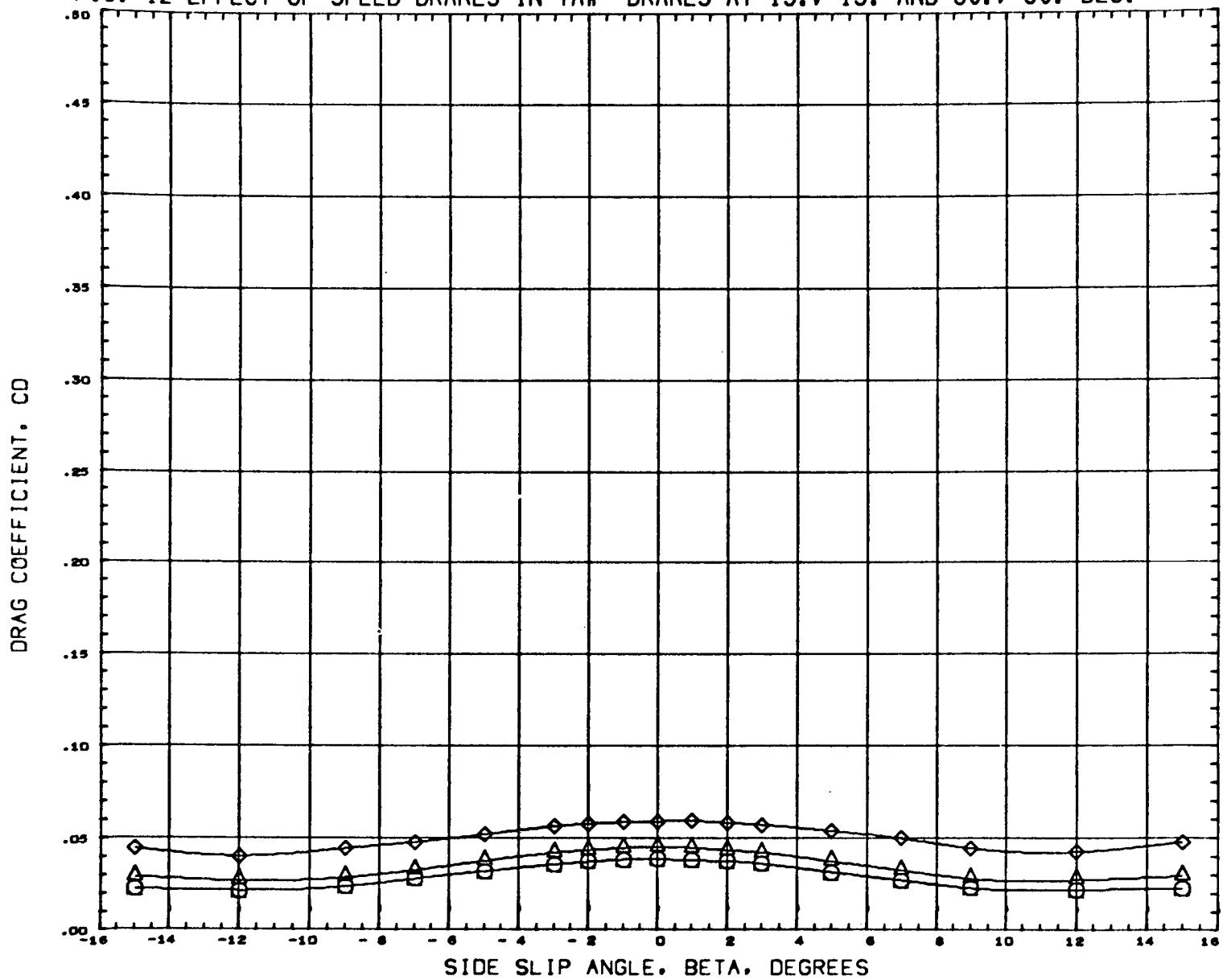
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) Q GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1039) C GWTT 292 CONF.H-33 ORBITER B5W4V5(0,+15,-15)
 (RD1034) D GWTT 292 CONF.H-33 ORBITER B5W4V5(0,+30,-30)

ALPHA
0.000
0.000
0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15., -15. AND 30., -30. DEG.



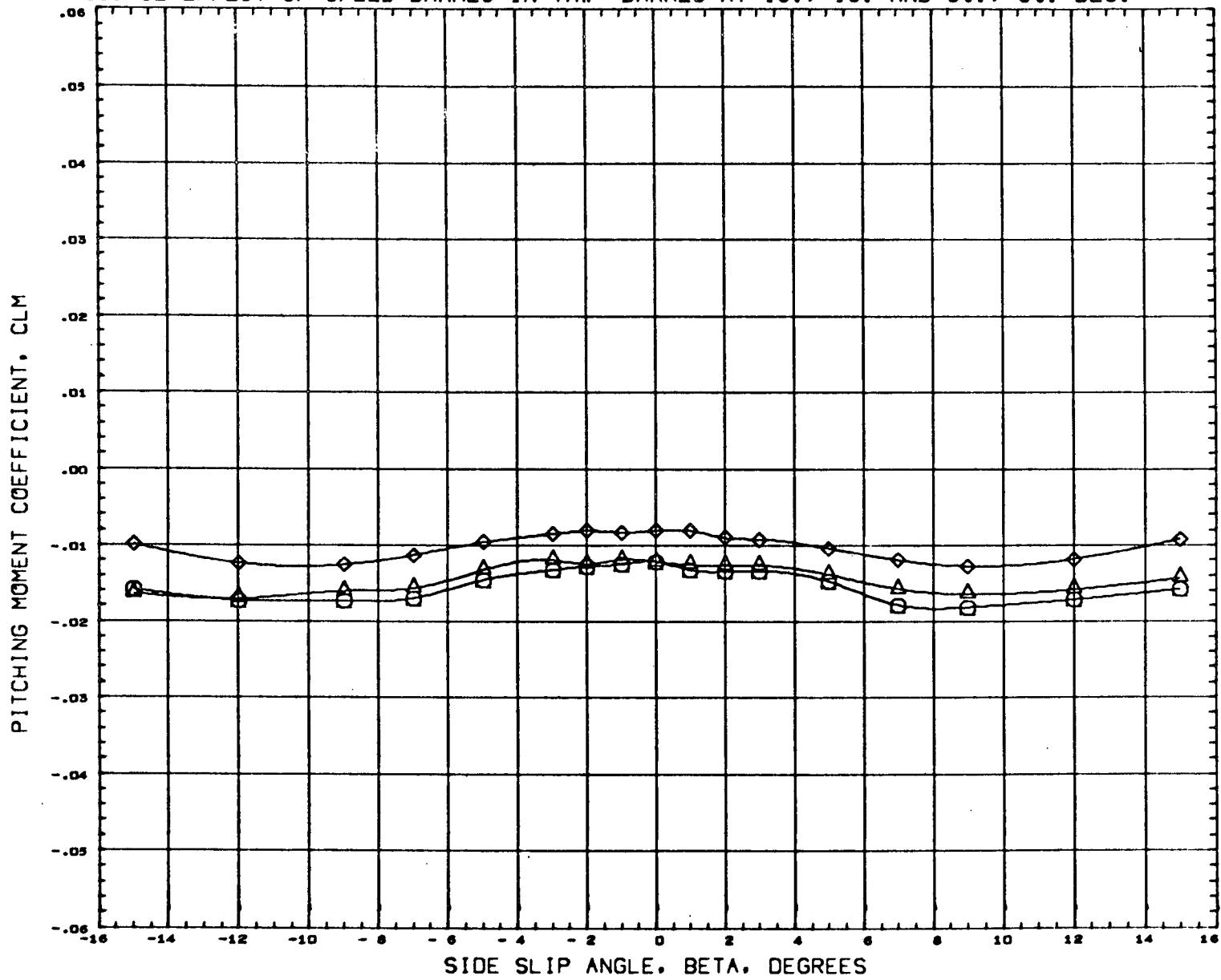
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.H-33 ORBITER BSW4V5
 (RD1039) GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+15,-15)
 (RD1034) GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+30,-30)

ALPHA
 0.000
 0.000
 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 12 EFFECT OF SPEED BRAKES IN YAW- BRAKES AT 15., -15. AND 30., -30. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.H-33 ORBITER BSW4V5
 (RD1039) GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+15,-15)
 (RD1034) GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+30,-30)

ALPHA:
 0.000
 0.000
 0.000

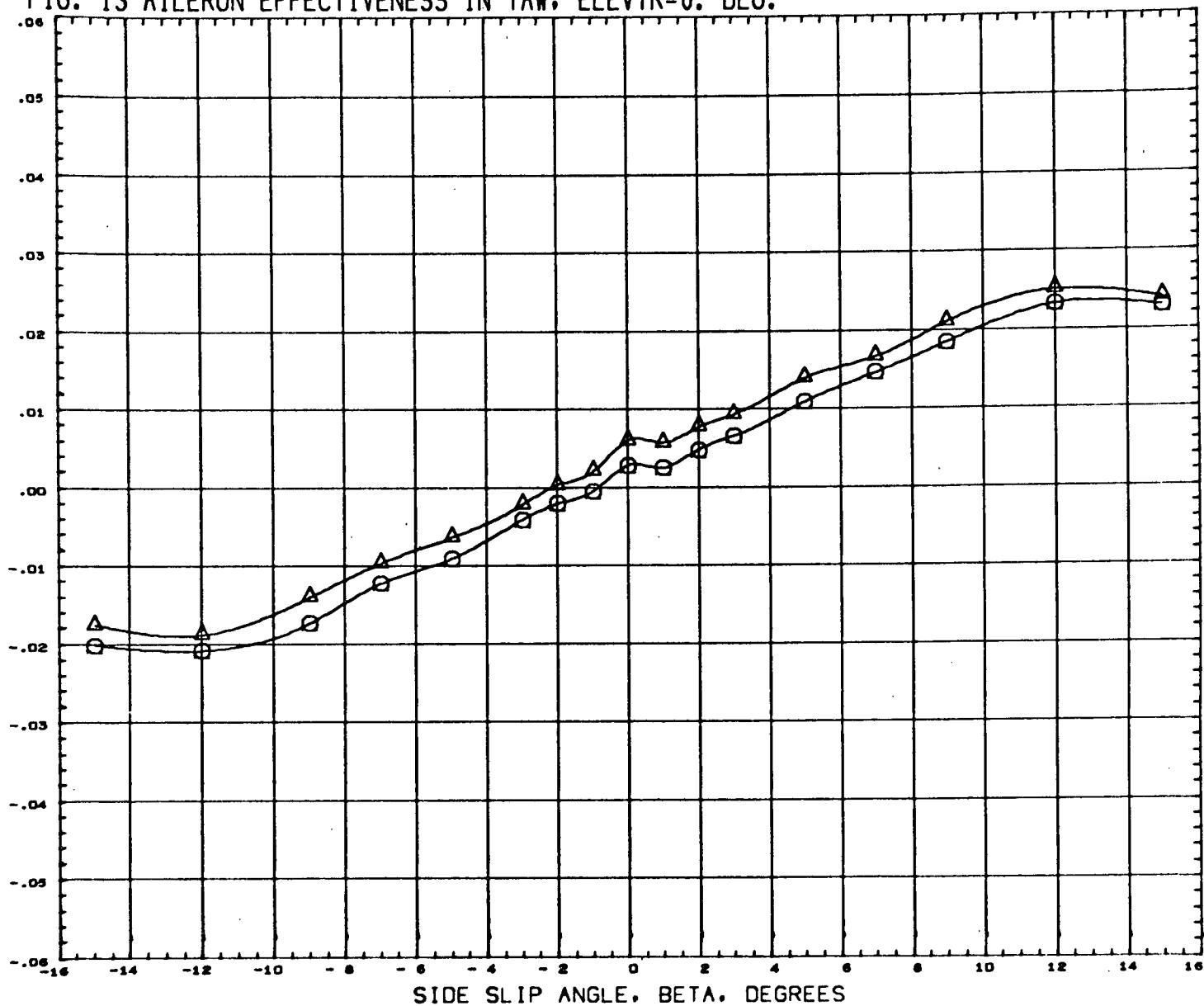
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 60

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.M-33 ORBITER B5W4V5
 (RD1032) GWTT 292 CONF.M-33 ORBITER B5W4 (+5,-5)V5

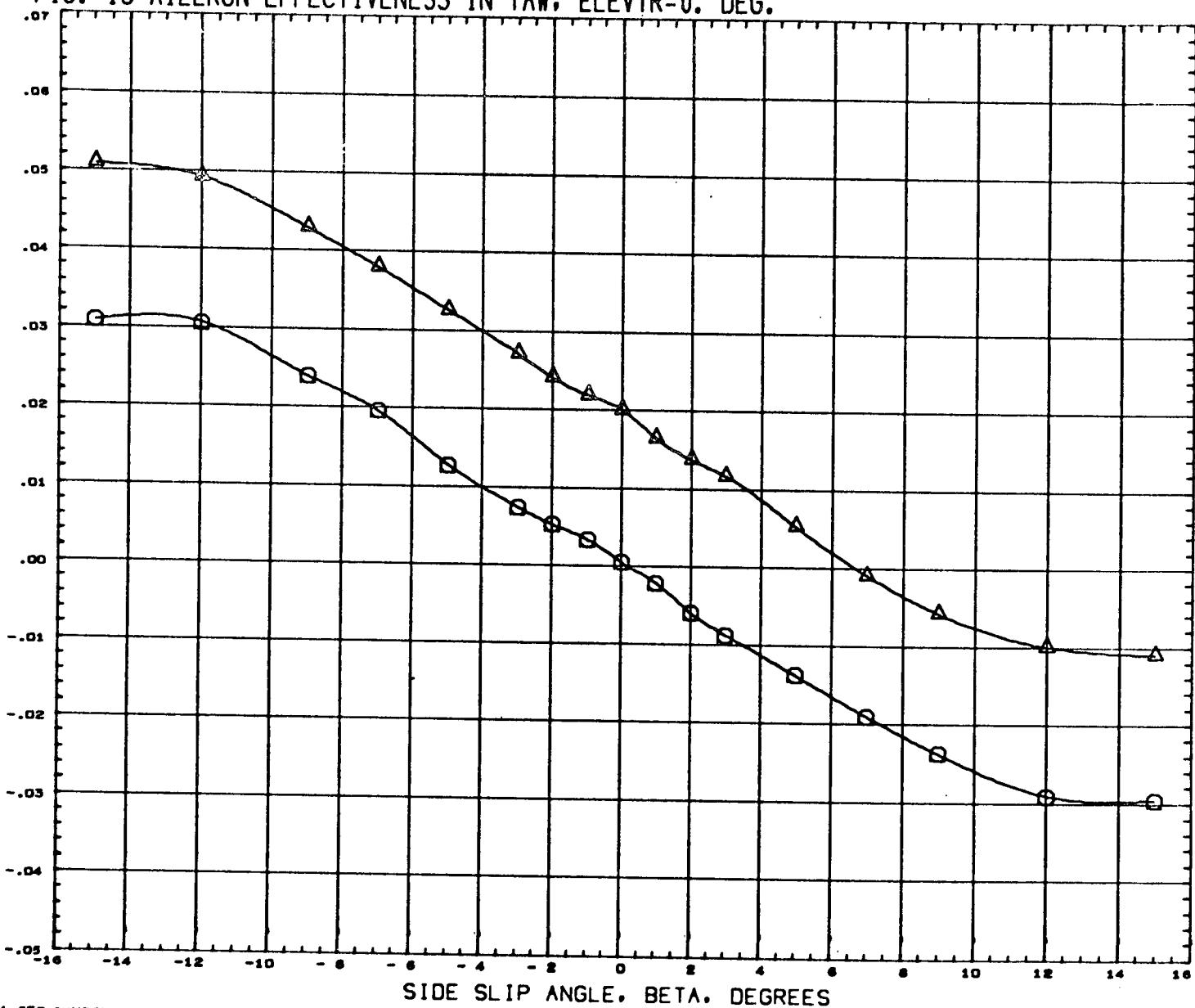
LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1027) GWT 292 CONF.H-33 CRBITER B5W4V5
 (RD1032) GWT 292 CONF.H-33 CRBITER B5W4 (+5,-5)V5

LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

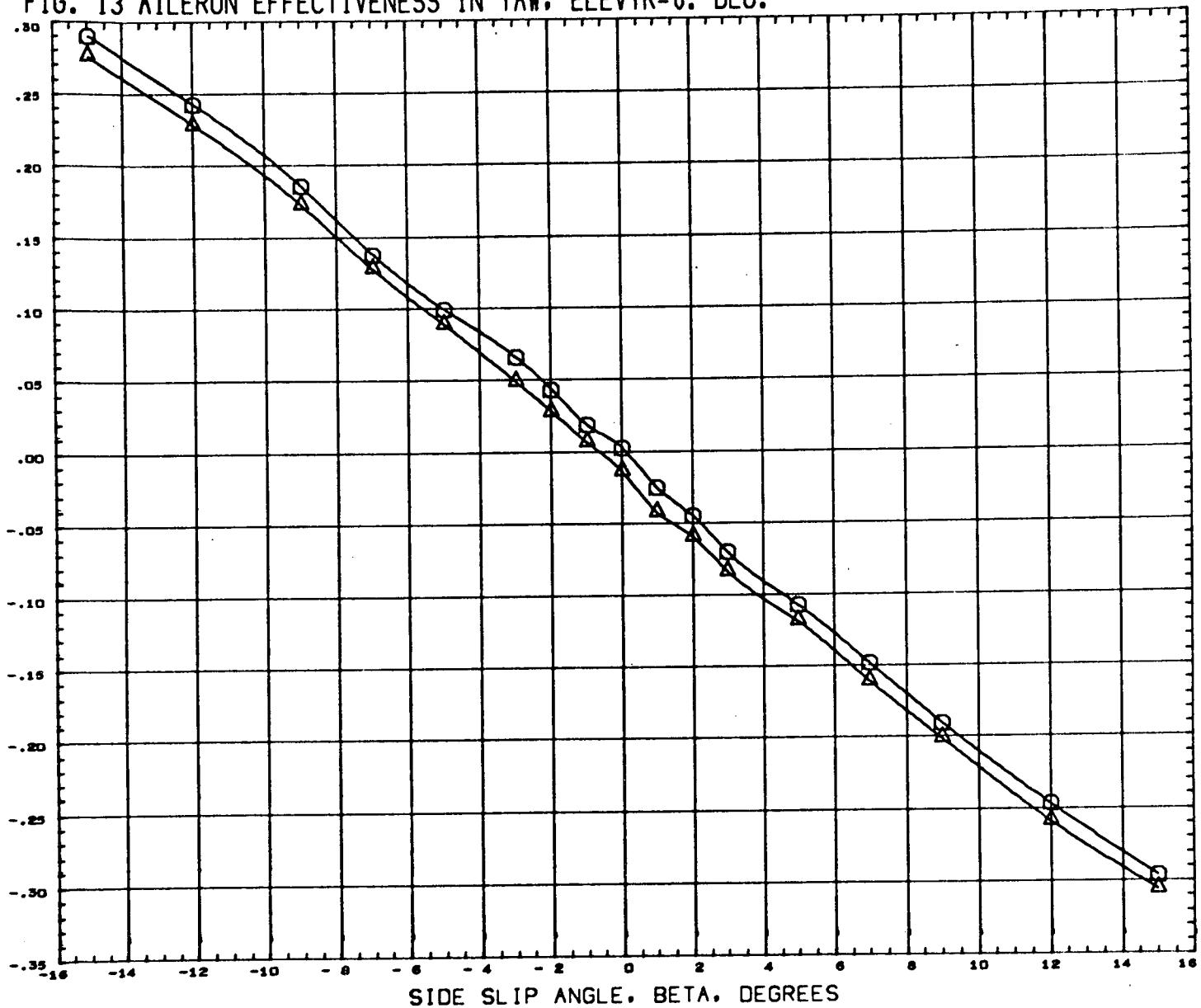
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 62

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0. DEG.

LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWT2 292 CONF.H-33 ORBITER B5W4V5
 (RD1032) GWT2 292 CONF.H-33 ORBITER B5W4(+5,-5)V5

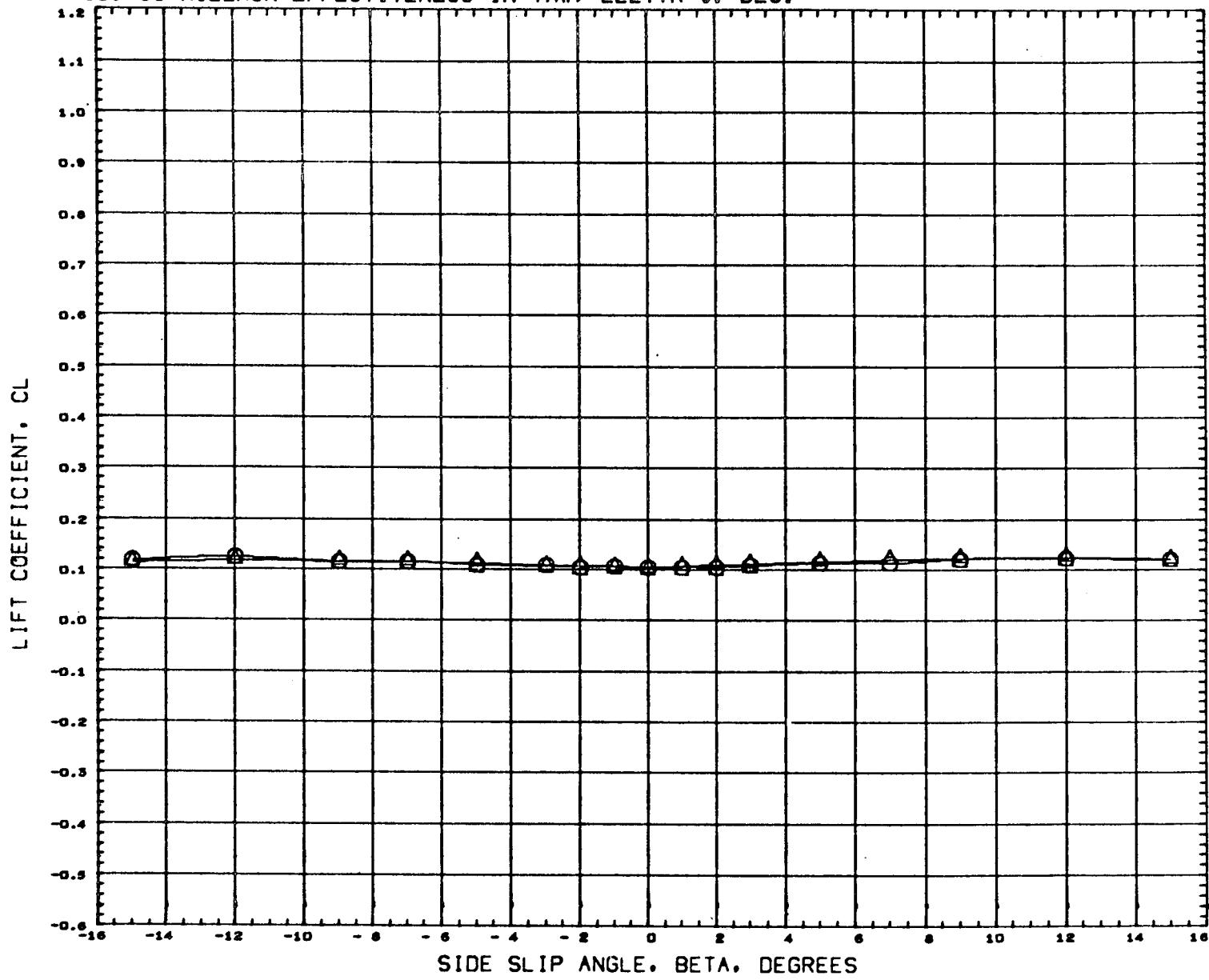
LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 63

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0, DEG.



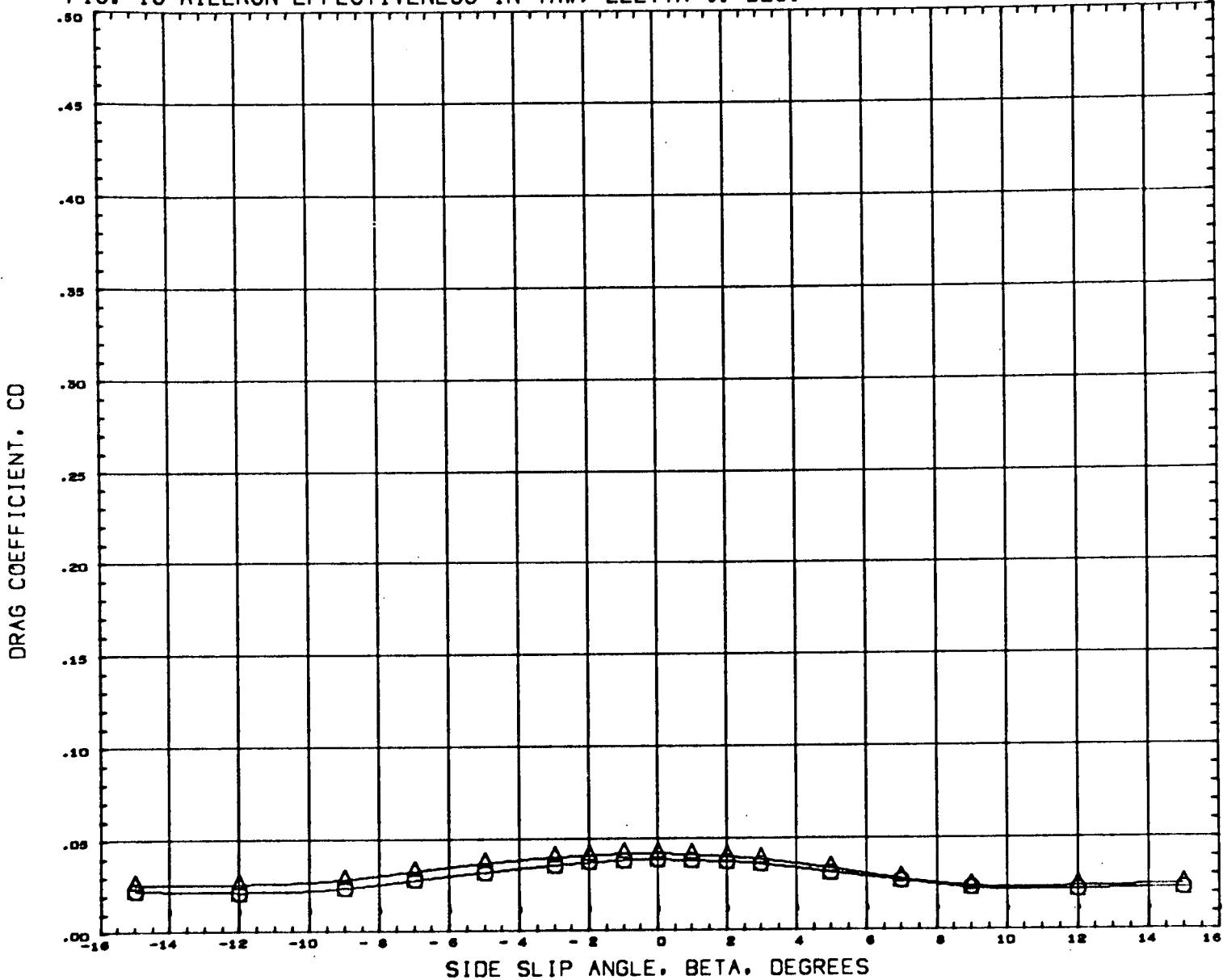
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) Q GWTT 292 CONF.H-33 ORBITER BSW4V5
 (RD1032) Q GWTT 292 CONF.H-33 ORBITER BSW4(+5,-5)V5

LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1032) GWTT 292 CONF.H-33 ORBITER B5W4 (+5,-5)V5

LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

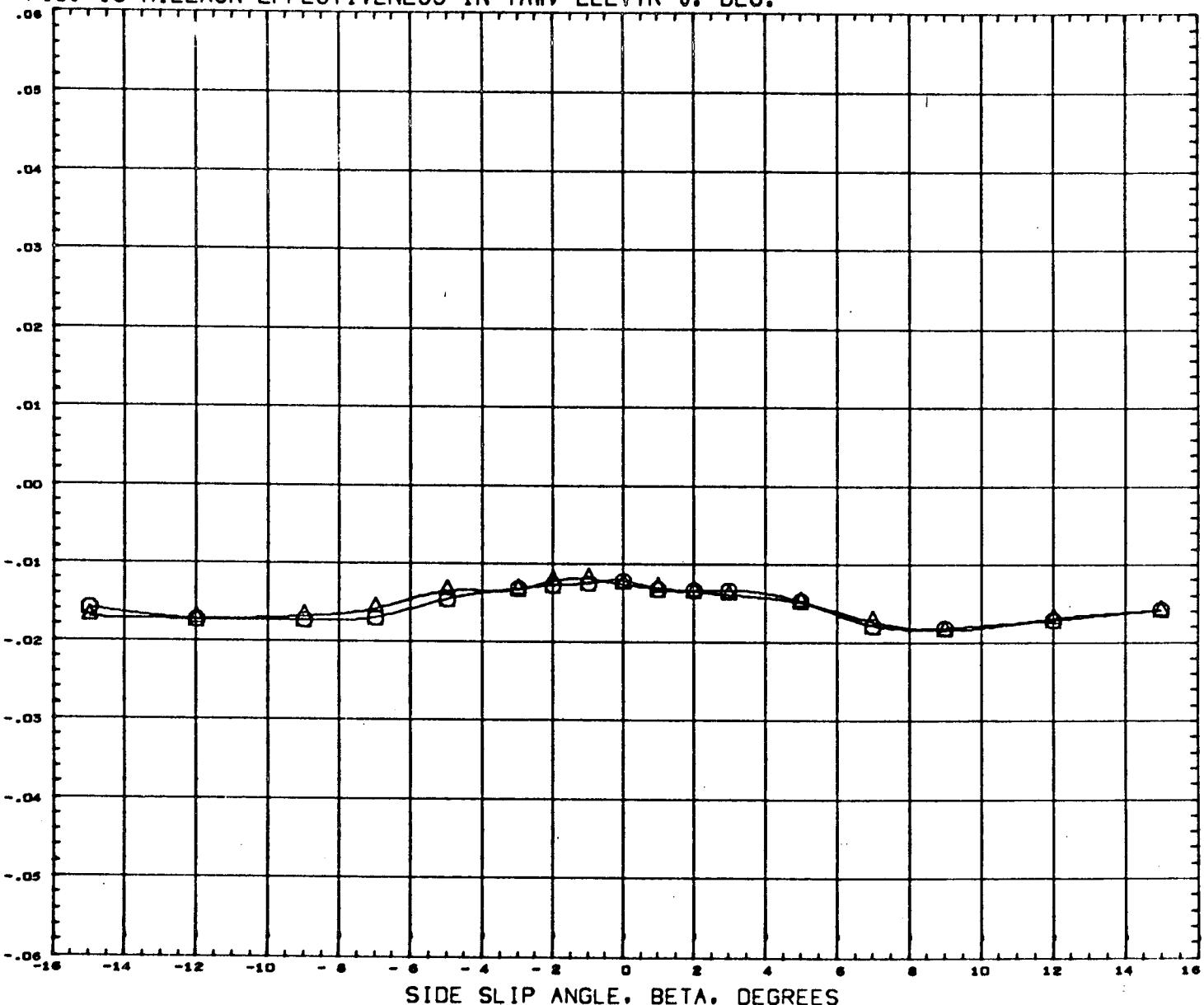
REFERENCE INFORMATION
 SREF 7.7440 3Q FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMNP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 65

FIG. 13 AILERON EFFECTIVENESS IN YAW, ELEVTR=0. DEG.

PITCHING MOMENT COEFFICIENT. CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) Q GWTT 292 CONF.H-33 CRBITER 85W4V5
 (RD1032) Q GWTT 292 CONF.H-33 CRBITER 85W4(+5,-5)V5

LELEVN RELEVN ALPHA
 0.000 0.000 0.000
 5.000 -5.000 0.000

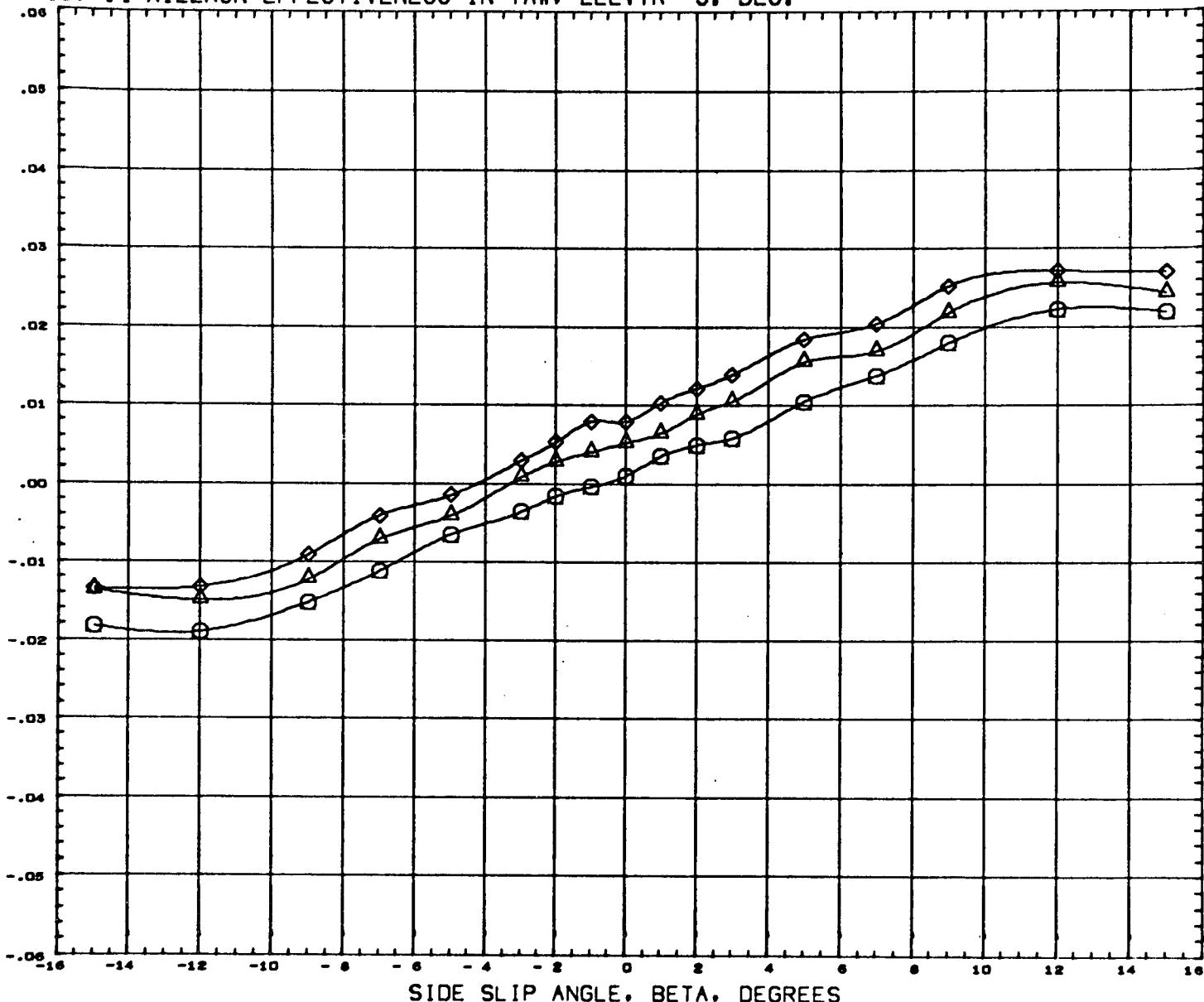
REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 66

FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5, DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1033) GWTT 292 CONF.H-33 CRBITER B5W4 (-5,-5)V5
 (RD1026) GWTT 292 CONF.H-33 CRBITER B5W4 (0,-10)V5
 (RD1031) GWTT 292 CONF.H-33 CRBITER B5W4 (+5,-15)V5

LELEVN RELEVN ALPHA

-5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION

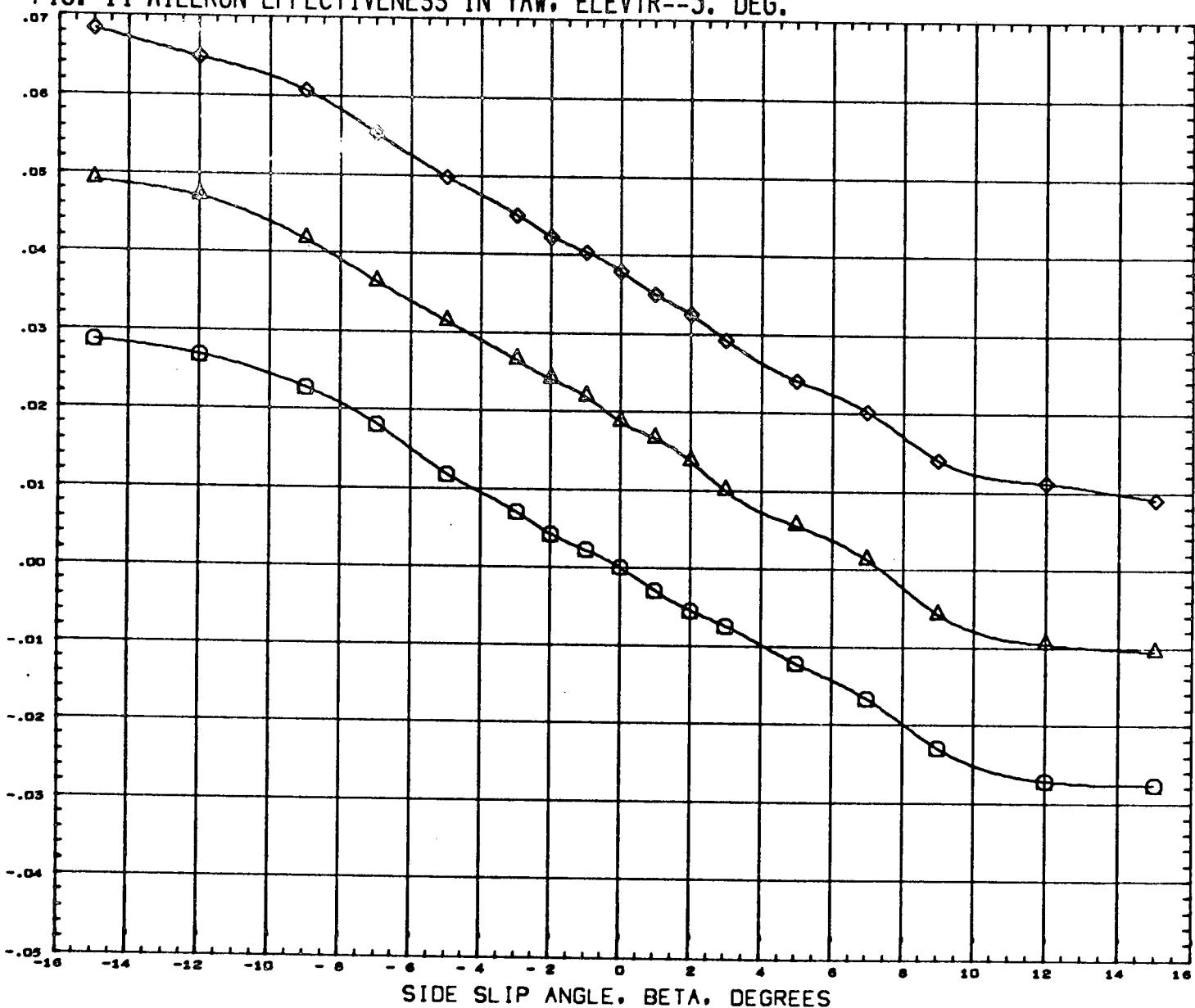
SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

PAGE 67

FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1033) \diamond GWT 292 CONF.H-33 ORBITER B5W4 (-5,-5)V5
 (RD1028) \triangle GWT 292 CONF.H-33 ORBITER B5W4 (0,-10)V5
 (RD1031) \square GWT 292 CONF.H-33 ORBITER B5W4 (+5,-15)V5

LELEVN RELEVN ALPHA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5. DEG.

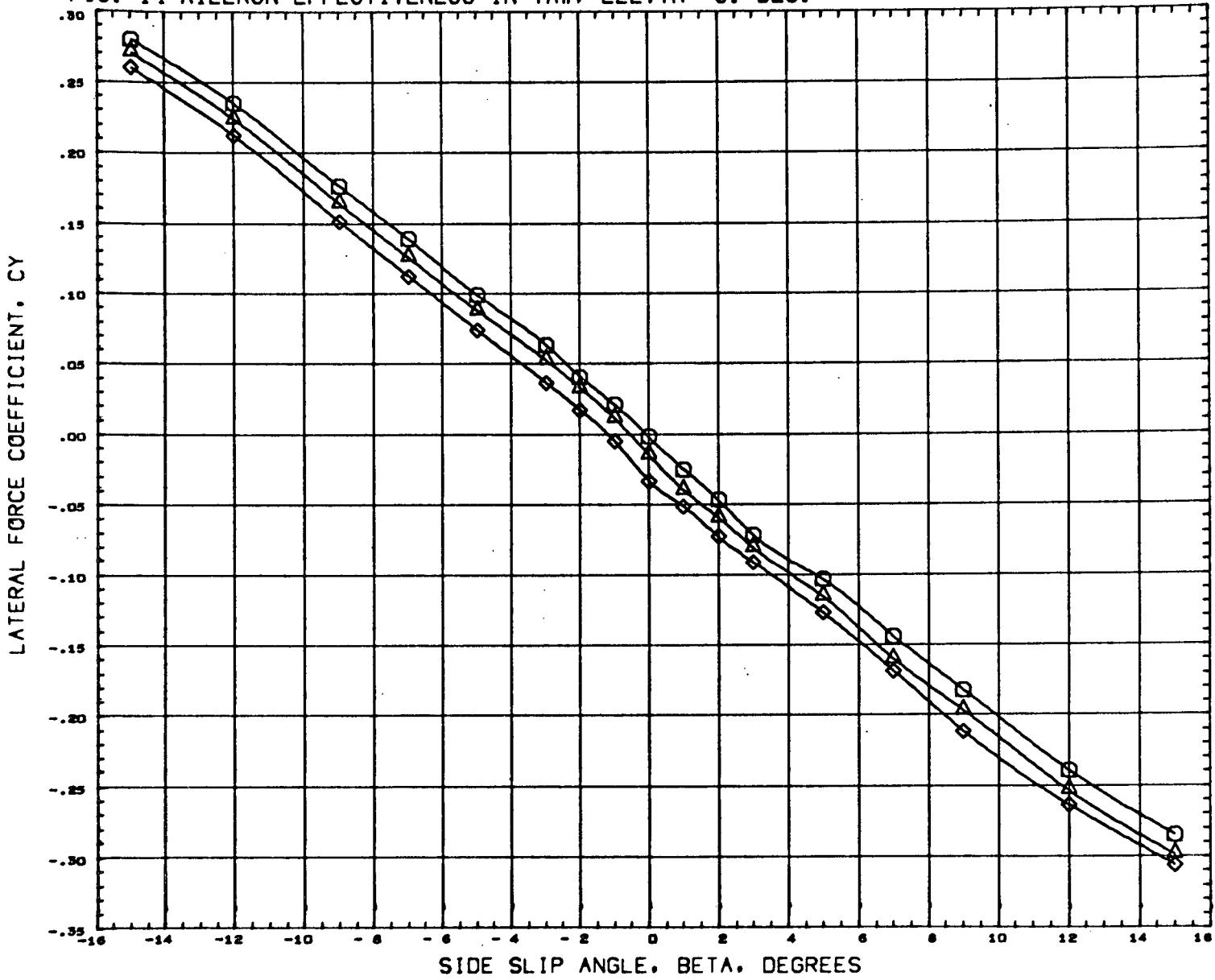
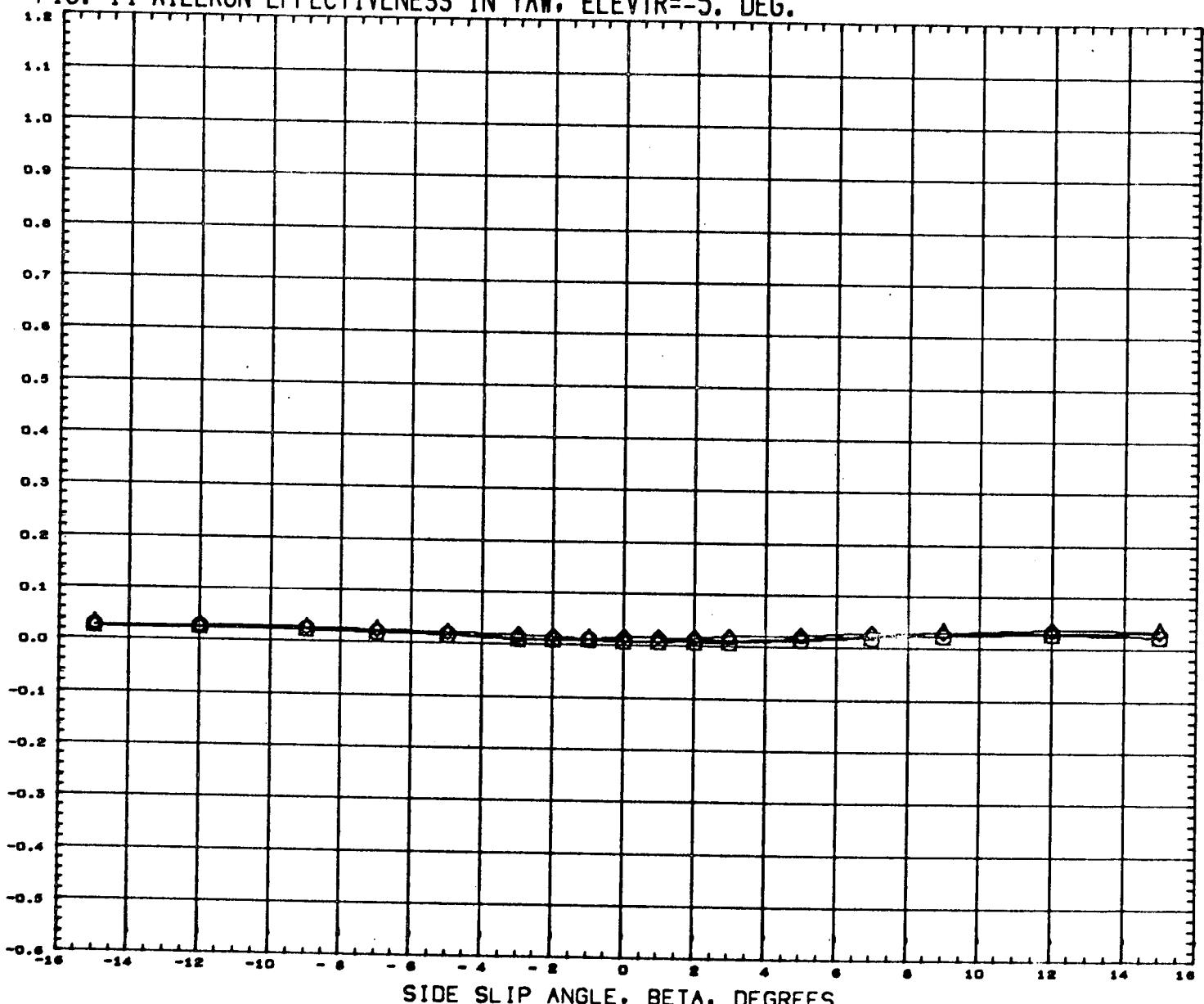


FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5, DEG.

LIFT COEFFICIENT, CL



DATA SET SYMBOL CONFIGURATION DESCRIPTION

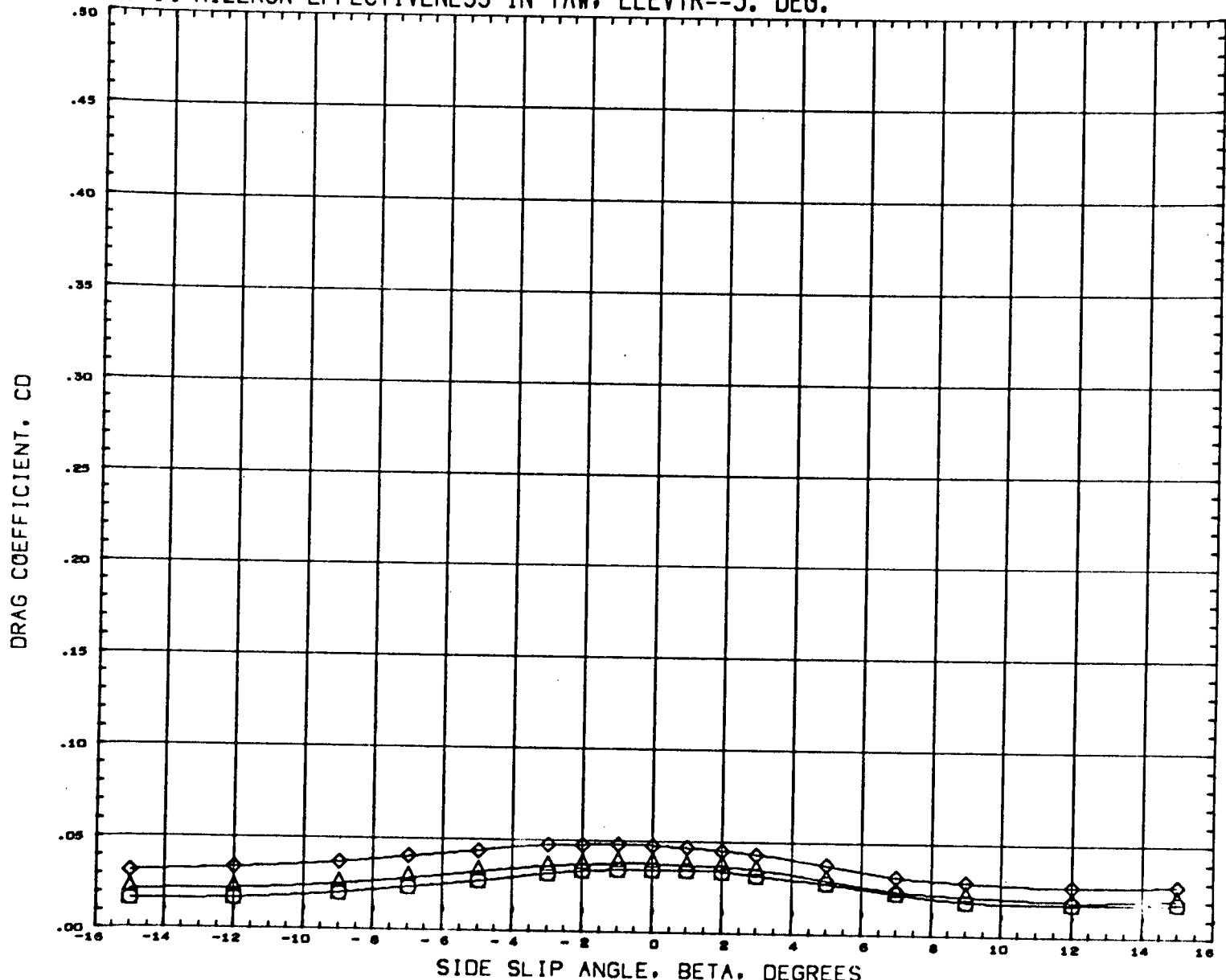
(RD1033) GWT 292 CONF.H-33 ORBITER B5W4 (-5,-5)V5
 (RD1026) GWT 292 CONF.H-33 ORBITER B5W4 (0,-10)V5
 (RD1031) GWT 292 CONF.H-33 ORBITER B5W4 (+5,-15)V5

LELEVN RELEVN ALPHA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRF 1285.0040 IN.
 YMRF 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1033) GWTT 292 CONF.H-33 ORBITER BSW4 (-5,-5)V5
 (RD1026) GWTT 292 CONF.H-33 ORBITER BSW4 (0,-10)V5
 (RD1031) GWTT 292 CONF.H-33 ORBITER BSW4 (+5,-15)V5

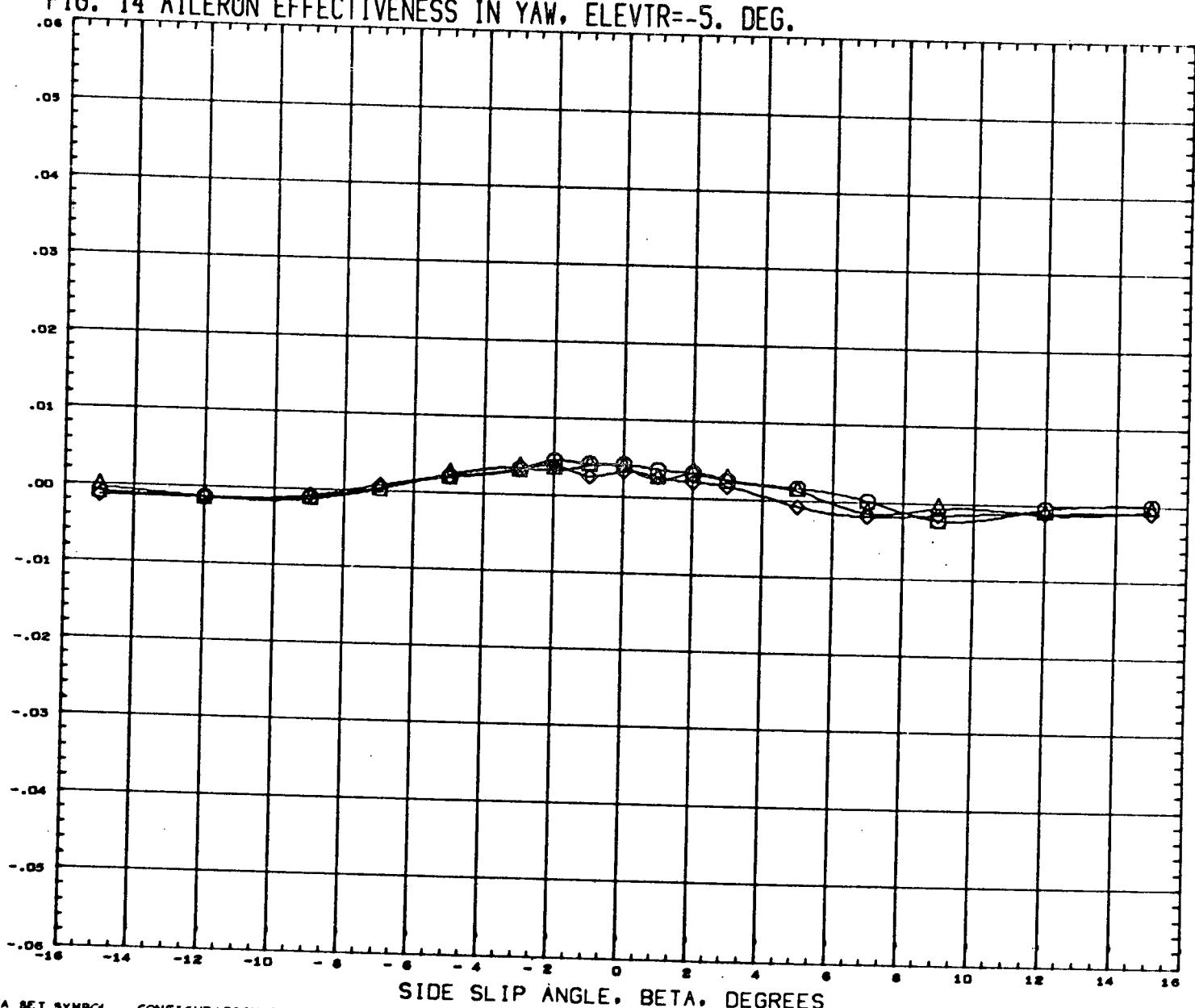
LELEVN RELEVN ALPHA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 14 AILERON EFFECTIVENESS IN YAW, ELEVTR=-5. DEG.

PITCHING MOMENT COEFFICIENT, CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1033) GWTT 292 CONF.H-33 CRBITER B5W4 (-5,-5)V5
 (RD1026) GWTT 292 CONF.H-33 CRBITER B5W4 (0,-10)V5
 (RD1031) GWTT 292 CONF.H-33 CRBITER B5W4 (+5,-15)V5

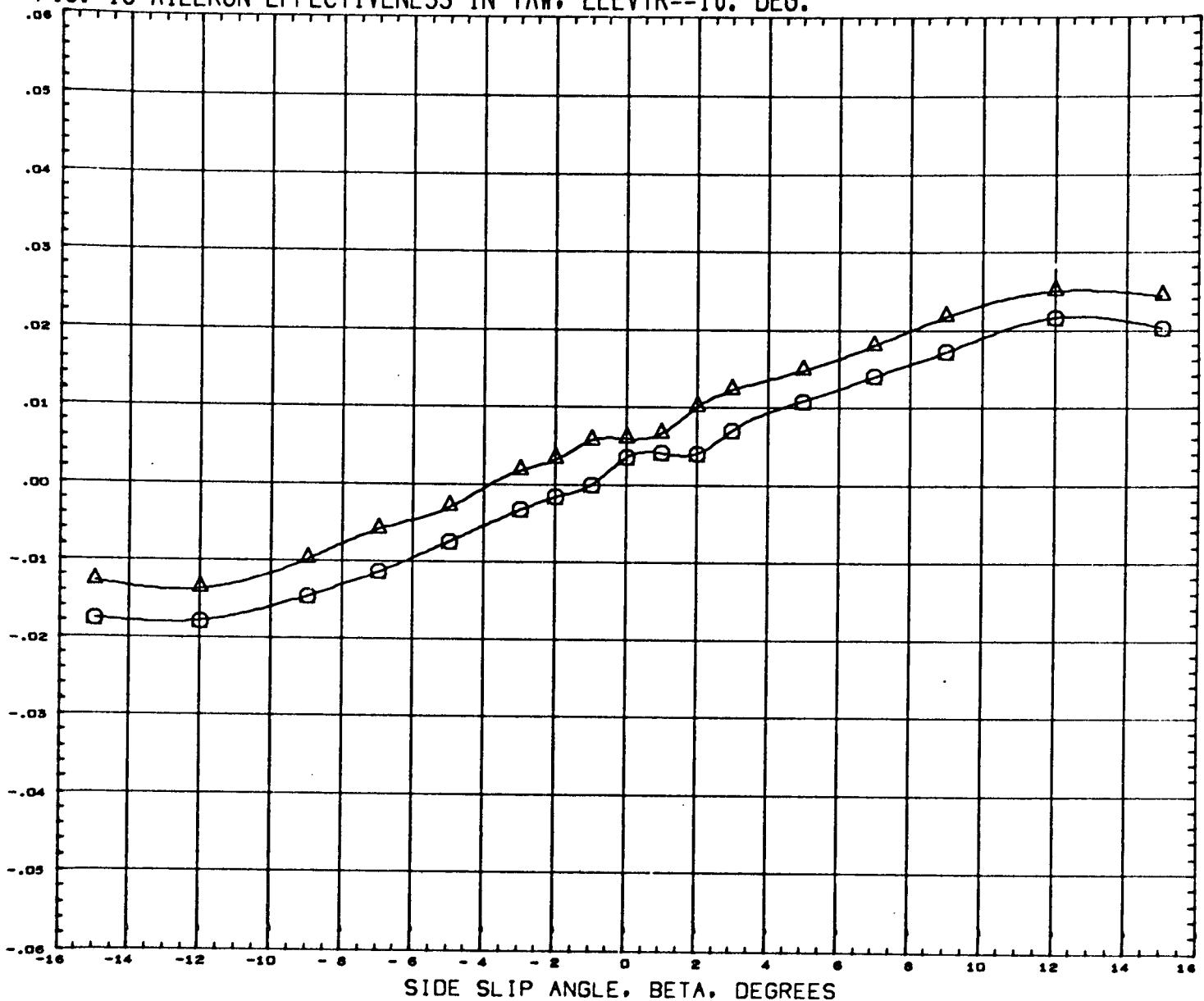
LELEVN RELEVN ALPHA
 -5.000 -5.000 0.000
 0.000 -10.000 0.000
 5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10, DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1029) GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5

(RD1030) GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-15)V5

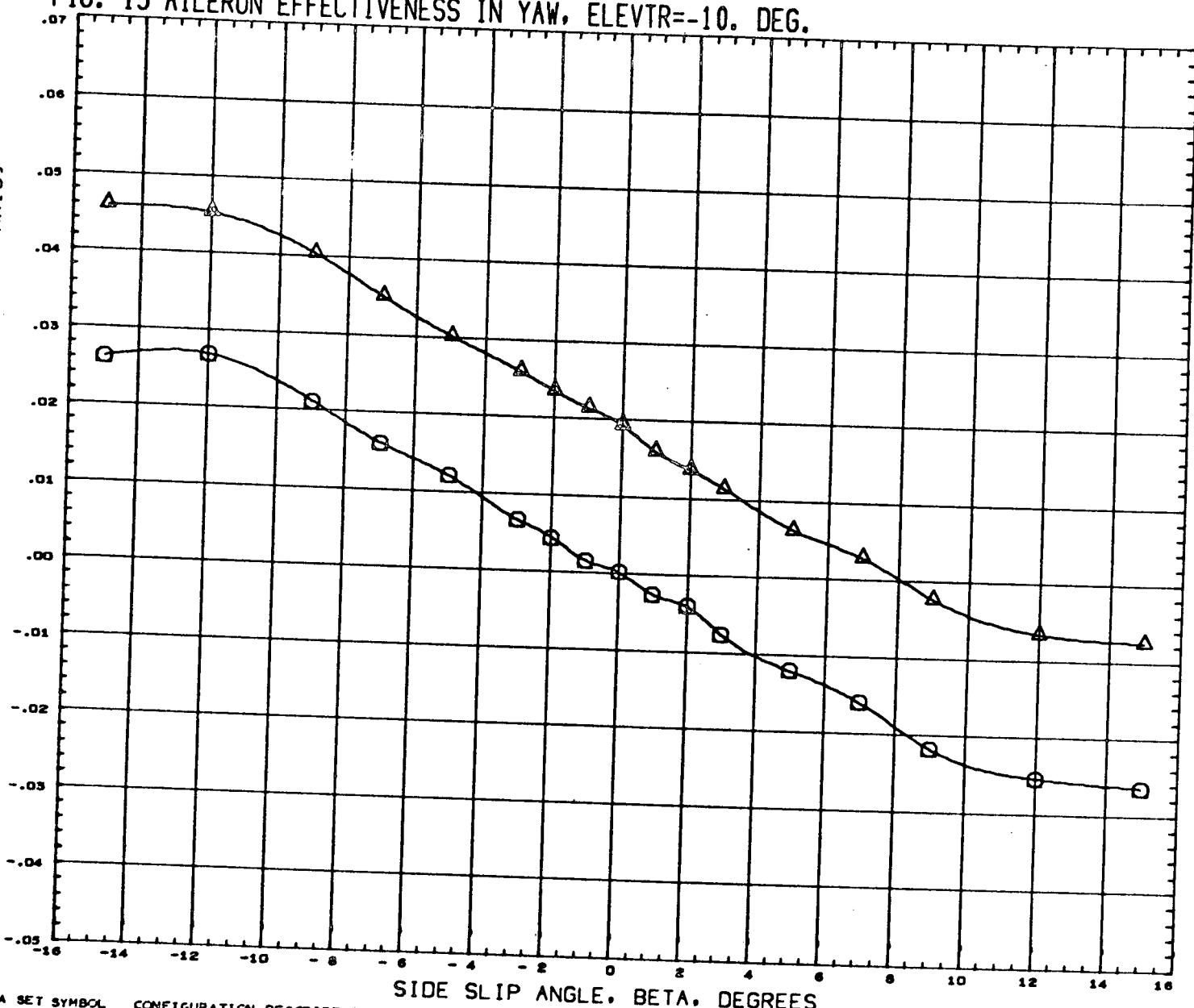
LELEVN RELEVN ALPHA
-10.000 -10.000 0.000
-5.000 -15.000 0.000

REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7600 FT.
XMRP 1265.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

MACH 0.170

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



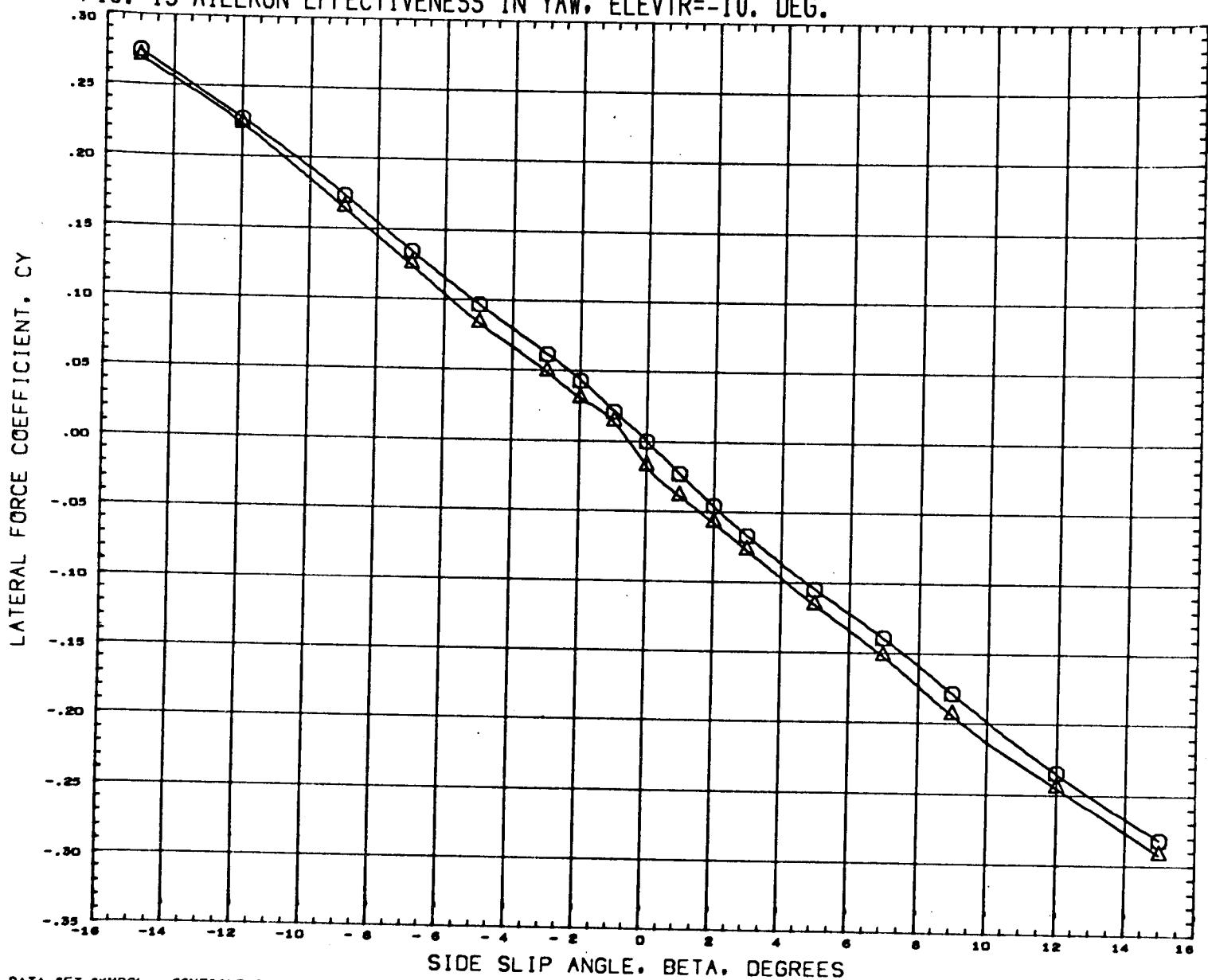
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1029) GWTI 292 CONF.M-33 ORBITER BSW4(-10,-10)V5
 (RD1030) GWTI 292 CONF.M-33 ORBITER BSW4(-5,-15)V5

LELEVN RELEVN ALPHA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1029) Q GWTT 292 CONF.M-33 CRBITER B5W4 (-10,-10)V5
 (RD1030) A GWTT 292 CONF.M-33 CRBITER B5W4 (-5,-15)V5

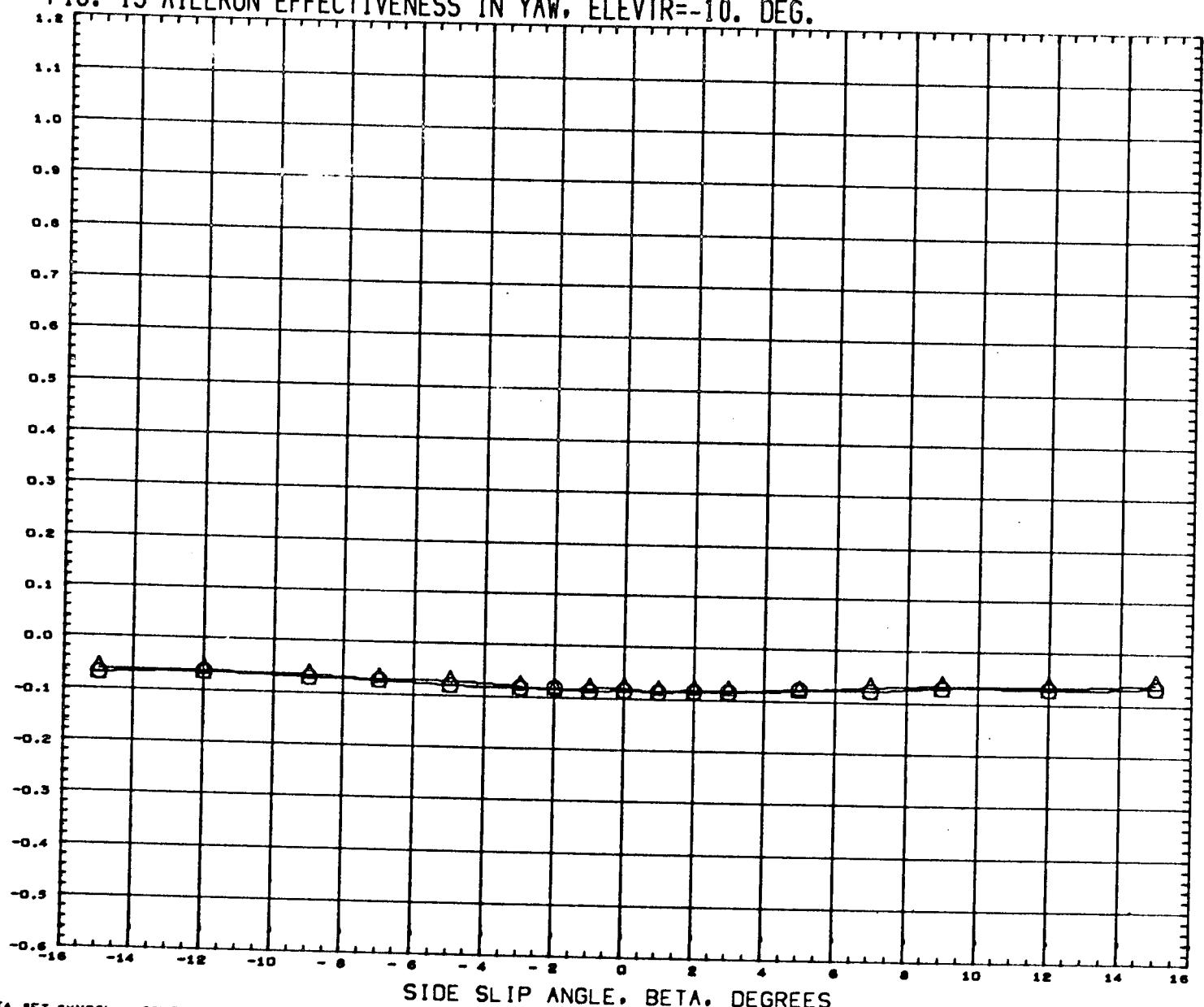
LELEVN RELEVN ALPHA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMPP 1265.0040 IN.
 YMPP 0.0000 IN.
 ZMPP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10. DEG.

LIFT COEFFICIENT, CL



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1029) GWTT 292 CONF.H-33 ORBITER BSW4(-10,-10)V5
(RD1030) GWTT 292 CONF.H-33 ORBITER BSW4(-5,-15)V5

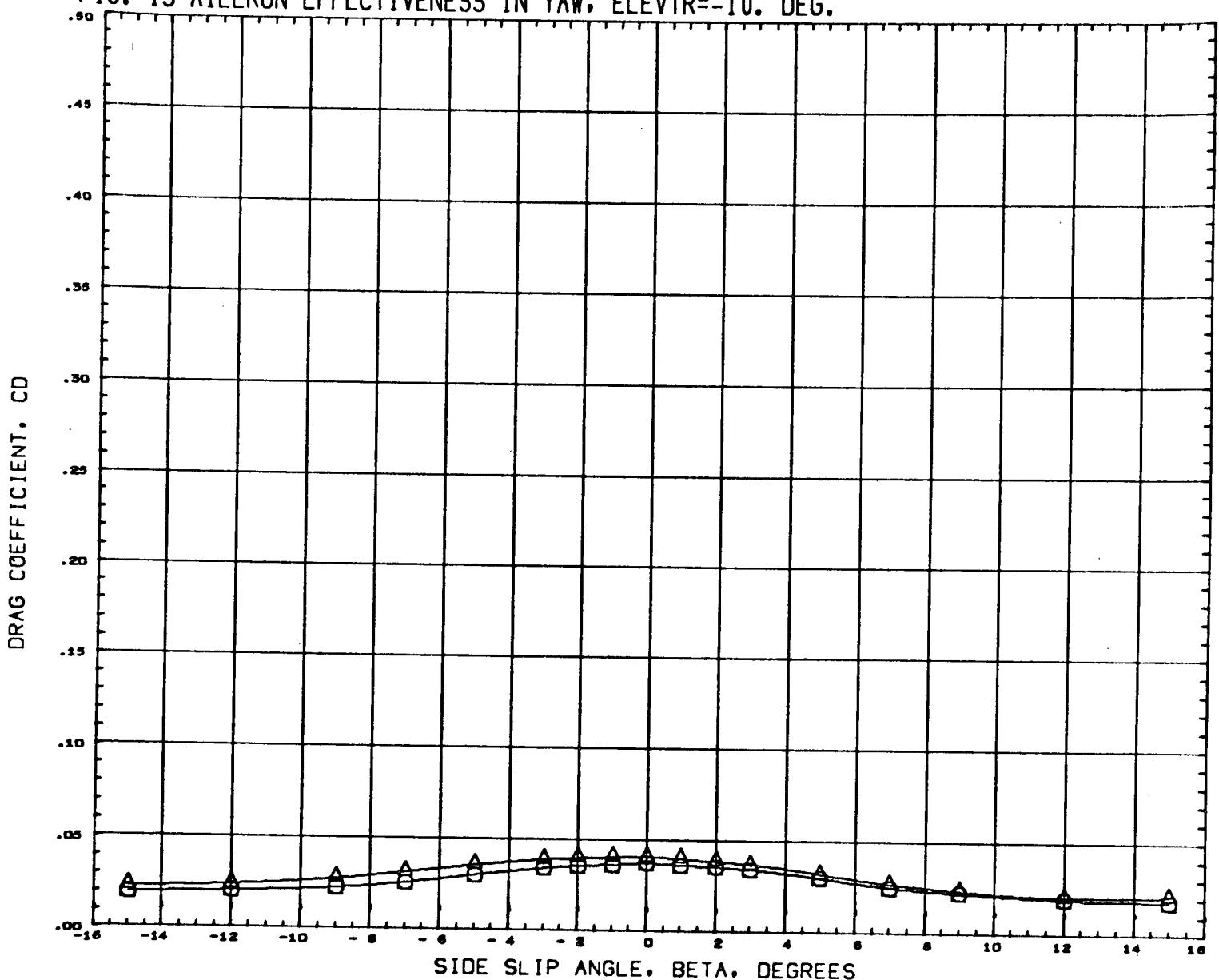
LELEVN RELEVN ALPHA
-10.000 -10.000 0.000
-5.000 -15.000 0.000

REFERENCE INFORMATION
SREF 7.7440 SQ FT
LREF 5.4000 FT.
BREF 3.7800 FT.
XMRP 1285.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

MACH 0.170

PAGE 76

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1029) GWTT 292 CONF.H-33 ORBITER B5W4 (-10,-10)V5
 (RD1030) GWTT 292 CONF.H-33 ORBITER B5W4 (-5,-15)V5

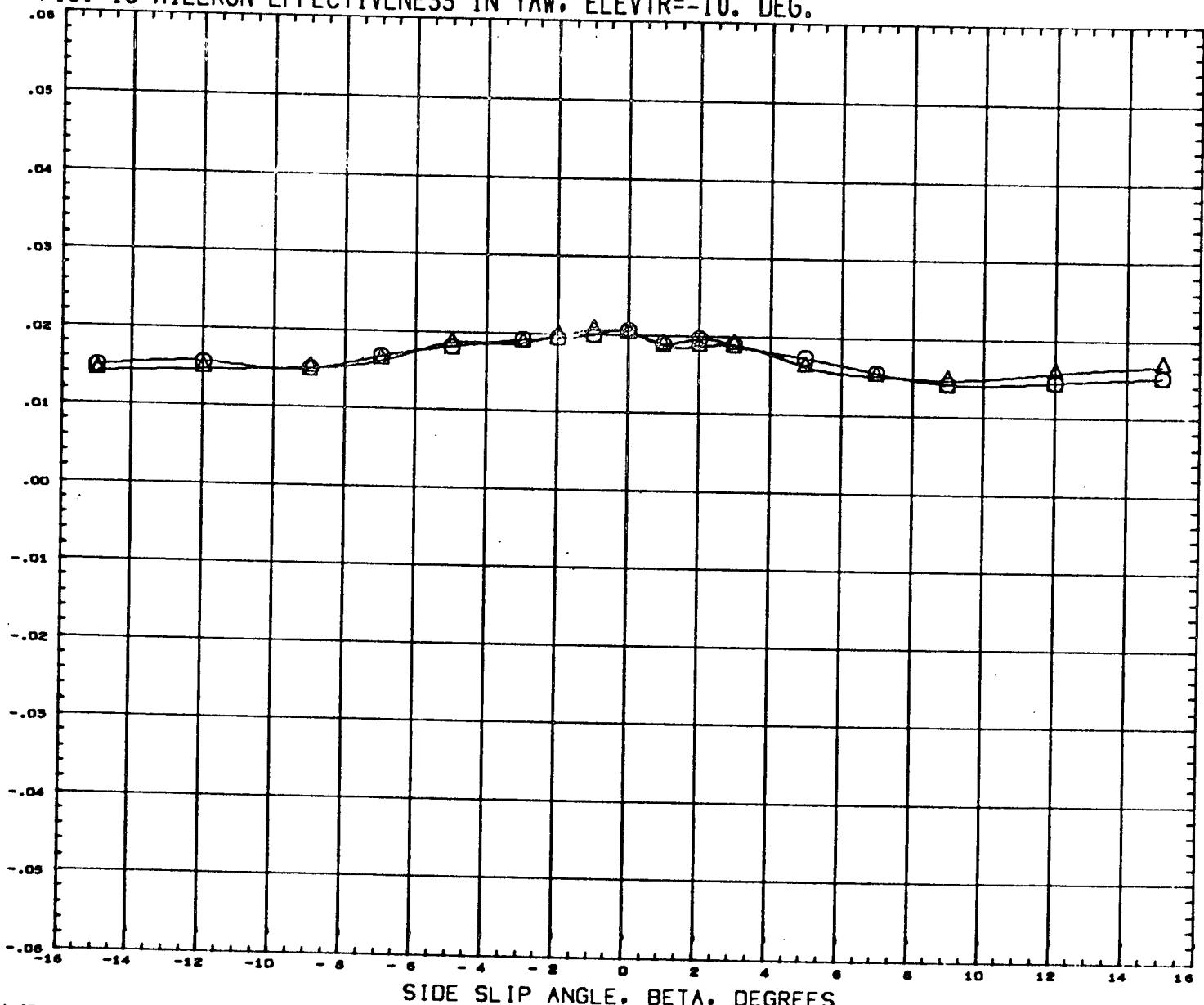
LELEVN RELEVN ALPHA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1265.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 15 AILERON EFFECTIVENESS IN YAW, ELEVTR=-10. DEG.

PITCHING MOMENT COEFFICIENT, CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1029) GWTT 292 CONF.H-33 ORBITER B5W4(-10,-10)V5
 (RD1030) GWTT 292 CONF.H-33 ORBITER B5W4(-5,-15)V5

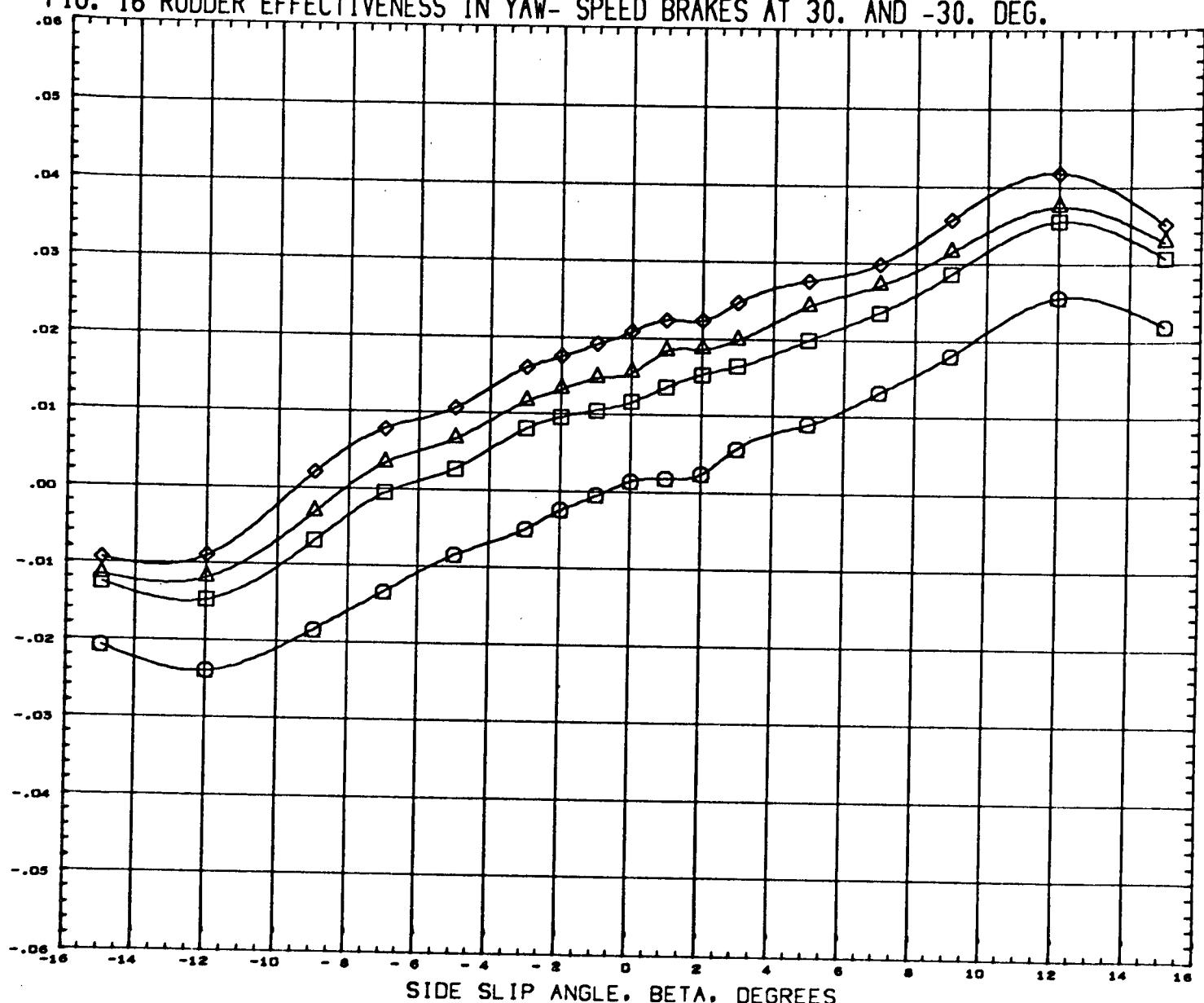
LELEVN RELEVN ALPHA
 -10.000 -10.000 0.000
 -5.000 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

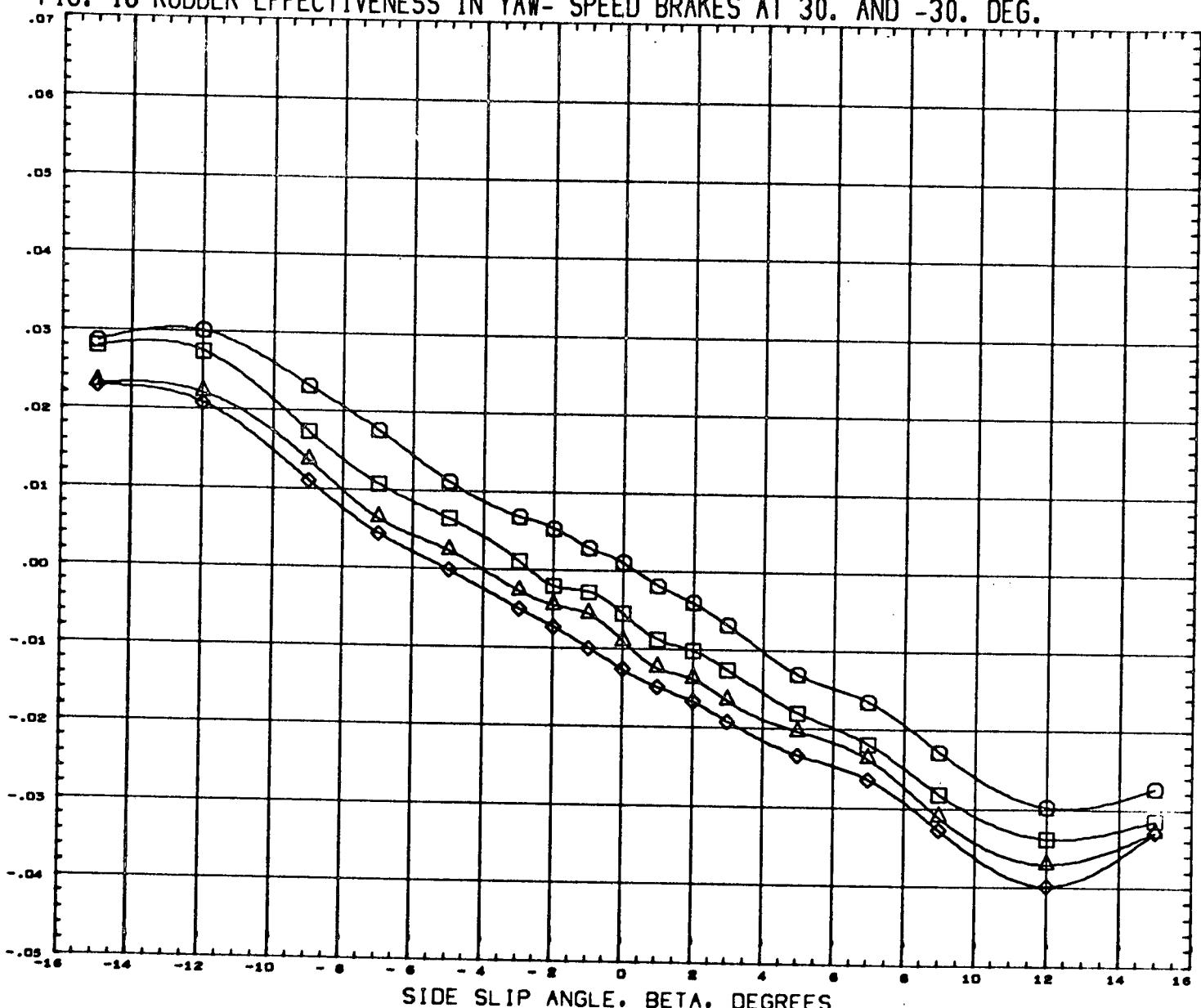
(RD1034) \odot GWTT 292 CONF.H-33 ORBITER B5W4V5 (0,+30,-30)
 (RD1035) \diamond GWTT 292 CONF.H-33 ORBITER B5W4V5 (-10,+25,-35)
 (RD1036) \triangle GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15,+25,-35)
 (RD1042) \square GWTT 292 CONF.H-33 ORBITER B5W4V5 (-5,+25,-35)

	TRUDDR	LRRUDD	LRRUDD	ALPHA	REFERENCE	INFORMATION
(RD1034)	0.000	30.000	-30.000	0.000	SREF	7.7440 SQ FT
(RD1035)	-10.000	25.000	-35.000	0.000	LREF	5.4000 FT.
(RD1036)	-15.000	25.000	-35.000	0.000	BREF	3.7800 FT.
(RD1042)	-5.000	25.000	-35.000	0.000	XMRP	1285.0040 IN.
					YMRP	0.0000 IN.
					ZMRP	403.0004 IN.
					SCALE	0.0400

MACH 0.170

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1034)	○	GWT 292 CONF.H-33 CRBITER B5W4V5 (0,+30,-30)
(RD1035)	◇	GWT 292 CONF.H-33 CRBITER B5W4V5 (-10,+25,-35)
(RD1036)	△	GWT 292 CONF.H-33 CRBITER B5W4V5 (-15,+25,-35)
(RD1042)	□	GWT 292 CONF.H-33 CRBITER B5W4V5 (-5,+25,-35)

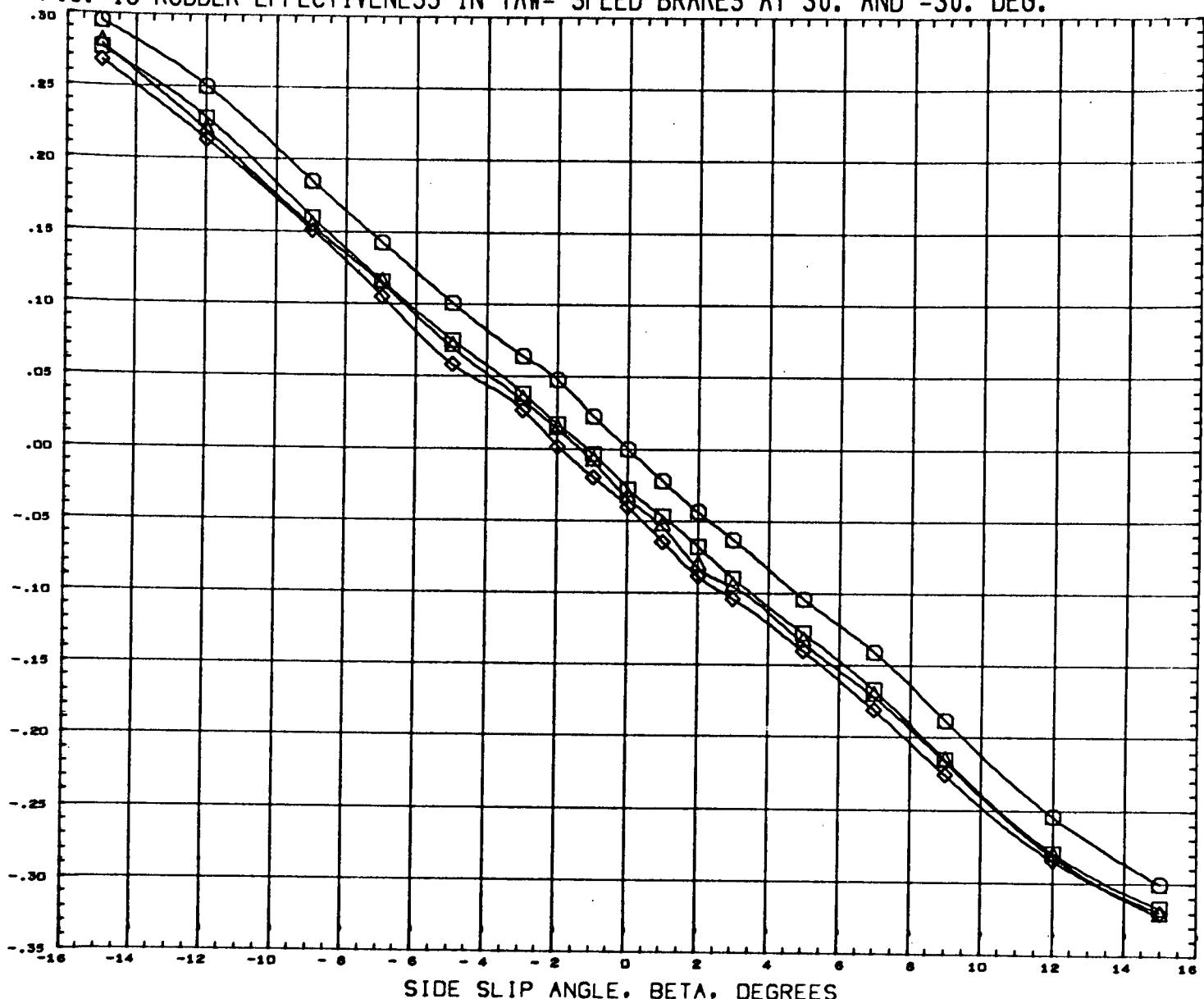
TRUDDR	LRLUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
-10.000	25.000	-35.000	0.000	LREF 5.4000 FT.
-15.000	25.000	-35.000	0.000	BREF 3.7800 FT.
-5.000	25.000	-35.000	0.000	XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

PAGE 80

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL CONFIGURATION DESCRIPTION

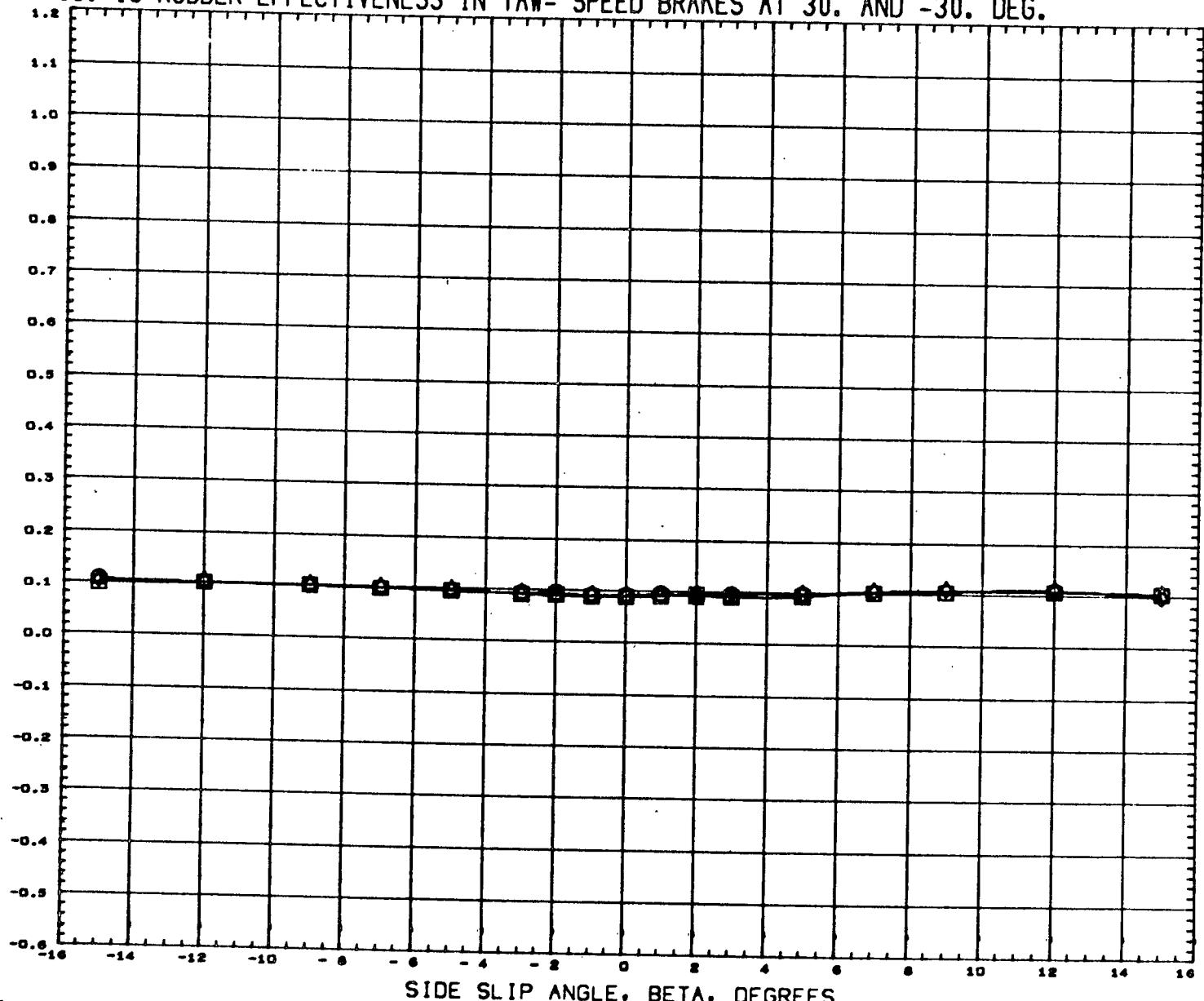
(RD1034) ○ GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+30,-30)
 (RD1035) □ GWTT 292 CONF.H-33 ORBITER BSW4V5 (-10,+25,-35)
 (RD1036) ▲ GWTT 292 CONF.H-33 ORBITER BSW4V5 (-15,+25,-35)
 (RD1042) ◇ GWTT 292 CONF.H-33 ORBITER BSW4V5 (-5,+25,-35)

TRUDDR	LRLUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
-10.000	25.000	-35.000	0.000	LREF 5.4000 FT.
-15.000	25.000	-35.000	0.000	BREF 3.7800 FT.
-5.000	25.000	-35.000	0.000	XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

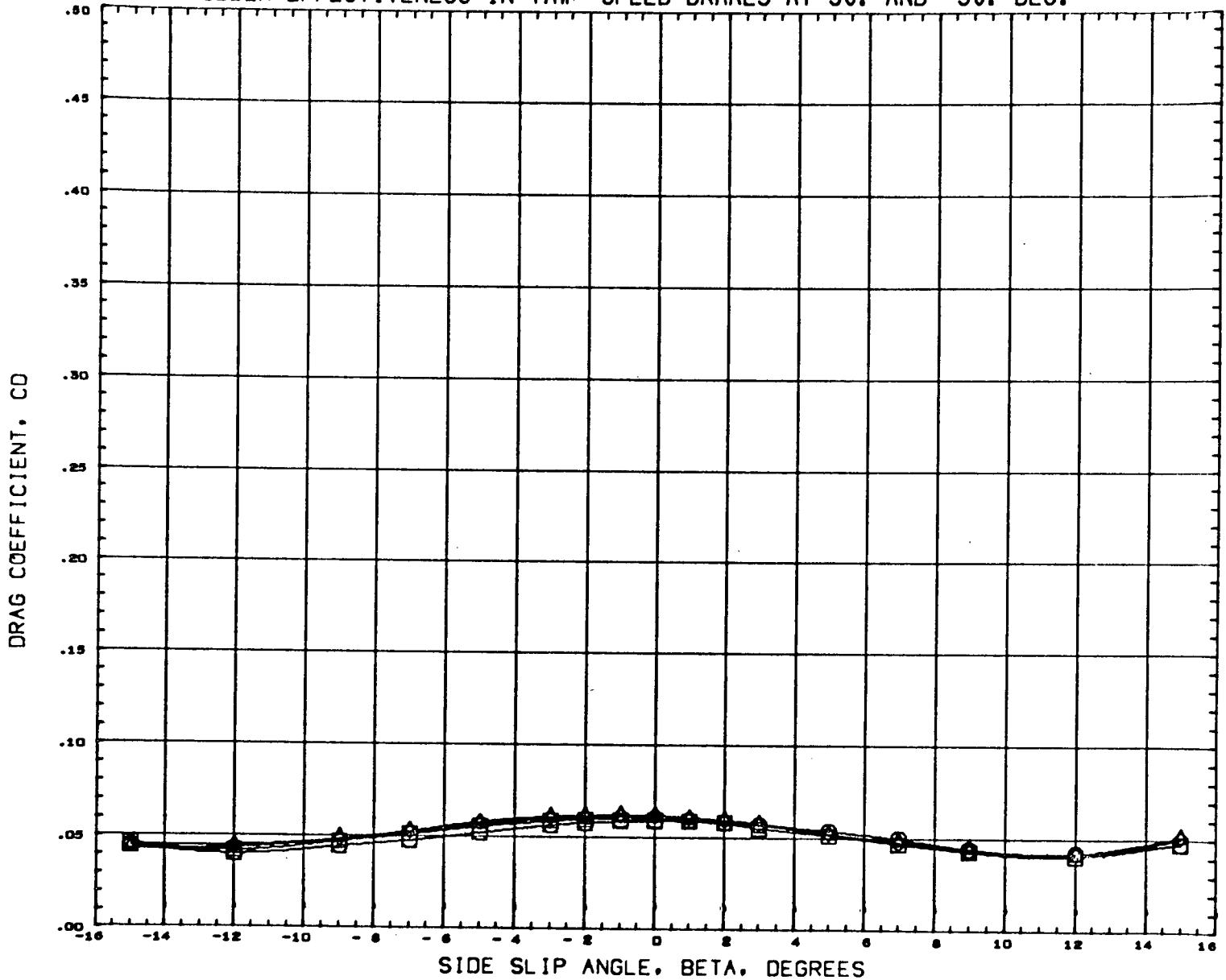
LIFT COEFFICIENT, CL



DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE	INFORMATION
(RD1034)	GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+30,-30)	0.000	30.000	-30.000	0.000	SREF	7.7440 SQ FT
(RD1035)	GWTT 292 CONF.H-33 ORBITER BSW4V5 (-10,+25,-35)	-10.000	25.000	-35.000	0.000	LREF	5.4000 FT.
(RD1036)	GWTT 292 CONF.H-33 ORBITER BSW4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	BREF	3.7800 FT.
(RD1042)	GWTT 292 CONF.H-33 ORBITER BSW4V5 (-5,+25,-35)	-5.000	25.000	-35.000	0.000	XMRP	1285.0040 IN.
						YMRP	0.0000 IN.
						ZMRP	403.0004 IN.
						SCALE	0.0400

MACH 0.170

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

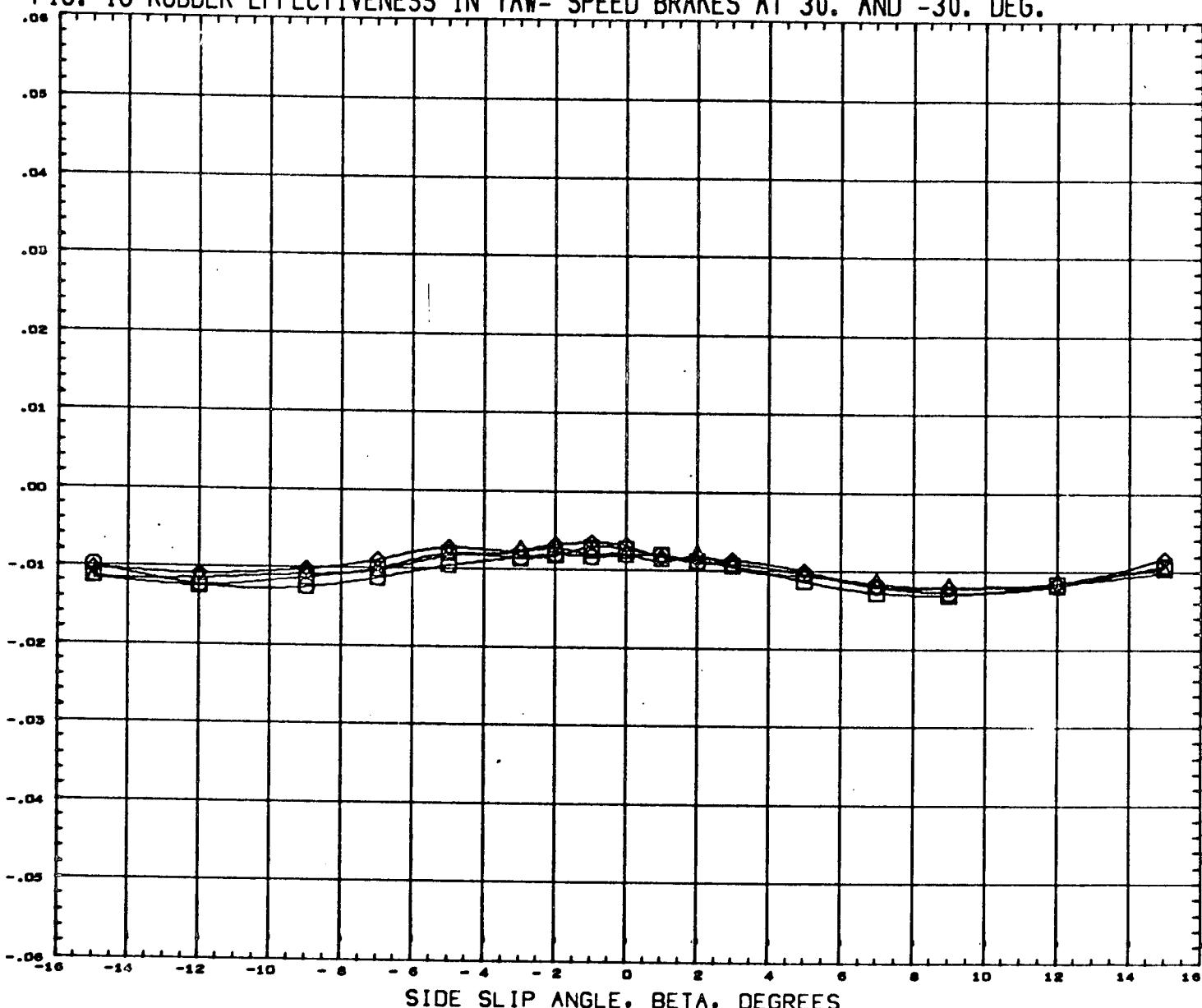


DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
(RD1034)	GWTT 292 CONF.H-33 ORBITER B5W4V5(0,+30,-30)	0.000	30.000	-30.000	0.000	SREF 7.7440 SQ FT
(RD1035)	GWTT 292 CONF.H-33 ORBITER B5W4V5(-10,+25,-35)	-10.000	25.000	-35.000	0.000	LREF 5.4000 FT.
(RD1036)	GWTT 292 CONF.H-33 ORBITER B5W4V5(-15,+25,-35)	-15.000	25.000	-35.000	0.000	BREF 3.7800 FT.
(RD1042)	GWTT 292 CONF.H-33 ORBITER B5W4V5(-5,+25,-35)	-5.000	25.000	-35.000	0.000	XMRP 1285.0040 IN.
					YMRP 0.0000 IN.	
					ZMRP 403.0004 IN.	
					SCALE 0.0400	

MACH 0.170

FIG. 16 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 30. AND -30. DEG.

PITCHING MOMENT COEFFICIENT. CLM



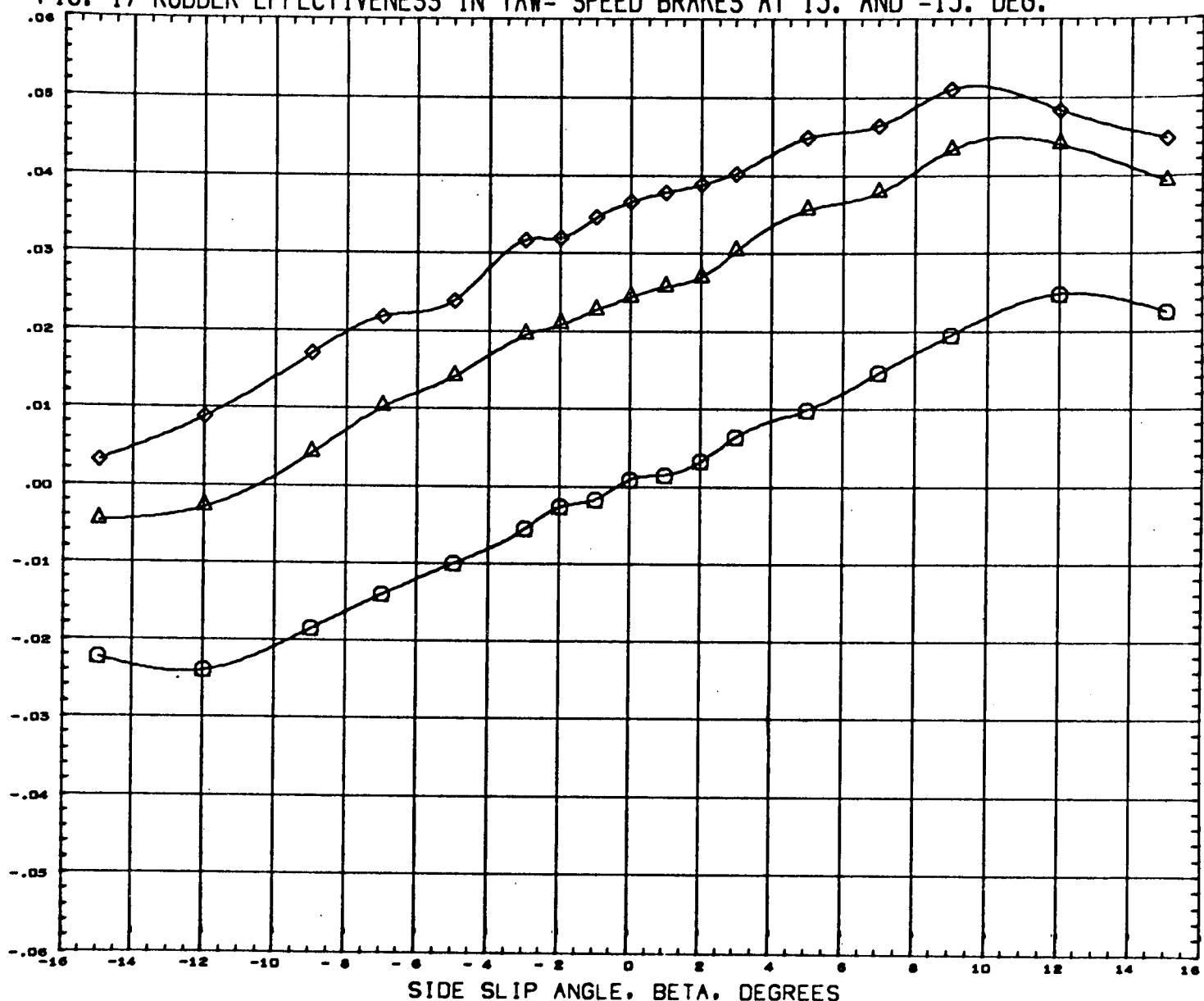
SIDE SLIP ANGLE, BETA, DEGREES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	TRUDDR	LRLRUDD	LRRUDD	ALPHA	REFERENCE	INFORMATION
(RD1034)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (0,+30,-30)	0.000	30.000	-30.000	0.000	SREF	7.7440 SQ FT
(RD1035)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-10,+25,-35)	-10.000	25.000	-35.000	0.000	LREF	5.4000 FT.
(RD1036)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-15,+25,-35)	-15.000	25.000	-35.000	0.000	BREF	3.7800 FT.
(RD1042)	GWTT 292 CONF.H-33 CRBITER B5W4V5 (-5,+25,-35)	-5.000	25.000	-35.000	0.000	XMRP	1285.0040 IN.
						YMRP	0.0000 IN.
						ZMRP	403.0004 IN.
						SCALE	0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.

YAWING MOMENT COEFFICIENT, CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

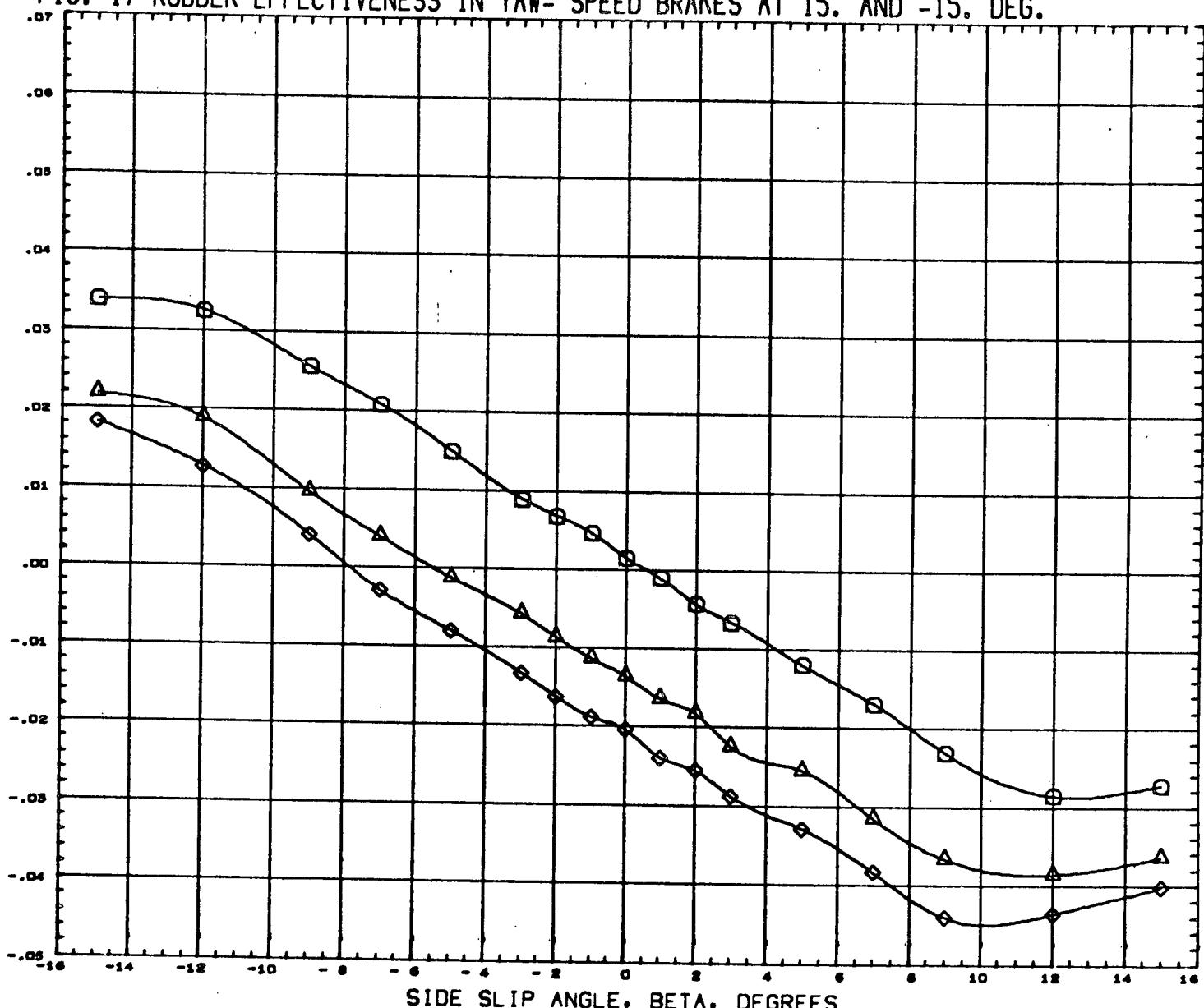
(RD1039) \circ GWTT 292 CONF.H-33 ORBITER 85W4V5 (0,+15,-15)
 (RD1038) \diamond GWTT 292 CONF.H-33 ORBITER 85W4V5 (-10,+5,-25)
 (RD1037) \triangle GWTT 292 CONF.H-33 CRBITER 85W4V5 (-15,0,-30)

TRUDD	LLRUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
-10.000	5.000	-25.000	0.000	LREF 5.4000 FT
-15.000	0.000	-30.000	0.000	BREF 3.7600 FT
				XMRP 1285.0040 IN
				YMRP 0.0000 IN
				ZMRP 403.0004 IN
				SCALE 0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.

ROLLING MOMENT COEFFICIENT. CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

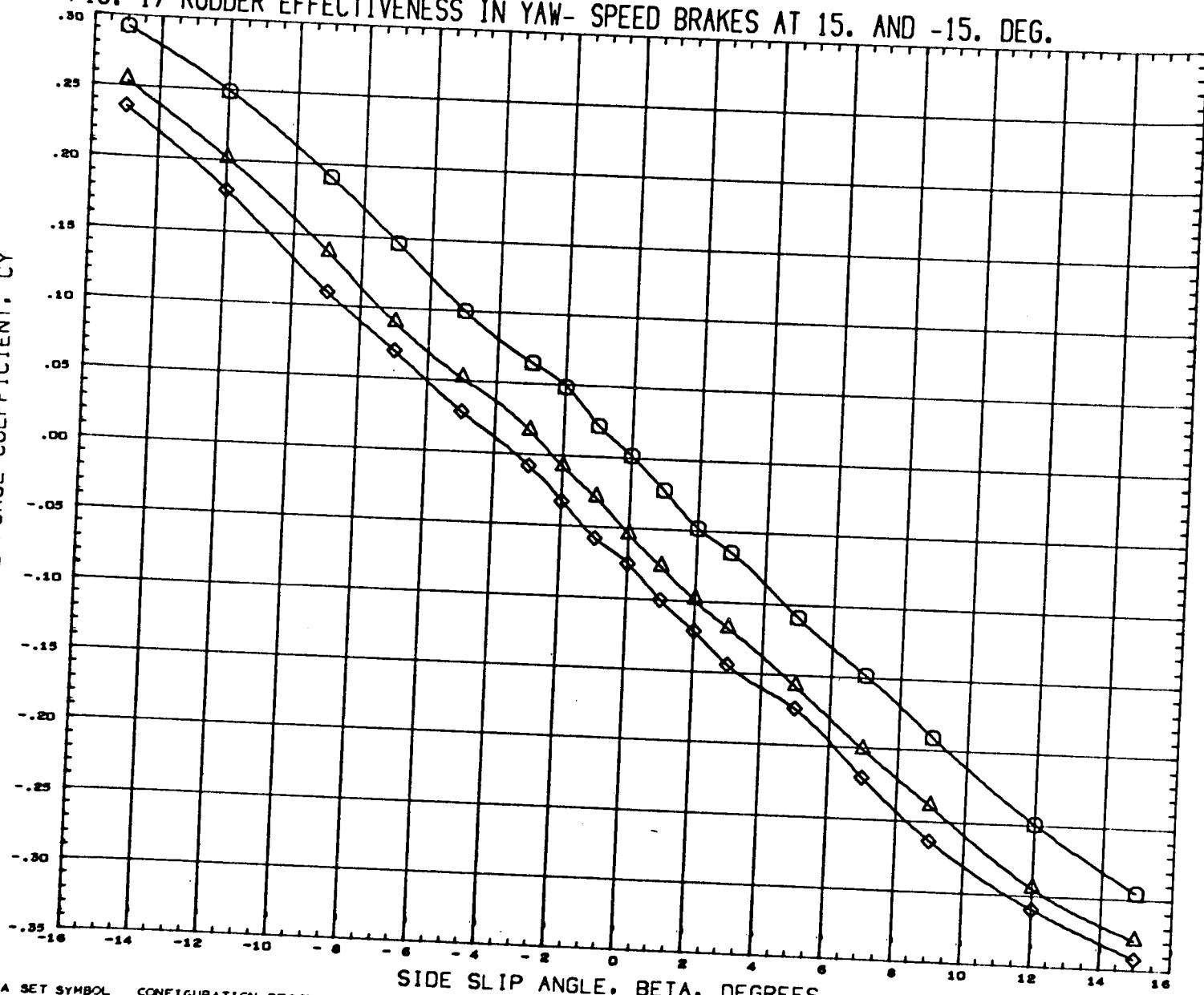
(RD1039) \circ GWTT 292 CONF.M-33 ORBITER BSW4V5 (0,+15,-15)
 (RD1038) \square GWTT 292 CONF.M-33 ORBITER BSW4V5 (-10,+5,-25)
 (RD1037) \diamond GWTT 292 CONF.M-33 ORBITER BSW4V5 (-15,0,-30)

	TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
(RD1039)	0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
(RD1038)	-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
(RD1037)	-15.000	0.000	-30.000	0.000	BREF 3.7600 FT. XMRP 1285.0040 IN. YMRP 0.0000 IN. ZMRP 403.0004 IN. SCALE 0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.

LATERAL FORCE COEFFICIENT, CY



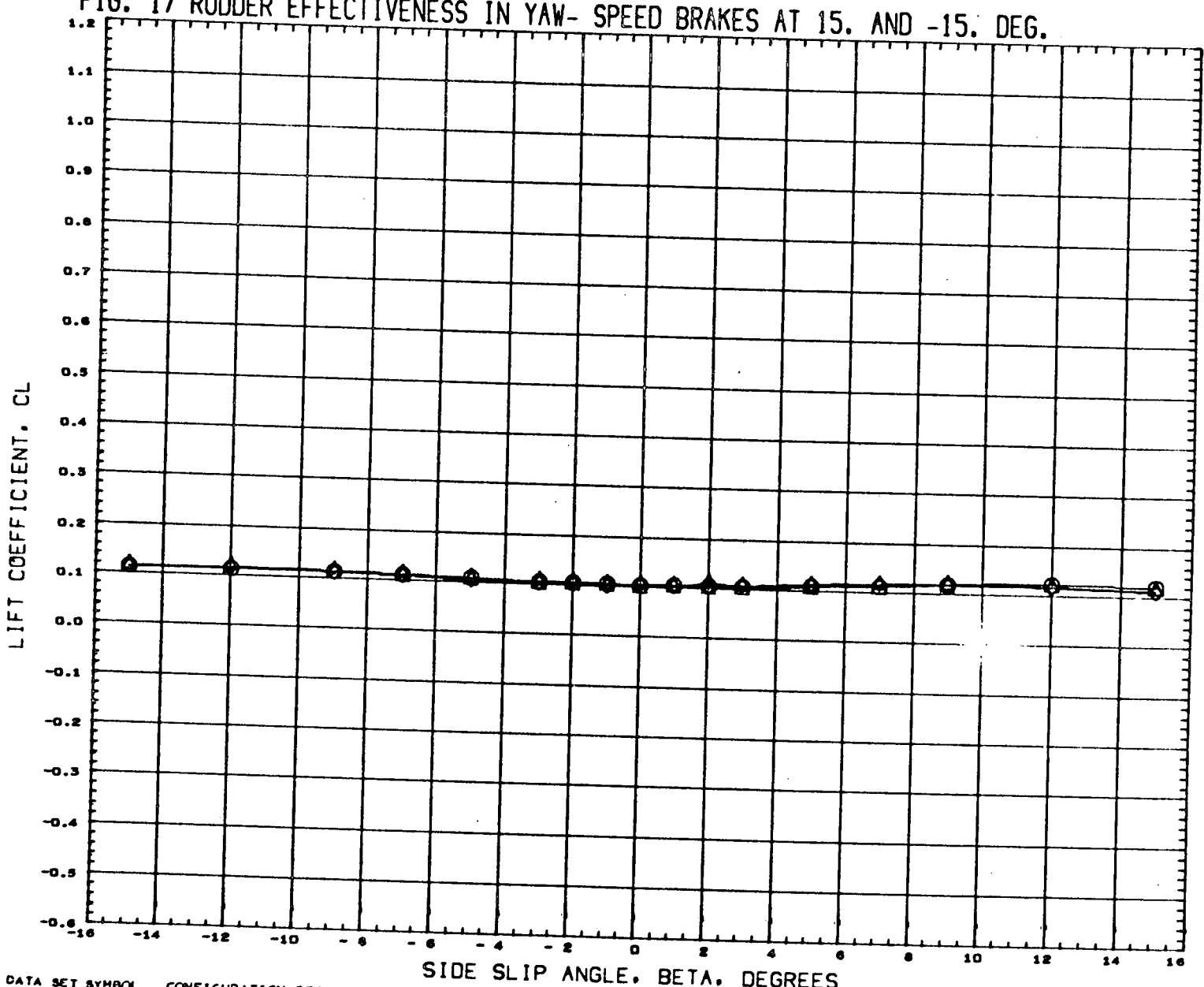
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1039) \textcircled{Q} GWTT 292 CONF.H-33 ORBITER BSW4V5 (0,+15,-15)
 (RD1038) \diamond GWTT 292 CONF.H-33 ORBITER BSW4V5 (-10,+5,-25)
 (RD1037) \triangle GWTT 292 CONF.H-33 ORBITER BSW4V5 (-15,0,-30)

TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE	INFORMATION
0.000	15.000	-15.000	0.000	SREF	7.7440 SQ FT
-10.000	5.000	-25.000	0.000	LREF	5.4000 FT.
-15.000	0.000	-30.000	0.000	BREF	3.7800 FT.
				XMRP	1285.0040 IN.
				YMRP	0.0000 IN.
				ZMRP	403.0004 IN.
				SCALE	0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.



DATA SET SYMBOL CONFIGURATION DESCRIPTION

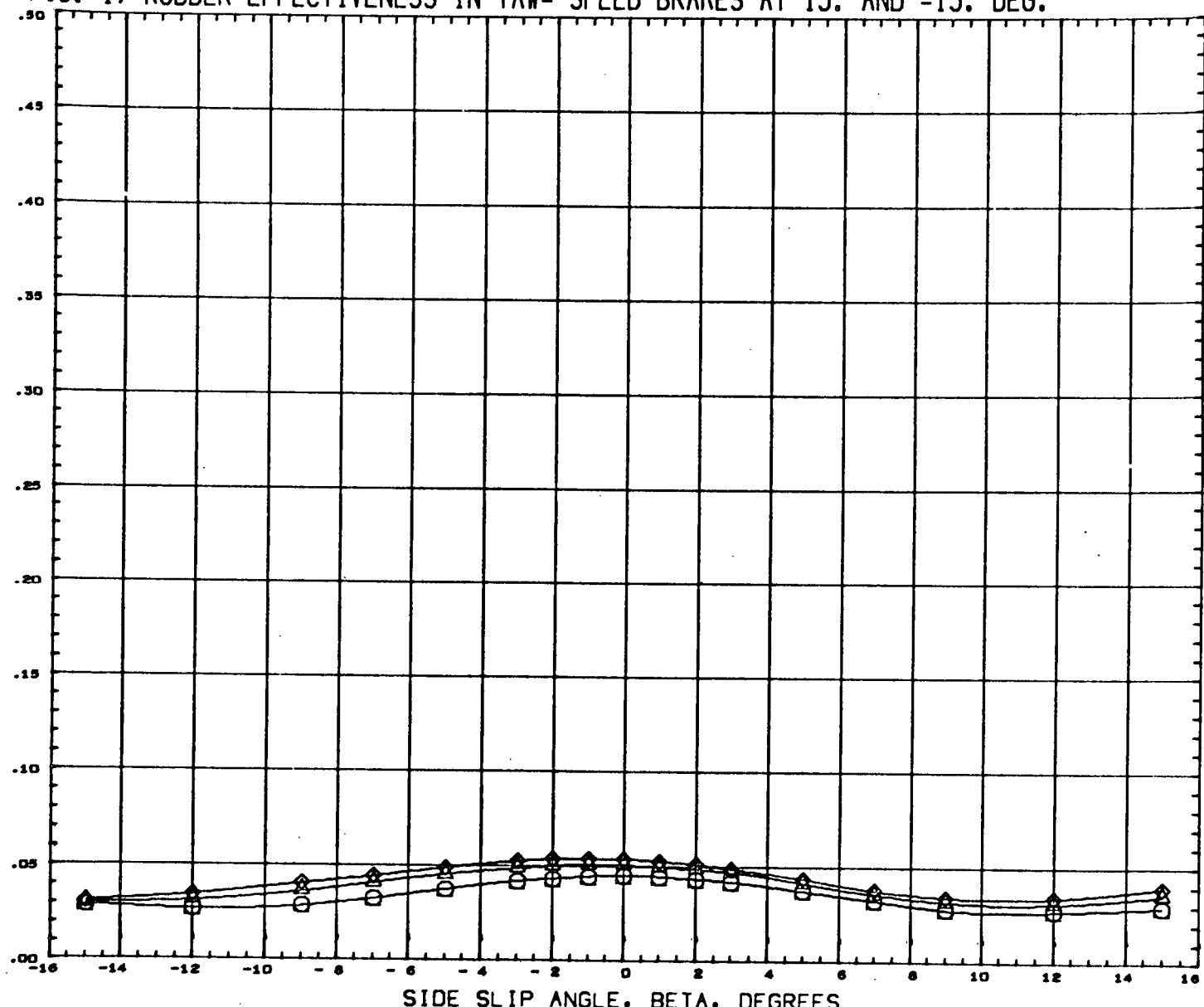
(RD1039) GWTT 292 CONF.H-33 ORBITER BSW4V5(0,+15,-15)
 (RD1038) GWTT 292 CONF.H-33 ORBITER BSW4V5(-10,+5,-25)
 (RD1037) GWTT 292 CONF.H-33 ORBITER BSW4V5(-15,0,-30)

TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
-15.000	0.000	-30.000	0.000	BREF 3.7800 FT.
				XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.

DRAG COEFFICIENT, CD



DATA SET SYMBOL CONFIGURATION DESCRIPTION

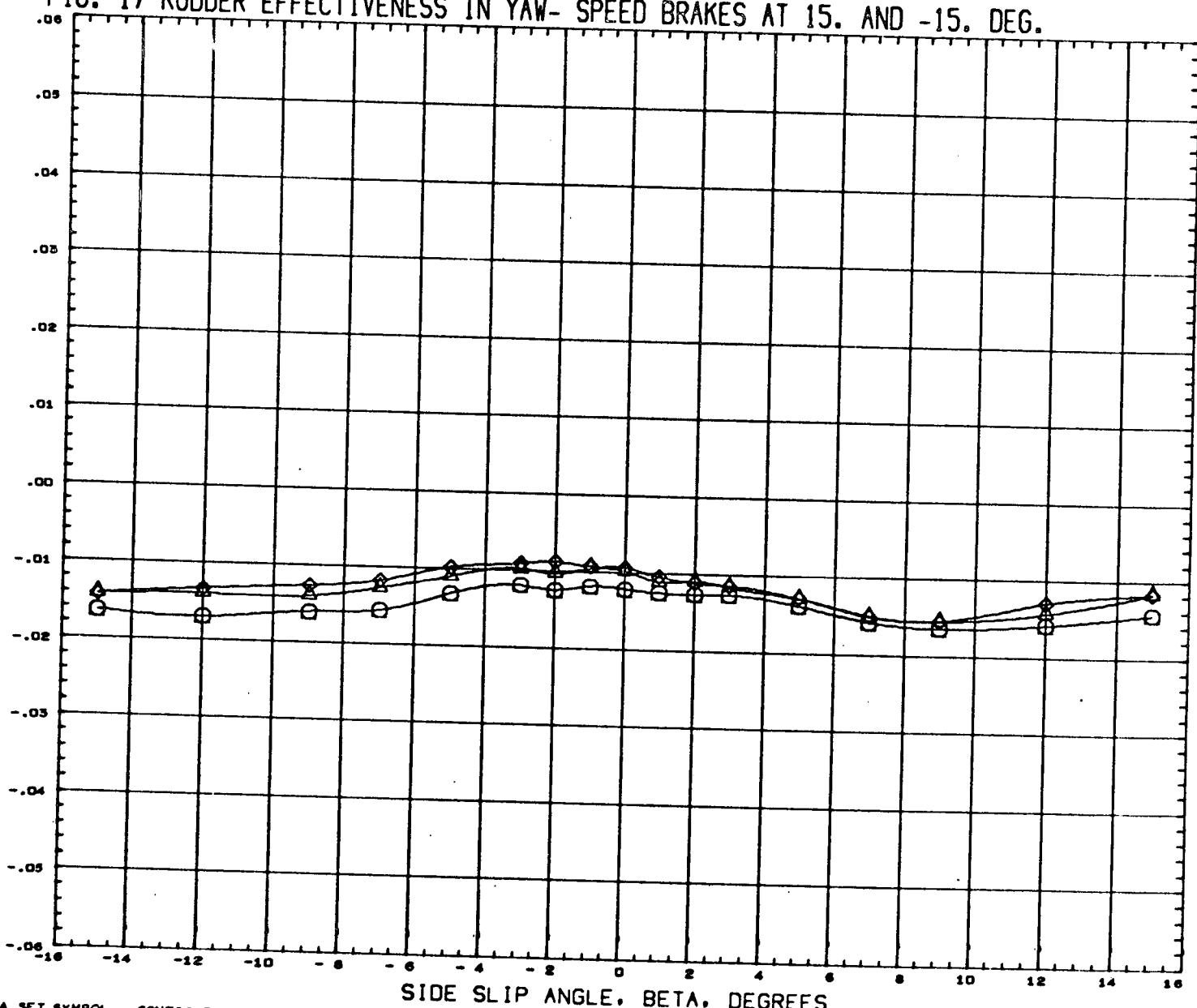
DATA SET	SYMBOL	CONFIGURATION	DESCRIPTION	TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE	INFORMATION
(RD1039)	○	GWTT 292 CONF.H-33 ORBITER	B5W4V5 (0,+15,-15)	0.000	15.000	-15.000	0.000	SREF	7.7440 SQ FT
(RD1038)	△	GWTT 292 CONF.H-33 ORBITER	B5W4V5 (-10,+5,-25)	-10.000	5.000	-25.000	0.000	LREF	5.4000 FT.
(RD1037)	◇	GWTT 292 CONF.H-33 ORBITER	B5W4V5 (-15,0,-30)	-15.000	0.000	-30.000	0.000	BREF	3.7800 FT.

XMRP 1265.0040 IN.
YMRP 0.0000 IN.
ZMRP 403.0004 IN.
SCALE 0.0400

MACH 0.170

FIG. 17 RUDDER EFFECTIVENESS IN YAW- SPEED BRAKES AT 15. AND -15. DEG.

PITCHING MOMENT COEFFICIENT. CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION

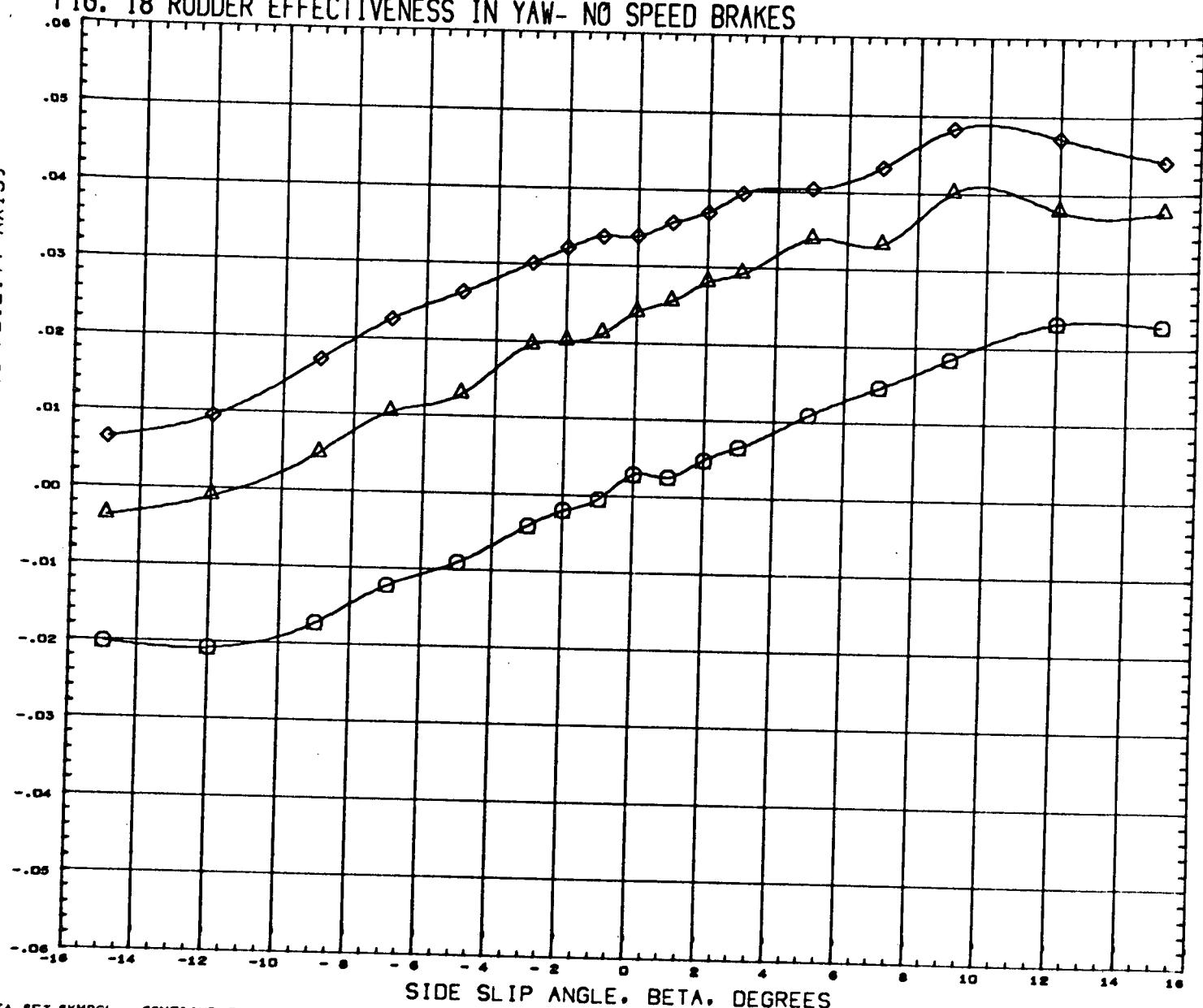
(RD1039) GWTT 292 CONF.H-33 ORBITER 85W4V5 (0,+15,-15)
 (RD1038) GWTT 292 CONF.H-33 ORBITER 85W4V5 (-10,+5,-25)
 (RD1037) GWTT 292 CONF.H-33 ORBITER 85W4V5 (-15,0,-30)

TRUDDR	LLRUDD	LRRUDD	ALPHA	REFERENCE INFORMATION
0.000	15.000	-15.000	0.000	SREF 7.7440 SQ FT
-10.000	5.000	-25.000	0.000	LREF 5.4000 FT.
-15.000	0.000	-30.000	0.000	BREF 3.7800 FT.
				XMRP 1285.0040 IN.
				YMRP 0.0000 IN.
				ZMRP 403.0004 IN.
				SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES

YAWING MOMENT COEFFICIENT. CLN (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1027) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1040) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-10)
 (RD1041) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15)

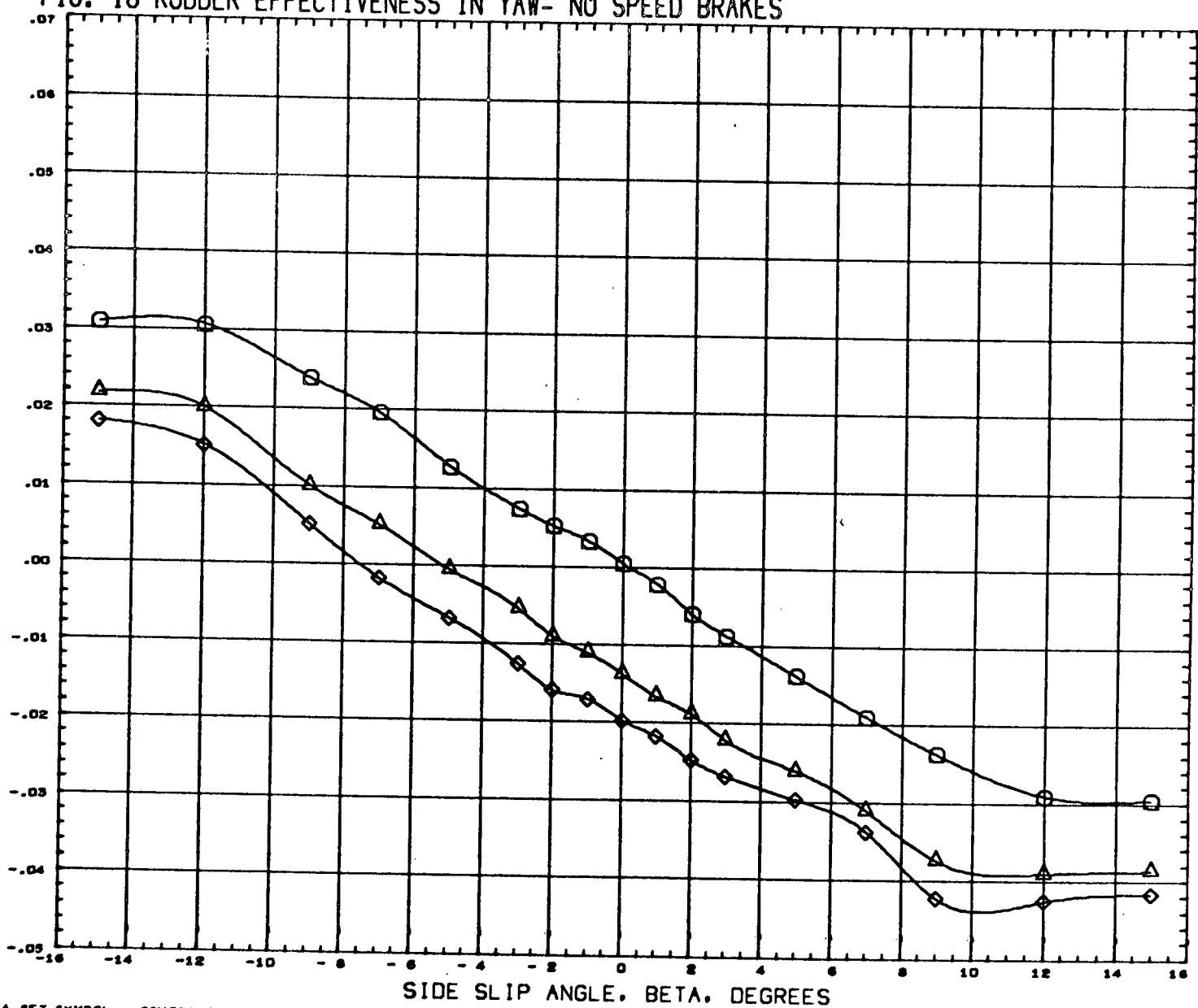
RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES

ROLLING MOMENT COEFFICIENT, CSL (STABILITY AXIS)



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1027) CIRCLE GWTT 292 CONF.H-33 CRBITER 85W4V5
 (RD1040) SQUARE GWTT 292 CONF.H-33 CRBITER 85W4V5 (-10)
 (RD1041) DIAMOND GWTT 292 CONF.H-33 CRBITER 85W4V5 (-15)

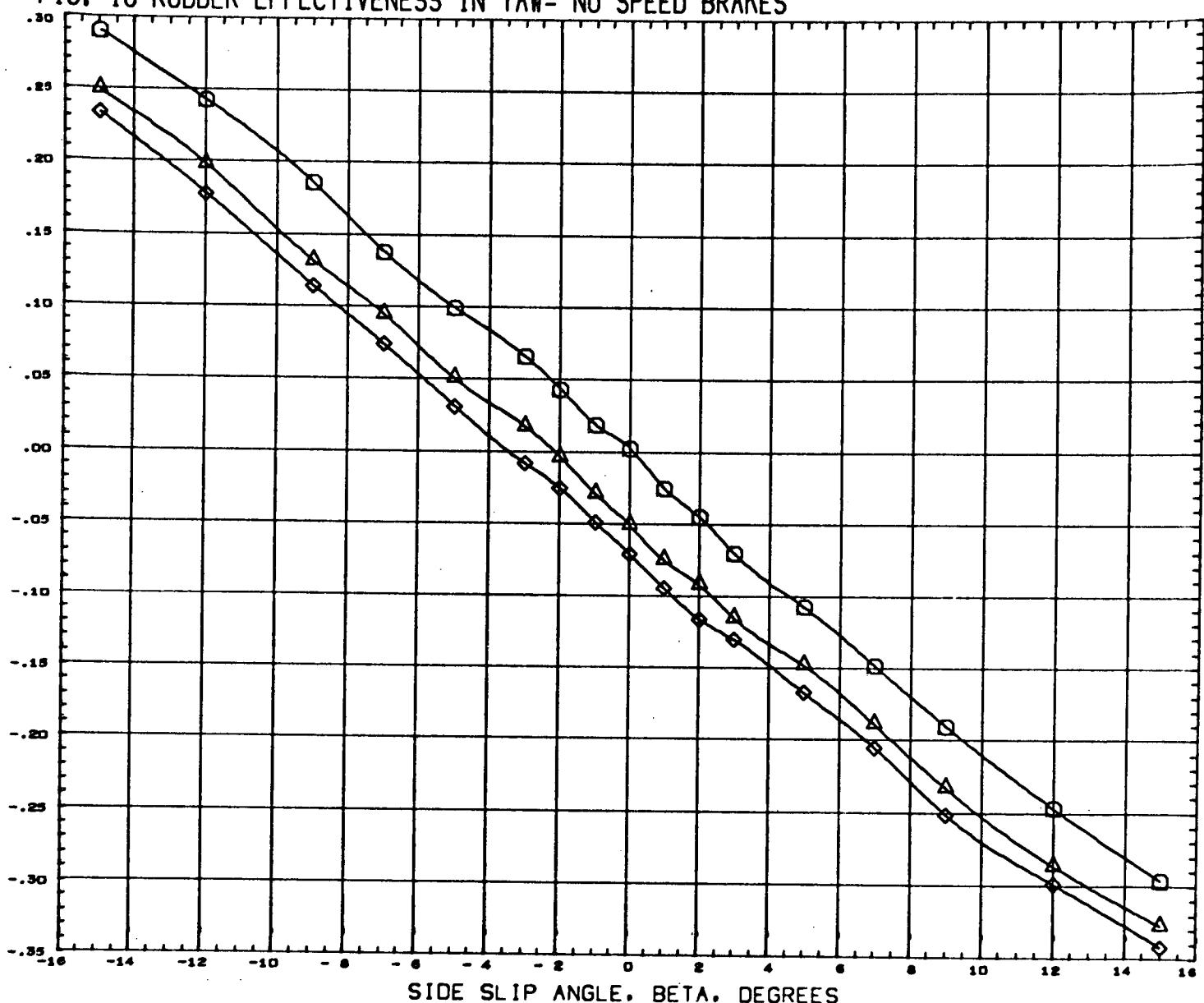
RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES

LATERAL FORCE COEFFICIENT, CY



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RD1027) GWTT 292 CONF.H-33 ORBITER B5W4V5
 (RD1040) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-10)
 (RD1041) GWTT 292 CONF.H-33 ORBITER B5W4V5 (-15)

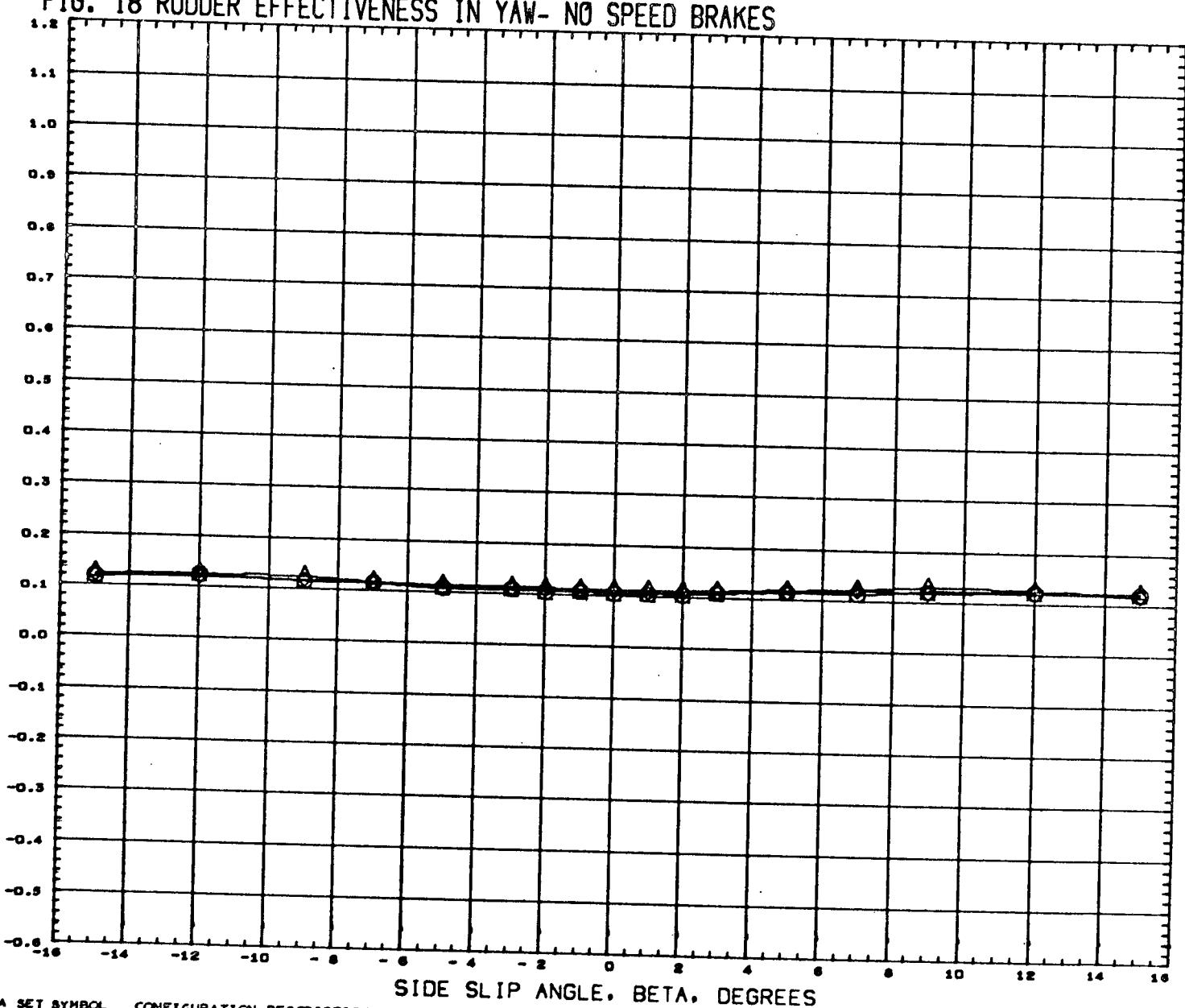
RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES

LIFT COEFFICIENT. CL



DATA SET SYMBOL CONFIGURATION DESCRIPTION

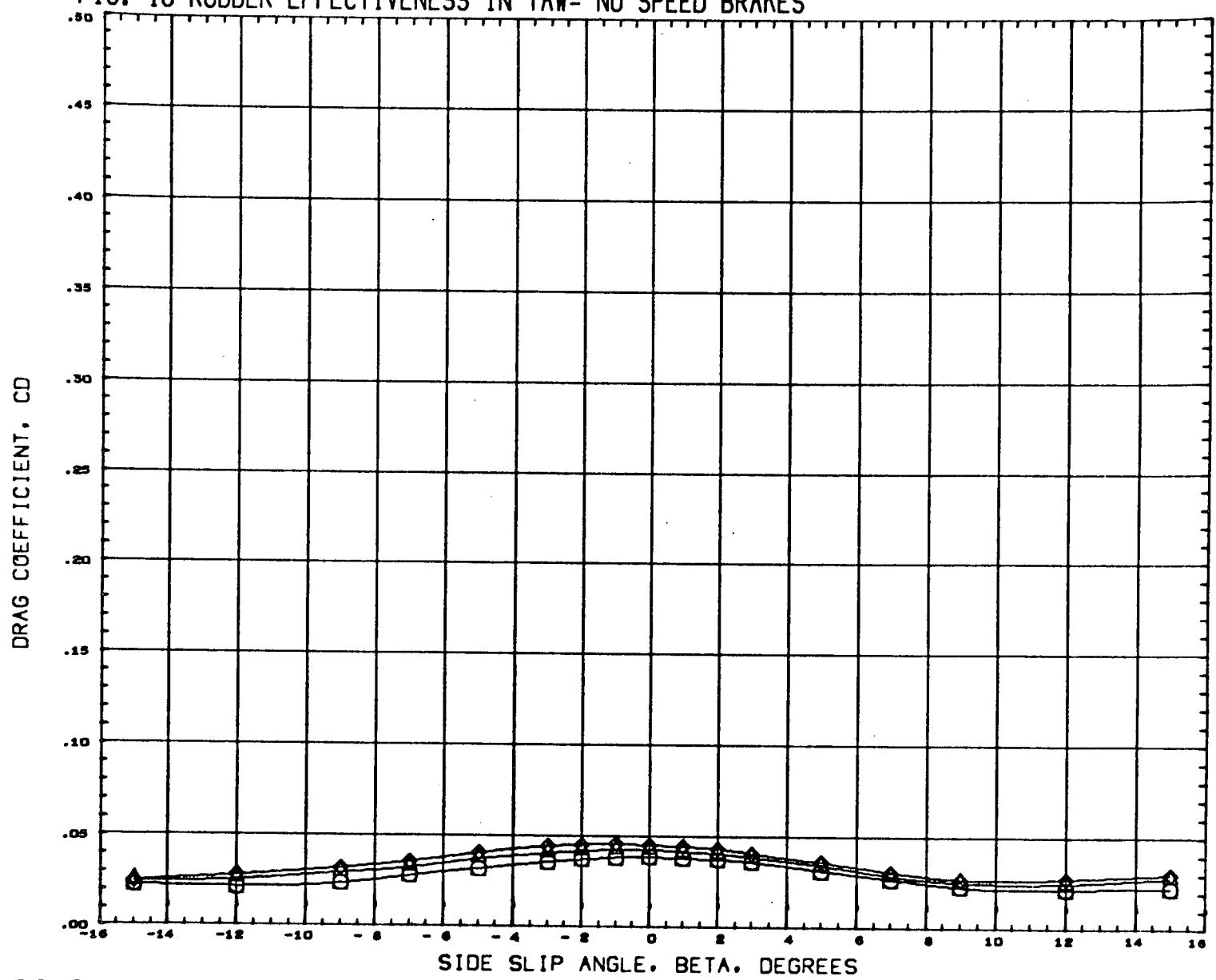
(RD1027) GWT 292 CONF.H-33 ORBITER B5W4V5
 (RD1040) GWT 292 CONF.H-33 ORBITER B5W4V5 (-10)
 (RD1041) GWT 292 CONF.H-33 ORBITER B5W4V5 (-15)

RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7600 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1027) GWT 292 CONF.H-33 ORBITER B5W4V5
 (RD1040) GWT 292 CONF.H-33 ORBITER B5W4V5 (-10)
 (RD1041) GWT 292 CONF.H-33 ORBITER B5W4V5 (-15)

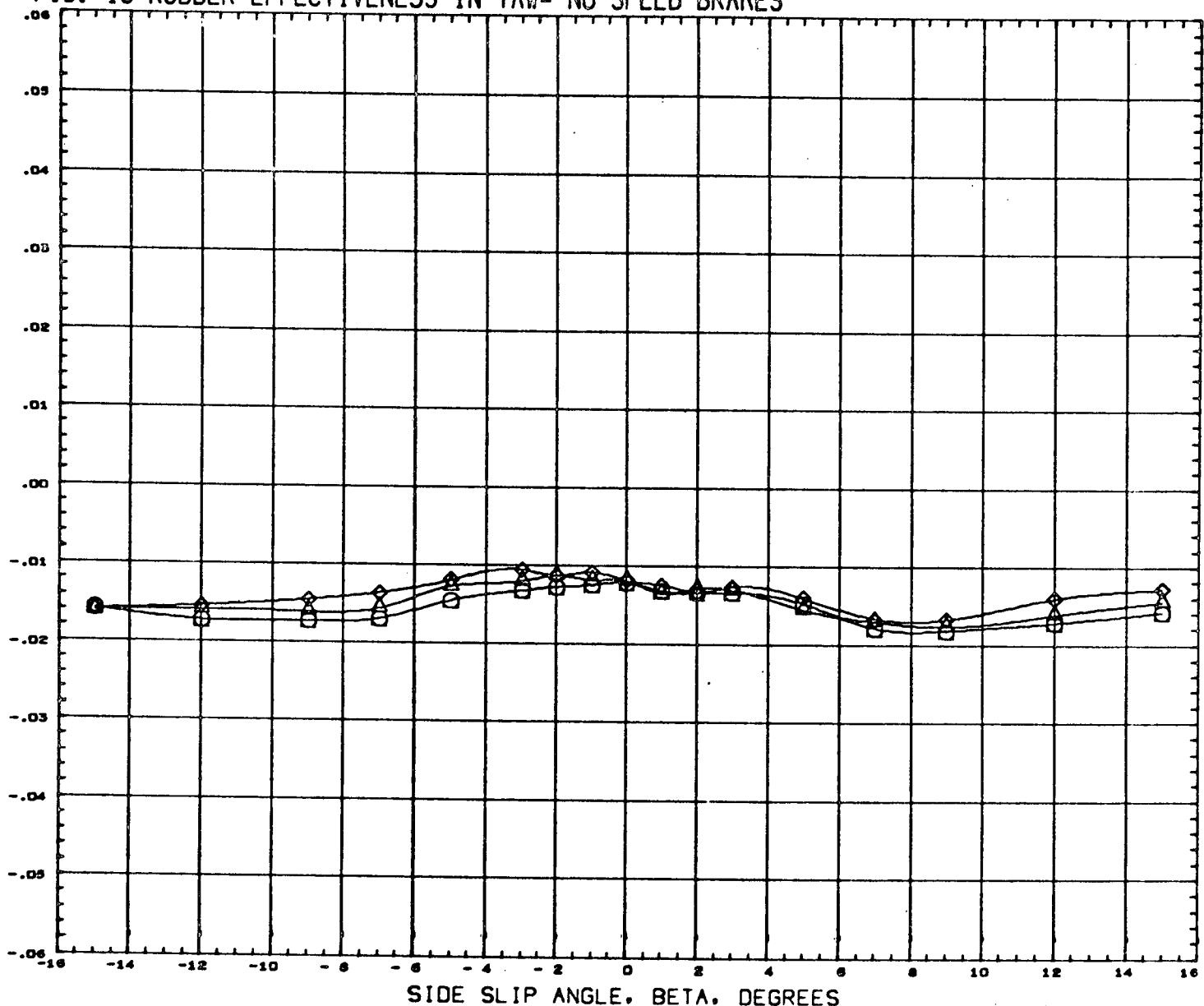
RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170

FIG. 18 RUDDER EFFECTIVENESS IN YAW- NO SPEED BRAKES

PITCHING MOMENT COEFFICIENT, CLM



DATA SET SYMBOL CONFIGURATION DESCRIPTION

(RD1027) GWTT 292 CONF.M-33 ORBITER BSW4V5
 (RD1040) GWTT 292 CONF.M-33 ORBITER BSW4V5(-10)
 (RD1041) GWTT 292 CONF.M-33 ORBITER BSW4V5(-15)

RUDDER ALPHA
 0.000 0.000
 -10.000 0.000
 -15.000 0.000

REFERENCE INFORMATION
 SREF 7.7440 SQ FT
 LREF 5.4000 FT.
 BREF 3.7800 FT.
 XMRP 1285.0040 IN.
 YMRP 0.0000 IN.
 ZMRP 403.0004 IN.
 SCALE 0.0400

MACH 0.170